

Commission of the European Communities

technical steel research

Practical design tools for unprotected steel columns submitted to ISO-Fire — Refao III



Report

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Commission of the European Communities

Practical design tools for unprotected steel columns submitted to ISO-Fire — Refao III

Arbed-Recherches 66, rue de Luxembourg

L-4221 Esch/Alzette

Contract No 7210-SA/505 (1.7.1986-31.12.1989)

Final report

Directorate-General Science, Research and Development

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C.E.C. Agreement N° 7210-SA/505

PRACTICAL DESIGN TOOLS FOR UNPROTECTED STEEL COLUMNS SUBMITTED TO ISO-FIRE

REFAO - III

Period from 01.07.1986 to 31.12.1989

FINAL REPORT

Parts I - П - П

RPS Report Nº 11/91

ARBED-Recherches 66, rue de Luxembourg

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FINAL REPORT

RPS Report Nº 11/91

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"Practical design tools for unprotected steel columns submitted to ISOfire"

Agreement Nº 7210 - SA/505 C.E.C. - ARBED

SUMMARY

The main parameters to be considered in this research programme, i.e. geometrical factors (shapes, buckling lengths), steel qualities and coefficients governing the heat exchanges are presented first.

The temperature dependent stress-strain relationships of steel as initially existing in the program CEFICOSS have been tested by simulation of bending tests described in the litterature. It has shown a necessity to improve these laws when pure steel elements have to be calculated.

New improved stress-strain relationships of steel have been carried out and calibrated thanks to transient state beam tests performed on small simply supported steel beams, subjected to a concentrated constant load, and submitted to a controlled temperature increase. These new laws have been established as well for commonly used construction steels as for high strength steel FeE 460.

The validity of these improved relationships has been next verified by simulating very well six full scale fire tests performed on unprotected steel columns in the laboratories of Braunschweig and Gent.

The possibility to take into account a distribution of residual stresses has been introduced in CEFICOSS. The simulation of the six column tests showed that residual stresses have a quite small influence of the fire resistance time of columns. It has been decided, however, to consider systematically a distribution of residual stresses in the calculations.

Practical design tools have been finally carried out and are proposed in form of tables as well as diagrams. "Outils pratiques de dimensionnement pour poutrelles-colonnes en acier non protégé soumises à l'incendie".

Contrat Nº 7210 - SA/505 C.C.E. - ARBED

RESUME

Dans une première phase sont définis les paramètres essentiels à introduire dans cette recherche, tels que les facteurs géométriques (sections, longueurs) et mécaniques (qualités d'acier), ainsi que les coefficients relatifs aux échanges thermiques par radiation et convection. Les lois de comportement thermomécanique de l'acier à haute température existant initialement dans le programme CEFICOSS ont été éprouvées par des simulations d'essais décrits dans la littérature, ce qui a montré la nécessité de les affiner dans le cas où des éléments purement métalliques doivent être simulés.

Des tests de flexion sur des petites poutres métalliques soumises à une charge constante et à une élévation de température (uniforme) régulière à vitesse contrôlée, ont permis de calibrer de nouvelles lois d'évolution des propriétés métalliques de l'acier en fonction de la température, aussi bien pour les aciers courants de construction que pour l'acier FeE 460 à haute limite élastique.

La validité de ces nouvelles lois a pu être ensuite vérifiée grâce à la simulation de six essais au feu en grandeur réelle réalisés sur des colonnes nues à Braunschweig et à Gand, et ensuite parfaitement simulés par CEFICOSS.

Ensuite, la possibilité de prendre en compte une répartition de contraintes résiduelles a été introduite dans CEFICOSS. Les simulations des six tests ont démontré que ces contraintes résiduelles n'ont pas une très grande importance sur le temps de ruine final, mais il a été néanmoins décidé de les prendre en compte dans tous les calculs.

Enfin, des outils pratiques de dimensionnement ont été établis par calcul et sont proposés sous forme de diagrammes aussi bien que de tables. Praktische Bemessungshilfen für die Interaktion von Normalkräften (N) und Biegemomenten (M) für Stahl-Beton Verbundelemente unter Feuerbeanspruchung (ISO - Kurve)

Vertrag Nº 7210-SA/505 KEG-ARBED

ZUSAMMENFASSUNG

Die erste Phase dieser Forschungsarbeit behandelt die Bestimmung der wesentlichen einzugebenden Parameter. Diese Parameter bestehen aus geometrischen Faktoren (Querschnitt, Länge) und mechanischen Faktoren (Stahlgüte), sowie aus den relativen thermischen Austauschkoeffizienten verursacht durch die Wärmeausstrahlung und Konvektion.

Die thermomechanischen Gesetze von Stahl bei hoher Temperatur, welche anfänglich im Programm CEFICOSS enthalten waren, wurden durch Simulationsversuche gemäss Beschreibung in Literatur überprüft. Diese ergaben die Notwendigkeit die Gesetze zu verfeinern im Falle der Simulation von ungeschützten Stahlelementen.

Biegeversuche von kleinen Stahlprofilträgern beansprucht durch eine konstante Einzellast und einer gleichmässig ansteigenden Temperatur haben es erlaubt, neue Gesetze über die metallischen Eigenschaften von Stahl unter Temperatureinfluss zu entwickeln, welche für geläufige Stahlgüten in der Baukonstruktion und ebenso für Stahl FeE 460 mit hoher Streckgrenze anwendbar sind.

Die Gültigkeit dieser neuen Gesetze kann auf Grund der Simulation von sechs Versuchen (Massstab 1:1) unter Feuerbeanspruchung an ungeschützten Stahlstützen in Braunschweig und in Gent bestätigt werden und konnten nachträglich mit CEFICOSS simuliert werden.

Ausserdem wurde im Programm CEFICOSS die Möglichkeit gegeben Eigenspannungen zu berücksichtigen. Die Simulation der sechs Versuche hat bewiesen, dass diese keinen grossen Einfluss auf das Endergebnis haben, sie wurden jedoch in allen Berechnungen berücksichtigt.

Schliesslich wurden praktische Bemessungshilfen, auf Grund von CEFI-COSS-Simulation, in Form von Diagrammen und Tafeln erstellt.

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PART I

REPORT

1. INTRODUCTION

1.1. Thermo-mechanical computer model CEFICOSS

During the C.E.C. research, agreement N° 7210-SA/502 [1], a computer program for the analysis of steel as well as composite structures under fire conditions has been developed. It is based on the finite element method using beam elements with subdivision of the cross section in a rectangular mesh. The structure submitted to increasing loads or temperatures is analysed step-by-step using the Newton-Raphson procedure. The thermal problem is solved by a finite difference method based on the heat balance between adjacent elements.

The numerical simulation of several full scale fire tests performed during various research projects ([1], [2], [3], [4]) has demonstrated that this numerical software CEFICOSS is able to simulate in a correct way the structural behaviour of elements submitted to fire and provides a pretty good estimation of the fire resistance times. CEFICOSS is a tool which allows most credible prediction of the fire resistance of structural elements, and which can be used particularly for steel columns, with or without fire protection.

1.2. <u>Aim of research</u>

Tests performed at the University of Gent [1] on thick flanged steel columns made clear that a high massivity – the section factor F/V of the steel profile was 27 m⁻¹ – provides a good fire resistance even to bare steel profiles. Only numerical models giving the temperature gradient through profile section are able to predict correctly the behaviour of such thick bare steel elements.

Indeed during the test of an unprotected column a fire resistance time of 45 minutes was measured while the simulation by CEFICOSS gives 46 minutes. This column was loaded at a level corresponding practically to the maximum allowable in normal service conditions, and would not have reached the fire resistance class F30 according to the usual simple calculation method based on the assumption of an uniform temperature inside of the steel section and on the conservative stress-strain-relationships of the ECCS-Recommendations [5]. In order to make the results of this computer code available for everybody, it has been decided to establish N-M interaction diagrams for unprotected steel columns made of massive steel H-shapes.

The research programme has been based on six fire tests performed in the furnaces of Gent and Braunschweig and, from the other hand, on the intensive use of the thermomechanical numerical code CEFICOSS to calculate massive shapes (HD and HEM series) in a parametrical way.

This Final Report summarizes the works performed during the previous research periods and described in the Technical Reports N^{\circ} 1 to 5 ([6], [7], [8], [9], [10]).

1.3. General scope on the parameters

The parameters to be introduced in this research are summarized as follows:

<u>Section</u>: The profiles from HEM and HD series with a flange thickness of at least 40 mm

Finally the following sections are concerned in this programme:

HEM 320 up to 1000 HD 210x210x198 to 249 HD 260x260x219 to 329 HD 310x310x283 to 500 HD 400x400x314 to 1086

Steel grades: Fe 510 and FeE 460

However sections with flange thickness higher than 40 mm are not actually usual in quality FeE 460 and therefore this steel quality will be reserved for HEM series.

Buckling lengths: from 2.00 m up to 8.00 m

Fire resistance classes: F30 and F60

Eccentricity of the load defined from first order bending moments:

- constant in a first step
- a simplified method will be proposed in a second step to cover other distributions.

2. FACTORS GOVERNING THE HEAT TRANSFER

2.1. Heating curve

All the calculations performed in this research have been made with the ISO-834 [11] standard heating curve, giving a gas temperature varying as follows around the heated element:

 $Tg = 20 + 345 \log_{10} (8t + 1)$

With t = fire time in minutes.

2.2. Coefficient of convection

Following the recommendations of Technical Committee 3 of the ECCS [5] it was decided to make all the calculations in this research programme with a value $\alpha = 25 \text{ W/m}^2.\text{K}$ for the convection heat transfer.

2.3. Resultant emissivity

The value of the resultant relative emissivity ε * to be introduced in CEFICOSS can usually vary between 0.45 and 0.7 depending of fire test conditions, and also normally varies during a test with temperature.

As suggested in the recommendations of the ECCS [5], one constant value $\varepsilon^* = 0.5$ could be used for steel surfaces.

However the full scale test done at the University of Gent [1] showed that the temperature in the middle of the web can only be satisfactorily calculated by choosing a resultant emissivity ε^* smaller for the inner surfaces in chambers as for the outer ones, thus simulating the radiative shadow effect (see figure 2.1).

The resultant relative emissivity ε^* in the concave part of a H-section can be calculated as follows according to [12]:

€* _{web}	*3 =	•	Fweb
^{£*} flanged	*3 =	•	Fflange

The coefficients Fweb and Fflange are given by:

Fweb =
$$\frac{-(b/2-a/2) + \sqrt{(h-2e)^2 + (b/2-a/2)^2}}{(h-2e)}$$

Fflange = $\frac{(h-2e) + (b/2-a/2) - \sqrt{(h-2e)^2 + (b/2-a/2)^2}}{(h-2e)^2}$

(b-a)

where h, b, a and e are dimensions of the steel shape height, width, thickness of the web and thickness of the flange.

The values of ε^* for the web and for the inside part of the flanges, and corresponding to $\varepsilon^* = 0.5$ for the outside face of the flanges, have been calculated for each shape concerned in this research.

For HD sections ε^* varies from 0.182 up to 0.188 for the inside face of the flange, whereas ε^* varies from 0.286 up to 0.301 for the web.

For HEM sections ε^* of the inside face of the flange increases regularly from 0.189 to 0.231 for shape increasing from 320 up to 1000, and an average value of 0.2 is not far away from the reality.

As concerns the web, the value of ε^* increases regularly from 0.305 to 0.43.

Therefore it seems reasonable as simplification to adopt for any section the following resultant relative emissivity:

 $\varepsilon^* = 0.5$ for outside faces of the flanges $\varepsilon^* = 0.2$ for inside faces of the flanges $\varepsilon^* = 0.3$ for the web.

and all the calculations have been performed according to figure 2.1.

2.4. Thermal properties of steel

In order to compute the time dependent temperature field in structural elements, the thermal conductivity λ (W/m.k) and the specific heat C (J/kg.K) of steel must be known as functions of temperature. These functions are presented in figure 2.2 for the thermal conductivity and in figure 2.3 for the specific heat, while the thermal expansion for steel is given in figure 2.4. These laws are the original ones introduced in the program CEFI-COSS [1].

3. THERMO-MECHANICAL STEEL PROPERTIES

3.1. Initial stress-strain relationships for steel

The laws describing the temperature dependent stress-strain relationships of steel are given in figure 3.1 to 3.6 as existing initially in CEFICOSS (see [1]). These laws have been established for usual construction steels like Fe 360 and Fe 510 and mainly for calculation of composite sections. It was to be examined, whether these temperature-dependent stress-strain relationships are too much simplified for simulating unprotected bare steel columns.

Furthermore the behaviour at high temperatures of steel FeE 460 had not yet been calibrated before, and it was reasonable to fear a different behaviour of this steel in fire owing to the fact that his properties are obtained by a thermomechanical treatment.

3.2. KRUPP tests

First of all it was interesting to try to find in the litterature some reports over tests performed on pure steel elements and covering as far as possible the field of strains interesting in this research. This possibility was given by bending tests performed by Rubert and Schaumann [13] in KRUPP Research Centre in order to investigate the properties of steel in fire, and it has been decided to simulate some of these tests with CEFICOSS.

These bending tests on profiles IPE 80 are schematically explained in figure 3.7. The beam with a span of 114 cm is situated inside of an electrical furnace, and subjected to a external point load F which is kept constantly during the test. After loading, the temperature inside of the small electrical furnace increases continuously with a given velocity. Because of the <u>small thickness of the profile IPE 80, the temperatures</u> can be considered as <u>uniform inside of the steel section</u>, and the thermal expansion of steel has practically no influence on the vertical deflection, which is registred at mid-span of the beam during the test. This type of test has been performed in KRUPP Research Centre for different loading rates F/Fplastic, and for different heating velocities. Process and results of these tests are described in [13] and [14].

3.3. Simulation of four tests by CEFICOSS

Four tests, called WK1 to WK4 and described in the reports [13] and [14] have been simulated with CEFICOSS with the existing stress-strain relationships, and with the following assumptions:

- 1) The dimensions of sections have been assumed to be constant and equal to the theoretical dimensions of an IPE 80. This assumption is justified by the very small differences measured by Rubert & Schaumann in tests and presented in the report [14]:
 - Average of differences on the inertia: 0.93 %
 - Average of differences on the plastic moment: 0.88 %
- 2) The curves giving the actual measured variation of temperature during tests are not given in [13]. The authors give just for each test a mean value of the heating velocities which are very close to the theoretical one (the highest difference for tests WK 1 to WK 4 reaches 5 %). Moreover, a conclusion of all the tests presented by the authors is that the heating velocity doesn't play an important role on results. Therefore the tests may logically be simulated with the theoretical temperature curve presented in figure 3.8.
- 3) All the temperatures mentioned in [13] are steel temperatures and not gas temperatures. Therefore, the statical calculations in CEFICOSS have been performed with effectively given steel temperatures.
- 4) The mechanical properties of steel used in CEFICOSS were taken from the report and are measured values:

As the actual tensile strengths σ_t have not been measured, the coefficient K of figure 3.1 was taken as a constant value 1.5 in these present CEFICOSS calculations.

3.4. Comparison with test results

The curves D=f(t) given by CEFICOSS in the four simulations have been transformed into the form D=f(T) using the relation T=f(t)defined earlier in figure 3.8. These curves D=f(T) are given in figure 3.9 together with the actual measured displacements.

For tests WK1 to WK3, CEFICOSS gives smaller displacements than measured up to a certain temperature. Over this critical point, the displacements given by CEFICOSS are larger than the measured ones. This critical temperature (for D=40 mm) increases when the rate of loading decreases, and the differences between CEFICOSS and tests go down progressively. For test WK4, which has the lowest load, CEFICOSS gives lower displacements for any temperature.

To sum up, the pseudo-vertical (assymptotic) curves given by CEFICOSS for high displacements go progessively from the left side of the measured curves to the right side when the rate of loading decreases. The crossing point can be roughly defined by $T \sim 600^{\circ}C$ or F/Fp ~ 0.50 .

Up to about 600° C, CEFICOSS gives results which are conservative for a design based on the plastic moment, but they seem to be unsafe for higher temperatures.

In the intermediate zone of lower temperature, where the curves turn from small to high displacements, the existing laws give deflections always smaller than measured in tests and seems to be clearly unsafe.

3.5. <u>Conclusion</u>

The last notice before is very important for the present research. As a matter of fact, the form of the $(\sigma - \epsilon)$ diagram in the intermediate zone just before to reach the plastic plateau has an important influence on the buckling behaviour of columns. The comparison done here shows that the $\sigma - \epsilon$ relationships included in CEFICOSS should be improved to perform calculation of pure steel columns.

4. BEHAVIOUR OF STEEL BEAMS UNDER TRANSIENT STATE BEAM TESTS

4.1. Structural steel qualities

Tests described in [13] have been performed on beams theoretically in Fe 360, but steels were rather of quality Fe 510 according to their yield strengths (see § 3.3). These tests can obviously be considered to cover the quality Fe 510, and the highest quality FeE460 was to be investigated too, in a same way, in the same testing device by KRUPP (figure 4.1).

In the previous bending tests, IPE 80 profiles were used; for steel FeE460, however, such profiles are not rolled and similar sections had to be manufactured. The test pieces have been extracted from a FeE460 steel beam W 360x410x314 in its 40 mm thick flanges (see figures 4.2). The tests take aim at measures of mechanical properties of steel at high temperatures, so that's why it was important to reduce as much as possible the heating of steel during the tooling, with adapted machine speed and cooling. The cuttings have been made preferably by sawing than with blowtorch.

The bending tests in themselves are schematically explained on the figure 4.1 and more detailed in Appendix A of part III. The simply supported beam with a span of 114.7 cm is situated inside an electrical heating furnace, and subjected to an external point load F which is applied at the middle of the span and kept constant during the test. After loading, the temperature induced by the electrical resistance increases continuously with a given velocity.

Because of the small thickness of the manufactured profile, the temperature can be considered as uniform inside the steel section and more, the thermal expansion of steel has practically no influence on the vertical displacement which is registered at the middle of the beam during the test.

To control the assumption of a uniform temperature, thermocouples have been placed on all the beams to record in different points of the steel cross section the time-temperature curves (see page A3 in Appendix A of PART III).

The next page A4 in this Appendix A shows the extrema values of steel temperatures capted by thermocouples for one of the most unfavorable tests and so proves the validity of this assumption.

The heating velocity has been chosen equal to 3.5 K/min; it has been shown in previous tests [13] that variations of velocity have no significant influence on results.

Nine tests of this type have been performed for different loading rates F/F_{pcold} , where F_{pcold} means the theoretical necessary applied force to obtain the middle-span section fully plastified (plastic hinge) with a bi-rectangular stress distribution (rigid-plastic theory). The nine transient state beam tests are called S1 to S7, S9 and S10; S8 is a cold test with loadings and unloadings up to collapse.

1.2. Results of the tests

The pages A5 to A13 of Appendix A, PART III, give the measured vertical displacements (mm) at the middle of the span in function of the temperature (°C), for the nine transient tests, in a decreasing order of loading levels.

The next page A14 shows the measured deflection at mid-span of the beam in function of the load F_r for the cold test S8.

The ten tests are summarized in figure 4.3 as well as in the table of page A15 in Appendix A of PART III, giving the following informations:

- The yield point β s has been determined with tensile test pieces extracted from the flanges of the beam W 360x410x314 as shown on the figure 4.2 (T = specimens for tensile tests). The different values of β s appears in the column Re_H, the superior elastic limits obtained by tensile tests.
- F_{pcold}, the theoretical necessary applied forces to obtain the middle-span section fully plastified (plastic hinge) with a bi-rectangular stresses distribution (rigid-plastic theory).
- F, the applied loads
- F/F_{pcold}, the loading levels
- θ_{m} , the mean velocities of heating
- $\theta_{\text{init.}}$, the initial temperatures during the cold loading

- $(\theta m)_{max}$, the maximal mean temperatures of the steel reached during the test (thermocouples measures)
- (D_{mes.})_{max}, the maximal vertical displacements at the middle of the beam measured during the test
- (t_{test})_{max}, the duration of the test (not fire resistance)

4.3. Simulation with CEFICOSS using the known Fe360 steel RS-LAW

The performed KRUPP tests have been simulated with the following assumptions:

1) All the dimensions of the cross section have been kept constant and equal to the theoretical dimensions of the tooled beams (see figure 4.1). This assumption is justified by the very small differences (lower than 3 %) produced on geometrical and mechanical characteristics of the profile by tooling tolerances of \pm 1/10 mm (see page A16 of Appendix A, PART III).

The modelisation of a quarter of a beam section is presented on the page A17 of Appendix A.

2) Temperature-time curves issued from mean measures $\theta_{m[^{\circ}C]}$ between thermocouples TH5 and TH12 (see page A.3), obtained for each test have been used for simulations. As explained before, differences between measured temperatures are so small that an uniform temperature can be considered every-where through the cross section and along the beams.

Moreover in the previous KRUPP tests, the authors showed that the heating velocity doesn't play an important role on results and so it's the same for differences between all the temperature-time curves with thermocouples for each test.

Therefore, the statical calculations have been performed with mean measured steel temperatures from TH5 and TH12 (see pages A13 and A19 of Appendix A).

The Fe360 steel RUBERT-SCHAUMANN laws (RS-LAWS) defined in [13] have been used to simulate the new tests. These laws are defined in figure 4.4 showing a simplified general σ - ϵ_{σ} diagram for steel. This diagram is characterized by 3 temperature dependant parameters: the elastic modulus $E_{\rm O}$, the proportional stress $\beta_{\rm P}$ and the yield point $\beta_{\rm S}$. Three domains are observed:

the linear elastic, the elliptical elasto-plastic and the plastic plateau.

Figures 4.5, 4.6 and 4.7 give respectively the reduction of the elastic modulus factor Eo(θ)/Eo(θ =20°C), the proportional stress factor $\beta_{\rm p}(\theta)/\beta_{\rm p}(\theta$ =20°C), and the yield point factor $\beta_{\rm S}(\theta)/\beta_{\rm S}(\theta$ =20°C), in function of temperature, for steel Fe 360. Figure 4.8 shows the resultant diagram with all the RS-LAW curves at different temperatures for steel FeE460 ($\beta_{\rm S}(\theta$ =20°C) = 460 N/mm², for example).

The results of the nine fire simulations S1 to S10 and the only cold one S8 are given in Appendix A, PART III, pages A20 to A89 in a decreasing loading level order. The fire simulation figures represent the curve $D = f(\theta)$, the simulated vertical displacement (-....) the middle of the beam in function of the temperature, compared with the measures (-----). The cold simulation figure shows W = f(F), the same type of displacement in function of the increasing load.

For the fire simulation of <u>S1</u> test (<u>loading level = 1.0</u>; that means a fully plastified middle-span section) CEFICOSS can't give any results because the cold loading ends already with problems of numerical convergence (plastic hing failure). Indeed, the RS-LAW ends with a plateau and so doesn't consider the strain-hardening (see page A20).

The same remark can be made for the <u>cold simulation of S8 test</u> because the collapse load cannot be reached without strain-hardening. The failure load obtained with CEFICOSS is 30.6 kN, different in about 2.0 % of the calculated value from the rigid-plastic theory, 31.3 kN. The difference between 30.6 kN calculated with RS-LAW and the measured real failure load 37.65 kN is about -18%.

For the other fire simulations (S2 to S7, S9 and S10) it can be observed that CEFICOSS with Fe 360 RS-LAW leads to a behaviour of the beams not too much different from reality. The results have especially good agreements in the field of usual loading level concerning this research, in other words $F/F_{pcold} = 0.30$ to 0.70.

For high or low loading level, for example 0.85 (S3 test) or 0.10 (S7 test) and 0.075 (S6 test), more important differences are found.

4.4. <u>Conclusions of the simulations</u>

As a matter of fact, the form of the $(\sigma-\beta)$ diagram in the intermediate zone just before to reach the plastic plateau has an important influence on the buckling behaviour of columns. All the comparisons done show that the $\sigma-\beta$ laws included in CEFICOSS should be improved, especially for steel FeE460, when pure steel columns are calculated.

4.5. Improvement of steel laws

- * A first possibility to improve the existing $(\sigma-\beta)$ laws in CEFICOSS, was to adapt the Fe360 Rubert-Schaumann law [13] to the FeE460 steel quality, because the simulations with the RS-LAW in CEFICOSS are not too bad as shown before.
- * Another approach could be to <u>take into account the strain-hardening reality</u>, by use of a simple type of diagram, a quadrilinear law defined by the following temperature-dependent parameters (see figure 4.9):
 - $E_{0,\theta}$, the elastic modulus. $\simeq 0,2$ %
 - $\sigma_{y,\theta}$, the yield point.
 - $E * \theta$, the elastic modulus relevant for strain-hardening.
 - $\sigma_{t,\theta}$, the ultimate stress.

Such a multilinear σ - ϵ idealization, correctly done, would have the advantage to cover conveniently all 10 tests (S1, S3, S7 and S8 included).

In other words:

- * the whole range of loading levels is covered i.e. $0.075 \leq F/F_{pcold} \leq 1.0$ corresponding to the critical temperature field $461^{\circ}C \leq \theta_{max} \leq 828^{\circ}C$ (see fig.4.3)
- * the cold test S8 could also be simulated in a correct way.

So this QUADRILINEAR LAW (QL-LAW), apparently less accurate than a curved law, would have a more general application field as ALL LOAD LEVELS and ALL TEMPERATURE SITUATIONS could be considered.

.6. Additional tests (S11, S12, V1 to V7)

Additional tests have been performed in the same testing device; two of them (S11 and S12) have been performed in the plastic domain for steel FeE460, while seven tests (V1 to V7) have been performed on small sections IPE80 in steel Fe360 (figure 4.10).

As mentionned before, the initial KRUPP tests [13] have been performed on steel having 386 MPa as mean value of yield strength at 20°C, while the specimens used here reach only 313 MPa. Thus the field of yield strengths has been fully covered.

These tests are reported with details in the Appendix A of PART III, pages A30 up to A40

For steel FeE460, the procedure of testing applied for S11 and S12 consisted in starting at room temperature to load the beam, with F/F_{pcold} greater than 1.0; afterwards heating until a definite temperature with a constant load and finally heating and increasing the load of the beam together until collapse.

For steel Fe360 two tests have been performed in the plastic domain with the same procedure as explained before (V1 and V2 tsts), while four tests have been performed in the elastic domain beginning with different F/F_{pcold} strictly smaller than 1.0 and being heated with constant load until collapse (V4, V4, V5 and V6 tests). The V7 test is a cold test with loading and unloading up to failure.

A sensitivity of beams to a well-known parasite phenomenon, the buckling (local buckling of the flanges; lateral-torsional buckling of the whole beam) has been observed, more especially for beams already fully plastified (F/Fcold \geq 1.0) before heating. Therefore the testing procedure has been adapted without changing its validity, in applying loads via a kind of knife edge and in welding by points some stiffeners on each side of the web, out of the middle-span of the beam (only this for V2 and S12 tests) (see details in Appendix A).

5. IMPROVED GL-LAWS

5.1. Definition of the new QL-laws

To define such a modelization, as suggested in § 4.5 by figure 4.9, steady state tensile tests (SSTT) for FeE460 steel quality have been performed in the laboratory of ARBED-Research. They consisted in tensile tests under <u>constant high temperature</u> induced by an electrical furnace around the specimen. Some curves (σ, ϵ) issued from these tests are shown in figures 5.1 and 5.2 with at the same time the <u>QL-IDEALIZATION</u>.

It can be seen on these figures how the four main parameters have been found, respectively for 20° C and 400° C:

$$\begin{bmatrix} \frac{E_{o,\theta}}{E_{o,20C}} \end{bmatrix} , \begin{bmatrix} \frac{\sigma_{y,\theta}^{-}, 2^{\ast}}{\sigma_{y,20^{\circ}C}} \end{bmatrix} , \begin{bmatrix} \frac{E_{\theta}^{\star}}{E_{o,20^{\circ}C}} \end{bmatrix} , \begin{bmatrix} \frac{\sigma_{t,\theta}}{\sigma_{y,20^{\circ}C}} \end{bmatrix}$$

A more precise evolution of these parameters in function of the temperature $\theta[^{\circ}C]$, for QL-law concerning steel FeE460, can be found in figures 5.3 to 5.6. The figures 5.7 and 5.8 show the diagrams (σ, ϵ) of QL-law for high strength steel with different curves corresponding to different given temperatures, respectively with ϵ increasing up to 4 % or 25 %.

5.2. Simulation of KRUPP test with the new QL-laws

All the KRUPP tests described in the previous chapter 4, called S1 to S12 for steel FeE460 and VI to V7 for steel Fe360, have been simulated in CEFICOSS with the new QL-laws proposed above and with the <u>following assumptions</u>:

1) For simulations of S1 to S10 tests (steel FeE460) the same geometrical dimensions have been taken for all the specimens because of the weak influence of the tooling tolerances on the geometrical and mechanical characteristics (see § 4.3).

For the simulations of tests S11, S12 (steel FeE460) and V1 to V7 (steel Fe360) all the specimens have been measured and the real mechanical characteristics of each section have been introduced in the simulations.

The modelisation of a quarter of the beam section is presented in figure 5.9, where the fillets between web and flanges are neglected. 2) For tests S1 to S10, the temperature-time curves issued from mean measures $\theta_{m[^{\circ}C]}$ between thermocouples TH5 and TH12 (see Appendix A page A3), obtained for each test, have been used.

For S11, S12 and V1 to V6 tests, the used temperature-time curves are issued from mean measures $\theta_{m[^{\circ}C]}$ between thermocouples TH6, TH8, TH10 and TH12 (see Appendix A page A33) obtained for each test.

As explained previously the differences of temperatures measured with thermocouples are so small that an uniform temperature can be considered all through the cross section and along the beams.

3) The yield point σ_y and the ultimate strength σ_t have been determined with tensile tests on pieces extracted from the shape W360x410x314 as shown in Appendix A of PART III, on pages A2 and A32 (T = specimens for tensile tests).

The same values σ_y and σ_t have been determined with tensile test pieces extracted from the flanges of the IPE 80 profile as shown also on that page A32 for V1 to V7 tests (P = specimens for tensile tests).

5.3. Comparison of the CEFICOSS results with the measures

** KRUPP tests for steel FeE460 (S1 to S10)

The comparative curves of the KRUPP tests measures and the QL simulations can be seen in Appendix A, PART III in pages A39 to A49, where the vertical middle-span deflections are given either in function of the applied load F for the cold test S8, or in function of the temperature θ in the steel profile in a decreasing order of loading level (F/F_{pcold}) for the other tests.

The cold bending test S8 can be quite well simulated, much better as with RS-law.

The bending test S1 can be simulated in a correcter way (428°C in place of 461°C measured) in spite of the fact that the load level ($F/F_{pcold} = 1.0$) doesn't give any result with RS-law which forgets completely strain-hardening.

The tests S3, S2, S4, S10, S5 and S9 can be simulated as well as with RS-laws.

For load levels $F/F_{pcold} = 0.10$ (S7 test) and 0.075 (S6 test), the simulation by QL-law is also quite acceptable, contrary to RS simulation giving always unsafe answers.

** KRUPP tests (S11 and S12)

For S11 and S12 tests, the procedure consisted in starting at room temperature to load the beam, with F/F_{pcold} equal to 1.05 (plastic domain), heating until ~ 70°C with a constant load and finally heating and increasing the load of the beam together until collapse. The variation of loading $\Delta F/\Delta \theta$ [kN/°C] was different in the two tests:

- 0.024 for S11,
- 0.009 (~ 38% of 0.024) for S12

to pass through the plastic domain by different paths.

For a first approach with the QL-law numerical simulations we increased the load until the measured failure value and kept it constant afterwards until the simulated failure (see pages A37 and A38 in the Appendix A of PART III).

Because of the lateral-torsional buckling especially more sensitive in the plastic domain an early failure occured for S11 test; S12 specimen has better results according to the welded stiffeners and the knife edge which transmits the load, both introduced to avoid as much as possible buckling problems without changing the validity of the test.

Pages A52 and A53 of Appendix A show the comparison between the measured and the simulated mid-span vertical deflection of the beam in function of temperature and also show the parameter $(E^{*}/E_{0.20^{\circ}C}) 20^{\circ}C$ influence of the defining the hardening in QL-law idealization. It can be seen for these particular cases (particular ranges of strains and load levels) that with the approximation of QL-law, the use coefficient $(E^{*}/E_{0.20^{\circ}C})_{20^{\circ}C} = 0.0172$ of а higher improves the results at cold conditions, but still provides higher deflections than reality. To approach better the reality, the law would have to leave (Quadri-Linear) QL model and to follow more closely the true (σ, ε) diagram with a multilinear idealization, for example.

** KRUPP tests for steel Fe360 (V1 to V7)

For numerical simulations with QL-law, used the same evolution of the four parameters in function of temperature as for steel FeE460 has been used. Only the values at room temperature corresponding to the real characteristics of this steel quality have been changed as follows, using the measured mean values:

- $\sigma_{y,20^{\circ}C}$ = 313 N/mm² (yield point) - $\sigma_{t,20^{\circ}C}$ = 505 N/mm² (tensile strength)

<u>V1 and V2 tests</u> are similar to tests S11 and S12 (FeE460 steel): the beam was first loaded at room temperature with $F/F_{pcold} = 1.10$ (plastic domain), and next heated until ~ 120°C (V1 test) or ~ 110°C (V2 test) with a constant load, and finally heated and subjected to an increasing load until collapse.

The variation of the loading $\Delta F/\Delta \theta$ [kN/°C] was different in the two tests:

- 0.030 for V1
- 0.012 (≃ 40% of 0.030) for V2



to pass through the plastic domain by different paths.

Appendix A, PART III, pages A54 and A55 show the comparison between the measured and the simulated mid-span vertical displacements of the beam in function of temperature for V1 and V2 tests.

Because of the lateral-torsional buckling an early failure occured for V1 and probably for V2 test .

More details about the performed simulations can be found in [10].

** <u>V3 to V6 tests</u> (see Appendix A, PART III, pages A56 to A59)

For the load levels 0.85 (V3 test), 0.60 (V4 test) and 0.50 (V5 test), a good agreement with the measures is found by using the QL-law. The calculated deflections follow quite well the evolution of the measured ones; in the ultimate conditions, the simulation is always in the safe side, and the failure temperatures are very similar.

For the low load level 0.10 (V6 test), the simulation with QL-law is also quite acceptable giving safe answer. The deflections are higher in the simulations but the failure temperature is similar.

For the <u>cold bending test V7</u>, similarly as made for test S8, two simulations have been performed by varying the value of QL-law parameter $(E_{20^{\circ}C}^{*}/E_{0,20^{\circ}C})$

The initial value 0.0061 leads to a quite good simulation.

5.4. Conclusion of the simulations

As a general conclusion, it can be said that the ARBED QL-LAW presented here allows to represent well the physical steel behaviour for COLD and for HOT conditions from 20°C to 900°C, for all load levels

$1.21 \ge F/F_{cold} \ge 0.075$

for FeE460 and Fe360 steel qualities (fig. 5.10 and 5.11). The undoubtful advantage of this idealization is that the calculation results are on the safe side. It's better to use the low slope of the QL-law hardening part $[(E*_{20^{\circ}C}/E_{0,20^{\circ}C}) = 0.0061]$ (see fig. 5.5) to stay more on the safe side.

6. FULL-SCALE TESTS OF COLUMNS

6.1. Description of the columns

The tested columns described in Appendix B of PART III have been selected in order to cover as far as possible the parameters of this research, according to the disponibility of shapes (for instance limited to a thickness of 40 mm for FeE460) and to the possibilities of testing devices (4.14 m in Gent and 5.70 m in Braunschweig).

End plates and stiffeners have been welded at both ends of unprotected steel columns, which have been loaded in the furnaces with a constant first order eccentricity. Rotations were not restrained at the ends thanks to cylindrical supports.

Details concerning the specimens are given in table 6.1 and in Appendix B, as well as in [15] and [16].
This table 6.1 shows that column specimens cover the field of the flange thickness from 39.6 (~ 40) mm up to 125 mm, with intermediate values of 45 and 75 mm. The section factor U/A varies from 20 up to a maximum of 58.4 m⁻¹.

Five specimens were in steel Fe510, while the sixth one was in steel FeE460, and a possible direct comparison was offered with tests n° 5 and 6, in spite of different slenderness ratios and rates of loading. This table shows also that five tests have been performed with bending about the minor axis, while one has been subjected to bending about the major axis. Four tests have been carried out in the furnace of GAND in Belgium with a length of 4.14 m, while two occured with a length of 5.70 m in the furnace of Braunschweig in Germany.

Moreover, it has to be pointed out that the six specimens have various slenderness ratios (from 0.3 up to 1.3), as well as loading rates and eccentricities (or eccentricity ratios e/d).

In order to compute the ultimate load in normal service conditions, the method proposed by the German Standard DIN 18800 part 2 [17] has been adopted here as reference just because it was immediately available on the computer.

6.2. Results of the tests

The six full scale fire tests have been performed during September and October 1988 in the furnaces of the Universities of Gent (Belgium) and Braunschweig (Germany), and are reported in [15] and [16]. The actual parameters and the results of these tests are summarized in table 6.1, showing that these unprotected steel columns reached fire resistance times varying from 37' up to 68'.

The following data have been measured during the tests:

- the temperatures in many points of the furnace,
- the temperature of steel measured with 32 to 40 thermocouples, located at the surface as well as inside of steel in two different cross sections of the column,
- the vertical and horizontal displacements.

The results of these measurements are given in the Appendix B of PART III on the same diagrams as the simulations made with CEFI-COSS.

6.3. Simulation of the six full scale fire tests with CEFICOSS

The simulations have been performed with CEFICOSS under consideration of the actual sizes of cross sections, which are given in table 6.2 as well as the initial imperfections measured on the column in the buckling direction.

The fillets between web and flanges have been neglected in the simulation carried out by following the process hereafter:

- 1) The actual measured geometrical sizes are introduced according to table 6.2. (The sizes given in this table are the mean of many measurements).
- 2) The heat transfer coefficients were given by the Fire Laboratories of Gent and Braunschweig $\alpha = 18$ and $\varepsilon = 0,45$ for tests performed in Gent and $\alpha = 25$ and $\varepsilon = 0,7$ for tests performed in Braunschweig.
- 3) The statical calculation started only after check of the good accordance between calculated and measured temperatures.

Moreover, these values given by the laboratoires have been modified for the web and the inner side of the flanges to take into account the shadow effect which is a physical reality. The physical meaning of the shadow effect and its influence on the emissivity factor ε are given in figure 6.4.

In reality at a given time, the temperature measured in one place of the section is not unique, but there is some difference between the values given by all the thermocouples situated in a same place (see details in Appendix B).

In figure 6.3 are indicated the heating parameters of each furnace and the <u>fire resistance times calculated by CEFICOSS with</u> <u>the new QL-8 steel law</u> presented in this report, and adapted according to the actual yield point and tensile strength of steels.

6.4. Conclusions

It has been pointed out that the programme CECLFOSS is able to simulate correctly the behaviour of thick pure steel profiles submitted to ISO-fire. The proposed QL-8 steel law gives very good concordance with the tests, and has been adopted to perform the calculations.

7. PARAMETERS

7.1. Section of steel shapes

As noted earlier, only shapes having a flange thickness of at least 40 mm are concerned by this research. Therefore only HEM and HD series could be interesting, and a priority has been given to the HD profiles which are particularly well adapted for columns.

7.2. Bending moment distribution

Except when transverse loads are applied between columns ends - which is not usual in normal buildings - the distribution of bending moments along the column may have any form presented in figure 7.1.

As shown in figure 7.2 by means of a diagram N-e, distribution type (1) ($\psi = -1$, bi-triangular distribution) is the most favourable one, allowing highest eccentricity for a given axial load, or the highest axial load for a given eccentricity. The distribution type (5) ($\psi = 1$, uniform distribution) is the most unfavourable one, whereas the unsymetrical distribution (2) is most usually encountered.

To go in the same way as Standards for design in normal service conditions, the columns have been calculated here with the most unfavourable distribution of moments which is uniform, having a constant eccentricity and therefore a constant first order bending moment. A transformation method is proposed next for other distributions, allowing to reduce the unfavourable effect of the uniform one.

7.3. Buckling lengths

The column buckling lengths considered in this project have been chosen between 2.00 meters and 8.00 meters by steps of 2.00 meters. Therefore, pin-ended columns of 2.00, 4.00, 6.00 and 8.00 meters have been calculated, whereas a simple linear interpolation method is sufficient for any intermediate length.

7.4. Design strength of steel

Any section considered in this program has a flange thickness equal or greater than 40 mm. This thickness is 40 mm for the HEM series, and varies between 40 and 125 mm for the HD series. Eurocode 3 [19], which has to be used for the design in normal service conditions, stipulates that the yield strength σ_y should be reduced according to the thickness of steel. This rule should be applied to Fe510, but steel FeE460 is not directly mentioned in this Eurocode.

To simplify and to remain in the safe side, the decision has been taken to use in the calculations a yield strength reduced as follows for steel Fe510, in accordance with EURONORM 25 [19]:

Flange thickness [mm]

from	<u>up to (≤)</u>	<u>σy [N/mm²]</u>
16	40	345
40	63	335
63	80	325
80	100	315
100	125	305

The tensile stength has been taken everytime equal to 510 $\rm N/mm^2.$

For steel FeE460 the following characteristic values have been used, in accordance with [20] for flange thickness of 40 mm:

 $\sigma_y = 450 \text{ N/mm}^2$ $\sigma_t = 560 \text{ N/mm}^2$

7.5. Calculation with CEFICOSS in normal service conditions

The static modulus included in the program CEFICOSS can be used to calculate by iterations the ultimate load at ambient temperature. That is made systematically because it furnishes the highest possible load which can be progressively reduced to establish curves N-t allowing to find loads corresponding to 30 and 60 minutes of fire resistance. In reality, however, it must be pointed out that the load given by CEFICOSS is a little different from the load calculated according to Eurocode 3 [18] for, mainly, the strain hardening effect is considered in the stress-strain relationships included in CEFICOSS. Of course, <u>FIRST OF ALL</u>, <u>DESIGNERS HAVE TO</u> <u>COMPLY WITH THE STANDARDS FOR NORMAL SERVICE</u> <u>CONDITIONS</u>. But for a quick and simple information, it was interesting to show together ultimate loads in normal service conditions and in fire.

From the other hand, some National Fire Codes accept clearly a load reduction in fire, and this concept will probably appear too in the final version of Eurocode 3. It is not logical indeed to consider for instance the full wind load on building together with a fire, or to have a crane in full action in an industrial building when fire occurs.

Therefore an interaction curve in fire has not to be directly compared with the interaction curve at ambient temperature where the values have been divided by 1.5. Such a comparison is not enough representative to qualify the economical interest for a construction system, and would not be sufficient for designers. It is necessary to give interaction curves in normal service conditions, as it has been made in this research.

Then, it has to be clearly noticed that results for normal service conditions given in this report are just furnished as information for scientific purpose.

7.6. Initial imperfection introduced in CEFICOSS

An initial column imperfection of the column has to be defined for CEFICOSS analysis, and has an influence for low bending moments. There is of course no specification in Standards to select this parameter, because a calculation in fire conditions is a quite new concept.

Eurocode 3 [18] defines a geometrical imperfection L/1000 to be applied with residual stresses in a second order analysis at ambient temperature. In another research [21] dealing with composite columns, the following initial imperfections have been adopted.

 e_0 = constant = L/500 for bending about the minor axis e_0 = constant = L/1000 for bending about the major axis To have a comparison with a few National Standards, several calculations have been carried out with CEFICOSS at ambient temperature, without considering the strain hardening effect of steel [7]. Some differences could be observed between the various standards and codes, and they are particularly significant for buckling about the minor axis. It appeared, however, that CEFI-COSS gives results which are in good concordance with the nonlinear method of second order proposed in the German DIN 18800 [17], as well for bending about minor axis as for bending about major axis, when calculations are run with geometrical imperfections L/500 and L/1000.

In order to keep uniformity in the calculations made with CEFI-COSS it has been decided to adopt here too these geometrical imperfections defined in [21], in spite of the fact that residual stresses will also be considered. The results will be a little in the safe side, especially for bending about the minor axis.

The constant initial imperfections defined above have been systematically introduced in the calculations <u>in addition on the</u> <u>indicated first order eccentricities</u>.

7.7. Failure criterion

The fire resistance time of a structure can be based on different criteria, usually depending on type of structure. For a column, failure corresponds practically to BUCKLING. In the program CEFICOSS, the mathematical simulation of this physical behaviour is given by the Determinant of the Structure Stiffness Matrix, which being positive becomes negative (DSSM = 0); from a practical point of view, it is sufficient to analyse the Minimum Proper Value (MPV) of the matrix, which also goes to zero.

The behaviour of the column can also be observed through horizontal displacement and displacement speed of the mid-height node, which rapidly increase when buckling occurs.

For columns many examples proved that a deflection criterion (D = L/10 for instance) has only a significant influence for very high bending moments, what means for very low axial loads N. In a practical point of view that zone of the interaction diagram is not really interesting as corresponding to a situation which never occurs in a building. Therefore it was decided to consider the <u>equilibrium failure as single criterion for columns</u>.

'.8. Influence of residual stresses

To be able to evaluate the influence of residual stresses on the fire resistance, the program CEFICOSS has been enlarged to accept a distribution of stresses in equilibrium inside of the cross section.

Using the distribution as well as the highest values suggested in Eurocode 3 [18], the stresses shown in figure 7.3 have been introduced in CEFICOSS to simulate the tests n° 2 and n° 3 described in Appendix B, PART III, where:

 $\sigma_{R(max)} = 0.3 \sigma_{V} = 0.3 \times 29.8 = 8.94 \text{ kN/cm}^2$

The results of both simulations are given in figure 7.4. The observed differences are very small, particularly for test 2 when the column is buckling about the major axis ($\overline{\lambda} = 0.3$). For the column of test n° 3 having a slenderness ratio of about 0.8 the <u>difference between the resistance times remain quite small</u> (1 minute, about 2 %).

It was also interesting to check the influence of residual stresses on the column of test 1 having the highest slenderness ratio for a low fire resistance time. The supposed residual stresses distribution is given in figure 7.3, with a maximum of 0.3x36.4 = 10.92 kN/cm². The table of figure 7.4 shows that the calculated fire resistance time decreases down to 35 minutes (~ 5 %) in this case.

Of course, actual residual stresses in the beam have not been measured, but these simulations show that their influence is probably quite small for thick flanged columns in the domain considered in this research. As this influence consists in a small reduction of the fire resistance time, and in order to have conservative results, it has been decided to perform <u>the</u> <u>calculations by taking into account residual stresses as proposed in Eurocode 3</u> [18] and explained on figure 7.5. The stresses will be kept constant on the whole flange or web thickness (as made in figure 7.3) and will be combined with the initial geometrical imperfection defined earlier in § 7.6 and given in figure 7.5.

8. DIAGRAMS

8.1. Calculation process

For any column cross section the calculation of temperatures in different patches of the discretized section is subordinated neither to the column length nor to the loading. Therefore the thermal analysis can be done first and the resulting time dependent temperatures can be used as data for every static calculation dealing with the same cross section.

The statical calculation process is described in figure 8.1; the first step is to find by iterations the ultimate axial load at ambient temperature (point A on vertical axis corresponding to a time t=0'). Then CEFICOSS is run by reducing progressively the load for a series of eccentricities. For instance a load corresponding to the point B applied with an eccentricity of 15 cm gives a fire resistance time $t_{B,15}$, and defines the point C in figure 8.1.

All the curves established with various eccentricities permit to read by interpolation the axial loads N for the required fire resistance times 30 and 60 minutes as shown on figure 8.1, and to create a file including pairs of values N-e for each fire resistance class.

In reality whole curves established with about 20 points as given in figure 8.1 are not necessary; it is sufficient to find points near the classified fire resistance times to allow a quite accurate interpolation. An average of not less than 3 simulations are needed, however, for each requested point.

For a practical purpose, constant <u>relative eccentricities</u> have been chosen: ratios e/h = 0, 0.10, 0.25, 0.50, 1.00, 2.00 and 4.00 have been used for bending about major axis, while ratios e/b = 0, 0.10, 0.25, 0.50 and 1.00 have been used for bending about minor axis. h and b are of course the height and the width of the steel shape.

8.2. Diagrams N-e

The presentation of results in a form N-e has been selected rather than N-M, first to have two independent variables, what leads to simplier mathematical function, and from the other hand in order to avoid the calculation of the pure bending moment corresponding to a fictitious situation. As the number of parameters is not too high, both tables or diagrams can be convenient for designers, and both forms are presented for the examples given in **PART II** of this report.

Tables are, however, the main data base which allows to build up the diagrams by means of any spreadsheet software with integrated graphic possibilities.

8.3. Interpolation on the buckling length

The simple linear interpolation method is proposed to evaluate the ultimate load corresponding to a buckling length between those given in the tables.

The examples of figure 8.2 show that by performing a calculation in CEFICOSS with the interpolated load for 5.00 m, a fire resistance time of 30 minutes is practically found in both examples. Moreover, if more accuracy was needed, the users have a possibility to improve the interpolation, for four buckling lengths are given in tables allowing to see the look of the curve, as shown in figure 8.3

8.4. Transformation method for non uniform moment distribution

As noted before, calculation have been performed with an uniform first order bending moment ($\psi = 1.00$ according to figure 7.1).

At ambient temperature, Eurocode 3 [18] proposes to apply a corrective factor β on the moment in the interaction formula, β depending only on the form of the moment distribution.

Figure 8.4 shows a column HD260x260x329 calculated for vairous lengths and eccentricities, as well as for bending about both axes and for an uniform or a bi-triangular bending moment distribution.

In the first of the three last columns of the table, the method proposed in Eurocode 3 has been applied as defined in figure 8.5, and leads sometimes to unsafety when eccentricity increases.

The β - ψ method used in the two last columns of the table has been established in another research [21] using CEFICOSS to calculate steel-concrete composite columns. In this method β is not only a function of the form of the moment distribution, but also of the slenderness ratio and of the relative eccentricity e/h (or e/b). Except for the calcul of β , this method is simular to the method of Eurocode 3, presented in figure 8.5.

The differences observed in these two last columns come from two possible ways to apply the method. In the penultimate column, N_{bitr} has been calculated exactly as in figure 8.5 only from the uniform value Nu, what means:

$$\begin{array}{c} \frac{Nc}{K_{u}} = \frac{Nc}{Nu} = 1\\ e \end{array}$$

and

 $\frac{N_{\text{bitr}}}{1 + \beta.\text{Ku.e}} = \frac{N_{\text{C}}}{1 - \beta (N_{\text{C}}/N_{\text{U}} - 1)}$

In the last column, however, an equivalent eccentricity has been calculated: $e_e = \beta e$ and $N_{\text{bitr}}^{\bullet}$ has been found by interpolation with this equivalent eccentricity in the table given in PART II for this section and for an uniform distribution.

The differences observed between the two possible ways to apply the β - ψ method are quite small, and depend only on the more or less good applicability of the basis formula presented first in figure 8.5.

It can be observed, however, that the values found in these last columns of figure 8.4 with this β - ψ method are more conservative that the other ones. This method could be used rather than the method of Eurocode 3.

9. CONCLUSIONS

Practical design tools for thick flanged steel columns are given here, allowing to take really benefit of the massivity of these shapes to save in some cases the costs of a fire protection. The differences of temperature existing inside of the cross section are taken into account to fully use this advantage, what could not be done by methods based on the assumption of an uniform steel temperature. This research will finally allow to use steel elements more economically. Moreover the validity of the program CEFICOSS has been improved to simulate whole structures including as well steel-concrete composite elements as pure steel elements ([22], [23], [24]).

In a next step, to improve the convenience of the results, it should perhaps be envisaged to put them on a floppy disk with a simple program allowing on a personal computer quick interpolations as well as the transformation of the bending moment distribution.

Moreover, the results of this research can be used as a basis to establish a simplified method or to improve some existing methods. For instance it can be observed in tables given in PART II that it is difficult to define exactly when the massivity could be sufficient to reach automatically the fire resistance class F30 with a correct design in normal service conditions. As a matter of fact, that occurs for F30 when:

$$N_{F30} \ge \frac{N(F0, EC3)}{1,5} = maximum service load with safety factor equal to 1,5$$

It can be observed in the tables that it depends not only on the flange thickness or on the section factor U/A, but also on the slenderness ratio of the column, and, moreover, on the eccentricity.

To simplify as far as possible, only the **flange thickness** and the **slenderness ratio** could be considered, with the minimum ratio N/N(F0,EC3) of any eccentricity. Moreover, the given ratios N/N(F0,EC3) could be multiplied by 1,5 to have a direct comparison with the maximum service load calculated according to EC3. This process leads directly to the **diagrams** of **figures 8.7 and 8.8** which are very easy to use and to interpretate, showing clearly what should be the **reduction of load in the worst case for the fire classes F30 and F60**.

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C.E.C. AGREEMENT

N° 7210-SA/505

FIGURES



Resultant emissivity f^* chosen for the numerical simulation of the column test 1.1.

Figure 2.1



Figure 2.2 : THERMAL CONDUCTIVITY



Figure 2.3 : SPECIFIC HEAT



Figure 2.4: Thermal expansion.







Fig.3.2: Reduction of the yield point of steel σy in function of temperature Θ.



Fig. 3.3 : Reduction of the elastic modulus E_0 of steel in function of the temperature.



Fig. 3.4 : Variation of the elastic modulus E^{*} of steel in function of the temperature.

----- "according to Anderberg"



Fig. 3.5 : $G - \mathcal{E}_{\sigma}$ - diagram for Fe 510 at different temperatures.



Fig. 3.6 : $\mathbf{C} - \mathcal{E}_{\mathbf{C}}$ - diagram for Fe 510 at different temperatures with elongations $\mathcal{E}_{\mathbf{C}}$ shown up to 20%.

KRUPP TESTS 1984

REFERENCES [13] and [14]





Figure 3.7





Tests: WK1 to WK4

Temperature dependent beam displacements as a function of utilization factor F/Fp (mean heating velocity T_m = 2,67 K/min).

* note : the test WK1 has been performed two times.

Figure 3.9

KRUPP TESTS 1988 FOR ARBED



Figure 4.1







Position s	of	test	pieces
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S= beams for transient state bending tests T= bars for tensile tests



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Krupp transient	state beam	tests parameters	(S1 to S10)
-----------------	------------	------------------	-------------

F										
TEST	ReH	FPCOLD	F	F/FPCOLD	Đ _m	O init	(Əm)max	(D _{mes})max	(t test) max	
	[N/mm ²]	[kN]	[kN]		[K/min]	[°C]	[°C]	[mm]	[min]	
S1	502	30.0	30.0	1.00	3.6	22.1	461	85.8	121	(1)
S2	504.5	30.2	22.7	0.75	3.4	22.5	525	53.9	146	(2)
S3	507	30.3	25.8	0.85	3.5	21.7	497	53.2	137	(2)
S4	516	30.8	18.5	0.60	3.5	21.4	566	53.1	155	(2)
S5	513	30.7	12.3	0.40	3.5	21.0	651	87.5	182	(2)
S6	529	31.6	2.4	0.075	3.4	31.2	828	87.8	235	(2)
S7	526	31.4	3.2	0.10	3.5	28	813	86.9	227	(2)
S8	523	31.3	37.65	1.2	1	1	1	75.0	/	(3)
S9	523.5	31.3	6.3	0.20	3.5	22.4	713	54.1	198	(2)
S10	522.5	31.2	15.6	0.50	3.4	20.1	605	83.2	175	(2)
	H			1						

REMARKS :

(1) after cold loading before the heating the middle-span section is already fully plastified.

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(2) after cold loading before the heating the middle-span section is partially plastified or still elastic

(3) only cold loading - unloadings.

Figure 4.3

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Figure 4.4



Figure 4.5



θ (°C)	20	100	200	300	400	500	600	700	800	900	1000	1100	1200
^β _{Ρ,θ} β _{Ρ,20} • _C	1,0	1,0	0,8	0,6	0,55	0,5	0,2	0,1	0,067	0,033	0,0	0,0	0,0

Figure 4.6



0107	20	100	200	000	400	000	600	100	000	900	1000	100	1200
β _{s,θ} β _{s,20} •c	1,0	1,0	1,0	1,0	1,0	0,74	0,47	0,2	0,13	0,07	0,0	0,0	0,0

Figure 4.7



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FIGURE 4.9

KRUPP TESTS 1988 FOR ARBED



Figure 4.10


Figure 5.1





θ(°C)	20	100	200	300	400	500	600	700	800	900	1000	1 100	1200
E _{0,θ} E _{0,20} • _C	1,0	0,96	0,94	0,88	0,79	0,68	0,37	0,20	0,09	0,06	0,04	0,02	0,0

Figure 5.3



θ(°C)	20	100	200	300	400	500	600	700	800	900	1000	1100	1200
<u>^σу,θ</u> ^σ у,20°С	1,0	1,0	0,80	0,71	0,63	0,54	0,34	0,15	0,07	0,05	0,03	0,015	0,0

Figure 5.4



Figure 5.5



θ(°C)	20	100	200	300	400	500	600	700	800	900	1000	1100	1200
[∞] t,θ [∞] y,20 °C	1,29	1,24	1,29	1,33	1,10	0,75	0,44	0,20	0,13	0,09	0,06	0,03	0,00

Figure 5.6







KRUPP TRANSIENT STATE BEAM TEST PARAMETERS (S1 TO S10) COMPARED TO QL-LAW SIMULATIONS

STEEL QUALITY : Fe E 460

ļī	EST	Sig y	Sig t	St/Sy	Fpcold	F	F/Fpcold	Tm	Tinit	(Tmeas.)max	(Tsim.)max	Dmeassim.	(Wmeas.)max	(Wsim.)max	(Esim.)max	
		[N/mm2]	[N/mm2]		[kN]	[kN]	 	[[ºC/min]	[[° C]	[°C]	[[[⁰ C]	[[X]	(mm)	 [mm]	 [X]	
											••••••				•••••	
	s1	502.00	653.00	1.30	30.00	30.00	<u>1.000</u>	3.60	22.10	461.00	428.00	-7.2	85.80	64.40	4.75	(1)
	s3	507.00	655.00	1.29	30.30	25.80	0.850	3.50	21.70	497.00	467.00	-6.0	53.20	44.80	2.96	(2)
	s2	504.50	654.00	 1.30	30.20	22.70	0.750	3.40	22.50	525.00	497.00	-5.3	53.90	39.30	2.79	(2)
	s4	516.00	658.00	1.28	30.80	18.50	0.600	3.50	21.40	566.00	543.00	-4.1	53.10	39.70	2.86	 (2)
	s10	522.50	650.00	1.24	31.20	15.60	0.500	3.40	20.10	605.00	574.00	-5.1	83.20	73.80	7.12	
	\$5	513.00	658.00	1.28	30.70	12.30	0.400	3,50	21.00	651.00	613.00	-5.8	87.50	89.90	7.96	(2)
	\$9 I	523.50	648.00	1.24	31.30	6.30	0.200	3.50	22.40	713.00	693.00	-2.8	54.10	108.60	10.14	(2)
	s7	526.00	652.00	1.24	31.40	3.20	0.100	3.50	28.00	813.00	813.00	0.0	86.90	135.90	7.78	(2)
	S6	529.00	655.00	1.24	31.60	2.40	<u>0.075</u>	3.40	31.20	828.00	827.00	-0.1	87.80	41.10	3.13	(2)
	ا 	ا 	ا 	 	 	 	1	l						 		
: 	58 	523.00 	649.00 	1.24	31.30 	37.65 	<u>, 1.200</u>	(Fmeas.)ma	1x=37.65 kN	; (Fsim.)max	= 36 kN ; Dn	nax = -4.3 %	75.00	74.50	7.13	(3)

• •

Remarks :

(1) after cold loading before the heating the middle-span section is already fully plastified

(2) after cold loading before the heating the middle-span section is already partially plastified or still elastic (3) only cold loadings - unloadings

FIGURE 5.10

KRUPP TRANSIENT STATE BEAM TEST PARAMETERS (S11,S12 ; V1 TO V7) COMPARED TO QL-LAW SIMULATIONS

STEEL QUALITY	TEST 	ProFiles	Sig y [N/mm2]	Sig t [N/mm2]	St/Sy 	/ Wplx [cm3]	Fpcold (1) [kN]	F [kn]	F/Fpcold 	(Tmeas.)max (7) [ºC]	(Tsim.)max [°C]	Dmeassim. [%]	(Wmeas.)max [mm]	(Wsim.)max [mm]	(Esim.)max [%]	(Fmeas.)max [kN]	(F/Fpcold)max measured	
Fe E 460	S11 S12	TooLED PROFILES	496.00 489.50	758.00 744.50	1.53 1.52	/ 17.470 17.750	30.22 30.30	31.73 31.82	<u>1.05</u> 1.05	120.10 422.80	460.0 440.0	(8) 4.07	21.34 54.70	135.20 98.00	12.43 7.41	32.80 35.06	1.085 1.157	(2),(6) (2),(5)
	V1 V2		 321.80 315.00	 508.30 505.30	 1.58 1.60	 24.150 25.270	 27.10 27.76	 29.81 30.54	 <u>1.10</u> 1.10	217.80	440.0 440.0	(8)	 37.90 58.90	84.00 63.70	8.67 5.68	 32.41 33.36	1.196 1.202	(2),(6) (2),(5)
360	V3 V4	TPE 80	 310.00 308.00	 501.50 503.30	 1.62 1.63	 24.780 25.100	26.79 26.96	 22.77 . 16.18	0.85	530.00 600.00	530.0 595.0	0.00 -0. 83	95.30 90.66	57.10 90.10	5.27 7.75			(3) (3)
 	V5 V6	H 	, 310.00 312.00	505.30 506.00	1.63 1.62	24.940 25.100	26.96	13.48	0.50	630.00 921.00	625.0 900.0	-0.79 -2.28	78.60 78.40	105.20 242.10	8.22 14.66			(3) (3)
	V7	- 	311.80 	504.30 	1.62 	25.100 	27.29 	33. 02	1.21 	(Fmeas.)max= (Fsim.)max=3	33.02 kN; 2.3 kN;Dmax	<=-2.2%	67.10	44.90	4.69	33.02	1.210	(4)

-

SUMMARY OF THE SIX FIRE TESTS

Nº	WIDE FLANGE SECTION	MAX THICKNESS t (mm)	SECTION FACTOR U/A (m ⁻¹)	STEEL GRADE ACTUAL <i>O</i> y (N/mm ²)	BUCKLING AXIS	Laboratorium	Column Length (m)	ACTUAL SLENDER- NESS RATIO Ā	ECCEN- TRICITY e (cm)	e/d(1)	TEST LOAD N (KN)	ULTIMATE LOAD Nut (kN) (2)	N utt	RESISTANCE TIME (minutes)
1	HD 210x210 x198	45	54	Fe 510 364	WEAK	BRAUN- SCHWEIG 2.09.1988	5,70	1,303	1,0	0,045	1100	3213	0,342	38'
2	HD 310x310 x500	75	33	Fe 510 298	STRONG	GAND 6.10.1988	4,14	0,305	8,5	0,2	2000	11716	0,171	58'
3	HD 310x310 x500	75	33	Fe 510 298	WEAK	BRAUN- SCHWEIG 7.09.1988	5,70	0,777	3,4	0,1	1650	9544	0,173	50'
4	HD 400x400 x1086	125	20	Fe 510 371	WEAK	GAND 13.10.1988	4,14	0,466	22,7	0,5	4000	13824	0,289	68'
5	W 360×410 x314	39,6	58,4	Fe 510 401	WEAK	GAND 29.9.1988	4,14	0,557	12,0	0,3	1800	6086	0,296	37'
6	W 360x410 x314	39,6	58,4	Fe E 460 496	WEAK	GAND 22.09.1988	4,14	0,620	12,0	0,3	1800	7326	0,246	39'

(1) d = width of the section for buckling around the weak axis.

d = hight of the section for buckling around the strong axis.

(2) Nult is the ultimate load in cold situation for the given eccentricity and has been calculated here according to the german DIN 18 800 (second order plastic theory) with the theoretical section sizes and the actual yield strengths.

FIGURE 6.1

ACTUAL SIZES OF SECTIONS AND MECHANICAL PROPERTIES

			Measured	actual sizes				
	WIDE	Hight	Width	Flange-	Web-	Initial	Yield	Tensile
TEST	FLANGE	h	b	thickness e	thickness a	Imperfection	Strength $\sigma_{ m y}$	Strength $\sigma_{ m t}$
N ^O	SECTION	(mm)	(mm)	(mm)	(mm)	(mm)	(N/mm ²)	(N/mm ²)
1	HD 210 x 210 x 198	270,3	222,4	44,6	27,5	0	364	523
2	HD 310 x 310 x 500	427	335	73	42	0	298	490
3	HD 310 x 310 x 500	428	335	73,6	45	0	298	490
4	HD 400 x 400 x 1086	568	446	125	71	o	371	574
5	W 360 x 410 x 314	402,8	401,7	39	20,8	1	401	536
6	W 360 x 410 x 314	401	400	38,7	23	- 4	496	676

-

Figure 6.2



PARAMETERS FOR HEAT TRANSFER AND CALCULATED FIRE RESISTANCE TIMES (QL-8LAW)

Furnace	Test	Convection	Resu	ltant em	issivity	Fire Resist	ance Times
of	N°	factor ∝ (W/m²K)	٤*	€ ₩ eb	\mathcal{E}_{FI}^*	in mi TEST	nutes CEFICOSS
Braun-	1	25	0,70	0,42	0,26	38'	37'
schweið	3	25	0,70	0,42	0,26	50'	48'
	2	18	0,45	0,26	0,16	58'	58'
Gand	4	18	0,45	0,26	0,16	68'	72'
Ganu	5	18	0,45	0,26	0,16	37'	38'
	6	18	0,45	0,26	0,16	39'	40'

Emissivity Coefficient affected by the Shadow Effect



Figure 6.4



FORM OF BENDING MOMENT DISTRIBUTION



FIGURE 7.1







INFLUENCE OF RESIDUAL STRESSES

ON FIRE RESISTANCE TIMES

Max. residual stresses : $\widetilde{O}_{R} = 0.3 \widetilde{O}_{y}$

	FIRE F	RESISTANCE TIMES	(MINUTES)
TEST N ⁰	TEST	SIMULATION WITHOUT RESIDUAL STRESSES	SIMULATION with RESIDUAL STRESSES 0,3 0 y
2	60'	58'	57,75'
3	50'	48,25'	47,25'
1	38'	37'	35'

FIGURE 7.4





DISTRIBUTION OF RESIDUAL STRESSES



FIGURE 7.5



EXAMPLE OF INTERPOLATION

SECTION : HD 260x260x329

Fire resistance class : F 30

		TABLE	S F 30	N [kN]	t /
Bending axis	e/h (or e/b)	N [kN] for L = 4 m	N [kN] for L = 6 m	interpolated for L = 5 m	(*)
Strong	0.5	3848	2996	3422	30'
Weak	0.5	1961	1487	1724	29.74'

(*) Time in minutes given by making a simulation with CEFICOSS for the interpolated load found for L = 5.00 m

Figure 8.2



HD 260x260x329

Bi-triangular moment distribution.

				CALCULATE	D BY CEFICOSS	APPRC	ACHED MET	THODS
						FOR BITRIA	NGULAR DIS	TRIBUTION
BENDING	SLENDERNESS	BUCKLING	ECCENTRICITY	N for diagr.	N for diagr.	N bitr.	N bitr.	N'bitr.
AXIS	RATIO	LENGTH [m]	RATIO e/H or e/B	UNIFORM	BITRIANGULAR	EC3 method	β-ψ method	β-ψ method
MAJOR	0.3854	4	0.5	3848	5400	5399.05	4543.80	4616
MAJOR	0.3854	4	· 1	2586	3660	4238.24	3020.95	3141
MAJOR	0.5781	6 .	0.5	2996	5000	4439.42	3931.31	3817
MAJOR	0.5781	6	1	2084	3390	3525.08	2649.45	2649
MINOR	0.6978	4	0.5	1961	3375	3061.83	2805.45	2859
MINOR	0.6978	4	1	1199	2167	2191,85	1654.46	1763
MINOR	1.047	6	0.5	1487	2636	2191.43	2279.65	2261
MINOR	1.047	6	1	980	1800	1679.3	1490.76	1487
							1	

Figure 8.4

GENERAL FORM OF INTERACTION FORMULAS FOR COLD SERVICE

$$\frac{N}{\chi \cdot N_{pl}} + \beta \cdot \frac{M}{M_{pl}} = 1$$
or $1 - \Delta$

$$\frac{M}{M_{pl}} = \frac{N \cdot e}{M_{pl}}$$
 and we can write : $M_{pl} = \alpha \cdot N_c$

$$\frac{N}{\chi \cdot N_{pl}} = \frac{N}{N_c}$$
 where N_c is the pure axial load (e = 0)
$$\Rightarrow \left(\frac{N}{N_c}\right) + \frac{\beta}{\alpha} \cdot \left(\frac{N}{N_c}\right) \cdot e = 1$$
 and for $K = \frac{1}{\alpha}$

$$\Rightarrow \frac{N}{N_c} = \frac{1}{1 + \beta \cdot K \cdot e}$$

(Eventual correction $:1-\Delta$ in place of 1)

Diagram uniform : $\beta = 1.1 \Rightarrow K_u = \frac{1 - \frac{N_u}{N_c}}{\frac{N_u}{N_c} \cdot 1.1 \cdot e} = \frac{\frac{N_c}{N_u} - 1}{1.1 \cdot e}$ (K_u can be calculated for each eccentricity)

Other distribution $:\beta = 0.66 + 0.44\psi > 0.44\psi$

and particularly for a bi-triangular distribution :

$$\psi = -1.00 \Rightarrow \beta = 0.44$$

$$\Rightarrow N_{bitr} = \frac{N_c}{1 + 0.44 \cdot K_u \cdot e} = \frac{N_u}{0.60 \frac{N_u}{N_c} + 0.4}$$

Figure 8.5





 $-1.0 < \Psi < 1.0$: linear interpolation :

$$\frac{\overline{AB}}{\overline{AC}} = \frac{\Psi + 1.0}{2.0}$$

$$\frac{\beta_{\rm B} - \beta_{\rm A}}{1.0 - \beta_{\rm A}} = \frac{\Psi + 1.0}{2.0}$$

$$\Longrightarrow \beta_{\rm B} = \beta_{\rm A} + (1.0 - \beta_{\rm A}) \frac{(\Psi + 1.0)}{2.0}$$

Figure 8.6



Figure 8.7



Figure 8.8

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PART II

DIAGRAMS

and TABLES

SUMMARY OF CALCULATED SECTIONS.

Steel	PROFILE	Flange thickness [mm]	Section factor U/A [m ⁻¹]	h/b	Reduced ^{oy} [kN/cm ²]	Max.Residual stress ores
	HE 550 M	40	64	1.87	34.5	0.3 ₀ y
	HD 210x210x198	45	54	1.21	33.5	0.3 oy
Fe	HD 260x260x219 HD 260x260x329	41 60	58 41	1.14 1.23	33.5 33.5	0.5 су 0.3 су
510	HD 310x310x283 HD 310x310x375 HD 310x310x500	44 57 75	54 42 33	1.13 1.18 1.26	33.5 33.5 32.5	0.5 σy 0.5 σy 0.3 σy
	HD 400x400x314 HD 400x400x678 HD 400x400x1086	40 82 125	58 30 20	1.0 1.13 1.25	34.5 31.5 30.5	0.5 σγ 0.5 σγ 0.3 σγ
FeE	HE 550 M	40	64	1.87	45	0.3 _о у
460	HD 400x400x314	40	58	1.0	45	0.5 _ơ y

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DIAGRAMS AND TABLES

FOR BENDING ABOUT

THE MAJOR AXIS


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FOR BENDING ABOUT

THE MAJOR AXIS

HE 550 M (Strong axis)	Sigma yield = 345 N/mm2	; $U/A = 64$; $t = 40$ mm
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I I F0 F30 F40 I Landback F0 F30 F40 I Landback F1 KM0 MCEG3/MpcI MCEG5/MpCI MCEF7/MpCI MCEF7/MpCI MCF0,EC3/MV/KF0,EC5/MV/KF0,E5/MV/KF0,EF/MV/KF0,E5/MV/KF0,E5/MV/K				 .	·		• • • • • • • •											
L Lemends Ber [ef] Kupi [Image: Section of the sect							FO		1		F30					F60		
Image: Construct on a structure of the construction of the cons		L amb da Bar	e/h	Npl				••••••		• • • • • • •	·····	·····					•••••	•••••
1 1	(m)			(KN)	N(EC3)	N(EC3)/Npl	IN(CEF)	N(CEF)/Npl	N(F30)	N/NC	N/N(FO,EC3)	N/N(FO,CEF)	N/Npl	N(F60)	N/Nc	N/N(FO,EC3)	N/N(FO,CEF)	N/Npl
2.00 0.1092 0.00 12213 11330 0.07 11996 0.98 1.400 0.34 0.33 0.26 1780 0.74 0.08 0.09 0.08 0.08 0.09 0.09 0.08 0.08 0.09 0.08 0.07 0.10 1323 0.10 1345 0.31 0.64 171 1.04 0.09 0.08 0.09 0.08 0.00 0.09 0.08 0.00 0.03 0.30 0.31 0.40 0.171 0.16 0.08 0.07 0.31 0.40 0.30 0.03 0.04 0.09 0.08					(KN)		(KN)		(KN)	ł	1	1		(kN)	ļ			ļ
12.00 0.0102 0.0102 11213 11005 0.79 11996 0.79 10002 10.02 0.0102 10.00 11213 10005 0.020 0.0002 0.0102 10.00 12213 1966 0.026 0.000 0.034 0.033 0.221 634 0.060 0.068 0.006 0.068 0.006 0.068 0.006 0.006 0.000 10002 1.001 12213 1355 0.455 5961 0.49 1956 0.484 0.335 0.33 0.101 1317 0.009 0.008 0.008 0.009 0.001	12 00	0 1002		12217	111970	0.07	111006	0.09		11.00	0.7/		10.77	1 10/0	11 00			
$ \begin{bmatrix} 1.00 \\ 0.1072 \\ 0.00 \\ 0.1092 \\ 0.20 \\ 1.0092 \\ 0.20 \\ 1.0092 \\ 0.20 \\ 1.0092 \\ 1.00 \\ 1.201 \\ 1.213 \\ 1.$	12.00	0.1072	10.00	112213		1 0.70	1 0052	0.70	1 7255	10.00	1 0.34 1 0.34	0.34	10.33	1 790	10.74		1 0.09	10.00
$ \begin{bmatrix} 1 & 0 & 0.102 \\ 2.00 & 0.1092 \\ 1.00 & 12213 \\ 1.00 & 1.00 & 12213 \\ 1.00 & 1.00 & 1.00 & 1.00 \\ 1.00 & 0.2183 & 1.00 & 12213 \\ 1.00 & 1.2213 & 1.00 & 1.00 & 1.00 \\ 1.00 & 0.2183 & 1.00 & 12213 \\ 1.00 & 1.2213 & 1.00 & 1.00 & 1.00 & 1.00 \\ 1.00 & 0.2183 & 1.00 & 12213 \\ 1.00 & 1.2213 & 1.00 & 1.00 & 1.00 & 1.00 & 1.00 \\ 1.00 & 0.2183 & 1.00 & 1.2213 & 1.00 & 0.00 & 1.00 & 1.00 & 1.00 & 0.00 \\ 1.00 & 0.2183 & 1.00 & 1.2213 & 1.00 & 0.00 & 1.00 & 1.00 & 1.00 & 1.00 & 0.00 & 0.00 \\ 1.00 & 0.2183 & 1.00 & 1.2213 & 1.00 & 0.00 & 1.00 & 1.00 & 1.00 & 0.00 & 0.00 & 0.00 \\ 1.00 & 0.2133 & 1.00 & 1.00 & 1.00 & 1.00 & 1.00 & 1.00 & 1.00 & 0.00 & $	12.00	0.1092	0.10	112213	7556	0.73	1 7064	0.01	1 2605	10.00	1 0.34 1 0.34	0.33 0.33	10.20	1 / 00 1 / 43/	10.14	0.08	1 0.08	10.00
	12.00	0.1092	0.50	112213	1 5553	0.45	5981	0.49	1 1956	10.48	0.34	0.33	10 16	480	10.46	1 0.00	1 0.08	10.03
2.00 0.1092 2.00 12213 2147 0.18 2496 0.20 778 0.19 0.36 0.31 0.06 177 0.16 0.08 0.07 0.01 2.00 0.1092 2.00 12213 1182 0.10 1353 0.11 413 0.10 0.35 0.30 0.03 104 0.10 0.09 0.08 0.00 4.00 0.2183 0.00 12213 11425 0.94 11761 0.96 3353 1.00 0.29 0.29 0.27 624 1.00 0.06 0.06 0.06 0.06 0.06 0.06 0.06 0.06 0.06 0.06 0.06 0.06 0.06 0.06 0.06 0.07	12.00	0.1092	1.00	12213	3630	0.30	4042	0.33	1 1286	10.32	0.35	0.32	10.10	1 317	10.30		1 0.00	10.03
2.00 0.1092 4.00 12213 1182 0.10 1363 0.11 413 0.10 0.35 0.30 0.03 104 0.10 0.09 0.08 0.00 4.00 0.2183 0.00 12213 11455 0.94 11761 0.96 1353 1.00 0.229 0.22 0.22 624 1.00 0.06 0.06 0.67 0.22183 10.25 12213 1541 0.60 7474 0.61 1238 0.67 0.33 0.32 0.118 1350 0.33 0.32 0.109 246 0.37 0.07 0.07 0.07 0.07 0.07 0.07 0.07 0.07 0.07 0.07 0.07 0.07 0.07	2.00	0.1092	2.00	12213	2147	0.18	2496	0.20	768	10.19	0.36	0.31	10.06	1 171	10.16	0.08		10.01
4.00 0.2183 0.00 12213 11455 0.94 11761 0.96 3353 1.00 0.29 0.27 624 1.00 0.05 0.05 0.05 0.06 0.07	2.00	0.1092	4.00	12213	1182	0.10	1363	0.11	413	10.10	0.35	0.30	10.03	1 104	10.10	0.09	0.08	10.00
4.00 0.2183 0.00 12213 11455 0.94 11761 0.96 3353 1.00 0.29 0.27 624 1.00 0.05 0.05 0.05 4.00 0.2183 0.10 12213 927 0.76 9510 0.78 2711 0.81 0.29 0.22 0.22 553 0.89 0.66 0.06 0.06 4.00 0.2183 0.25 12213 5413 0.44 554 0.451 2248 0.67 0.31 0.30 0.18 472 0.76 9.07 0.07 <td>i i</td> <td></td> <td></td> <td>i i</td> <td>1</td> <td></td> <td> </td> <td> </td> <td></td> <td></td> <td>1</td> <td> </td> <td>1</td> <td></td> <td>1</td> <td></td> <td> </td> <td> </td>	i i			i i	1						1		1		1			
4.00 0.2183 0.10 12213 9327 0.76 9510 0.78 2711 0.81 0.29 0.22 553 0.89 0.06 0.07 <td>4.00</td> <td>0.2183</td> <td>0.00</td> <td> 12213 </td> <td>11455</td> <td>0.94</td> <td> 11761</td> <td>0.96</td> <td>3353</td> <td>1.00</td> <td>0.29</td> <td>0.29</td> <td>0.27</td> <td>624</td> <td>1.00</td> <td>0.05</td> <td>0.05</td> <td>10.05</td>	4.00	0.2183	0.00	12213	11455	0.94	11761	0.96	3353	1.00	0.29	0.29	0.27	624	1.00	0.05	0.05	10.05
4.00 0.2183 [0.25 12213 7319 0.60 7474 0.61 2238 0.67 0.31 0.30 0.18 472 0.76 0.06 0.06 0.03 4.00 0.2183 0.50 12213 5413 0.44 5544 0.45 1778 0.53 0.33 0.32 0.14 369 0.59 0.07 0.08 0.80 0.33 0.33 0.33 0.33 0.03 0.33 0.03 0.33 0.03 0.33 0.33 0.33 0.33 0.33 0.33 0.33 0.33 0.33 0.33 0.33 0.33	4.00	0.2183	0.10	12213	9327	0.76	9510	0.78	2711	0.81	0.29	0.29	0.22	553	0.89	0.06	0.06	0.04
4.00 0.2183 0.50 12213 5413 0.44 5544 0.45 1778 0.53 0.33 0.32 0.14 369 0.59 0.07 0.07 0.03 4.00 0.2183 1.00 12213 3566 0.29 3689 0.30 1183 0.35 0.33 0.32 0.09 246 0.39 0.07 0.07 0.07 0.07 0.01 4.00 0.2183 2.00 12213 1174 0.10 1193 0.18 713 0.21 0.33 0.33 0.05 153 0.25 0.07 0.07 0.07 0.07 0.01 4.00 0.2183 4.00 12213 1174 0.10 193 0.11 101 101 101 101 101 101 101 101 101 103 1133 0.75 1223 1035 0.73 1035 105 106 101 101 101 103 101 101	4.00	0.2183	0.25	12213	7319	0.60	7474	0.61	2238		0.31	0.30	0.18	472	0.76	0.06	0.06	0.03
4.00 0.2183 1.00 12213 3566 0.29 3689 0.30 1183 0.35 0.33 0.32 0.09 246 0.39 0.07 0.07 0.07 0.07 4.00 0.2183 2.00 12213 2124 0.17 2173 0.18 713 0.21 0.33 0.05 153 0.15 0.08 0.007 0.07 0.01 4.00 0.2183 4.00 12213 11035 0.90 1583 0.95 3265 1.00 0.33 0.33 0.15 0.08 0.08 0.01 4.00 0.2183 4.00 12213 11035 0.90 1583 0.95 3265 0.00 0.328 0.21 4.99 0.81 0.06 0.05 0.04 4.00 0.3275 0.10 12213 1873 0.98 2622 0.80 0.25 0.27 2.91 0.47 0.66 0.05 0.47 4.00 0.3275 0.21 2	4.00	0.2183	0.50	12213	5413	0.44	5544	0.45	1778	0.53	0.33	0.32	0.14	369	0.59	0.07	0.07	0.03
4.00 0.2183 2.00 12213 217 2173 0.18 713 0.21 0.34 0.33 0.05 153 0.25 0.07 0.07 0.01 4.00 0.2183 4.00 12213 1174 0.10 193 0.10 391 0.12 0.33 0.03 91 0.15 0.08 0.08 0.00 6.00 0.3275 0.00 12213 11035 0.90 1583 0.95 2265 0.00 0.28 0.26 617 0.06 0.05 0.05 6.00 0.3275 0.10 12213 8976 0.73 9302 0.67 2262 0.80 0.29 0.28 0.21 499 0.81 0.06 0.05 0.03 6.00 0.3275 0.50 12213 347 0.28 3529 0.29 0.21 0.31 0.31 0.03 0.21 0.33 0.01 0.23<	4.00	0.2183	1.00	12213	3566	0.29	3689	0.30	1183	0.35	0.33	0.32	0.09	246	0.39	0.07	0.07	0.02
[4.00] 0.2183 [4.00] 12213 1174 0.10 193 0.10 391 [0.12] 0.33 [0.03] 91 [0.15] 0.08 [0.00] [6.00] 0.3275 [0.00] 12213 11035 0.90 [11583] 0.95 3265 [1.00] 0.30 0.28 [0.26] [617] 1.00 0.06 0.055 [0.05] [6.00] 0.3275 [0.10] 1213 8976 0.73 9302 0.76 [2262 [0.80] 0.28 [0.26] [617] 1.00 0.06 0.055 [0.05] [6.00] 0.3275 [0.25] 12213 [5259] 0.43 [555] [0.48] 0.30 0.29 [0.17] [391] 0.46 0.06 0.06 [0.00] [0.02] [6.00] 0.3275 [0.00] 12213 [529] 0.43 [555] [0.48] 0.30 0.29 [0.17] [291] 0.46 0.30 0.29 [0.17] [291] 0.66 0.06 [0.00] [6.00] 0.3275 [1.00] 12213 [0.37]	4.00	0.2183	2.00	12213	2124	0.17	2173	0.18	713	0.21	0.34	0.33	0.05	153	0.25	0.07	0.07	0.01
6.00 0.3275 0.00 12213 11035 0.90 11583 0.95 3265 1.00 0.30 0.28 0.26 617 1.00 0.066 0.05 0.05 6.00 0.3275 0.10 12213 8976 0.73 9302 0.76 2622 0.80 0.29 0.28 0.21 499 0.81 0.06 0.05 0.04 6.00 0.3275 0.25 12213 5239 0.43 5291 0.43 1555 0.48 0.30 0.29 0.29 0.17 91 0.64 0.06 0.05 0.06 6.00 0.3275 10.50 12213 1539 0.43 5291 0.43 1555 0.48 0.30 0.29 0.17 0.66 0.06 0.06 0.02 6.00 0.3275 1.00 12213 13477 0.28 3529 0.29 1048 0.33 0.31 0.31 0.03 144 0.23 0.07 0.07 0.07 0.07 0.07 0.07 0.07 0.07 0.07	4.00	0.2183	4.00	12213	1174	0.10	1193	0.10	391	0.12	0.33	0.33	0.03	91	0.15	0.08	0.08	0.00
6.00 0.3275 0.00 12213 11035 0.90 11583 0.95 3265 1.00 0.30 0.28 0.26 617 1.00 0.06 0.05 0.05 6.00 0.3275 0.10 12213 8976 0.73 9302 0.76 2622 0.80 0.29 0.28 0.21 499 0.81 0.06 0.05 0.03 6.00 0.3275 0.25 12213 7063 0.58 7202 0.59 2080 0.64 0.29 0.29 0.17 391 0.63 0.06 0.05 0.03 6.00 0.3275 0.50 12213 3477 0.28 3529 0.29 1068 0.33 0.31 0.30 0.08 201 0.33 0.06					1								1	1				
6.00 0.3275 0.10 12213 8976 0.73 9302 0.76 2622 0.80 0.29 0.28 0.21 499 0.81 0.06 0.05 0.04 6.00 0.3275 10.25 12213 7063 0.58 7202 0.59 2080 0.64 0.29 0.29 0.17 391 0.63 0.06 0.05 0.03 6.00 0.3275 10.50 12213 5239 0.43 5291 0.43 1555 0.48 0.30 0.29 0.12 291 0.47 0.06 0.06 0.06 0.02 6.00 0.3275 1.00 12213 3477 0.28 3529 0.29 1068 0.33 0.31 0.30 0.08 201 0.33 0.06 0.06 0.06 0.06 0.06 0.06 0.07 <td>6.00</td> <td>0.3275</td> <td>0.00</td> <td> 12213 </td> <td>11035</td> <td>0.90</td> <td>11583</td> <td>0.95</td> <td>3265</td> <td>1.00</td> <td> 0.30</td> <td>0.28</td> <td>0.26</td> <td>617</td> <td>1.00</td> <td>0.06</td> <td>0.05</td> <td>0.05</td>	6.00	0.3275	0.00	12213	11035	0.90	11583	0.95	3265	1.00	0.30	0.28	0.26	617	1.00	0.06	0.05	0.05
6.00 0.3275 0.25 12213 7063 0.58 7202 0.59 2080 0.64 0.29 0.29 0.17 391 0.63 0.06 0.05 0.03 6.00 0.3275 0.50 12213 5239 0.43 5291 0.43 1555 0.48 0.30 0.29 0.12 291 0.47 0.06 0.07	6.00	0.3275	0.10	12213	8976	0.73	9302	0.76	2622	0.80	0.29	0.28	0.21	499	0.81	0.06	0.05	0.04
6.00 0.3275 0.50 12213 5239 0.43 5291 0.43 1555 0.48 0.30 0.29 0.12 291 0.47 0.06 0.02 6.00 0.3275 1.00 12213 3477 0.28 3529 0.29 1068 0.33 0.31 0.30 0.08 201 0.33 0.06 0.06 0.06 0.07 6.00 0.3275 2.00 12213 2089 0.17 2093 0.17 657 0.20 0.31 0.31 0.05 144 0.23 0.07 0.07 0.07 0.07 0.07 0.07 0.07 0.00 12213 1163 0.10 1116 0.09 377 0.12 0.32 0.34 0.03 72 0.12 0.06 0.06 0.00 6.00 0.4367 10.00 12213 110578 0.87 11353 0.93 3122 1.00 0.30 0.27 0.25 604 1.00 0.06 0.05 0.04 8.00 0.4367 0.10 12213 8955 <td>6.00</td> <td>0.3275</td> <td>0.25</td> <td>12213</td> <td>7063</td> <td>0.58</td> <td>7202</td> <td>0.59</td> <td>2080</td> <td>0.64</td> <td>0.29</td> <td>0.29</td> <td>0.17</td> <td>391</td> <td>0.63</td> <td>0.06</td> <td>0.05</td> <td>0.03</td>	6.00	0.3275	0.25	12213	7063	0.58	7202	0.59	2080	0.64	0.29	0.29	0.17	391	0.63	0.06	0.05	0.03
6.00 0.3275 1.00 12213 3477 0.28 3529 0.29 1068 0.33 0.31 0.30 0.08 201 0.33 0.06 0.06 0.01 6.00 0.3275 2.00 12213 2089 0.17 2093 0.17 657 0.20 0.31 0.31 0.05 144 0.23 0.07 0.07 0.07 0.07 0.01 6.00 0.3275 4.00 12213 1163 0.10 1116 0.09 377 0.12 0.32 0.34 0.03 72 0.12 0.06 0.06 0.06 0.00 6.00 0.4367 0.00 12213 10578 0.87 11353 0.93 3122 1.00 0.30 0.27 0.25 604 1.00 0.06 0.05 0.04 8.00 0.4367 0.10 12213 8562 0.70 8935 0.73 2519 0.81 0.29 0.28 0.20 482 0.80 0.06 0.05 0.03 8.00 0.4367 0.25	6.00	0.3275	0.50	12213	5239	0.43	5291	0.43	1555	0.48	0.30	0.29	0.12	291	0.47	0.06	0.06	0.02
18.00 0.3275 12.00 12213 2089 0.17 2093 0.17 657 0.20 0.31 0.31 0.05 144 0.23 0.07 0.06 0.06 0.06 0.06 0.06 0.06 0.06 0.06 0.06 0.06 0.06 0.06 0.05 0.04 0.07 0.32 0.34 0.31 0.27 0.25 0.44 1.00 0.06 0.05 0.04 8.00 0.4367 0.10 12213 8562 0.70 8935 0.73 2519 0.81 0.29 0.28 0.20 482 0.80 0.06 0.05 0.03 8.00 0.4367 0.55	16.00	0.3275	1.00	12213	3477	0.28	3529	0.29	1068	0.33	0.31	0.30	0.08	201	0.33	0.06	0.06	0.01
8.00 0.3275 4.00 12213 1163 0.10 1116 0.09 377 0.12 0.32 0.34 0.03 72 0.12 0.06 0.05 0.04 8.00 0.4367 0.10 12213 8562 0.70 8935 0.73 2519 0.81 0.29 0.28 0.20 482 0.80 0.06 0.05 0.03 8.00 0.4367 0.25 12213 6750 0.55 6918 0.57 1998 0.64 0.30 0.29 0.12 279 0.46 0.06 0.05 0.03 0.80 0.4367 0.50	10.00	0.3275	2.00	12213	2089	0.17	2093	0.17	657	0.20	0.31	0.31	0.05	144	0.23	0.07	0.07	10.01
8.00 0.4367 0.00 12213 10578 0.87 11353 0.93 3122 1.00 0.30 0.27 0.25 604 1.00 0.06 0.05 0.04 8.00 0.4367 0.10 12213 8562 0.70 8935 0.73 2519 0.81 0.29 0.28 0.20 482 0.80 0.06 0.05 0.03 8.00 0.4367 0.25 12213 6750 0.55 6918 0.57 1998 0.64 0.30 0.29 0.16 376 0.62 0.06 0.05 0.03 8.00 0.4367 0.50 12213 5092 0.42 470 0.47 0.29 0.29 0.12 279 0.46 0.06 0.05 0.02 8.00 0.4367 1.00 12213 3376 0.28 996 0.29 0.30 0.29 0.12 279 0.46 0.06 0.06 0.05 0.02 <td>10.00 </td> <td>0.32/5</td> <td>4.00</td> <td>12213</td> <td> 1165</td> <td> 0.10</td> <td>1116</td> <td>0.09</td> <td>377</td> <td>0.12</td> <td>0.32</td> <td> 0.34</td> <td>0.03</td> <td>72</td> <td>0.12</td> <td>0.06</td> <td>0.06</td> <td>0.00</td>	10.00	0.32/5	4.00	12213	1165	0.10	1116	0.09	377	0.12	0.32	0.34	0.03	72	0.12	0.06	0.06	0.00
18.00 0.4367 0.00 12213 10578 0.87 11353 0.95 13122 1.00 0.30 0.27 0.25 604 1.00 0.06 0.05 0.04 [8.00 0.4367 [0.10 12213 8562 0.70 8935 0.73 2519 0.81 0.29 0.28 0.20 482 0.80 0.06 0.05 0.03 [8.00 0.4367 [0.25 12213 6750 0.55 6918 0.57 1998 0.64 0.30 0.29 0.16 376 0.62 0.06 0.05 0.03 [8.00 0.4367 [0.50 12213 5042 0.41 5092 0.42 1470 0.47 0.29 0.29 0.12 279 0.46 0.06 0.05 0.03 [8.00 0.4367 1.00 12213 3376 0.28 3390 0.28 996 0.32 0.30 0.29 0.12 279 0.46 0.06 0.06 0.06 0.06 0.06 0.06 0.06 0.06 0.06 <td< td=""><td></td><td>0 / 7 4 7</td><td>0.00</td><td> </td><td>1405.79</td><td></td><td> </td><td></td><td>1</td><td></td><td></td><td></td><td></td><td></td><td></td><td></td><td></td><td></td></td<>		0 / 7 4 7	0.00		1405.79				1									
18.00 0.4367 10.10 12213 1352 0.70 6353 0.73 12519 10.81 0.29 0.28 10.20 482 10.80 0.06 0.05 10.05 18.00 0.4367 10.25 12213 6750 0.55 6918 0.57 1998 10.64 0.30 0.29 10.16 376 10.62 0.06 0.05 10.03 18.00 0.4367 10.50 12213 5042 0.41 5092 0.42 1470 0.47 0.29 0.29 0.12 279 0.46 0.06 0.05 10.02 18.00 0.4367 1.00 12213 3376 0.28 3390 0.28 996 0.32 0.30 0.29 10.12 279 10.46 0.06 0.05 10.02 18.00 0.4367 1.00 12213 3376 0.28 3390 0.28 996 0.32 0.30 0.29 10.08 187 0.31 0.06 0.06 0.06 0.06 0.06 0.06 0.06 0.06 0.06		0.4307	0.00	12213	1 9543	0.8/ 0.70	11333	0.93 0.77	1 3122	11.00		0.27	10.25	004	11.00	U.06		10.04
8.00 0.4367 0.50 12213 5092 0.41 5092 0.42 1470 0.47 0.29 0.29 0.12 279 0.46 0.06 0.05 0.02 8.00 0.4367 1.00 12213 3376 0.28 3996 0.32 0.30 0.29 0.12 279 0.46 0.06 0.05 0.02 8.00 0.4367 2.00 12213 3376 0.28 3996 0.32 0.30 0.29 0.08 187 0.31 0.06 0.06 0.06 <td< td=""><td>10.00</td><td>0.4367</td><td>0.10</td><td> 12213 12213 </td><td>1 6750</td><td>U.70</td><td>1 6933 1 4018</td><td>0.73</td><td>1 1009</td><td>10.01</td><td>0.29</td><td> 0.28 0.20</td><td>10.20</td><td>482</td><td>10.80</td><td></td><td></td><td>10.03</td></td<>	10.00	0.4367	0.10	12213 12213	1 6750	U.70	1 6933 1 4018	0.73	1 1009	10.01	0.29	0.28 0.20	10.20	482	10.80			10.03
8.00 0.4367 1.00 12213 3376 0.28 396 0.32 0.30 0.29 0.08 187 0.31 0.06	18.00	0.4367	0.50	12213 12213	1 50/2	0.33	0710 5002	0.27 0.27	1770	10.04	U.3U 0.20	; U.29 0.20	10.10	015	10.02		U.US	10.03
8.00 0.4367 2.00 12213 2046 0.17 2021 0.17 606 0.19 0.30 0.30 0.04 127 0.21 0.06 0.06 0.01 8.00 0.4367 4.00 12213 1149 0.09 1357 0.11 0.31 0.32 0.02 65 0.11 0.06 0.06 0.00	18.00	0.4367	1.00	12213	3376	0.41 ∩ 29	1 3300	V.++2 0.29	004	10.47 10 72	0.27 0.30	0.29 0.20	10.12	21Y	10.40	U.UO	U.US	10.02
8.00 0.4367 4.00 12213 1149 0.09 1113 0.09 357 0.11 0.31 0.32 0.02 65 0.11 0.06 0.06 0.00	18.00	0,4367	2.00	12213	2046	0.23	2021	0.20		10.32	1 0.30 1 0.30	0.29 0.30	10.00	107	10.31	0.00 0.04	0.00 0.04	0.01
	18.00	0.4367	4.00	12213	1149	0.09		0.09	357	10.11	0.31	0.30	10.02	1 65	10.21	1 0.00 1 0.04	I 0.00	
				· · · •									10.05			l v.w		10.00



HD	210x210x198	(Strong	axis)	Sigma	yield =	335	N/mm2	: U/A	= 54	:	t =	45	
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	 embrie: Poo					FO				F 3 0					F60	
(m)	iramooa sar	e/n	Mpl (kN)	N(EC3)	N(EC3)/Nol	IN(CEF)	IN(CEF)/Nol	 N(F30)	N/Nc	IN/N(FO.EC3)	N/N(FO.CEF)		 N(F60) N/Nc		
	i	i		(kN)		(kN)		(kN)			,,,, 		(kN)	1		
j	·	j					ii							.		
2.00	0.2435	0.00	8442	7837	0.93	8222	0.97	2930	1.00	0.37	0.36	0.34	463	1.00	0.06	0.0
2.00	0.2435	0.10	8442	6318	0.75	6584	0.78	2542	0.87	0.40	0.39	0.30	376	10.81	0.06	0.0
2.00	0.2435	0.25	8442	4907	0.58	5049	0.60	2024	0.69	0.41	0.40	0.23		0.66	0.06	0.0
2.00	0.2435	0.50	8442	3592	0.43	3705	0.44	1556	0.53	0.43	0.42	0.18		0.52	0.07	0.0
2.00	0.2435	1.00	8442	2347	0.28	2426	0.29	1050	0.36	0.45	0.43	0.12	167	0.36	0.07	0.0
2.00	0.2435	2.00	8442	1388	0.16	1426	0.17	620	0.21	0.45	0.43	0.07	108	0.23	0.08	0.0
2.00	0.2435	4.00	8442	764	0.09	770	0.09	343	0.12	0.45	0.45	0.04	67	0.14	0.09	0.0
1	1					ĺ		Ì	Ì	l	ĺ	İ	ÌÌ	i -	i	1
4.00	0.4870	0.00	8442	7144	0.85	7841	0.93	2687	1.00	0.38	0.34	0.31	439	1.00	0.06	0.0
4.00	0.4870	0.10	8442	5703	0.68	6075	0.72	2082	0.77	0.37	0.34	0.24	342	0.78	0.06	0.0
4.00	0.4870	0.25	8442	4454	0.53	4612	0.55	1610	0.60	0.36	0.35	0.19	262	10.60	0.06	0.0
4.00	0.4870	0.50	8442	3309	0.39	3362	0.40	1175	0.44	0.36	0.35	0.13	192	0.44	0.06	0.0
4 .0 0	0.4870	1.00	8442	2206	0.26	2200	0.26	784	0.29	0.36	0.36	0.09	127	0.29	0.06	0.0
4.00	0.4870	2.00	8442	1331	0.16	1323	0.16	472	0.18	0.35	0.36	0.05	76	0.17	0.06	0.0
4.00	0.4870	4.00	8442	745	0.09	728	0.09	281	0.10	0.38	0.39	0.03	42	0.10	0.06	0.0
	1					1		1	1		ł		11	1		
6.00	0.7305	0.00	8442	6190	0.73	6986	0.83	2386	1.00	0.39	0.34	0.28	395	1.00	0.06	0.0
6 .0 0	0.7305	0.10	8442	4944	0.59	5313	0.63	1821	0.76	0.37	0.34	0.21	303	0.77	0.06	0.0
6.00	0.7305	0.25	8442	3904	0.46	4042	0.48	1413	0.59	0.36	0.35	0.16	233	0.59	0.06	0.0
6.00	0.7305	0.50	8442	2954	0.35	2996	0.35	1047	0.44	0.35	0.35	0.12	172	0.44	0.06	0.0
6.00	0.7305	1.00	8442	2022	0.24	2010	0.24	702	0.29	0.35	0.35	0.08	116	0.29	0.06	0.0
6.00	0.7305	2.00	8442	1257	0.15	1233	0.15	431	0.18	0.34	0.35	0.05	70	0.18	0.06	0.0
16.00	0.7305	4.00	8442	716	0.08	692	0.08	242	0.10	0.34	0.35	0.02	39	0.10	0.06	0.0
				[]			!!!				1	ļ				l
8.00	0.9740	0.00	8442	5024	0.60	5822	0.69	1945	1.00	0.39	0.33	0.23	332	1.00	0.07	0.0
8.00	0.9740	0.10	8442	4099	0.49	4383	0.52	1530	0.79	0.37	0.35	0.18	257	0.77	0.06	0.0
8.00	0.9740	0.25	8442	3316	0.39	3408	0.40	1191	0.61	0.36	0.35	0.14	200	0.60	0.06	, 0.0
8.00	0.9740	0.50	8442	2575	0.31	2584	0.31	903	0.46	0.35	0.35	0.10	151	0.46	0.06	0.0
8.00	0.9740	11.00	8442	1824	0.22	1791	0.21	626	0.32	0.34	0.35	0.07	104	0.31	0.06	0.0
18.00	0.9740	2.00	8442	1171	0.14	1130	0.13	395	0.20	0.34	0.35	0.04	65	0.20	0.06	0.0





1 = 2 m

+

1 = 4 m

HD 200X200X219 (Strong axis) Sigma yield = 335 N/mm2; U/A = 36; t = 41	HD 260	0x260x219	(Strong	axis)	Sigma	yield =	335	N/mm2	;	U/A	=	58	;	t	=	41	m	Π
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						FO	l	1		F30			 . 		F60		
(m)	Lancoa par	[=/n 	(kN)	N(EC3)	N(EC3)/Npl	N(CEF)	N(CEF)/Npl	N(F30)	N/Nc	N/N(F0,EC3)	N/N(FO,CEF)	N/Npl	N(F60)	N/Nc	N/N(F0,EC3)	N/N(FO,CEF	
		1	İ	(kN)	ĺ	(kN)	İ	(kN)	ĺ	l	Í 1	l	(kN)	1	1		
	•														1		•
2.00	0.2072	0.00	9347	8784	0.94	9135	0.98	3075	1.00	0.35	0.34	0.32	524	1.00	0.06	0.06	j
2.00	0.2072	0.10	9347	7134	0.76	7301	0.78	2602	0.85	0.36	0.36	0.27	461	0.88	0.06	0.06	į
2.00	0.2072	0.25	9347 -	5579	0.60	5658	0.61	2161	0.70	0.39	0.38	0.23	390	0.74	0.07	0.07	1
2.00	0.2072	0.50	9347	4098	0.44	4158	0.44	1667	0.54	0.41	0.40	0.17	306	0.58	0.07	0.07	ļ
2.00	0.2072	1.00	9347	2689	0.29	2713	0.29	1118	0.36	0.42	0.41	0.11	209	0.40	0.08	0.08	5
2.00	0.2072	2.00	9347	1594	0.17	1628	0.17	671	0.22	0.42	0.41	0.07	132	0.25	0.08	0.08	1
2.00	0.2072	4.00	9347	881	0.09	917	0.10	372	0.12	0.42	0.41	0.03	79	0.15	0.09	0.09)
										1							
4.00	0.4144	0.00	9347	8155	0.87	8733	0.93	2788	1.00	0.34	0.32	0.29	481	1.00	0.06	0.06	j
4.00	0.4144	0.10	9347	6591	0.71	6873	0.74	2194	0.79	0.33	0.32	0.23	379	0.79	0.06	0.06	j
4.00	0.4144	0.25	9347	5168	0.55	5301	0.57	1693	0.61	0.33	0.32	0.18	293	0.61	0.06	0.06	į
4.00	0.4144	0.50	9347	3839	0.41	3876	0.41	1266	0.45	0.33	0.33	0.13	216	0.45	0.06	0.06	į
4.00	0.4144	1.00	9347	2558	0.27	2563	0.27	851	0.31	0.33	0.33	0.09	144	0.30	0.06	0.06	1
4.00	0.4144	2.00	9347	1544	0.17	1525	0.16	519	0.19	0.34	0.34	0.05	88	0.18	0.06	0.06	i
4.00	0.4144	4.00	9347	863	0.09	847	0.09	302	0.11	0.35	0.36	0.03	51	0.11	0.06	0.06	i
		1	1	H		I	[lł	1	1			11		1		
6.00	0.6216	0.00	9347	7362	0.79	8000	0.86	2554	1.00	0.35	0.32	0.27	441	1.00	0.06	0.06	j
6.00	0.6216	0.10	9347	5914	0.63	6198	0.66	1979	0.77	0.33	0.32	0.21	342	0.78	0.06	0.06	1
6.00	0.6216	0.25	9347	4671	0.50	4802	0.51	1533	0.60	0.33	0.32	0.16	267	0.61	0.06	0.06	1
6.00	0.6216	0.50	9347	3518	0.38	3559	0.38	1146	0.45	0.33	0.32	0.12	198	0.45	0.06	0.06	5
6.00	0.6216	1.00	9347	2391	0.26	2377	0.25	777	0.30	0.32	0.33	0.08	133	0.30	0.06	0.06	1
6.00	0.6216	2.00	9347	1476	0.16	1441	0.15	471	0.18	0.32	0.33	0.05	81	0.18	0.05	0.06	1
6.00	0.6216	4.00	9347	839	0.09	810	0.09	265	0.10	0.32	0.33	0.02	45	0.10	0.05	0.06	1
	1				1	1			1	1	l				1		
8.00	0.8288	0.00	9347	6350	0.68	6861	0.73	2190	1.00	0.34	0.32	0.23	385	1.00	0.06	0.06	1
8.00	0.8288	0.10	9347	5135	0.55	5366	0.57	1713	0.78	0.33	0.32	0.18	301	0.78	0.06	0.06	;
8.00	0.8288	0.25	9347	4113	0.44	4208	0.45	1343	0.61	0.33	0.32	0.14	237	0.61	0.06	0.06	1
8.00	0.8288	0.50	9347	3160	0.34	3165	0.34	1034	0.47	0.33	0.33	0.11	179	0.46	0.06	0.06	1
8.00	0.8288	1.00	9347	2202	0.24	2178	0.23	702	0.32	0.32	0.32	0.07	122	0.32	0.06	0.06	1
8.00	0.8288	2.00	9347	1395	0.15	1350	0.14	441	0.20	0.32	0.33	0.04	75	0.20	0.05	0.06	1
8.00	0.8288	4.00	9347	809	0.09	778	0.08	251	10.11	0.31	0.32	0.02	43	10.11	0.05	0.06	4



HD 260X260X329	(Strong axis)) Sigma yield =	335 N/mm2 ;	; U/A = 41 ;	t = 60 mm
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İ.	 			F0 						F30			ļ.		F60			ļ
	Ilamoda Bar	ie∕n I	INPL			1									IN (N/ EQ. EC7)		IN (N=1	!
1 (90)	1		(KR)		#(EC2)/#PC	I (LEF)	INCCEPJ/MPL		JIN/NC	IN/N(FU,ECS)	N/N(FU,LEF)	#/ #pi	I due	V)[N/NC	[N/W(FU,EL3)	W/W(FU,CEF)	levebr 1	1
1	1] 1		(KN)	1	(KN) 	1	(KW) 		1		 	L (KM	7	1	1	1	1
12 00	0 1027		114070			1 470/4		1110957	11 00	0.83	0 79	10 77		7 11 00	0.07	0.07	10.07	1
12.00	0.1927	10.00	14070	1110441	0.74 0.76	111041	1 0.70	11 0144	10.75	0.02 0.77	0.78	0.77	1 94	7 10 97			10.04	1
12.00	0.1727	10.75	114070	8267	1 0.70	1 8573	0.77	11 6/39	10.75	1 0.78	0.74	0.30	1 72	2 10.07		0.00	10.05	1
12.00	0.1927	10.50	114070	1 6032	0.37		1 0.51	0430 4731	10.35	0.78	0.75	10.75	1 55	3 10 56			10.03	ł
12.00	0.1727		114070	1 3024	i 0.45	4152		ון בויד ון אאמצ ון	10.28	0.78		10.33	1 77	7 10 38		1 0.09		ł
12.00	0.1927		114070	1 2310	0.20	1 2466	0.50	3000 1700	10.20	0.70	0.74	10.12	1 27	7 10 24		0.09 0.10	10.02	1
12.00	0.1927	14.00	14070	1 1268		1 1350		079			0.73		1 14	1 10.14	0.11	0.10	10.01	i
1			1						1			1	1	1	1		1	ì
 4.00	0.3854	10.00	14070	12405	0.88	113365	0.95	 7383	11.00	, I 0.60	, I 0.55	10.52	, 90	2 11.00	0.07	0.07	0.06	i
14.00	0.3854	0.10	14070	9933	0.71	10414	0.74	6147	0.83	0.62	0.59	0.43	1 70	6 0.78	0.07	0.07	0.05	i
4.00	0.3854	0.25	14070	7727	0.55	7952	0.57	 5048	0.68	0.65	0.63	0.35	54	2 0.60	0.07	0.07	0.03	i
4.00	0.3854	0.50	14070	5693	0.40	5813	0.41	 3848	0.52	0.68	0.66	0.27	39	6 0.44	0.07	0.07	0.02	i
4.00	0.3854	1.00	14070	3755	0.27	3806	0.27	 2586	0.35	0.69	0.68	0.18	j 26	2 0.29	0.07	0.07	0.01	i
4.00	0.3854	2.00	14070	2247	0.16	2250	0.16	 1575	0.21	0.70	0.70	0.11	j 15	6 0.17	0.07	0.07	0.01	İ
4.00	0.3854	4.00	14070	1245	0.09	1221	0.09	888	0.12	0.71	0.73	0.06	j 9	2 0.10	0.07	80.0	0.00	Ì
1	ĺ	ĺ	İİ	ii	ĺ	ĺ	Ì	İİ	İ	ĺ	ĺ	İ	İ	Ì	İ	1	Ì	İ
6.00	0.5781	0.00	14070	11341	0.81	12583	0.89	6540	1.00	0.58	0.52	0.46	85	6 1.00	0.08	0.07	0.06	I
6.00	0.5781	0.10	14070	9037	0.64	9679	0.69	5031	0.77	0.56	0.52	0.35	65	9 0.77	0.07	0.07	0.04	I
6.00	0.5781	0.25	14070	7070	0.50	7349	0.52	3929	0.60	0.56	0.53	0.27	50	2 0.59	0.07	0.07	0.03	I
6.00	0.5781	0.50	14070	5270	0.37	5338	0.38	2996	0.46	0.57	0.56	0.21	36	7 0.43	0.07	0.07	0.02	I
6.00	0.5781	1.00	14070	3543	0.25	3531	0.25	2084	0.32	0.59	0.59	0.14	24	2 0.28	0.07	0.07	0.01	1
6.00	0.5781	2.00	14070	2161	0.15	2140	0.15	1311	0.20	0.61	0.61	0.09	14	6 0.17	0.07	0.07	0.01	I
6.00	0.5781	4.00	14070	1214	0.09	1179	0.08	788	0.12	0.65	0.67	0.05	8	1 0.09	0.07	0.07	0.00	I
					l	l		11	1	1		1	l		1	1		ļ
8.00	0.7708	0.00	14070	9993	0.71	11412	0.81	5788	1.00	0.58	0.51	0.41	78	7 1.00	0.08	0.07	0.05	ļ
8.00	0.7708	0.10	14070	7984	0.57	8550	0.61	4444	0.77	0.56	0.52	0.31	61	5 0.78	0.08	0.07	0.04	ļ
8.00	0.7708	0.25	14070	6325	0.45	6548	0.47	3500	0.60	0.55	0.53	0.24	45	5 0.58	0.07	0.07	0.03	İ
8.00	0.7708	0.50	14070	4792	0.34	4844	0.34	2617	0.45	0.55	0.54	0.18	33	5 [0.43	0.07	0.07	0.02	ļ
18.00	0.7708	1.00	14070	3294	0.23	3267	0.23	1791	10.31	0.54	0.55	0.12	22	6 0.29	0.07	0.07	0.01	ļ
18.00		2.00	14070	2057	0.15	2005	0.14	1120	0.19	0.54	0.56	0.07	13	8 0.17	0.07	0.07	0.00	!
8.00	0.7708	4.00	14070	1177	0.08	1132	0.08	681	0.12	0.58	0.60	0.04	7	7 0.10	0.07	0.07	0.00	I
																		-

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HD 260X260X329 (Strong axis) Fe 510



HD 310X310X283	(Strong axis) Sigma yield =	335 N/mm2 ;	U/A = 54	; t = 44 mm
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 	i	 	 	 		 F0				F30					 F60		
jι	Lambda Bar	e/h	Npl	 											••••••	•••••	•••••
(m)	1	ļ	(kN)	N(EC3)	N(EC3)/Npl	N(CEF)	N(CEF)/Npl	N(F30)	N/Nc	N/N(FO,EC3)	N/N(FO,CEF)	N/Npl	NCF	50) N/Nc	N/N(FO,EC3)	N/N(FO,CEF)	N/Npl
I	1			(kN)		(kN)	l	(kW)		l		ļ	(k	N		ļ	
				[[·[-		
2.00	0.1721	0.00	12060	11460	0.95	11876	0.98	5291	1.00	0.46	0.45	0.43	8	26 1.00	0.07	0.07	0.06
2.00	0.1721	10.10	12060	9323	0.77	9587	0.79	4413	0.83	0.47	0.46	0.36	7	17 0.87	0.08	0.07	10.05
2.00	0.1721	0.25	12060	7304	0.61	7430	0.62	3594	0.68	0.49	0.48	0.29	5	96 0.72	0.08	1 0.08	10.04
2.00	0.1721	0.50	12060	5370	0.45	5452	0.45	2706	0.51	0.50	0.50	0.22	4	68 0.57	0.09	0.09	0.03
2.00	0.1721	1.00	12060	3524	0.29	3563	0.30	1814	0.34	0.51	0.51	0.15	3	14 0.38	0.09	0.09	10.02
2.00	0.1721	2.00	12060	2089	0.17	2138	0.18	1061	10.20	0.51	0.50	0.08	1	94 0.23	0.09	0.09	10.01
12.00	0.1721	4.00	12060	1151	0.10	1247	0.10	597	10.11	0.52	0.48	0.04	1	14 0.14	0.10	0.09	0.00
4.00	0.3442	0.00	12060	110814	0.90	11583	0.96	4243	11.00	0.39	0.37	0.35	6	60 1.00	0.06	0.06	
4.00	0.5442	10.10	12060	8//6 (and	0.75	9117	0.76	3452	10.81	0.39	0.38	10.28	>	24 0.79	0.06	0.06	10.04
14.00	0.5442	10.25	112060		0.57	1 7066	0.59	2728	10.64	0.40	0.39	10.22	4	07 U.02	0.06	0.06	10.03
14.00	0.5442	10.50	112060	1 5115	0.42	5192	0.43	2159	10.51	0.42	0.42	10.17	1 3	US U.40	0.06	0.06	
14.00	0.3442	11.00	112060	1 2070	0.28	3400	0.28	1471 010	10.35	0.43	0.43	10.12	4	00 0.31	1 0.06		
14.00	0.3442	12.00	112060	2039	į 0.17	2043			10.21	0.45	0.45		ינו	25 JU.19	0.06	0.00	
14,00	0.3442	14.00	112060	1135	1 0.09	1 1113	0.09	224	10.12	U.40	0.47	10.04		/0 U.12	0.07	1 0.07	10.00
	1 0 51/7		1			 4000E		 700/						1			
10.00	0.5163	10.00	112060	10051			0.90	11 3005	10.70	0.40	0.37	10.25	º	27 1.00 01 0.79		0.06	
10.00	0.5163	10.10	112000	1 4705	0.07	04/3		11 2444	10.10	0.30		10.25	4	91 10.10 93 10 41		1 0.06	10.04
10.00		10.25	112000	11 4701		1 / 945	0.04	2440 1971	10.01	0.30	0.37	10.20	3	02 U.OI		1 0.06	10.03
16.00		11 00	112000	1 7770	0.40	1 7219		1021 1220	10.40	0.30	0.37	10.15	"	92 10 70 99 10 70			10.02
14.00		12 00	112000	1 1072	0.27	1 1050		1 740	10.31		0.30	10.04	11 '	16 JO.30			
16.00	0.5163	14 00	112000	1772		1 1080		<i>109</i> 440	10.17	0.39		10.00	64 ' 41	14 [0.10 63 [0 10			10.00
10.00	1 0.5105	14.00	1	···· · 	1 0.07	1 1000	1 0.07	** /	10.11	1 0.40	0.72	1		00 0.10 		1 0.00	1
18.00	1 0 6886	10.00	I 112060	 0117	I 0.76	l 0701	1 I 0.81	3444 	1	1 0.40	I I 037	1 10 30	 5	ו די 11 00		1 0.06	
18.00	1 0.6884	10.10	112060		0.61	1 7703	1 0.64	1 2814	10.77	0.38	0.37	10.23	11 4	49 10.78			10.03
18.00	0.6884	10.25	12060	1 5823	0.48	1 5967	0.49	1 2223	10.61	87.0	1 0.37	10.18	11 7 11 7	51 10.41		0.06	10.02
18.00	1 0,6884	10.50	112060	4410	1 0.37	4450	0.37	11 1660	10,44	0.38	0.37	10.13	2	62 10.44		1 0.04	10.02
18.00	0,6884	11.00	112060	1 3034	0.25	2000	0.25	1 1122	10.31	0.37	0.37	10.09		76 10.31	0.04	0.04	0.01
18.00	0.6884	12.00	12060	1890	0.16	1854	0.15	698	10.19	0.37	0.38	10.05	 	08 0.19	0.06	0.06	10.00
18.00	0.6884	14.00	12060	1 1083	0.09	1 1042	0.09	402	10.11	0.37	0.39	10.03		61 0.11	0.06	0.06	10.00
										·							



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HD 310X310X283 (Strong axis) Fe 510 F60



HD 310X310X375	(Strong	axis)	Sicana	vield =	: 3 35	N/mm2	; U/	A =	42	:	t	= !	57
	(• • · · • · · · · · · · · · · · · · ·	un 107	• • 9 114	,			,	~			-	•	

				!!		FO	1	I		F30		ļ	I		F60			ļ
	Lamoda Bar /	e/n	Npl		····					· · · · · · · · · · · · · · · · · · ·		••••••					•••••	1
(m)	1	1	(KN)	W(EC3)	IN(EC5)/NPL	IN(CEF)	IN(CEF)/NPL	N(F3U)	N/NC	N/N(FU,EC3)	N/N(FU,CEF)	N/Npl	N(F60)	N/NC	N/N(FU,EC3)	N/N(FO,CEF)	[N/Npl	ļ
!	i	1		(KN) 	1	(KN)		(KN)		1			(KN)	ļ		1	ļ	-
12.00	0 4/5/																	
12.00			10013	15220	כיייה ו	15040	0.98	112459	11.00	0.82	0.80		1260	11.00	0.08	0.08	10.07	1
12.00	0.1000	10.10	10013	1112324	[0.77	1 2003		9402 7402	10.75	0.76	0.74 0.75	10.50		10.00	0.09		10.00	
12.00	0.1000	10.25	114013	1 7010	0.00	1 7/22	0.02 0.44	744) 6614	10.00	0.78 0.70	0.75 0.7/	0.40 10.7/	700	10.75				
12.00	0.1000	11 00	14013	1 /574	0.44 0.20	1 / 900	0.40		10.44	0.79 0.79	0.74 0.77	10.34	/09 /70	10.70			10.04	
12.00	0.1000	12 00	114013	1 2700	0.29	1 2010	0.31	3303	10.29	0.70 0.79	0.73 0.77	10.22 10.17	4/C 207	10.37			10.02	
12.00	0.1050	14 00	114013	1 1/8/	0.17	1 1444	0.10 0.10	2111	10.00	0.78 0.77	0.73 0.40	10.13	293	10.25	0.11 0.12	0.10	10.01	
12.00		14.00 I		404	0.07	1 1004	0.10	44	10.07	0.77	1 0.09	10.07	17 2 	0.14 	0.12	1 0.10	10.01	
	 0 3312	1 In nn	I 16013	 14404	l I 1001	1	I 104	 0031		 0.63	 0.50	 0 56	1 1020	 1_00_		 0.07		
		10.00		14400	0.70 0.73	12164	0.76 0.76	1 7587	10.8/	0.03 0.65	0.37 0.62	0.30	027	10.70			10.05	
	0.3312	10.10	116013	1 0081	0.75	1 0334	0.70 0.58	1 6050	10.67	0.65 0.67	1 0.62	10.47		10.77			10.03	
14.00	0.3312	10.50	16013		1 0.27	1 6790	0.20	1 4667	10.52		0.00	10.20	465	10.01			10.03	
14.00	0.3312	11.00		1 4422	0.72 0.28	1 4401	0.72	1 3144	10.35	0.70	0.09	in 10	4 05 312	10.30				
14.00	0.3312	2.00	16013	1 2638	0.16	2672	0,17	1 1871	10.21	0.71			11 188	10.30				
14.00	0.3312	4.00	16013	1 1464	l 0.09	1 1440	0.09	1 1063	10.12	0.73	0.74	10.06	1 114	10.11	0.08	0.08		1
				11		1			1	1	1		11 11	1	1		1	
6.00	0.4968	0.00	16013	113454	، ا 0.84	' 14550	0.91	1 7661	11.00	l 0.57	0.53	0.47	982	I 1.00	۱ 0.07	0.07	10.06	
6.00	0.4968	0.10	16013	10794	0.67	111305	0.71	5952	10.78	0.55	0.53	0.37	766	10.78	0.07	0.07	10.04	
6.00	0.4968	0.25	16013	8457	0.53	8745	0.55	4762	10.62	0.56	0.54	0.29	591	10.60	0.07	0.07	10.03	i
6.00	0.4968	0.50	16013	6301	0.39	6426	0.40	3763	0.49	0.60	0.59	0.23	435	10.44	0.07	0.07	10.02	i
6.00	0.4968	1.00	16013	4216	0.26	4208	0.26	2624	0.34	0.62	0.62	0.16	287	0.29	0.07	0.07	10.01	i
j6.00 j	0.4968	2.00	16013	2559	. 0.16	2525	0.16	1631	0.21	0.64	0.65	0.10	173	0.18	0.07	0.07	10.01	
j6.00 j	0.4968	4.00	16013	1435	0.09	1398	0.09	961	0.13	0.67	0.69	0.05	95	0.10	0.07	0.07	0.00	1
i i			i i	i	İ	i	i i	i	i	Ì	i	i	i i	i	i	İ	i	
8.00	0.6624	0.00	16013	12285	0.77	13329	0.83	6928	1.00	0.56	0.52	0.43	914	1.00	0.07	0.07	0.05	j
8.00	0.6624	0.10	16013	9835	0.61	10327	0.64	5354	0.77	0.54	0.52	0.33	708	0.77	0.07	0.07		İ
8.00	0.6624	0.25	16013	7750	0.48	8000	0.50	4212	0.61	0.54	0.53	0.26	546		0.07	0.07	0.03	į
8.00	0.6624	0.50	16013	5845	0.37	5891	0.37	3149	0.45	0.54	0.53	0.19	405	0.44	0.07	0.07	0.02	i
8.00	0.6624	1.00	16013	3981	0.25	3961	0.25	2188	0.32	0.55	0.55	0.13	270	0.30	0.07	0.07	0.01	i
8.00	0.6624	2.00	16013	2461	0.15	2425	0.15	1392	0.20	0.57	0.57	0.08	164	0.18	0.07	0.07	0.01	İ
8.00	0.6624	4.00	16013	1400	0.09	1363	0.09	850	0.12	0.61	0.62	0.05	92	0.10	0.07	0.07	0.00	Í
									,									

mm



HD 310X310X375 (Strong axis) Fe 510



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HD	310x310x500	(Strong axis)	Siama	vield =	325	N/mm2	: U/	'A =	33 :	t	: =	75	m
		(ottong avita)	a 1 9a	,			, -,	••			•		

İ.				1	[]		FO	l	1		F30					F60		ļ
	, i	Lamoda Bar	e/n 	I APR			144055		1.445703							IN /N/FO FC7)		
1 (m)	'			(60)		#(EC3)/#pt 	I (LEF)	(1 (130)	N/NC 	M/M(FU,EL3)	W/W(FU,CEF)	In/npc	[R(FOU)	N / NC	1 1	M/M(FU,CEF) 	Тихиргі
	 		 	 		 	1	 	(KM)	 	 	 	1 	(KM) 	 	 	 	
12.00	h	0.1538	I 0.00	1 20721	 19725	I 0.95	120582	 	l 118041		l n 91	! 0.88	 0.87	1 2205	I 1.00	1 0 11	1 0.11	10 10 1
12.00		0,1538	0.10	20721	115867	0.77	116568	0.80	114181	10.79	0.89	0.86	0.68	1 1860	10.84	0.12	0.11	10.08
12.00		0.1538	0.25	20721	112276	0.59	13045	0.63	111166	10.62	0.91	0.86	0.53	1498	10.68	0.12	0.11	
12.00		0.1538	0.50	20721	8936	0.43	9591	0.46	8209	10.46	0.92	0.86	0.39	1 1125	10.51	0.13	0.12	10.05
12.00		0.1538	1.00	20721	5790	0.28	6331	0.31	1 5092	10.28	0.88	0.80	0.24	1 750	10.34	i 0.13	0.12	10.03
12.00	i	0.1538	2.00	20721	3404	0.16	3798	0.18	2988	10.17	0.88	0.79	0.14	442	10.20	0.13	0.12	10.02 1
12.00) İ	0.1538	4.00	20721	1864	0.09	2100	0.10	1615	0.09	0.87	0.77	0.07	247	0.11	0.13	0.12	0.01
i	i		İ	i i	ii	İ	i	i i	i	i		I	Ì	ii	i	j	i	i i
4.00	۱i	0.3077	0.00	20721	18753	0.91	19971	0.96	15005	1.00	0.80	0.75	0.72	1957	1.00	0.10	0.10	0.09
j4.00	۱Ì	0.3077	0.10	20721	15050	0.73	15800	0.76	12151	0.81	0.81		0.58	1539	0.79	0.10	0.10	0.07
4.00	Ì	0.3077	0.25	20721	11693	0.56	12124	0.59	9538	0.64	0.82	0.79	0.46	1191	0.61	0.10	0.10	0.05
4.00	1	0.3077	0.50	20721	8572	0.41	8747	0.42	7036	0.47	0.82	0.80	0.33	869	0.44	0.10	0.10	0.04
4.00		0.3077	1.00	20721	5614	0.27	5726	0.28	4606	0.31	0.82	0.80	0.22	575	0.29	0.10	0.10	0.02
4.00		0.3077	2.00	20721	3338	0.16	3370	0.16	2710	0.18	0.81	0.80	0.13	342	0.17	0.10	0.10	0.01
4.00	1	0.3077	4.00	20721	1841	0.09	1816	0.09	1528	0.10	0.83	0.84	0.07	198	0.10	0.11	0.11	0.00
1								1 1	1			1			1	1		
6.00	1	0.4615	0.00	20721	17638	0.85	19378	0.94	11819	1.00	0.67	0.61	0.57	1880	1.00	0.11	0.10	0.09
6.00	1	0.4615	0.10	20721	14100	0.68	14876	0.72	9757	0.83	0.69	0.66	0.47	1458	0.78	0.10	0.10	0.07
6.00	1	0.4615	0.25	20721	10984	0.53	11414	0.55	8122	0.69	0.74	0.71	0.39	1120	0.60	0.10	0.10	0.05
6.00		0.4615	0.50	20721	8127	0.39	8221	0.40	6249	0.53	0.77	0.76	0.30	821	0.44	į 0.10	0.10	0.03
6.00		0.4615	1.00	20721	5394	0.26	5437	0.26	4229	0.36	0.78	0.78	0.20	535	0.28	0.10	0.10	0.02
6.00		0.4615	2.00	20721	3251	0.16	3228	0.16	2539	0.21	0.78	0.79	0.12	319	0.17	0.10	0.10	0.01
6.00		0.4615	4.00	20721	1808	0.09	1779	0.09	1486	0.13	0.82	0.84	0.07	176	0.09	0.10	0.10	0.00
1						1	1	! !	1							1	1	
8.00		0.6154	0.00	20721	16301	0.79	18155	0.88	11009	1.00	0.68	0.61	0.53	1804	1.00	0.11	0.10	0.08
8.00		0.6154	0.10	20721	12971	0.63	13825	0.67	8440	0.77	0.65	0.61	0.40	1374	0.76	0.11	0.10	0.06
8.00	۱.	0.6154	0.25	20721	10161	0.49	10497	0.51	6657	0.60	0.66	0.63	0.32	1055	0.58	0.10	0.10	0.05
8.00		0.6154	0.50	20721	7599	0.37	7702	0.37	5242	0.48	0.69	0.68	0.25	775	0.43	0.10	0.10	0.03
8.00		0.6154	1.00	20721	5129	0.25	5146	0.25	3608	0.33	0.70	0.70	0.17	513	0.28	0.10	0.10	0.02
18.00		0.6154	2.00	20721	3139	0.15	3100	0.15	2329	0.21	0.74	0.75	0.11	307	0.17	0.10	0.10	0.01
18.00	I	0.6154	4.00	20721	1769	0.09	1727	0 .0 8	1343	0.12	0.76	0.78	0.06	171	0.09	0.10	0.10	10.00
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HD 310X310X500 (Strong axis) Fe 510 -



HD 400X400X314 (Strong axis) Sigma yield = 345 N/mm2; U/A = 58; t = 40 mm

	••••••••••																
		l				FO	I	1		F30		l	·		F60		
	L am bda Bar	e/h	Npl														
(m) 			(kN)	N(EC3)	N(EC3)/Npl	N(CEF)	N(CEF)/Npl	N(F30)	N/NC	N/N(FO,EC3)	N/N(FO,CEF)	N/Npl	N(F60) N/Nc	N/N(FO,EC3)	N/N(FO,CEF)	N/Npl
				(KN)		(KN) 		(KN)		1	1		(kN)	ļ	1	1	!
12 00 1	0 15/9	0.00	117900	117200		117707		 E444	11 00					-			
12 00 1	0.1548	10.00	113000	110804	0.70 0.78	110051	0.99 0.70	1 /517			0.41	10.40	1 903				
12 00 1	0.1548	10.10	113800	1 8513	0.70 0.42	1 8440	0.79 0.43	1 7445	10.00	0.42 0.43	0.41	10.32	1 707	10.00	1 0.08	0.08	10.00
	0.1548		113800	6200	0.02 0.46	6321	0.03 0.46	275/	10.00	0.43 0.46	0.42 0.46	10.20	1 557	10.72		0.08	10.02
2.00	0.1548	1.00	113800	4143	0.40 0.30	1 4173	0.40 0.30	1 1850	10.47	0.44 0.45	0.44	10.17 10.17	1 372	10.30	0.09 0.09	1 0.09	10.04
2.00	0.1548	2.00	13800	2462	0.18	2642	0.30 0.19	1118	10.35	0.45			1 227	10.30			10.02
2.00	0.1548	4.00	13800	1361	0.10	1480	0.11	615	10.11	0.45	0.42	10.04	1 131	10.13	0.10	0.09	10.00
i						1							1				
4.00	0.3096	0.00	13800	12554	0.91	13231	0.96	4421	1.00	0.35	0.33	0.32	, 739	1.00	0.06	0.06	0.05
4.00	0.3096	0.10	13800	10266	0.74	10626	0.77	3550	0.80	0.35	0.33	0.25	591	0.80	0.06		0.04
4.00	0.3096	0.25	13800	8101	0.59	8235	0.60	2862	0.65	0.35	0.35	0.20	464	0.63	0.06	0.06	0.03
4.00	0.3096	0.50	13800	6029	0.44	6113	0.44	2179	0.49	0.36	0.36	0.15	349	0.47	0.06	0.06	0.02
4.00	0.3096	1.00	13800	4010	0.29	4002	0.29	1545	0.35	0.39	0.39	0.11	244	0.33	0.06	0.06	0.01
4.00	0.3096	2.00	13800	2414	0.17	2406	0.17	949	0.21	0.39	0.39	0.06	155	0.21	0.06	0.06	0.01
4.00	0.3096	4.00	13800	1344	0.10	1323	0.10	545	0.12	0.41	0.41	0.03	92	0.12	0.07	0.07	0.00
				1	1			1	1	l			1	1	1	1	1
6.00	0.4644	0.00	13800	11817	0.86	12583	0.91	4204	1.00	0.36	0.33	0.30	709	1.00	0.06	0.06	10.05
6.00	0.4644	0.10	13800	9603	0.70	10056	0.73	3360	0.80	0.35	0.33	0.24	559	0.79	0.06	0.06	0.04
6.00	0.4644	0.25	13800	7601	0.55	7794	0.56	2604	0.62	0.34	0.33	0.18	438	0.62	0.06	0.06	0.03
6.00	0.4644	0.50	13800	5709	0.41	5755	0.42	1933	0.46	0.34	0.34	0.14	324	0.46	0.06	0.06	0.02
6.00	0.4644	1.00	13800	3846	0.28	3806	0.28	1326	0.32	0.34	0.35	0.09	216	0.31	0.06	0.06	0.01
6.00 / 00	0.4644	2.00	13800	2345	0.17	2311	0.17	824	0.20	0.35	0.36	0.05	132	0.19	0.06	0.06	0.00
6.00	0.4644	4.00	13800	1321	0.10	1284	0.09	477	0.11	0.36	0.37	0.03	74	0.10	0.06	0.06	0.00
	A (400								1					l			
10.00	0.6192	0.00	115800	10927	0.79	11644		3977	11.00	0.36	0.34	0.28	663	1.00	0.06	0.06	0.04
10.00 10.00	0.6192	0.10	13800	8842 7005	0.64 0.54	9204		3075	10.77	0.35		0.22	520	0.78	0.06	0.06	0.03
10.00 10.00	0.6192	0.20	112800	/UZD	U.51	1 7202	0.52	2406	10.61	[0 .3 4		0.17	408	10.62	1 0.06	0.06	0.02
0.00 8.00	0.0192	1 00	113800	3327	0.39 0.27	0392	U.39 0.37	1802	10.45	į 0.34	0.33	0.13	304	10.46	0.06	0.06	10.02
	0 4102	2.00	13000 13000	3044 3344	U.26	1 2205	0.26	1250	10.51	0.34	0.34	0.08	205	10.31	0.06		10.01
13.00 18.00	0.0172	2.00 6 00	13000 13800	1 120/	0.16	1 2205	U.16	(53	10.19	U.53	0.34	0.05	1 125	10.19		U.06	10.00
	0.0192	4.00	13000	1294	0.09	1240	0.09	454	10.11	∕ U.54	0.35	0.03	71	10.11	0.05	0.06	10.00



HD 400X400X314 (Strong axis) Fe 510 F60



	1					FO				F30			1		F60		
l ()	Lambda Bar I	e/h 	Npl					 N/E301				 N/Nol	 N(E60)		IN/N(E0 EC3)		
(m)	1	1		(kN)	(:::::::::::::::::::::::::::::::::::	(kN)		(kN)					(kN)				
	' -•				 										j		• • • • • •
2.00	0.1321	0.00	27216	26112	0.96	27030	0.99	26800	1.00	1.03	0.99	0.98	4280	1.00	0.16	0.16	0.15
2.00	0.1321	0.10	27216	21071	0.77	22092	0.81	21896	0.82	1.04	0.99	0.80	3400	0.79	0.16	0.15	0.12
2.00	0.1321	0.25	27216	16343	0.60	17656	0.65	17497	0.65	1.07	0.99	0.64	2753	0.64	0.17	0.16	0.10
2.00	0.1321	0.50	27216	11903	0.44	13245	0.49	13121	0.49	1.10	0.99	0.48	2088	0.49	0.18	0.16	0.07
2.00	0.1321	1.00	27216	7717	0.28	8920	0.33	8834	0.33	1.14	0.99	0.32	1391	0.32	0.18	0.16	0.05
2.00	0.1321	2.00	27216	4535	0.17	5344	0.20	5292	0.20	1.17	0.99	0.19	828	0.19	0.18	0.15	0.03
2.00	0.1321	4.00	27216	2488	0.09	3377	0.12	3009	0.11	1.21	0.89	0.11	471	0.11	0.19	0.14	0.01
						1				1	ł	I		1	1		
4.00	0.2642	0.00	27216	25018	0.92	26492	0.97	23880	1.00	0.95	0.90	0.87	3305	1.00	0.13	0.12	0.12
4.00	0.2642	0.10	27216	20152	0.74	20960	0.77	20754	0.87	1.03	0.99	0.76	2615	0.79	0.13	0.12	0.09
4.00	0.2642	0.25	27216	15685	0.58	16083	0.59	15931	0.67	1.02	0.99	0.58	2006	0.61	0.13	0.12	0.07
4.00	0.2642	0.50	27216	11503	0.42	11818	0.43	11708	0.49	1.02	0.99	0.43	1474	0.45	0.13	0.12	0.05
4.00	0.2642	1.00	27216	7536	0.28	7738	0.28	7667	0.32	1.02	0.99	0.28	996	0.30	0.13	0.13	0.03
4.00	0.2642	2.00	27216	4463	0.16	4558	0.17	4517	0.19	1.01	0.99	0.16	601	0.18	0.13	0.13	10.02
4.00	0.2642	4.00	27216	2465	0.09	2457	0.09	2435	10.10	0.99	0.99	0.08	350	0.11	0.14	0.14	10.01
															1 0.17		
6.00	0.3963		27216	23820	0.88	25581	0.94	119140	10.05		0.75	10.70	11 2/02	10.70	0.13	0.12	10.11
6.00	0.3963	10.10	27216	19120		20154	U.74	110295	10.85	1 0.85	0.01	10.59	1 2402	10.70	0.13	0.12	10.07
6.00	0.3903	10.25	27210	14921	0.00	15374	0.50	1110076	10.70	1 0.90	1 0.00	10.49	1/10				10.05
6.00	0.3703	11 00	27210		0.41	11239	U.41	11 7041	10.33		1 0.90	10.57	1411 018	10.99			10.03
6.00 4 00	0,3903	12.00	27216	1 1 292	0.27	1 4350	0.27	11 / 308	10.37	1 0.97	1 0.90	10.25	551	10.27		0.12	10.03
6.00	0.3903	14 00	127216	1 2/33	1 0.10	1 238/	0.10 0.00	11 2362	10.25	0.90	1 0.99		11 302		0.13	0.13	10.01
0.00	0.3903	14.00		2455	0.07	1 2304	0.07	11	1	1		1	502	1	1	0115	1
8.00	I I 0.5284	10.00	127216	122437	I I 0.82	I 124327	L 1 0.89	LI 116006	1	0.71	I 0.66	۱ 10.58	 3035	1	0.14	0.12	10.11
8.00	0.5284	10.10	27216	17921	0.66	18901	0.69	112561	10.78	0.70	0.66	10.46	2358	0.78	0.13	0.12	0.08
8.00	0.5284	10.25	127216	14047	0.52	14496	0.53	110489	10.66	0.75	0.72	0.38	1808	0.60	0.13	0.12	0.06
8.00	0.5284	0.50	27216	10457	0.38	10636	0.39	8251	0.52	0.79	0.78	0.30	1346	0.44	0.13	0.13	10.04
8.00	0.5284	1.00	27216	7006	0.26	7035	0.26	5790	0.36	0.83	0.82	0.21	878	0.29	0.13	0.12	0.03
8.00	0.5284	2.00	27216	4262	0.16	4221	0.16	3771	0.24	0.88	. 0.89	0.13	527	0.17	0.12	0.12	0.01
8.00	0.5284	4.00	27216	2389	0.09	2326	0.09	1 2226	0.14		0.96	0.08	290	0.10	0.12	0.12	0.01

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MD 400X400X1086 (Strong axis) Sigma yield = 305 N/mm2 ; U/A = 20 ; t = 125 mm

	••••••••••••••••••••••••••••••••••••••				•••••	•••••				•••••							
				!!		FO				F30			l l		F60		l
	Lambda Bar	e/h	Npl			•••••	•••••										
(m) 	1	 5	(KN) 	N(EC3)	IN(EC3)/NPL	IN(CEF)	N(CEF)/Npl	N(F30)	N/NC	N/N(FO,EC3)	N/N(FO,CEF)	N/Npl	N(F60)	N/NC	N/N(FO,EC3)	N/N(FO,CEF)	N/Npl
	 	l 	1	(KN)	1	(KN)		[(KN)	ļ	1			(kN)	1			1
12 00	1 0 1170		142273	 30500	1 0.04	1/10/7	0.00		11 00	1.05		10.00					
12.00		10.00	142273	1131685	0.74	135776	0.99	17552/	10.95	1 1.05	0.99	10.90	145080	11.00	0.49	0.46	10.45
12.00	0.1170	10.25	42273	124410	1 0.58	127123	0.67	124079	10.65	1 1.12	1 0.99	10.04	15088	10.78	0.48 0.48	0.42	10.35
12.00		10.50	142273	117650		120878	1 0.04	120736	10.05	1.10	i 0.99		0557	10.01	0.48	0.43	10.27
12.00	0.1170	11.00	42273	11374	0.72	114256	0.47	113167	10.30	1.17	0.99	10.47		10.44	0.40 0.78	0.41 0.70	10.20
12.00	0.1170	2.00	42273	1 6655		1 9000	0.21	1 7452	10.32	1.10	0.92 0.83	10.31	1 3105	10.20	0.40 0.78	0.39	
12.00	0.1170	4.00	42273	1 3620	0.09	1 5005			10.10	1.12	0.85 0.88	10.10	1 1725		0.40 0.48	0.36	
1	1		1		0.07					'	1 0.00			10.07	0.40	0.34 	10.04
4.00	0.2340	0.00	 42273	37071	0.88	' 41160	0.97	140346	I 1.00	I 1.09	I I 0.98	10.95.1	113765	1	I I 037	0.33	
4.00	0.2340	0.10	42273	29891	0.71	32524	0.77	32287	10.80	1.08		10.76	111410	10.83	[0.37 0.38	0.35	10.32
4.00	0.2340	0.25	42273	23232	0.55	24848	0.59	24671	0.61	1.06	0.99	10.58	9225	10.67	0.40		10.20
4.00	0.2340	0.50	42273	16984	0.40	17905	0.42	17779	0.44	1.05	0.99	10.42	6872	10.50	0.40	0.38	10.16
4.00	0.2340	1.00	42273	11065	0.26		0.28	111610	0.29	1.05	0.99	10.27	4568	10.33	0.41	0.39	10.10
4.00	0.2340	2.00	42273	6549	0.15	6779	0.16	6731		1.03	0.99	0.15	2741	0.20	0.42	0.40	10.06
4.00	0.2340	4.00	42273	3579	0.08	3705	0.09	3679		1.03	0.99	0.08	1556	0.11	0.43	0.42	10.03
1			1	1	ĺ	İ	İ	i	i	i	İ	i i	i	i			
6.00	0.3510	0.00	42273	34585	0.82	40320	0.95	36533	1.00	1.06	0.91	0.86	12600		0.36	0.31	0.29
6.00	0.3510	0.10	42273	27994	0.66	31558	0.75	28932	0.79	1.03	0.92	0.68	9774	0.78	0.35	0.31	0.23
6.00	0.3510	0.25	42273	21957	0.52	23800	0.56	22052	0.60	1.00	0.93	0.52	7589	0.60	0.35	0.32	jo. 17 j
6.00	0.3510	0.50	42273	16208	0.38	17289	0.41	17152	0.47	1.06	0.99	0.40	5709	0.45	0.35	0.33	0.13
6.00	0.3510	1.00	42273	10716	0.25	11200	0.26	11114	0.30	1.04	0.99	0.26	3992	0.32	0.37	0.36	jo.09 j
6.00	0.3510	2.00	42273	6406	0.15	6512	0.15	6464	0.18	1.01	0.99	0.15	2321	0.18	0.36	0.36	0.05
6.00	0.3510	4.00	42273	3535	0.08	3523	0.08	3498	0.10	0.99	0.99	0.08	1360	0.11	0.38	0.39	0.03
						I	†	1		1		1 1	1	ĺ	Ì	Ì	i i
8.00	0.4680	0.00	42273	32050	0.76	39204	0.93	28818	1.00	0.90	0.74	0.68	12067	1.00	0.38	0.31	0.28
8.00	0.4680	0.10	42273	26089	0.62	30096	0.71	23707	0.82	0.91	0.79	0.56	9142	0.76	0.35	0.30	0.21
8.00	0.4680	0.25	42273	20574	0.49	22862	0.54	19409	0.67	0.94	0.85	0.45	7037	0.58	0.34	0.31	0.16
8.00	0.4680	0.50	42273	15392	0.36	16466	0.39	14945	0.52	0.97	0.91	0.35	5170	0.43	0.34	0.31	0.12
8.00	0.4680	1.00	42273	10304	0.24	10759	0.25	9907	0.34	0.96	0.92	0.23	3403	0.28	0.33	0.32	0.08
18.00	0.4680	2.00	42273	6256	0.15	6336	0.15	6285	0.22	1.00	0.99	0.14	2050	0.17	0.33	0.32	0.04
8.00	0.4680	4.00	42273	3480	0.08	3453	0.08	3427	0.12	0.98	0.99	0.08	1158	0.10	0.33	0.34	0.02
1																	





I = 6 m

T.21

l = 8 m

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HE 550 M (Strong axis) Sigma yield = 450 N/mm2 ; U/A = 64 ; t = 40 mm

				•••••	• • • • • • • • • • • • •										•••••		
	l					FO		1		F30		l	<u> </u> .		F 6 0		
L	Lambda Bar	e/h	Npl		• • • • • • • • • • • •									·			
(m)		ļ	(kN)	N(EC3)	N(EC3)/Npl	N(CEF)	N(CEF)/Npl	N(F30)	N/NC	N/N(F0,EC3)	N/N(FO,CEF)	N/Npl	N(F60)	N/NC	N/N(FO,EC3)	N/N(FO,CEF)	N/Npl
!		ł		(kN)		(kN)		(kN)	ļ	1	ļ		(kN)				
12.00	0.1247	10.00	15930 45070	15442	0.97	15530		5032	11.00	0.33	0.32	0.31	11160	11.00			10.07
12.00	0.1247	10.10	15930	112209	0.79	12027	0.79	4091	10.81	į 0.33	0.32	10.25	7 00	10.82	0.08		10.06
12.00	0.1247	10.25	115070		0.02	9942 77 9/		0 3248	10.65		0.33 0.77	10.20	780 505	10.01		[0.08	10.04
12.00	0.1247	10.50	15070	(234 (773	0.40	1 / 304	0.46 0.70	2472 1415	10.49	U.34	i 0.33	10.15	CYC	10.51			
12.00	0.1247	12 00	115030	1 2709	0.30	1 2805	0.30		10.32	0.34 0.7/	j 0.33		1 371	10.34			10.02
12.00	0.1247	14 00	15030	2170	0.10 0.10	1 1773	0.10	901 577	10.19	1 0.34	į 0.55		1 120	10.20			
12.00		4.00 	0.560	24 	0.10	1 1752	0.11		10.11	i 0.33	i 0.31	10.03	127	10.11	1 0.00	0.07	10.00
14.00	0.2494	1 10.00	I 15930	 14912	I 1 04	 15375	 0.07	 	1	 0.28	I I 0.28	1 26 1	I I 816		1 0.05	1 0.05	
14.00	0.2494	10.10	15930	112127	0.74	112411	0.78	1 3530	10.83	1 0.20	0.20	10 22 1	663	10.81	1 0.05	0.05	
14.00	0.2494	10.25	15930	9518	0.60	9619	0.60	2857	10.67	0.30	0.30	10.17	563	10.69	0.06	0.06	10.03
14.00	0.2494	10.50	15930	7035	0.44	7069	0.44	2204	10.52	0.31	i 0.31	10.13	443	10.54	0.06	1 0.06	10.02
4.00	0.2494	1.00	15930	4629	0.29	4712	0.30	1512	10.36	0.33	0.32	10.09	303	10.37	0.07	0.06	10.01
4.00	0.2494	2.00	15930	2764	0.17	2824	0.18	915	10.22	0.33	0.32	10.05	1 188	10.23	0.07	0.07	10.01
4.00	0.2494	4.00	15930	1528	0.10	1497	0.09	497	10.12	0.33	0.33	10.03	1 110	10.13	0.07	0.07	10.00
i		Ì	i i	Í	Ì	i			[1 1	1	1			
6.00	0.3741	0.00	15930	14352	. 0.90	14846	0.93	4189	1.00	0.29	0.28	0.26	801	1.00	0.06	0.05	0.05
6.00	0.3741	0.10	15930	11602	0.73	12042	0.76	3398	0.81	0.29		j0.21 j	641	0.80		0.05	0.04
6.00	0.3741	0.25	15930	9110	0.57	9240	0.58	2658	0.63	0.29	0.29	0.16	502	0.63	0.06	0.05	0.03
6.00	0.3741	0.50	15930	6771	0.43	6790	0.43	1996	0.48	0.29	0.29	0.12	j 372	0.46	0.05	0.05	0.02
6.00	0.3741	1.00	15930	4505	0.28	4527	0.28	1371	0.33	0.30	0.30	0.08	252	0.31	0.06	0.06	0.01
6.00	0.3741	2.00	15930	2707	0.17	2720	0.17	837	0.20	0.31	0.31	0.05	151	0.19	0.06	0.06	0.00
6.00	0.3741	4.00	15930	1512	0.09	1468	0.09	471	0.11	0.31	0.32	0.02	86	0.11	0.06	0.06	0.00
1						1	1 1		Í	ĺ	[i i	I	Ì	1	1	İ
8.00	0.4988	0.00	15930	13673	0.86	14697	0.92	4038	1.00	0.30	0.27	0.25	777	1.00	0.06	0.05	0.04
8.00	0.4988	0.10	15930	11002	0.69	11512	0.72	3248	0.80	0.30	0.28	0.20	615	0.79	0.06	0.05	0.03
8.00	0.4988	0.25	15930	8655	0.54	8833	0.55	2558	0.63	0.30	0.29	0.16	482	0.62	0.06	0.05	0.03
8.00	0.4988	0.50	15930	6459	0.41	6457	0.41	1883	0.47	0.29	0.29	0.11	357	0.46	0.06	0.06	0.02
8.00	0.4988	1.00	15930	4343	0.27	4305	0.27	1290	0.32	0.30	0.30	0.08	238	0.31	0.05	0.06	0.01
8.00	0.4988	2.00	15930	2647	0.17	2613	0.16	783	0.19	0.30	0.30	0.04	144	0.19	0.05	0.06	0.00
8.00	0.4988	4.00	15930	1489	0.09	1428	0.09	440	0.11	0.30	0.31	0.02	78	0.10	0.05	0.05	0.00

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HE 550 M (Strong axis) FeE 460



T.23

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$\frac{1}{100} + \frac{1}$	HD 400X400X314	(Strong axis)	Sigma yield =	450 N/mm2 ;	U/A = 58 ; t = 40 m
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1				• •						• • • • • • • • • • • • • • • • • •						•••••	•••••
						FO				F30			<u> </u> .		F60		
	[Lambda Bar	} e/h	Npl													- 	
(m) 	1	1	(KN)	N(EC3)	N(EC3)/Npl	IN(CEF)	N(CEF)/Npl	N(F30)	N/NC	N/N(FO,EC3)	N/N(FO,CEF)	N/Npl	N(F60)	N/Nc	N/N(FO,EC3)	N/N(FO,CEF)	N/Npl
 	1	 		(KN)	1] (KN)		(KN)	1	1	l		(kN)	ļ	1	l	ļ
12 00	0 1768		118000	117107	0.04	117700	0.00		11 00	0.77		10.75					
12.00	0.1768	10.10	118000	114074	0.70 0.78	11//09	0.90 0.90	1 5307	10.94	0.37 0.79	0.30 0.37	10.35	1 077	10.94		0.06	10.06
12.00	0.1768	10.25	118000	111071		111102	0.00	1 6238	10.04	0.36 0.38	[0.37 0.38	10.29	1 905	10.00			10.05
12.00	0.1768	10.50	118000	8188	0.02	1 8420	0.02 0.47	1 3168	10.50	i 0.30 I ∩.30	0.38 0.38	10.25	1 627	10.71			10.04
12.00	0.1768	11.00	118000	5392	0.30	5444			10.30	1 0.39	0.38 0.38	10.17	1 426	10.30	0.08	0.07 0.08	10.03
2.00	0.1768	2.00	18000	3209	0.18	3236	0.18	1 1281	10.20	0.40	0.40		255	10.30		0.08	
2.00	0.1768	4.00	18000	1774	0.10	1916	0.11	683	10.11	0.39	0.36		1 147	10.13	0.08		
Ì	Ì	İ	i i	i	i								1				1
4.00	0.3536	0.00	18000	16325	0.91	17183	0.95	5741	1.00	0.35	0.33	0.31	 956	1.00	, I 0.06	0.06	10.05
4.00	0.3536	0.10	18000	13292	0.74	13732	0.76	4588	0.80	0.35	0.33	0.25	763	0.80	0.06	0.06	0.04
4.00	0.3536	0.25	18000	10472	0.58	10643	0.59	3556	0.62	0.34	0.33	0.19	597	0.62	0.06	0.06	0.03
4.00	0.3536	0.50	18000	7797	0.43	7821	0.43	2724	0.47	0.35	0.35	0.15	442	0.46	0.06	0.06	0.02
4.00	0.3536	1.00	18000	5195	0.29	5162	0.29	1840	0.32	0.35	0.36	0.10	303	0.32	0.06	0.06	0.01
4.00	0.3536	2.00	18000	3130	0.17	3093	0.17	1137	0.20	0.36	0.37	0.06	185	0.19	0.06	0.06	0.01
4.00	0.3536	4.00	18000	1748	0.10	1725	0.10	634	0.11	0.36	0.37	0.03	111	0.12	0.06	0.06	0.00
				1	1	1			1	1	[1	1	1		1
6.00	0.5304	0.00	18000	15258	0.85	16262	0.90	5434	1.00	0.36	0.33	0.30	908	1.00	0.06	0.06	0.05
6.00	0.5304	0.10	18000	12316	0.68	12737	0.71	4256	0.78	0.35	0.33	0.23	712	0.78	0.06	0.06	0.03
16.00	0.5304	0.25	18000	9722	0.54	9872	0.55	3299	0.61	0.34	0.33	0.18	557	0.61	0.06	0.06	0.03
10.00	0.5304	0.50	18000	7299	0.41	7290	0.41	2490	0.46	0.34	0.34	0.13	413	0.45	0.06	0.06	0.02
10.00	0.5304		18000 48000	4938		1 4860	0.27	1660	0.31	0.34	0.34	0.09	276	0.30	0.06	0.06	0.01
10.00	0.5304	2.00	18000 48000	1 3024	0.17	2950	0.16	1027	0.19	0.34	0.35	0.05	168	10.19	0.06	0.06	0.00
10.00	0.5504	4.00		1713	0.10 	1657	ן ט.טין	577	10.11	0.34	0.35	0.03	94	0.10	0.05	0.06	0.00
 2:00	0 7072		 19000		0.77												
8 00		0.00	18000 18000	111177		114749	0.02 0.47	4920	10.70	1 0.36		10.27	832	11.00	0.06	0.06	10.04
8.00	0.7072	0.25	18000	8838	0.02	8041	U.04 ೧೯೧۱	1 2002	10.70	j U.35	U.33	10.21 10.44	049 E00	10.78	U.06	U.06	10.03
8.00		0.50		6718	ן ט.47 ∩ ז7	1 6606	0.50 0.37	2284	10.01	∪.34 ∫ ∩.7/	U.34	10.10 10.12	1 797	10.01 10.74	1 0.06	U.US	10.02
8.00	0.7072	1.00	18000	1 4625	0.26	4560	0.57	1 1557	10.32	0.34 0.7/	0.34 0.7/	וייין וח הפו	1 250	10.40	0.06	0.00 0.04	10.02
8.00	0.7072	2.00	18000	2893	0.16	2805	0.16	058	0.10	ן ט.שיי ן ∩ זז	0.34 ∩ 3/.	10.00	2.59	10.31 10.10	0.00 0.05	0.00 0.04	10.01
8.00	0.7072	4.00	18000	1665	0.09	1 1596	0.09	1 545	10.11	0.33	0.34 0.34	10.03	1 00	10.17	1 0.05	0.00 0.04	
	· · · · · · · · · · · · · · · · · · ·		·••									10.00	1 70	10.11	1 0.05	0.00	10.00





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FOR BENDING ABOUT

THE MINOR AXIS

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	 esta Rec	 				FO	ļ			F30			1		F60		
(m)		=/11 	Mpt (kN) 	 N(EC3) (kN)	N(EC3)/Npl 	N(CEF)	N(CEF)/Npl	 N(F30) (kN)	N/Nc 	N/N(F0,EC3) 	N/N(FO,CEF)	 N/Npl 	 N(F60)	 N/Nc 	N/N(F0,EC3)	N/N(FO,CEF)	 > N
										, 	 	 		 	 	 	- -
2.00	0.3509	0.00	12213	11141	0.91	11127	0.91	3060	1.00	0.27	0.28	0.25	593	1.00	0.05	0.05	i
2.00	0.3509	0.10	12213	8291	0.68	8473	0.69	2388	0.78	0.29	0.28	0.19	459	0.77	0.06	0.05	l
2.00	0.3509	0.25	12213	5427	0.44	5967	0.49	1708	0.56	0.31	0.29	0.13 j	326	0.55	0.06	0.05	i
2.00	0.3509	0.50	12213	3083	0.25	3604	0.30	1087	0.36	0.35	0.30	0.08	200		0.06	0.06	i
2.00	0.3509	1.00	12213 1	1620 	0.13	1959	0.16	606	0.20	0.37	0.31	0.04	110	0.19	0.07	0.06	j
4.00	0.7019	0.00	 12213	9286	0.76	8654	0.71	2320	1.00	 0.25	 0.27	 0.18	463	 1.00	0.05	0.05	
4.00	0.7019	0.10	12213	6528	0.53	6216	0.51	1752	0.76	0.27	0.28	0.14	342	0.74	0.05	0.06	j
4.00	0.7019	0.25	12213	4307	0.35	4376	0.36	1294	0.56	0.30	0.30	0.10 j	246	0.53	0.06	0.06	i
4.00	0.7019	0.50	12213	2638	0.22	2883	0.24	872	0.38	0.33	0.30	0.07	161	0.35	0.06	0.06	j
4.00	0.7019	1.00	12213 	1484	0.12	1699 	0.14	512	0.22	0.35	0.30	0.04	94	0.20	0.06	0.06	j
6.00	1.0528	0.00	12213	6430	0.53	5998	0.49	 1649	1.00	 0.26	0.27	 0.13	326	 1.00	0.05	0.05	 0
6.00	1.0528	0.10	12213	4653	0.38	4314	0.35	1246	0.76	0.27	0.29	0.10	246	0.75	0.05	0.06	j
6.00	1.0528	0.25	12213	3268	0.27	3249	0.27	960	0.58	0.29	0.30	0.07	185	0.57	0.06	0.06	i
6.00	1.0528	0.50	12213	2195	0.18	2313	0.19	684	0.41	0.31	0.30	0.05	130	0.40	0.06	0.06	j
6.00	1.0528	1.00	12213 	1331	0.11	1457	0.12	441	0.27	0.33	0.30	0.03	81	0.25	0.06	0.06	ļ
8.00	1.4038	0.00	 12213	4174	0.34	4077	0.33	 1149	 1.00	0.28	0.28	 0.09	227	 1.00	0.05	0.06	 (
8.00	1.4038	0.10	12213	3273	0.27	3108	0.25	897	0.78	0.27	0.29	0.07	i 177		0.05	0.06	i
8.00	1.4038	0.25	12213	2501	0.20	2447	0.20	723	0.63	0.29	0.30	0.05	141	0.62	0.06	0.06	j
8.00	1.4038	0.50	12213	1811	0.15	1854	0.15	551	0.48	0.30	0.30	0.04	105	0.46	0.06	0.06	j
8.00	1.4038	1.00	12213	1174	0.10	1249	0.10	378	0.33	0.32	0.30	10.03 I	1 70	0.31	i 0.06	0.06	i

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HE 550 M (Weak axis)	Sigma vield = 345 N/m	$n^2 : U/A = 64 : t = 40 mm$



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I Lambda Bar e/h NpL F0 F30 F60 (m) (kN) NRCE3) [NCCE3) [NCCE5] [NCCE5] [NCCE5] [NCCE5] [NCF30] [N/Nc N/NCF0, EC3] [N/NCF0, EC3] [N/NcF0, EC5] [N/NcF0, EC5] [N/NcF0, EC5] [N/Nc N/NcF0, EC3] [N/NcF0, EC3] [N/NcF0, EC5] [N/NcF0, EC5] [N/NcF0, EC5] [N/NcF0, EC5] [N/Nc 2.00 0.4390 [0.00 8442 7382 0.87 7382 0.87 10.0 0.34 0.34 0.29 111 1.00 0.06 2.00 0.4390 [0.00 8442 7375 0.45 4001 0.47 1367 0.34 0.34 0.34 0.29 1411 1.00 0.06 2.00 0.4390 [0.50 8442 12514 0.65 1531 0.66 1889 0.76 0.34 0.34 0.16 230 0.56 0.06 2.00 0.4390 [0.50 8442 1231 0.28 2610 0.31 1867 0.36 0.34 0.34 0.16 230 0.55 2.00 0.442 1285 <t< th=""><th></th><th></th><th></th><th></th><th>•••••</th><th></th><th></th><th></th><th></th><th></th><th></th><th></th><th></th><th></th><th></th><th></th><th></th><th></th></t<>					•••••													
(m) (km)	 _	ambda Barl	 e/h	 Nol	 		F0				F30					F60		
1 (kN) <t< th=""><th>m) </th><th></th><th> -/</th><th> "P' (kN) </th><th> N(EC3)</th><th>N(EC3)/Npl</th><th>IN(CEF)</th><th>N(CEF)/Npl</th><th>N(F30)</th><th>N/Nc</th><th>N/N(FO.EC3)</th><th>IN/N(FO.CEF)</th><th></th><th> N(F60)</th><th>IN/Nc</th><th>IN/N(E0.EC3)</th><th>IN/NCED CEET</th><th></th></t<>	m)		-/	"P' (kN)	 N(EC3)	N(EC3)/Npl	IN(CEF)	N(CEF)/Npl	N(F30)	N/Nc	N/N(FO.EC3)	IN/N(FO.CEF)		 N(F60)	IN/Nc	IN/N(E0.EC3)	IN/NCED CEET	
	į		ĺ	i i	(kN)	ļ	(kN)		(kN)	i				(kN)				
$ \begin{bmatrix} 2.00 \\ 0.4390 \\ 0.00 \\ 0.4390 \\ 0.10 \\ 0.4390 \\ 0.10 \\ 0.4390 \\ 0.50 \\ 0.4390 \\ 0.50 \\ 0.4390 \\ 0.50 \\ 0.4390 \\ 0.50 \\ 0.4390 \\ 0.50 \\ 0.4390 \\ 0.50 \\ 0.4390 \\ 0.50 \\ 0.4390 \\ 1.00 \\ 0.4390 \\ 1.00 \\ 0.4390 \\ 1.00 \\ 0.4390 \\ 1.00 \\ 0.4390 \\ 1.00 \\ 0.4390 \\ 1.00 \\ 0.4390 \\ 1.00 \\ 0.4390 \\ 1.00 \\ 0.4390 \\ 1.00 \\ 0.4390 \\ 1.00 \\ 0.4390 \\ 1.00 \\ 0.442 \\ 1 \\ 1264 \\ 0.15 \\ 1485 \\ 0.15 \\ 1485 \\ 0.16 \\ 1 \\ 1 \\ 1 \\ 1 \\ 1 \\ 1 \\ 1 \\ 1 \\ 1 \\$	••••		[·1								•
$ \begin{bmatrix} 2.00 \\ 0.4390 \\ 0.55 \\ 0.64 \\ 0.4390 \\ 0.55 \\ 0.4390 \\ 0.55 \\ 0.4390 \\ 0.55 \\ 0.4390 \\ 0.55 \\ 0.4390 \\ 0.55 \\ 0.4390 \\ 0.55 \\ 0.4390 \\ 0.55 \\ 0.4390 \\ 0.55 \\ 0.4390 \\ 0.55 \\ 0.442 \\ 0.4390 \\ 0.55 \\ 0.442 \\ 0.4390 \\ 0.55 \\ 0.442 \\ 0.4390 \\ 0.55 \\ 0.442 \\ 0.4390 \\ 0.55 \\ 0.442 \\ 0.4390 \\ 0.55 \\ 0.442 \\ 0.4390 \\ 0.55 \\ 0.442 \\ 0.4390 \\ 0.55 \\ 0.442 \\ 0.4390 \\ 0.56 \\ 0.4390 \\ 0.58 \\ 0.00 \\ 0.4390 \\ 0.10 \\ 0.4390 \\ 0.10 \\ 0.4390 \\ 0.10 \\ 0.4390 \\ 0.10 \\ 0.4390 \\ 0.10 \\ 0.4390 \\ 0.10 \\ 0.4390 \\ 0.10 \\ 0.442 \\ 0.10 \\ 0.4390 \\ 0.10 \\ 0.442 \\ 0.10 \\ 0.4390 \\ 0.10 \\ 0.442 \\ 0.10 \\ 0.4390 \\ 0.10 \\ 0.442 \\ 0.10 \\ 0.4390 \\ 0.10 \\ 0.442 \\ 0.10 \\ 0.442 \\ 0.10 \\ 0.442 \\ 0.44 \\ 0.15 \\ 0.44 \\ 0.44 \\ 0.43 \\ 0.66 \\ 0.58 \\ 0.58 \\ 0.32 \\ 0.33 \\ 0.33 \\ 0.33 \\ 0.33 \\ 0.10 \\ 0.11 \\ 0.41 \\$	00	0.4390	0.00	8442	7382	0.87	7382	0.87	2479	1.00	0.34	0.34	0.29	411	1.00	0.06	0.06	0.04
$ \begin{bmatrix} 2.00 \\ 0.4390 \\ 0.50 \\ 0.4390 \\ 0.50 \\ $	00	0.4390	0.10	8442	5514	0.65	5531	0.66	1889	0.76	0.34	0.34	0.22	315	0.77	0.06	0.06	0.03
$ \begin{bmatrix} 2.00 \\ 0.4390 \\ 0.50 \\ 0.4390 \\ 1.00 \\ 8422 \\ 1 \\ 1 \\ 1 \\ 1 \\ 1 \\ 1 \\ 1 \\ 1 \\ 1 \\$	00	0.4390	0.25	8442	3775	0.45	4001	0.47	1367	0.55	0.36	0.34	0.16	230	0.56	0.06	0.06	0.02
2.00 0.4390 1.00 8442 1264 0.15 1485 0.18 507 0.20 0.40 0.34 0.06 84 0.21 0.07 4.00 0.8780 0.00 8442 5337 0.64 5033 0.60 1570 1.00 0.29 0.31 0.18 269 1.00 0.05 4.00 0.8780 0.10 8442 3929 0.47 3414 0.43 1181 10.75 0.30 0.33 0.10 1188 269 1.00 0.05 4.00 0.8780 0.05 8442 2794 0.33 2754 0.33 1906 0.55 0.33 0.01 1188 0.59 0.06 4.00 0.8780 0.50 8442 1862 0.22 1938 0.23 647 0.41 0.35 0.33 0.07 1111 0.41 0.06 4.00 0.8780 1.00 8442 1321 0.38 3136 0.37 931 1.00 0.33 0.011 111 0.06 0.06 0.37	00	0.4390	0.50	8442	2331	0.28	2610	0.31	891	0.36	0.38	0.34	0.10	149	0.36	0.06	0.06	0.01
4.00 0.8780 0.00 8442 5387 0.64 5033 0.60 1570 1.00 0.29 0.31 0.18 269 1.00 0.055 4.00 0.8780 0.10 8442 3929 0.47 3614 0.43 1181 0.75 0.30 0.33 0.13 209 0.78 0.05 4.00 0.8780 0.25 8442 2794 0.33 2754 0.33 906 0.58 0.32 0.33 0.01 158 0.59 0.06 4.00 0.8780 1.00 8442 1862 0.22 1938 0.23 647 0.41 0.35 0.33 0.07 111 0.41 0.06 4.00 0.8780 1.00 8442 1362 0.22 1938 0.23 647 0.41 0.35 0.33 0.04 69 0.26 0.06 4.00 0.8780 1.00 8442 3221 0.38 3136 0.37 931 1.00 0.29 0.30 0.11 174 1.00 0.05 <td>00</td> <td>0.4390</td> <td>1.00</td> <td>8442</td> <td>1264</td> <td>0.15</td> <td> 1485</td> <td>0.18</td> <td> 507</td> <td>0.20</td> <td>0.40</td> <td>0.34</td> <td>0.06</td> <td>84</td> <td>0.21</td> <td>0.07</td> <td>0.06</td> <td>0.01</td>	00	0.4390	1.00	8442	1264	0.15	1485	0.18	507	0.20	0.40	0.34	0.06	84	0.21	0.07	0.06	0.01
4.00 0.8780 0.00 8442 5387 0.64 5033 0.60 1570 1.00 0.29 0.31 0.18 269 1.00 0.05 4.00 0.8780 0.10 8442 3929 0.47 3614 0.43 1181 0.75 0.30 0.33 0.13 209 0.78 0.05 4.00 0.8780 0.25 8442 2794 0.33 2754 0.33 906 0.58 0.32 0.33 0.01 158 0.59 0.06 4.00 0.8780 0.50 8442 1862 0.22 1938 0.23 647 0.41 0.35 0.33 0.07 111 0.41 0.06 4.00 0.8780 1.00 8442 13221 0.13 1218 0.14 407 0.26 0.37 0.33 0.04 69 0.26 0.06 4.00 0.8780 1.00 8442 3221 0.38 3136 0.37 931 1.00 0.29 0.30 0.11 177 0.79 0.05 <td>1</td> <td></td> <td></td> <td></td> <td></td> <td>1</td> <td>4</td> <td>1</td> <td></td> <td></td> <td>1</td> <td>ļ</td> <td>1 1</td> <td> </td> <td>1</td> <td>1</td> <td> </td> <td>1</td>	1					1	4	1			1	ļ	1 1		1	1		1
4.00 0.8780 0.10 8442 3929 0.47 3614 0.43 1 1181 0.75 0.30 0.33 0.13 1 209 0.78 0.05 4.00 0.8780 0.25 8442 2794 0.33 2754 0.33 906 0.58 0.32 0.33 0.01 1 158 0.59 0.06 4.00 0.8780 0.50 8442 1862 0.22 1938 0.23 647 0.41 0.35 0.33 0.07 1 111 0.41 0.06 4.00 0.8780 1.00 8442 1105 0.13 1218 0.14 407 0.26 0.37 0.33 0.04 69 0.26 0.06 4.00 0.8780 1.00 8442 3221 0.38 3136 0.37 931 1.00 0.29 0.30 0.11 174 1.00 0.05 0.31 0.32 1.00 1.11 174 1.00 0.05 0.31 0.32 0.07 1.00 1.17 1.41 0.17 1.41 <td>00 </td> <td>0.8780</td> <td>0.00</td> <td>8442</td> <td>5387</td> <td>0.64</td> <td>5033</td> <td>0.60</td> <td> 1570</td> <td>1.00</td> <td>0.29</td> <td>0.31</td> <td>0.18</td> <td>269</td> <td>1.00</td> <td>0.05</td> <td>0.05</td> <td>0.03</td>	00	0.8780	0.00	8442	5387	0.64	5033	0.60	1570	1.00	0.29	0.31	0.18	269	1.00	0.05	0.05	0.03
4.00 0.8780 0.25 8442 2794 0.33 2754 0.33 906 0.58 0.32 0.33 0.10 158 0.59 0.06 4.00 0.8780 0.50 8442 1862 0.22 1938 0.23 647 0.41 0.35 0.33 0.07 111 0.41 0.06 4.00 0.8780 1.00 8442 1105 0.13 1218 0.14 407 0.26 0.37 0.33 0.04 69 0.26 0.06 4.00 0.8780 1.00 8442 3221 0.38 3136 0.37 931 1.00 0.29 0.30 0.11 174 1.00 0.05 6.00 1.3170 0.00 8442 2562 0.30 2391 0.28 746 0.80 0.29 0.31 0.08 137 0.79 0.05 6.00 1.3170 0.25 8442 1978 0.23 1882 0.22 606 0.65 0.31 0.33 0.33 0.07 109 0.63 0	00	0.8780	0.10	8442	3929	0.47	3614	0.43	1181	0.75	0.30	0.33	0.13	209	0.78	0.05	0.06	0.02
4.00 0.8780 0.50 8442 1 1862 0.22 1938 0.23 647 0.41 0.35 0.33 0.07 111 0.41 0.06 4.00 0.8780 1.00 8442 1 1105 0.13 1218 0.14 407 0.26 0.37 0.33 0.07 111 0.41 0.06 6.00 1.3170 0.00 8442 3221 0.38 3136 0.37 931 1.00 0.29 0.30 0.11 111 1.00 0.05 6.00 6.00 1.3170 0.010 8442 2562 0.30 2391 0.28 746 0.80 0.29 0.31 10.08 1137 0.79 0.05 6.00 1.3170 0.25 8442 1978 0.23 1882 0.22 666 0.65 0.31 0.32 0.07 109 0.63 0.06 6.00 1.3170 0.50 8442 1438 0.17 1441 0.17 471 0.51 0.33 0.33 0.05 83 0.48	00	0.8780	0.25	8442	2794	0.33	2754	0.33	906	0.58	0.32	0.33	0.10	158	0.59	0.06	0.06	0.01
4.00 0.8780 1.00 8442 1105 0.13 1218 0.14 407 0.26 0.37 0.33 0.04 69 0.26 0.06 6.00 1.3170 0.00 8442 3221 0.38 3136 0.37 931 1.00 0.29 0.30 0.11 1 174 1.00 0.05 6.00 1.3170 0.10 8442 2562 0.30 2391 0.28 746 0.80 0.29 0.31 0.08 1137 0.79 0.05 6.00 1.3170 0.25 8442 1978 0.23 1882 0.22 606 0.65 0.31 0.32 0.07 109 0.63 0.06 6.00 1.3170 0.50 8442 1438 0.17 1441 0.17 471 0.51 0.33 0.05 83 0.64 0.06 6.00 1.3170 1.00 8442 936 0.11 990 0.12 323 0.35 0.33 0.03 1.56 0.32 0.06 8.00 </td <td>00 </td> <td>0.8780</td> <td>0.50</td> <td> 8442 </td> <td> 1862</td> <td>0.22</td> <td>1938</td> <td>0.23</td> <td>647</td> <td>0.41</td> <td>0.35</td> <td>0.33</td> <td>0.07</td> <td> 111</td> <td>0.41</td> <td>0.06</td> <td>0.06</td> <td>0.01</td>	00	0.8780	0.50	8442	1862	0.22	1938	0.23	647	0.41	0.35	0.33	0.07	111	0.41	0.06	0.06	0.01
6.00 1.3170 0.00 8442 3221 0.38 3136 0.37 931 1.00 0.29 0.30 0.11 1 174 1.00 0.05 6.00 1.3170 0.10 8442 2562 0.30 2391 0.28 746 0.80 0.29 0.31 0.08 1 177 0.79 0.05 6.00 1.3170 0.25 8442 1978 0.23 1882 0.22 606 0.65 0.31 0.32 0.07 109 0.63 0.06 6.00 1.3170 0.50 8442 1438 0.17 1441 0.17 471 0.51 0.33 0.33 0.05 83 0.48 0.06 6.00 1.3170 1.00 8442 936 0.11 990 0.12 323 0.35 0.35 0.33 0.03 1.56 0.32 0.06 8.00 1.7560 0.00 8442 936 0.11 990 0.23 845 0.29 0.30 0.31 0.62 0.32	00	0.8780	1.00	8442	1105	0.13	1218	0.14	407	0.26	0.37	0.33	0.04	69	0.26	0.06	0.06	jo.oo
6.00 1.3170 0.00 8442 3221 0.38 3136 0.37 931 1.00 0.29 0.30 0.11 174 1.00 0.05 6.00 1.3170 0.10 8442 2562 0.30 2391 0.28 746 0.80 0.29 0.31 0.08 137 0.79 0.05 6.00 1.3170 0.25 8442 1978 0.23 1882 0.22 606 0.65 0.31 0.32 0.07 109 0.63 0.06 6.00 1.3170 0.50 8442 1438 0.17 1441 0.17 471 0.51 0.33 0.33 0.33 0.33 0.33 0.33 0.33 0.33 0.33 0.33 0.33 0.33 0.33 0.48 0.66 0.66 0.33 0.33 0.33 0.33 0.33 0.33 0.48 0.66 0.66 0.60 1.37 0.20 1497 0.20 1197 <td></td> <td> </td> <td></td> <td>1 1</td> <td>1</td> <td> </td> <td> </td> <td> </td> <td> </td> <td>1</td> <td>1</td> <td>1</td> <td>i i</td> <td>I</td> <td>Ì</td> <td>Ì</td> <td>i</td> <td>i</td>				1 1	1					1	1	1	i i	I	Ì	Ì	i	i
6.00 1.3170 0.10 8442 2562 0.30 2391 0.28 746 0.80 0.29 0.31 0.08 137 0.79 0.05 6.00 1.3170 0.25 8442 1978 0.23 1882 0.22 606 0.65 0.31 0.32 0.07 109 0.63 0.06 6.00 1.3170 0.50 842 1438 0.17 441 0.17 471 0.51 0.33 0.03 0.05 83 0.48 0.06 6.00 1.3170 0.00 8442 936 0.11 990 0.12 323 0.35 0.33 0.03 56 0.32 0.06 <td>00 </td> <td>1.3170 </td> <td>0.00</td> <td>8442 </td> <td>3221</td> <td>0.38</td> <td>3136</td> <td>0.37</td> <td>931</td> <td> 1.00</td> <td>0.29</td> <td>0.30</td> <td>0.11</td> <td>174</td> <td>1.00</td> <td>0.05</td> <td>0.06</td> <td>0.02</td>	00	1.3170	0.00	8442	3221	0.38	3136	0.37	931	1.00	0.29	0.30	0.11	174	1.00	0.05	0.06	0.02
[6.00] 1.3170 [0.25] 8442 [1978] 0.23 1882 0.22 [] 606 [] 0.31 0.32 [] [] 0.06 [6.00] 1.3170 [] 0.50 8442 [] 1438 0.17 1441 0.17 [] 471 [] 0.33 [] 0.33 [] 0.33 [] 0.33 [] 0.33 [] 0.33 [] 0.33 [] 0.64 [] 0.06 [] 0.33 [] 0.33 [] 0.33 [] 0.33 [] 0.33 [] 0.33 [] 0.33 [] 0.33 [] 0.33 [] 0.66 [] 0.66 [] [] [] [] 11 990 0.12 [] 323 [] 0.35 0.33 [] 0.33 [] 0.66 0.32 0.06 [] [] [] 0.11 [] 0.12 [] 323 [] 0.35 0.33 [] 0.33 [] 0.33 [] 0.33	00	1.3170	0.10	8442	2562	0.30	2391	0.28	746	0.80	0.29	0.31	0.08	137	0.79	0.05		0.01
6.00 1.3170 0.50 8442 1438 0.17 1441 0.17 471 0.51 0.33 0.05 83 0.48 0.06 6.00 1.3170 1.00 8442 936 0.11 990 0.12 323 0.35 0.33 0.05 83 0.48 0.06 6.00 1.3170 1.00 8442 936 0.11 990 0.12 323 0.35 0.35 0.33 0.03 56 0.32 0.06 <t< td=""><td>00 </td><td>1.3170</td><td>0.25</td><td>8442</td><td> 1978</td><td>0.23</td><td>1882</td><td>0.22</td><td>606</td><td>0.65</td><td>0.31</td><td>0.32</td><td>0.07</td><td>109</td><td>0.63</td><td>0.06</td><td>0.06</td><td>0.01</td></t<>	00	1.3170	0.25	8442	1978	0.23	1882	0.22	606	0.65	0.31	0.32	0.07	109	0.63	0.06	0.06	0.01
6.00 1.3170 1.00 8442 936 0.11 990 0.12 323 0.35 0.35 0.33 0.03 56 0.32 0.06	00	1.3170	0.50	8442	1438	0.17	1441	0.17	471	0.51	0.33	0.33	0.05	83	0.48	0.06	İ 0.06	10.00
	00	1.3170	1.00	8442	936	0.11	990	0.12	323	0.35	0.35	0.33	jo.03 j	56	0.32	0.06	I 0.06	10.00
8.00 1.7560 0.00 8442 2006 0.24 1957 0.23 610 1.00 0.30 0.31 0.07 116 1.00 0.06 8.00 1.7560 0.10 8442 1717 0.20 1643 0.19 495 0.81 0.29 0.30 0.05 95 0.82 0.06 8.00 1.7560 0.25 8442 1421 0.17 329 0.16 424 0.70 0.30 0.32 0.05 95 0.82 0.06 8.00 1.7560 0.25 8442 1421 0.17 329 0.16 424 0.70 0.30 0.32 0.05 79 0.68 0.06 8.00 1.7560 0.50 8442 1112 0.13 0.76 0.13 339 0.56 0.31 0.32 0.04 64 0.55 0.06 8.00 1.7560 1.00 8442		1			1	1	ĺ	i i	Í.	i	Ì	İ	i i	i	i	Ì	i	1
8.00 1.7560 0.10 8442 1717 0.20 1643 0.19 495 0.81 0.29 0.30 0.05 95 0.82 0.06 8.00 1.7560 0.25 8442 1421 0.17 1329 0.16 424 0.70 0.30 0.32 0.05 79 0.68 0.06 8.00 1.7560 0.50 8442 1112 0.13 1076 0.13 339 0.56 0.31 0.32 0.04 64 0.55 0.06 8.00 1.7560 1.00 8442 782 0.09 796 0.09 260 0.43 0.33 0.33 0.03 46 0.40 0.06	00	1.7560	0.00	8442	2006	0.24	1957	0.23	610	1.00	0.30	0.31	10.07	116	11.00	0.06	0.06	10.01
8.00 1.7560 0.25 8442 1421 0.17 1329 0.16 424 0.70 0.30 0.32 0.05 79 0.68 0.06 8.00 1.7560 0.50 8442 1112 0.13 1076 0.13 339 0.56 0.31 0.32 0.04 64 0.55 0.06 8.00 1.7560 1.00 8442 782 0.09 796 0.09 260 0.43 0.33 0.33 0.03 46 0.40 0.06	00	1.7560	0.10	8442	1717	0.20	1643	0.19	495	jo.81	0.29	0.30	0.05	95	10.82	0.06	0.06	10.01
8.00 1.7560 0.50 8442 1112 0.13 0.13 339 0.56 0.31 0.32 0.04 64 0.55 0.06 8.00 1.7560 1.00 8442 782 0.09 796 0.09 260 0.43 0.33 0.33 0.03 46 0.40 0.06	00	1.7560	0.25	8442	1421	0.17	1329	0.16	424	0.70	0.30	0.32	10.05	79	0.68	0.06	0.06	10.00
8.00 1.7560 1.00 8442 782 0.09 796 0.09 260 0.43 0.33 0.33 0.03 46 0.40 0.06	00	1.7560	0.50	8442	1112	0.13	1076	0.13	339	0.56	0.31	0.32	10.04	64	0.55	0.06	0.06	10.00
	00	1.7560	1.00	8442	782	0.09	796	0.09	260	0.43	0.33	0.33	0.03	46	10.40	0.06	0.06	10.00
	• • • • • •															••••••••••••		

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HD 210X210X198 (Weak axis) Sigma yield = 335 N/mm2; U/A = 54; t = 45 mm



HD 210X210X198 (Weak axis) Fe 510



	 			[]		FO				F30					F60	
l ()	iraun coa sa r	∣e/n I				1440553			1.1.1.1.				114/5603		IN/N/ E0 EC3)	
(m)	1	!	(KM) 	M(EC3) (EN)	1 (EC2)/MPL	THUER)	INCCEFJ/NPC	M(F3U)	INTAC	W/W(FU,ECS)	W/W(FU,CEF)	luvubr I	[]#(F00) [] (FN)	1		/ (0;00)
	 	1 	 	(KM) 	1	(KN) 	 	(KM) 	 	 	1 	 		 	 	
2.00	0.3658	10.00	، 9347	8419	0.90	8305	0.89	2651	1.00	0.31	0.32	0.28	461	1.00	0.05	0.04
2.00	0.3658	0.10	9347	6359	0.68	6324	0.68	2049	0.77	0.32	0.32		357	0.77	0.06	0.0
2.00	0.3658	0.25	9347	4354	0.47	4659	0.50	1472	0.56	0.34	0.32	0.15	259	0.56	0.06	0.0
2.00	0.3658	0.50	9347	2656	0.28	2996	0.32	946	0.36	0.36	0.32	0.10	168	0.36	0.06	0.0
2.00	0.3658	1.00	9347	1420	0.15	1691	0.18	534	0.20	0.38	0.32	0.05	94	0.20	0.07	0.00
	1	Ì	Ì	ÎÎ.	Ì	ĺ	1		Ì	Ì	1		11	1	1	
4.00	0.7316	0.00	9347	6854	0.73	6000	0.64	1894	1.00	0.28	0.32	0.20	321	1.00	0.05	0.0
4.00	0.7316	0.10	9347	4961	0.53	4472	0.48	1361	0.72	0.27	0.30	0.14	248	0.77	0.05	0.00
4.00	0.7316	0.25	9347	3453	0.37	3332	0.36	1026	0.54	0.30	0.31	0.10	187	0.58	0.05	0.0
4.00	0.7316	0.50	9347	2231	0.24	2325	0.25	725	0.38	0.33	0.31	0.07	129	0.40	0.06	0.00
4.00	0.7316	1.00	9347	1283	0.14	1426	0.15	445	0.23	0.35	0.31	0.04	79	0.25	0.06	0.0
	1	1	1	11		1	1		1	1	ł	l	II			
6.00	1.0974	0.00	9347	4647	0.50	4029	0.43	1111	1.00	0.24	0.28	0.11	216	1.00	0.05	0.0
6.00	1.0974	0.10	9347	3520	0.38	3047	0.33	905	0.81	0.26	0.30	0.09	168	0.78	0.05	0.0
6.00	1.0974	0.25	9347	2603	0.28	2401	0.26	731	0.66	0.28	0.30	0.07	134	0.62	0.05	0.00
6.00	1.0974	0.50	9347	1809	0.19	1797	0.19	547	0.49	0.30	0.30	0.05	100	0.46	0.06	0.0
6.00	1.0974	1.00	9347	1127	0.12	1188	0.13	366	0.33	0.32	0.31	0.03	66	10.31	0.06	0.00
			ļ	[]			ļ			1	1					1
8.00	1.4632	0.00	9347	3017	0.32	2713	0.29	725	1.00	0.24	0.27	0.07	147	11.00	0.05	
8.00	1.4632	0.10	9347	2466	0.26	2147	0.23	622	0.86	0.25	0.29	0.06	120	10.81	0.05	
8.00	1.4632	10.25	9347	1952	0.21	1732	0.19	11 522	10.72	0.27	0.30	10.05	11 100	10.67		1 0.00
8.00	1.4632	0.50	9347	1457	0.16	1386	0.15	418	0.58	0.29	1 0.30	0.04	11 78	10.55		0.00

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ן (m)	Lambda Bar	 e/h 	 Npl (kN)	F0				F30					F60 				
				N(EC3)	N(EC3)/Npl	N(CEF)	N(CEF)/Npl	N(F30)	N/Nc	N/N(F0,EC3)	N/N(FO,CEF)	N/Npl	N(F60)	N/Nc	N/N(F0,EC3)	N/N(FO,CEF)	N/Np
				(kN)	ļ	(kN)		(kN)					(kN)	!	1	1	ļ
2 00	0 3480		 14070	12720	 0 00	12068	0 92	7700	11 00	 0.61	0 4 0	 0 55	870	11 00		0.07	10.06
2 00	0.3407	10.00		1 9621	8A.0	9776	0.72	1 5073	10 77		0.50	0.22	1 672	10 76			
2.00	0.3489	10.25	14070		0.00		0.07	1 4251	10.55	0.62	0.59	0.72		10.56			
2.00	0.3489	10.50		4041	0.29	4631	0.33	1 2770	10.36	0.69	0.60	0.30 0 10	1 311	10.35	1 0.08		10.02
2.00	0.3489	1.00	14070	2159	0.15	2587	0.18	1 1671	10.21		0.65		1 175	10.20	0.08		10.01
								1	1	1					1		1
4.00	0.6978	0.00	14070	110564	0.75	9985	0.71	4893	11.00	, I 0.46	0.49	0.34	685	 1.00	, I 0.06	0.07	10.04
4.00	0.6978	0.10	14070	7661	0.54	7245	0.51	3808	0.78	0.50	0.53	0.27	500	0.73	0.07	0.07	10.03
4.00	0.6978	0.25	14070	5323	0.38	5348	0.38	2859	0.58	0.54	0.53	0.20	368	0.54	0.07	0.07	0.02
4.00	0.6978	0.50	14070	3433	0.24	3668	0.26	1961	0.40	0.57	0.53	0.13	248	0.36	0.07	0.07	0.01
4.00	0.6978	1.00	14070	1963	0.14	2205	0.16	1199	0.24	0.61	0.54	0.08	149	0.22	0.08	0.07	jo.01
		ĺ		i i	Ì			Ì	Ì	j		i	ii	i	i	I	i
5.00	1.0467	0.00	14070	7405	0.53	6990	0.50	3203	1.00	0.43	0.46	0.22	486	1.00	0.07	0.07	10.03
5.00	1.0467	0.10	14070	5563	0.40	5155	0.37	2547	0.80	0.46	0.49	0.18	357	0.73	0.06	0.07	0.02
5.00	1.0467	0.25	14070	4084	0.29	3959	0.28	2011	0.63	0.49	0.51	0.14	273	0.56	0.07	0.07	10.01
5.00	1.0467	0.50	14070	2818	0.20	2878	0.20	1487	0.46	0.53	0.52	0.10	198	0.41	0.07	0.07	0.01
5.00	1.0467	1.00	14070	1738	0.12	1864	0.13	980	0.31	0.56	0.53	0.06	127	0.26	0.07	0.07	0.00
l				i I				1	1	I	1				l .	I	1
B.OO	1.3956	0.00	14070	4893	0.35	4835	0.34	2181	1.00	0.45	0.45	0.15	345	1.00	0.07	0.07	0.02
8.00	1.3956	0.10	14070	3956	0.28	3733	0.27	1774	0.81	0.45	0.48	0.12	263	0.76	0.07	0.07	0.01
3.00	1.3956	0.25	14070	3099	0.22	2969	0.21	1467	0.67	0.47	0.49	0.10	209	0.61	0.07	0.07	0.01
8.00	1.3956	0.50	14070	2289	0.16	2302	0.16	1169	0.54	0.51	0.51	0.08	159	0.46	0.07	0.07	0.01
8.00	1.3956	1.00	14070	1515	0.11	1585	0.11	819	0.38	0.54	0.52	0.05	108	0.31	0.07	0.07	0.00

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HD 260X260X329 (Weak axis) Sigma yield = 335 N/mm2; U/A = 41 ; t = 60 mm




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				11		FO		1		F30					F60		ļ
	L amb da Bar	e/h	Npl	}	····		•••••		•	·····	·····						· · · · · · · · · ·
(m) 		!	(KN)	N(EC3)	N(EC3)/Npl 	IN(CEF)	IN(CEF)/NPL	[N(F3U)	N/NC	N/N(FU,EC3)	ĮN/N(FU,CEF) 1	N/Npl	N(F6U)	N/NC	N/N(FU,EC3)	N/N(FU,CEF)	N/Npl
 	 -1	 		[] (KN)	 	(KN)	 	(KN)		1	 		(KN)	1	1	1	! !
12 00	1 0 3080	10.00	112060		0.02	144427			11 00	0.79		0 75	4.7	11 00	0.04	1 0.04	
12.00	0.3080	10.00	112060		0.72	1 9550	0.72	4247 7774	10.70		0.30	0.37	507	10.70		1 0.06	
12.00		10.25	112060	0420 5768	0.70 0.48	1 6262	0.71	1 234/	10.19	0.40	0.39 0.38	10.27	1 744	10.70			10.04
12.00	0.3080	10.25	112060	1 3/88	1 0.40	1 4014	0.52	2.304	10.30	1 0.41	0.30	10 12	27/	10.36			10.03
12.00		11 00	112060	18/3	0.27	2220	0.55	10 1401 11 800	10.21	0.42 0.48		10.12	1 136	10.30			10.01
1	1 0.5000	1.00	1		0.15	2220	0.10	0,0	10.21	0.40 	0.40	10.07		10.21	1 0.07	1 0.00	10:01
14.00	I 0.6160	10.00	112060		I 0.80	I I 8654	I 0.72	11 3020	I 1_00_	 031	I 0.35	I I0 25 I	I 500	1	1 0.05	I 0.06	
14.00	0.6160	10.10	12060		0.58	1 6536	0.72	1 2285	10 76			10 18	370	10 76			
14.00	0.6160	10.25	112060	4823	0.40	1 4777	0.40		10.56	0.35	0.36	10.14	280	10.56			
14.00	0.6160	10.50	112060	3044	0.25	3243	0.27	1 1156	10.38	0.38	0.36	10.09	1 189	10.38	0.06	0.06	10.01
4.00	0.6160	11.00	112060	1704	0.14	1931	0.16	688	0.23	0.40	0.36	10.05	112	10.22	0.07	0.06	10.00
i	1	1													1		
6.00	0.9240	0.00	12060	7339	0.61	6307	0.52	2087	1.00	0.28	0.33	0.17	355	1.00	0.05	, I 0.06	10.02
6.00	0.9240	0.10	12060	5376	0.45	4698	0.39	1605	0.77	. 0.30	0.34	0.13	274	0.77	0.05	0.06	0.02
6.00	0.9240	0.25	12060	3843	0.32	3598	0.30	1255	0.60	0.33	0.35	0.10	211	0.59	0.05	0.06	jo.01 j
6.00	0.9240	0.50	12060	2571	0.21	2596	0.22	925	0.44	0.36	0.36	0.07	153	0.43	0.06	0.06	j0.01 j
6.00	0.9240	1.00	12060	1540	0.13	1665	0.14	593	0.28	0.39	0.36	0.04	97	0.27	0.06	0.06	jo.o0 j
1	1	1	i i	1	ĺ	Ì			İ	ĺ	ĺ	1	Í	İ	İ	ĺ	i i
8.00	1.2320	0.00	12060	5096	0.42	4496	0.37	1457	1.00	0.29	0.32	0.12	258	1.00	0.05	0.06	0.02
8.00	1.2320	0.10	12060	3969	0.33	3453	0.29	1160	0.80	0.29	0.34	0.09	202	0.78	0.05	0.06	[0.01]
8.00	1.2320	0.25	12060	3009	0.25	2753	0.23	943	0.65	0.31	0.34	0.07	163	0.63	0.05	0.06	0.01
8.00	1.2320	0.50	12060	2145	0.18	2098	0.17	732	0.50	0.34	0.35	0.06	123	0.48	0.06	0.06	0.01
8.00	1.2320	1.00	12060	1373	0.11	1430	0.12	510	0.35	0.37	0.36	0.04	83	0.32	0.06	0.06	0.00
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HD 310X310X283 (Weak axis) Sigma yield = 335 N/mm2 ; U/A = 54 ; t = 44 mm



T.36

 l	 L ambd a Bar	 e/h	 Npl	 		F0		 		F30			 !		F60	
(m)	1	i i	(kN)	N(EC3)	N(EC3)/Npl	IN(CEF)	N(CEF)/Npl	 N(F30)	N/Nc	IN/N(FO.EC3)	N/N(FO.CEF)	N/Npl	 N(F60)	IN/Nc		IN/NCEO.CE
i	i	i	i	(kN)	1	(kN)		(kN)				1	ll (kn)	1		
							[]		j		, 					
2.00	0.3003	0.00	16013	14743	0.92	14846	0.93	9425	1.00	0.64		0.58	1012	1.00	0.07	, I 0.0
2.00	0.3003	0.10	16013	11202	0.70	11512	0.72	7180	0.76	0.64	0.62	0.44	784	0.77	0.07	0.0
2.00	0.3003	0.25	16013	7667	0.48	8328	0.52	5088	0.54	0.66		0.31	567	0.56	0.07	0.0
2.00	0.3003	0.50	16013	4624	0.29	5355	0.33	3272	0.35	0.71	0.61	0.20	359	0.35	0.08	, 0.0
2.00	0.3003	1.00	16013	2440	0.15	2932	0.18	1927	0.20	0.79	0.66	0.12	204	0.20	80.0	, j 0.0
ł			1	11	1	1	1			l	1	i i	Í	i	i	i
4.00	0.6006	0.00	16013	12885	0.80	11663	0.73	6004	1.00	0.47	0.51	0.37	804	1.00	0.06	j 0.0
4.00	0.6006	0.10	16013	9398	0.59	8763	0.55	4685	0.78	0.50	0.53	0.29	599	0.74	0.06	j 0.0
4.00	0.6006	0.25	16013	6444	0.40	6437	0.40	3441	0.57	0.53	0.53	0.21	437	0.54	0.07	0.0
4.00	0.6006	0.50	16013	4051	0.25	4348	0.27	2325	0.39	0.57	0.53	0.14	292	0.36	0.07	0.0
4.00	0.6006	1.00	16013	2261	0.14	2588	0.16	1413	0.24	0.62	0.55	0.08	172	0.21	0.08	0.0
									l	1	1	1 1	1			1
6.00	0.9009	0.00	16013	9975	0.62	8556	0.53	3988	1.00	0.40	0.47	0.24	592	1.00	0.06	0.0
6.00	0.9009	0.10	16013	7276	0.45	6395	0.40	3191	0.80	0.44	0.50	0.19	439	0.74	0.06	0.0
6.00	0.9009	0.25	16013	5174	0.32	4898	0.31	2487	0.62	0.48	0.51	0.15	333	0.56	0.06	0.0
6.00	0.9009	0.50	16013	3446	0.22	3499	0.22	1808	0.45	0.52	0.52	0.11	237	0.40	0.07	0.0
6.00	0.9009	11.00	16013	2053	0.13	2222	0.14	1166	0.29	0.57	0.52	0.07	150	0.25	0.07	0.0
						ļ		1	ł		l			1	1	
8.00	1.2012	0.00	16013	7006	0.44	6212	0.39	2717	1.00	0.39	0.44	0.16	435	1.00	0.06	0.0
8.00	1.2012	0.10	16013	5417	0.34	4701	0.29	2262	0.83	0.42	0.48	0.14	327	0.75	0.06	0.0
8.00	1.2012	0.25	16013	4077	0.25	3748	0.23	1870	0.69	0.46	0.50	0.11	258	0.59	0.06	0.0
8.00	1.2012	10.50	16013	2893	0.18	2856	0.18	1425	0.52	0.49	0.50	0.08	194	0.45	0.07	0.0
8.00	1.2012	11.00	16013	1836	0.11	1924	0.12	977	0.36	0.53	0.51	0.06	130	0.30	0.07	i 0.0

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D 310X310X375 (Weak axis) Sigma yield = 335 N	N/mm2;U/A=42;t=57mm
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T.38

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	 ambda Bar	 e/h	 Nol	 		F0		 		F30					F60		
(m)	 	•/"	(kN)	[N(EC3)	N(EC3)/Npl	IN(CEF)	N(CEF)/Npl	 N(F30)	N/Nc	N/N(F0.EC3)	N/N(FO,CEF)	N/Npl	1 N(F60) N/Nc	N/N(FO,EC3)	N/N(F0,CEF)	N/Npl
Ì		İ		(kN)		(kN)		(kN)					(kN)	1			i .
j									Ì				j	-			· [
2.00	0.2844	0.00	20721	19102	0.92	19573	0.94	15052	1.00	0.79	0.77	0.72	1920	1.00	0.10	0.10	0.09
2.00	0.2844	0.10	20721	14570	0.70	15178	0.73	11672	0.78	0.80	0.77	0.56	1462	0.76	0.10	0.10	0.07
2.00	0.2844	0.25	20721	9997	0.48	11022	0.53	8281	0.55	0.83	0.75	0.39	1040	0.54	0.10	0.09	0.05
2.00	0.2844	0.50	20721	6075	0.29	7060	0.34	5140	0.34	0.85	0.73	0.24	655	0.34	0.11	0.09	0.03
2.00	0.2844	1.00	20721	3207	0.15	3915	0.19	3010	0.20	0.94	0.77	0.14	361	0.19	0.11	0.09	0.01
1	l					1	,	1			1		1	T	1		1
4.00	0.5689	0.00	20721	16899	0.82	16466	0.79	10096	1.00	0.60	0.61	0.48	1602	1.00	0.09	0.10	0.07
4.00	0.5689	0.10	20721	12392	0.60	12073	0.58	7968	0.79	0.64	0.66	0.38	1160	0.72	0.09	0.10	0.05
4.00	0.5689	0.25	20721	8526	0.41	8836	0.43	5892	0.58	0.69	0.67	0.28	822	0.51	0.10	0.09	0.03
4.00	0.5689	0.50	20721	5372	0.26	5880	0.28	3921	0.39	0.73	0.67	0.18	539	0.34	0.10	0.09	0.02
4.00	0.5689	1.00	20721	2987	0.14	3465	0.17	2412	0.24	0.81	0.70	0.11	316	0.20	0.11	0.09	0.01
1	l									1			1	1	1		1
6.00	0.8533	0.00	20721	13458	0.65	12709	0.61	6965	1.00	0.52	0.55	0.33	1202	1.00	0.09	0.09	0.05
6.00	0.8533	0.10	20721	9812	0.47	9112	0.44	5587	0.80	0.57	0.61	0.26	871	0.72	0.09	0.10	0.04
6.00	0.8533	0.25	20721	6959	0.34	6859	0.33	4329	0.62	0.62	0.63	0.20	641	0.53	0.09	0.09	0.03
6.00	0.8533	0.50	20721	4620	0.22	4834	0.23	3081	0.44	0.67	0.64	0.14	446	0.37	0.10	0.09	0.02
6.00	0.8533	1.00	20721	2731	0.13	3026	0.15	1947	0.28	0.71	0.64	0.09	277	0.23	0.10	0.09	0.01
1	ľ				1		I			1	I		1	1	1	1	1
8.00	1.1377	0.00	20721	9718	0.47	9356	0.45	4901	1.00	0.50	0.52	0.23	891	[1.00	0.09	0.10	0.04
8.00	1.1377	0.10	20721	7450	0.36	6969	0.34	4035	0.82	0.54	0.58	0.19	661	0.74	0.09	0.09	0.03
8.00	1.1377	0.25	20721	5563	0.27	5432	0.26	3258	0.66	0.59	0.60	0.15	509	0.57	0.09	0.09	0.02
8.00	1.1377	0.50	20721	3911	0.19	3988	0.19	2468	0.50	0.63	0.62	0.11	374	0.42	0.10	0.09	0.01
8.00	1.1377	1.00	20721	2460	0.12	2636	0.13	1661	0.34	0.68	0.63	0.08	243	0.27	0.10	0.09	0.01
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HD 310X310X500 (Weak axis) Sigma yield = 325 N/mm2; U/A = 33; t = 75 mm





1	l embrie Bar	 h				F0		 		F30			 		F60		
(m)		e/n 	(kN) (kN)	N(EC3) (kN)	N(EC3)/Npl	N(CEF) (kn)	N(CEF)/Npl	N(F30) (kN)	N/Nc 	N/N(FO,EC3) 	N/N(FO,CEF)	N/Npl	N(F60) (kN)	N/Nc 	N/N(FO,EC3) 	N/N(FO,CEF)) N/N
						1	 		 	, 		 				 	·
2.00	0.2498	0.00	13800	12940	0.94	13099	0.95	4668	1.00	0.36	0.36	0.33	757	1.00	0.06	0.06	10.0
2.00	. 0.2498	0.10	13800	9940	0.72	10157	0.74	3620	0.78	0.36	0.36	0.26	597	0.79	0.06	0.06	10.0
2.00	0.2498	0.25	13800	6853	0.50	7498	0.54	2551	0.55	0.37	0.34	0.18	431	0.57	0.06	0.06	jo.0
2.00	0.2498	0.50	13800	4154	0.30	4796	0.35	1667	0.36	0.40	0.35	0.12	277	0.37	0.07	0.06	jo.0
2.00	0.2498	1.00	13800	2183	0.16	2614	0.19	970	0.21	0.44	0.37	0.07	177	0.23	0.08	0.07	10.0
	ĺ	ĺ		1		İ		i i	Ì	Ì	ĺ	İ	Í Í	Ì	1		1
4.00	0.4996	0.00	13800	11786	0.85	11073	0.80	3666	1.00	0.31	0.33	0.26	620	1.00	0.05	0.06	10.0
4.00	0.4996	0.10	13800	8752	0.63	8380	0.61	2715	0.74	0.31	0.32	0.19	470	0.76	0.05	0.06	0.0
4.00	0.4996	0.25	13800	6020	0.44	6124	0.44	2001	0.55	0.33	0.33	0.14	346	0.56	0.06	0.06	0.0
4.00	0.4996	0.50	13800	3757	0.27	4035	0.29	1340	0.37	0.36	0.33	0.09	230	0.37	0.06	0.06	0.0
4.00	0.4996	1.00	13800	2059	0.15	2354	0.17	782	0.21	0.38	0.33	0.05	133	0.21	0.06	0.06	0.0
						1				1					1		1
6.00	0.7494	0.00	13800	10031	0.73	8645	0.63	2661	1.00	0.27	0.31	0.19	478	1.00	0.05	0.06	0.0
6.00	0.7494	0.10	13800	7275	0.53	6444	0.47	2057	0.77	0.28	0.32	0.14	365	0.76	0.05	0.06	0.0
6.00	0.7494	0.25	13800	5084	0.37	4850	0.35	1549	0.58	0.30	0.32	0.11	276	0.58	0.05	0.06	0.0
6.00	0.7494	0.50	13800	3306	0.24	3402	0.25	1111	0.42	0.34	0.33	0.08	193	0.40	0.06	0.06	0.0
6.00	0.7494	1.00	13800	1910	0.14	2097	0.15	685	0.26	0.36	0.33	0.04	118	0.25	0.06	0.06	0.0
						1	ļ			1	1				1		
B.00	0.9992	0.00	13800	7768	0.56	6616	0.48	2021	1.00	0.26	0.31	0.14	364	1.00	0.05	0.06	10.0
8.00	0.9992	0.10	13800	5777	0.42	4978	0.36	1552	0.77	0.27	0.31	0.11	283	0.78	0.05	0.06	10.0
8.00	0.9992	0.25	13800	4201	0.30	3880	0.28	1239	0.61	0.29	0.32	0.08	221	0.61	0.05	0.06	0.0
8.00	0.9992	0.50	13800	2870	0.21	2835	0.21	919	0.45	0.32	0.32	0.06	162	0.44	0.06	0.06	10.0
8.00	0.9992	1.00	13800	1750	0.13	1845	0.13	603	0.30	0.34	0.33	0.04	105	0.29	0.06	0.06	0.0

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HD 400X400X314 (Weak axis) Sigma yield = 345 N/mm2 ; U/A = 58 ; t = 40 mm



T.42

ι	 Lambda Bar	 e/h	 Npt	 		FO 		 		F30		 			F60		
(m)	ł	1	(kN)	N(EC3)	N(EC3)/Npl	N(CEF)	N(CEF)/Npl	 N(F30)	N/Nc	N/N(F0,EC3)	N/N(FO,CEF)	N/Npl	N(F60)	N/Nc	N/N(F0,EC3)	N/N(FO,CEF)) N/Npl
		ļ	[]	(kN)	ļ	(kN)		(kN)	l	1		i i	(kN)	İ	İ	İ	İ
2.00	0.2218	10.00	27216		 1 0 94	 26100	 0 06		 1_00	 1 01	0.00						
2.00	0.2218	10.10	127216	119690		120543	0.75	203/1	10.70	1.01	0.99	10.70	1 3572	10.77	0.13	0.13	10.12
2.00	0.2218	10.25	27216	113594	0.50	115068	0.75	1120341	10.77			10.74	1 2002		0.13	0.12	10.09
2.00	0.2218	10.50	27216	8250	0.30 0.30	1 9750	0.35	05/5	10.30	1.07	0.70	10.25	1 1120	10.55		0.12	10.06
2.00	0.2218	1.00	27216	4340	0.16	5352	0.20	5300	0.20	1.13	0.98	0.19	655	0.20	0.14	0.12	10.04
						1			1			i i	i	İ	i	i	i
4.00	0.4436	0.00	27216	23622	0.87	22738	0.84	17221	1.00	0.73	0.76	0.63	2778	1.00	0.12	0.12	0.10
4.00	0.4436	0.10	27216	17674	0.65	17357	0.64	13403	0.78	0.76	0.77	0.49	2036	0.73	0.12	0.12	0.07
4.00	0.4436	0.25	27216	12163	0.45	12659	0.47	9700	0.56	0.80	0.77	0.35	1485	0.53	0.12	0.12	0.05
4.00	0.4436	0.50	27216	7570	0.28	8345	0.31	6635	0.39	0.88	0.80	0.24	967	0.35	0.13	0.12	0.03
4.00	0.4436	1.00	27216	4129	0.15	4854	0.18	4238	0.25	1.03	0.87	0.15	555	0.20	0.13	0.11	0.02
6.00	0.6654	10.00	 27216	 20791	0.76	 18544	 8A 0	1	 1_00_	0.57	0.44						
6.00	0.6654	10.10	27216	115164	0.56	13866	0.51	1 0404	10.80	0.57	0.04	0.43 0.7/	1 1500	10.75			
6.00	0.6654	10.25	27216	10560	0.39	110339	0.38	7165	10.00	84.0	0.00	0.34 10 34	1 119/	10.75	0.10		
6.00	0.6654	10.50	27216	6797	0.25	7163		1 4000	10.00	0.00	0.09	0.20 0.18	1 921	10.30			10.04
6.00	0.6654	1.00	27216	3876	0.14	4327	0.16	3250	0.27	0.84	0.75	0.11	501	10.24	0.12	0.11	10.03
				1 1			İ	i				i i	i	i	1		1
8.00	0.8872	0.00	27216	17017	0.63	14744	0.54	8682	1.00	0.51	0.59	0.31	1620	11.00	0.10	0.11	10.05
8.00	0.8872	0.10	27216	12498	0.46	11020	0.40	7096	0.82	0.57	0.64	0.26 j	1227	0.76	0.10	0.11	10.04
8.00	0.8872	0.25	27216	8957	0.33	8439	0.31	5586	0.64	0.62	0.66	0.20 j	956	0.59	0.11	0.11	10.03
8.00	0.8872	0.50	27216	6016	0.22	6090	0.22	4097	0.47	0.68	0.67	0.15	. 696		0.12	0.11	10.02
8.00	0.8872	1.00	27216	3595	0.13	3895	0.14	2667	0.31	0.74	0.68	10.09	446	0.28	0.12	0.11	10.01

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HD 400X400X678 (Weak axis) Sigma yield = 315 N/mm2; U/A = 30; t = 82 mm





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			•														
i	 ambda Bar	 b				F0				F30					F60		
		0/11	I (KN)					1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1		IN /N/ E0 EC3)		1 / / No. 1	 w/e401				
1		1 1	((KN)	lu(rem)/ubr			(FN)	/		/ (0,021)	1 m/ mpr [(FN)			W/W(FU,CEF)	Inter
1	 	 	 		 	1	 		 	 	 	1 1	(
1	I 0.2039	I 0.00	 42273	 38937	I 0.92	140850	0 97	140260		I I 103	I 1.000	 0.95	 14400	1	I I 0.37	1 0 35	1034
12.00	0.2039	10.10	42273	29964	0.71	32524	0.77	132278	i0.80	1.08		0.76	110739	10.75	0.36	0.33	10.25
2.00	0.2039	0.25	42273	20742	0.49	23800	0.56	23615	10.59	1.14	0.99	0.55	1 7289	0.51	0.35	0.31	10,17
2.00	0.2039	0.50	42273	12690	0.30	15347	0.36	15228	0.38	1.20	0.99	0.36	4649	0.32	0.37	0.30	10.10
2.00	0.2039		42273	6722	0.16	8468	0.20	8405	0.21	1.25	0.99	0.19	2829	0.20	0.42	0.33	0.06
i		İ	i i	i i	İ	i	İ	i	i	Ì	i	ii	ii	i	i	i	i
4.00	0.4078	0.00	42273	35149	0.83	37659	0.89	31957	1.00	0.91	0.85	0.75	10865	1.00	0.31	0.29	0.25
4.00	0.4078	0.10	42273	26543	0.63	28335	0.67	25078	0.78	0.94	0.89	0.59	8079	0.74	0.30	0.29	0.19
4.00	0.4078	0.25	42273	18445	0.44	20576	0.49	18268	0.57	0.99	0.89	0.43	5683	0.52	0.31	0.28	0.13
4.00	0.4078	0.50	42273	11582	0.27	13448	0.32	12611	0.39	1.09	0.94	0.29	3655	0.34	0.32	0.27	0.08
4.00	0.4078	1.00	42273	6377	0.15	7762	0.18	7694	0.24	1.21	0.99	0.18	2075	0.19	0.33	0.27	0.04
1						1			1	l				1	1	1	1
6.00	0.6117	0.00	42273	30424	0.72	32400	0.77	22627	1.00	0.74	0.70	0.53	8179	1.00	0.27	0.25	0.19
6.00	0.6117	0.10	42273	22684	0.54	23701	0.56	17950	0.79	0.79	0.76	0.42	6117	0.75	0.27	0.26	0.14
6.00	0.6117	0.25	42273	16013	0.38	17460	0.41	13396	0.59	0.84	0.77	0.31	4506	0.55	0.28	0.26	0.10
6.00	0.6117	0.50	42273	10418	0.25	11744	0.28	9287	0.41	0.89	0.79	0.21	3102	0.38	0.30	0.26	0.07
6.00	0.6117	1.00	42273	5978	0.14	7057	0.17	6263	0.28	1.05	0.89	0.14	1864	0.23	0.31	0.26	0.04
								1	l					ļ	ļ	1	ļ
8.00	0.8156	0.00	42273	24988	0.59	26897	0.64	17028	1.00	0.68	0.63	0.40	6115	1.00	0.24	0.23	0.14
8.00	0.8156	0.10	42273	18824	0.45	19319	0.46	13771	0.81	0.73	0.71	0.32	4705	0.77	0.25	0.24	0.11
18.00	0.8156	U.25	42273	15658	0.32	14550	0.34	10687	10.63	0.78		0.25	3621	0.59	0.27	0.25	0.08
18.00	0.8156	10.50	42275	9260	0.22	10231	0.24	7687	10.45	0.83		0.18	2609	0.43	0.28	0.26	0.06
8.00 	0,8156	1.00	42273	>>63	J 0.13	6400	0.15	4942	10.29	0.89	0.77	0.11	1670	0.27	Į 0.30	0.26	10.03
1			•••••									•••••		•••••			

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HD 400X400X1086 (Weak axis) Sigma yield = 305 W/mm2; U/A = 20; t = 125 mm



HD 400X400X1086 (Weak axis) Fe 510



T.46

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 	 		 I		1		F0		 			F30	•••••					F60			
L (m	 (Lambda Bar	e/h 	Npl (kN)	 N(EC3)	 N(EC3)/Npl	 N(CEF)	N(CEF)/Npl	 n((F30)	 N/Nc	N/N(F0,EC3)	N/N(FO,CEF)	N/Npl		N(F60)	 N/Nc	N/N(F0,EC3)	N/N(FO,CEF)	N/Np	ן- וו
I			ĺ	İ	(kN)		((kN)			(kN)		l		i	İİ	(kN)	ļ	l	ļ	ļ	į
															Π						- [
2.0	0	0.4008	0.00	15930	14473	0.91	14261	0.90	3	3927	1.00	0.27	0.28	0.24	П	757	1.00	0.05	0.05	10.04	
2.0	0	0.4008	0.10	15930	10675	0.67	10751	0.67	2	2970	0.76	0.28	0.28	0.18	П	578	0.76	0.05	0.05	0.03	1
12.0	0	0.4008	0.25	15930	6965	0.44	7496	0.47	2	2186	0.56	0.31	0.29	0.13	11	411	0.54	0.06	0.05	0.02	
[2.0	οΙ	0.4008	0.50	15930	3977	0.25	4572	0.29	1	1371	0.35	0.34	0.30	0.08		254	0.34	0.06	0.06	0.01	1
2.0	0	0.4008	1.00	15930	2099	0.13	2510	0.16		773	0.20	0.37	0.31	0.04		139	0.18	0.07	0.06	0.00	
1			I			1	1				1		Į	1	Π		l	l	1		1
4.0	οļ	0.8016	0.00	15930	11533	0.72	10392	0.65	2	2862	1.00	0.25	0.28	0.17	Н	565	1.00	0.05	0.05	0.03	
4.0	0	0.8016	0.10	15930	7996	0.50	7377	0.46	2	2091	0.73	0.26	0.28	0.13	П	416	0.74	0.05	0.06	0.02	
4.0	0	0.8016	0.25	15930	5314	0.33	5280	0.33	1	1568	0.55	0.30	0.30	0.09	11	300	0.53	0.06	0.06	0.01	-
4.0	οΙ	0.8016	0.50	15930	3312	0.21	3534	0.22	1	1076	0.38	0.32	0.30	0.06	11	200	0.35	0.06	0.06	0.01	
4.0	0	0.8016	1.00	15930	1895	0.12	2133	0.13		649	0.23	0.34	0.30	0.04	11	118	0.21	0.06	0.06	0.00	
I	1			1 1					11		1	1	ł	1	11			l I	1	1	1
6.0	οĮ	1.2024	0.00	15930	7259	0.46	6816	0.43	1	1923	1.00	0.26	0.28	0.12	11	385	1.00	0.05	0.06	0.02	
6.0	0	1.2024	0.10	15930	5379	0.34	4915	0.31	1	1460	0.76	0.27	0.30	0.09	11	. 285	0.74	0.05	0.06	0.01	1
16.0	0	1.2024	0.25	15930	3886	0.24	3786	0.24	1	1135	0.59	0.29	0.30	0.07	ŧI	219	0.57	0.06	0.06	0.01	1
6.0	οι	1.2024	0.50	15930	2680	0.17	2743	0.17	H	835	0.43	0.31	0.30	0.05	11	156	0.41	0.06	0.06	10.00	1
6.0	οΪ	1.2024	1.00	15930	1665	0.10	1781	0.11	II.	542	0.28	0.33	0.30	0.03	П	101	0.26	0.06	0.06	0.00	1
Ì	Ì		I	i i	1	Ì	l	l	ÎÎ.		Ì	I		Ì			1	I	I	1	I
18.0	0	1.6032	0.00	15930	4520	0.28	4499	0.28	11 1	1303	1.00	0.29	0.29	0.08	ÌÌ	260	1.00	0.06	0.06	0.01	1
18.0	0	1.6032	0.10	15930	3667	0.23	3470	0.22		1031	0.79	0.28	0.30	0.06	ÌÌ	202	0.78	0.06	0.06	[0.01	1
8.0	οİ	1.6032	0.25	15930	2888	0.18	2776	0.17	İİ.	845	0.65	0.29	0.30	0.05	ÌÌ	163	0.63	0.06	0.06	0.01	1
8.0	οj	1.6032	0.50	15930	2152	0.14	2152	0.14	ii -	655	0.50	0.30	0.30	0.04	Ì	124	0.48	0.06	0.06	[0.00	Í
8.0	οi	1.6032	1.00	15930	1439	0.09	1506	0.09	ii -	451	0.35	0.31	0.30	0.02	Ì	85	0.33	0.06	0.06	[0.00	Í
						••••••															

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HE 550 M (Weak axis) Sigma yield = 450 N/mm2 ; U/A = 64 ; t = 40 mm



T.48

 	1					 F0				F30		 	 !		F60		
ļι	Lambda Bar	e/h	Npl		• • • • • • • • • • • • • • • • • • •						· · · ·						
(m)	l	ł	(kN)	N(EC3)	N(EC3)/Npl	N(CEF)	N(CEF)/Npl	N(F30)	N/Nc	N/N(FO,EC3)	N/N(FO,CEF)	N/Npl	N(F60) N/Nc	N/N(FO,EC3)	N/N(FU,CEF)	N/Npl
		1		(kN)		(kN)		(kN)	ļ				(kN)	ļ		1	1
														-			
2.00	0.2853	0.00	18000	16853	0.94	16929	0.94	5759	1.00	0.34	0.34	0.31	951	11.00	0.06	0.06	10.05
2.00	0.2853	0.10	18000	12890	0.72	12929	0.72	4503	0.78	0.35	0.35	0.25	750	0.79	0.06	0.06	10.04
2.00	0.2853	0.25	18000	8864	0.49	9593	0.53	3264	0.57	0.37	0.34	0.18	546	0.57	0.06	0.06	10.03
2.00	0.2853	0.50	18000	5377	0.30	6136	0.34	2088	0.36	0.39	0.34	0.11	350	0.37	0.07	0.06	0.01
2.00	0.2853	1.00	18000	2838	0.16	3386	0.19	1177	0.20	0.41	0.35	0.06	207	0.22	0.07	0.06	0.01
1	1			1		1			l				1				
4.00	0.5706	0.00	18000	15166	0.84	13680	0.76	4440	1.00	0.29	0.32	0.24	762	1.00	0.05	0.06	0.04
4.00	0.5706	0.10	18000	11111	0.62	10299	0.57	3337	0.75	0.30	0.32	0.18	578	0.76	0.05	0.06	0.03
4.00	0.5706	0.25	18000	7620	0.42	7526	0.42	2459	0.55	0.32	0.33	0.13	428	0.56	0.06	0.06	0.02
4.00	0.5706	0.50	18000	4783	0.27	5059	0.28	1653	0.37	0.35	0.33	0.09	286	0.38	0.06	0.06	0.01
4.00	0.5706	1.00	18000	2652	0.15	2980	0.17	973	0.22	0.37	0.33	0.05	168	0.22	0.06	0.06	0.00
1	1				l						l		1	1	Į	1	ļ
6.00	0.8559	0.00	18000	12318	0.68	10310	0.57	3139	1.00	0.25	0.30	0.17	566	1.00	0.05	0.05	0.03
6.00	0.8559	0.10	18000	8868	0.49	7571	0.42	2417	0.77	0.27	0.32	0.13	430	0.76	0.05	0.06	0.02
6.00	0.8559	0.25	18000	6249	0.35	5814	0.32	1856	0.59	0.30	0.32	0.10	330	0.58	0.05	0.06	0.01
6.00	0.8559	0.50	18000	4129	0.23	4130	0.23	1349	0.43	0.33	0.33	0.07	235	0.42	0.06	0.06	0.01
6.00	0.8559	1.00	18000	2425	0.13	2602	0.14	850	0.27	0.35	0.33	0.04	147	0.26	0.06	0.06	0.00
1	1					1	1	11		1		1	l	1	1	1	
8.00	1.1412	0.00	18000	8915	0.50	7655	0.43	2274	1.00	0.26	0.30	0.12	419	1.00	0.05	0.05	0.02
8.00	1.1412	0.10	18000	6745	0.37	5718	0.32	1783	0.78	0.26	0.31	0.09	325	0.78	0.05	0.06	0.01
8.00	1.1412	0.25	18000	5002	0.28	4516	0.25	1442	0.63	0.29	0.32	80.0	258	0.62	0.05	0.06	0.01
8.00	1.1412	0.50	18000	3498	0.19	3381	0.19	1079	0.47	0.31	0.32	0.05	193	0.46	0.06	0.06	0.01
8.00	1.1412	1.00	18000	2185	0.12	2257	0.13	737	0.32	0.34	0.33	0.04	129	0.31	0.06	0.06	0.00
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PART III



CONTENTS OF PART III: TESTS

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Appendix B: Six full scale steel column fire tests	
Test 1: HD 210x210x198	1.1 to 1.15
Test 2: HD 310x310x500	2.1 to 2.15
Test 3: HD 310x310x500	3.1 to 3.15
Test 4: HD 400x400x1086	4.1 to 4.15
Test 5: W 360x410x314	5.1 to 5.14
Test 6: W 360x410x314	6.1 to 6.14

PART III

APPENDIX A

Transient state beam tests



KRUPP TESTS 1988 FOR ARBED





_	SI	T)	S2	Т2	53	
			Superior fla	nge		
T3	54	-	T4			TS

S6	;	т6	57	T7	58	
			Inferior flang	e		
TB	59		T9	S 10		T 10

Positions of test pieces

S= beams for transient state bending tests T= bars for tensile tests



✿ THERMOCOUPLES



A.4





















A.14
FEST	R _{ell}	FPCOLD	F	F F _{FCOLD}	ė"	θ _{init}	19 m max	D mes.'max	(^t test ⁾ max]
	[11,*mm²]	[kN}	[k:1]		(K/min)	[*C]	[°C]	{ ביניה }	[min]	
S1	502	30,0	30,0	1,00	3.6	22,1	461	85,8	121	11
52	501.5	30,2	22,7	0,75	3,4	22,5	525	53,9	146	(2)
S3	507	30,3	25,8	0,85	3,5	21,7	497	53,2	137	:2)
د۲	516	30,8	19,5	0,60	3,5	21.4	566	53,1	155	121
s5	513	30.7	12,3	0,40	2,5	21.0	651	37,3	182	2)
56	512	31,5	2, 1	0,975	3,1	31,2	823	37,8	235	12)
S 7	326	31,4	3,2	Ū,10	3,5	23	813	36,9	227	21
58	523	31,3	37,65	1,20	1	/	1	75,0	1	(3)
S9	523,5	31,3	6,3	0,20	3,5	22,4	713	54,1	198	(2)
S10	522,5	31,2	15,6	0,50	3,4	20,1	605	83,2	175	(2)

Erupp transient state beam tests parameters (S1 to S10)

REMARKS: (1) after cold loading before the heating the middle-span section is already fully plastified

(2) after cold loading before the heating the middle-span section is partially plastified or still elastic (3) only cold loadings - unloadings

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			SUPPOSED VALUES							
meaning of notations		dimensions	- 1/10mm	exact	+1/10mmm	max∕min in ≹				
depth of section	h .	mm	69,90	70,0	70,1	±0,14				
width of flange	ь	ותחת	36,9 37,0		37,1	±0,27				
thickness of web										
and flange	t,e	mm	4,9	5,0	5,1	±2,00				
radius of fillet	r	mm	4,9	5,0	5,1	±2,00				
cross-section area	л	Cm²	6,77	6,91	7,06	±2,17				
moment of inertia	1 xx	Cm	48,86	49,95	51,04	±2,18				
elastic section modulus	w xx	Cw ,	13,90	14,27	14,56	±2,03				
plastic section modulus	W xxpl	Cw ,	16,77	17,14	17,52	±2,22				
(W _{xxpl} /W _{xx})factor	$lpha_{p1}$	/	1,20	1,20	1,20					

INFINENCE OF THE TOOLING TOLERANCE GEOMETRICAL AND MECHANICAL CHARACTERISTICS DIMENSIONS

Values: * exactly : all the cross-section values are nominal

 * -1/10mm : all the cross-section values are situated on the minimal tolerance

* +1/10mm : all the cross-section values are situated on the maximal tolerance

3,25 30 3,25 3,25 3,25 2,5 2,5 1 1 1 1 1 1 1 1 1 1 1 2,5 1 3,0 1 4,0 1 5,0 1 1 5,5 5,0 1 5,0 1 1 5,0 Dimensions in [mm] ARBED-RECHERCHES / RPS DEPARTMENT CEFICOSS A / CEF7DPi PROJECT TITLE PROJECT NUMBER TESTS S 1 TO S 10 - STE 460 REFAO III

ESCH/ALZETTE : 17-AUG-1988 SHEET :

MCCELISATION OF THE SECTION

<u>scale</u> : 5/1



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A.18



A.19

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KRUPP TESTS 1988 FOR ARBED



TOOLING OF I-BEAMS 70x40x5x5



TRUNCATED BASIS BEAM

POSITION OF TESTPIECES S,V,T,P

S,V: BEAMS FOR TRANSIENT STATE BENDING TESTS T.P. BARS FOR TENSILE TESTS

A. FeE 460







B. Fe 360



ARBED BEAM TEST : POSITION OF THERMOCOUPLES (SPECIMEN S11, S12, V1-V6)



EXTREMA VALUES OF THE PROFILE TEMPERATURES CAPTED WITH THERMOCOUPLES



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STIFFENING AGAINST BUCKLING (S12 and V2 Tests)







IPE 80 PROFILE (Fe360)

12 STIFFENERS (Fe360)

KRUPP TRANSIENT STATE BEAM TEST PARAMETERS (S11,S12 ; V1 TO V7)

Geometrical characteristics

	Measures of test specimens												
	Tooled Fe E 46	profile O steel	IPE 80 profile Fe 360 steel										
Variables [mm]	 s 11 	 s12 	 V 1 	 V 2 	V 3	 V 4	V 5	 V 6	 v 7 				
h	 70.00	70.00	 80.20	 80.40	80.20	80.30	80.30	 80.30	80.30				
ь	 38.00	38.00	44.50	 45.30	45.00	45.00	45.00	45.00	45.00				
a	5.00	5.10	4.30	4.30	 4.30	4.30	4.40	4.30	4.30				
e	5.00	5.10	5.50	 5.80	5.70	5.80	5.70	5.80	5.80				
Г	 5.00 	5.00 	 5.00 	 4.50 	 4.50 	4.50	4.50	4.50	4 . 50 [°]				
F [cm2]	 7.01 	 7.14 	 8.09 	8.39	 8.26 	8.35	8.34	 8.35 	 8.35 				
Wplx [cm3]	17.47 	17.75 17.75	24.15	25.27 	24.78	25.10	24.94	25.10	25.10				
		۱ I	I 	 	 	 	 	I 	 				



A.36

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Evolution of the forces in function

of the temperature (S 11 test; Fe E 460)



----: evolution of the test data

- \mathbf{x} : failure of the beam test
- ----: evolution of the simulation data

failure temperature of the simulation : 460,0 °C

Evolution of the forces in function

of the temperature (S 12 test; Fe E 460)



× : failure of the beam test

----: evolution of the simulation data

failure temperature of the simulation : 440,0 °C

KRUPP TRANSIENT STATE BEAM TEST PARAMETERS (S1 TO S10) COMPARED TO QL-LAW SIMULATIONS

STEEL QUALITY : Fe E 460

 TE	st	Sig y	Sig t	St/Sy	Fpcold	F	F/Fpcold	Tm	Tinit	(Tmeas.)max	(Tsim.)max	Dmeassim.	(Wmeas.)max	(Wsim.)max	(Esim.)max		ļ
 		[N/mn2]	[N/mn2]	 	 [kn]	(kN)	 	[[⁰C/min]	[° C]	[°C]	 [ºC]	ן נאז	 (mm)	 [mm]	 [X]		
••••																	
\$	1	502.00	653.00	1.30	30.00	30.00	<u>1.000</u>	3.60	22.10	461.00	428.00	-7.2	85.80	64.40 	4.75	(1) 	
 s	 3	507.00	655.00	 1.29	30.30	25.80	0.850	 3.50	21.70	497.00	 467.00	-6.0	53.20	44.80	2.96	(2)	
 s	;2	504.50	 654.00	 1.30	30.20	22.70	 0.750	 3.40	22.50	525.00	l 497.00	-5.3	53.90	39.30	2.79	 (2)	
 s	 4	516.00	658.00	 1.28	 30.80	 18.50	 0.600	 3.50	 21.40	 566.00	543.00	-4.1	53.10	 39.70	2.86	(2)	
 s	 ;10	522.50	650.00	 1.24	 31.20	 15.60	 0.500	 3.40	 20.10	 605.00	574.00	-5.1	83.20	 73.80	 7.12	(2)	
 s	 55	513.00	658.00	 1.28	 30.70	 12.30	0.400	3.50	 21.00	 651.00	613.00	-5.8	87.50	 89.90	7.%	(2)	
 s	 9	523.50	 648.00	1.24	 31.30	 6.30	 0.200	 3.50	 22.40	 713.00	693.00	-2.8	54.10	108.60	10_14	(2)	
 s	 ;7	526.00	 652.00	 1.24	 31.40	 3. 20	 0.100	 3.50	 28.00	 813.00	813.00	0.0	 86.90	135.90	 7.78	(2)	
 s	 6	529.00	655.00	 1.24	 31.60	 2.40	 <u>0.075</u>	 3.40	 31.20	 828.00	 827.00	-0.1	 87.80	 41.10	3.13	(2)	
Ì	ĺ		1	I	 	I	l	l	l	I	l	1					
s	8	523.00	649.00	1.24	31.30	37.65	<u>1.200</u>	(Fmeas.)m	ax=37.65 k	N ; (Fsim.)max	= 36 kN ; D	max = -4.3 %	75.00 	74.50	7.13 	(3)	
1					l 	 	l 	l 					l 	l 	I 	۱ 	

Remarks :

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(1) after cold loading before the heating the middle-span section is already fully plastified

(2) after cold loading before the heating the middle-span section is already partially plastified or still elastic

(3) only cold loadings - unloadings

A.39



188



















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	TEST	I ES	Sig y	Sig t	St/Sy	Wplx	Fpcold	F	F/Fpcold	(Tmeas.)max	(Tsim.)max	Dmeassim.	(Wmeas.)max	(Wsim.)max	(Esim.)max	(Fmeas.)max	(F/Fpcold)max	
STE SVAL		102	 [N/mm2]	 [N/mm2]		[cm3]	(1) [kN]	I [[kn]		(?) [°C]	[°C]	[*]	[mm]	[mm]	[%]	[kN]	measur eu	1
::· -::·																		
12	S 11		496.00	758.00	1.53	17.470	30.22	31 .73	<u>1.05</u>	120.10	460.0	(8)	21.34	135.20	12.43	32.80	1.085	(2),(6)
1	S12	1002	489.50	 744.50	1.52	17.750	30.30	 31.82	1.05	422.80	440.0	4.07	54.70	98.00	7.41	35.06	1.157	 (2),(5)
	V1		 321.80	 508.30	 1.58	24.150	 27.10	 29.81	 <u>1.10</u>		440.0	(8)	37.90	84.00	8.67	32.41	1.1%	 (2),(6)
	V2	i I	 315.00	 505.30	 1.60	25.270	 27.76	 30.54	 1.10	336.90	440.0	(8)	58.90	63.70	5.68	33.36	1.202	 (2),(5)
	V 3	80	 310.00	 501.50	 1.62	24.780	 26.79	22.77	0.85	530.00	530.0	0.00	95.30	57.10	5.27			 (3)
[0]	₩4	Ч. Ш. Ш.	308.00	 503.30	1.63	25.100	26.96	 16.18	0.60	600.00	595.0	-0.83	90 .66	90.10	7.75			(3)
14	V5	H	 310.00	 505.30	1.63	24.940	 26.96	 13.48	0.50	630.00 	625.0	-0.79	78.60	105.20	8.22			(3)
	V 6	1	 312.00 	 506.00 	1.62	25.100	 27.31 	 2.73 	0.10		900.0	-2.28	78.40	242.10	14.66			(3)
 - 	 v7	17 311.80 504.30 1.62 25.100 27.29 33 n9 4.9 4						(Fmeas.)max=33.02 kN;			67.10	44.90	4.69	33.02	1.210	 (4)		
i		i								(Fsim.)max=3	2.3 kN;Dmax	=-2.2%						

-

KRUPP TRANSIENT STATE BEAM TEST PARAMETERS (S11,S12 ; V1 TO V7) COMPARED TO QL-LAW SIMULATIONS

REMARKS ON THE KRUPP TRANSIENT BEAM TEST PARAMETERS (S11,S12;V1 TO V7)

- (1) Fpcold, the theoretical necessary applied force to obtain the middle-span section fully plastified (plastic hinge) with a bi-rectangular stress distribution (rigid-plastic theory) = 4*Sigy*Wplx/L (for a simply-supported beam, with a mid-span concentrated load)
- (2) Testing procedure (plastic domain) :
 - a) cold loading till F/Fpcold level
 - b) heating with constant load
 - c) heating and load increasing together till collapse
- (3) Testing procedure (elasto-plastic domain) :
 - a) cold loading till F/Fpcold levelb) heating with constant load F till collapse
- (4) Testing procedure (cold test) :

cold loading till collapse with unloading

- (5) Specimens with stiffeners welded by points like shown on figure 2.6
- (6) The load is applied to these beams (V1;S11) via a ball (between the plunger of the actuator and the upper flange) instead of a kind of knife edge for the other beams (V2 to V7;S12) better against buckling
- (7) Mean value of the temperature measured with numbers 6, 8, 10 and 12 thermocouples
- (8) Early lateral-torsional buckling failure problem



















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PART III

APPENDIX B

SIX FULL SCALE STEEL COLUMN FIRE TESTS

III TAAA

APPENDIX B

SIX FULL SCALE STEEL COLUMN FIRE

TEST 1

COLUMN HD 210X210X198 - Fe 510

BUCKLING LENGTH 5.7 m

TEST PERFORMED IN BRAUNSCHWEIG





HD 210X210X198 Fe 510 e=1,0cm WEAK AXIS MEASURED TEMPERATURES : 1 --> ◇ 19--> + 20--> × 38--> □ T [DEG.] CALCULATED TEMPERATURES : -----1000.0 500.0 0.0 ₽ 90. 120. 60. 30. t imin]

1.3









HD 210X210X198 Fe 510 e=1,0cm WEAK AXIS 6-25 MEASURED TEMPERATURES : 6 --> ◇ 16--> + 25--> × 35--> □ T [DEG.] CALCULATED TEMPERATURES :--16-35 1000.0 500.0 0.0 ∖ 30. 601 90. 120. t (min)

222









HD 210X210X198 Fe 510 e=1,0cm WEAK AX1S 11.30 T [DEG.] CALCULATED TEMPERATURES : --Δ 1000.0 500.0 0.0 ᡰ₽ 30. 60. 120. 90. t (min)

1.13





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TEST 2

COLUMN HD 310x310x500 - Fe 510

BUCKLING LENGTH 4.14 m

TEST PERFORMED IN GAND






2.3

(mm HD 310X310X500 Fe 510 e=8.5 cm STRONG AXIS 3-23 MEASURED TEMPERATURES : 3 --> ◇ 12--> + 23--> × 32--> □ T DEG.) CALCULATED TEMPERATURES : Δ 11-32 1000.0. 500.0 0.0 +Ð 30. 60[.] 90. 120. t (min)

4-24 HD 310X310X500 Fe 510 e=8.5 cm STRONG AXIS MEASURED TEMPERATURES : 4 --> ◇ 13--> + 24--> × 33--> □ T (DEG.) CALCULATED TEMPERATURES :---13-33 1000.0 500.0 0.0 +₽ 120. 30. 60. 90. t (min)

2.5

HD 310X310X500 Fe 510 e=8.5 cm STRONG AXIS 49 mm MEASURED TEMPERATURES : 20--> \diamond 40--> + T (DEG.) CALCULATED TEMPERATURES :--Δ 1000.0 500.0 0.0 ᡰ₽ 30. 60'. 90. 120. t (min) ;

238

n

HD 310X310X500 Fe 510 e=8.5 cm STRONG AX1S T (DEG.) CALCULATED TEMPERATURES :---1000.0 500.0 0.0 ₽ 120. 30. 60. 90. t (min)

HD 310X310X500 Fe 510 e=8.5 cm STRONG AXIS - 5-25 MEASURED TEMPERATURES : 5 --> ◇ 14--> + 25--> × 34--> □ T (DEG.) CALCULATED TEMPERATURES :---14-34 1000.0 500.0 0.0 +₽ 30. 60[.]. 90. 120. t (min)

240



2.9

HD 310X310X500 Fe 510 e=8.5 cm STRONG AXIS 7-28 MEASURED TEMPERATURES : 7 ---> ◇ 16---> + 28---> ≍ 36---> □ T DEG.J CALCULATED TEMPERATURES : --16-36 Λ 1000.0 500.0 0.0 +Þ 30. 60. 120. 90. t (min)

242

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HD 310X310X500 Fe 510 e=8.5 cm STRONG AXIS MEASURED TEMPERATURES : 17--> ◇ 19--> + 37--> × 39--> □ 19-39 T (DEG.) CALCULATED TEMPERATURES : 17-17 Л 1000.0 500.0 0.0 +₽ 30. 60. 90. 120. t fmin] .

2.11



244



2.13





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TEST 3

COLUMN HD 310x310x500 - Fe 510

BUCKLING LENGTH 5.70 m

TEST PERFORMED IN BRAUNSCHWEIG



. HD 310X310X500 Fe 510 e = 3.4cm WEAK AXIS THEORE TICAL ISO - CURVE : . . . T [DEG.] EFFECTIVELY MEASURED HEATING CURVE : . . . SIMULATED HEATING CURVE IN CEFICOSS : ----1000.0 ···×·····×····× 500.0 0.0 ₽ 30. 60, 90. 120. t (min)

с

252

HD 310X310X500 Fe 510 e = 3.4cm WEAK AXIS 1-20 MEASURED TEMPERATURES : 1 ---> ◆ 19---> + 20---> × 38---> □ T [DEG.] CALCULATED TEMPERATURES :--13-38 1000.0 ~ 500.0 0.0 ╊ 30. 60. 90. 120. t imin] : .

3.3



254



ა.5

. 130 mm HD 310X310X500 Fe 510 e = 3.4cm WEAK AXIS . MEASURED TEMPERATURES : 4 ---> \diamond 23---> + 4-23 T [DEG.] CALCULATED TEMPERATURES :--4 1000.0_ 500.0 0.0 ╊ 30. 60. 90. 120. t (min)

256 °

HD 310X310X500 Fe 510 e = 3.4cm WEAK AXIS 5.24 MEASURED TEMPERATURES : 5 --> \diamond 24--> + T [DEG.] CALCULATED TEMPERATURES : --1000.0. 500.0 DEFECTIVE THERMOCOUPLE 0.0 ╊ 120. 30. 60. 90. t [min] 1

HD 310X310X500 Fe 510 e = 3.4cm WEAK AXIS 6-25 MEASURED TEMPERATURES : 6 ---> ◇ 16---> + 25---> × 35---> □ T [DEG.] CALCULATED TEMPERATURES :----16-35 o[Δ 1000.0 500.0 0.0 +₽ 30. 60. 90. 120. t (min)





260

HD 310X310X500 Fe 510 e = 3.4cm WEAK AXIS MEASURED TEMPERATURES : 9 ---> ◇ 13---> + 28---> ≫ 32---> □ 9-28 T [DEG.] CALCULATED TEMPERATURES : -----1-19 Δ 1000.0. 500.0 0.0 ╊ 30. 60. 120. 90. t (min)

HD 310X310X500 Fe 510 e = 3.4cm WEAK AXIS MEASURED TEMPERATURES : 10--> \diamond 12--> + 29--> \times 31--> \Box 10-25 12-31 T [DEG.] CALCULATED TEMPERATURES :-----Δ 1000.0 500.0 0.0 ╊ 30. 60. 90. 120. t (min)

. HD 310X310X500 Fe 510 e = 3.4cm WEAK AXIS MEASURED TEMPERATURES : 11--> \diamond 30--> + 11-30 T [DEG.] CALCULATED TEMPERATURES : --1000.0 500.0 0.0 +₽ 30. 60. 90. 120. t (min)





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TEST 4

COLUMN HD 400x400x1086 - Fe 510

BUCKLING LENGTH 4.14 m

TEST PERFORMED IN GAND



. HD 400X400X1086 Fe510 e=2.7cm WEAK AXIS THEORETICAL ISO-CURVE :- · · × · · · T DEG.1 EFFECTIVELY MEASURED HEATING CURVE :--@---SIMULATED HEATING CURVE IN CEFICOSS : ----1000.0 Θ 500.0 0.0 +₽ 60, 120. 30. 90. t (min)
1,21 HD 400X400X1086 Fe510 e=2.7cm WEAK AXIS MEASURED TEMPERATURES : 1 --> ◇ 11--> + 21--> × 31--> □ T (DEG.) CALCULATED TEMPERATURES : --Δ ·11,34 1000.0 500.0 0.0 +₽ 30. 60. 120. 90. t (min)

271

3,23 HD 400X400X1086 Fe510 e=2.7cm WEAK AXIS 90 mm MEASURED TEMPERATURES : 3 ---> ◆ 12--> + 23--> ¥ 32--> □ T (DEG.) CALCULATED TEMPERATURES : -- ---Δ 42,32 : 1000.0. 500.0 0.0 ╊ 30. 60. 90. 120. t (min)

4,24 HD 400X400X1086 Fe510 e=2.7cm WEAK AXIS MEASURED TEMPERATURES : 4 --> ◇ 13--> + 24--> × 33--> □ T (DEG.) CALCULATED TEMPERATURES : -----Λ 43,33 1000.0. 500.0 0.0 ╊ 30. 60. 90. 120. t (min)

کمړسه ez,s* HD 400X400X1086 Fe 510 e=22.7cm WEAK AX1S 50 T (DEG.) CALCULATED TEMPERATURES :---Λ 1000.0. 500.0 0.0 ᡰ₽ 120. 30. 60: 90. t (min]

274

:2,22 . -HD 400X400X1086 Fe510 e=2.7cm WEAK AX1S T (DEG.) CALCULATED TEMPERATURES : ----1000.0 500.0 0.0 ∣⊳ 90. 120. 30. 60. t [min] ; -

4.7

275

HD 400X400X1086 Fe510 e=2.7cm WEAK AXIS 0 5,25 MEASURED TEMPERATURES : 5 --> ◇ 14--> + 25--> × 34--> □ T (DEG.) CALCULATED TEMPERATURES :--A535 0 Л\ 1000.0 500.0 0.0 ₽ 30. 60: 90. 120. t (min)

276

HD 400X400X1086 Fe510 e=2.7cm WEAK AXIS 6,26 MEASURED TEMPERATURES : 6 --> ♦ 15--> + 26--> × 35--- ₽ T (DEG.) CALCULATED TEMPERATURES :---15,35 1000.0 500.0 Ū Ø 0.0 +₽ 120. 30. 60. 90. t (min) 277

HD 400X400X1086 Fe510 e=2.7cm WEAK AXIS 7,28 MEASURED TEMPERATURES : 7 ---> ◇ 16---> + 28---> ≍ 36---> □ T (DEG.) CALCULATED TEMPERATURES :-----4,X 0 Δ 1000.0. • 🖸 500.0 . 🖸 0.0 +₽ 30. 60. 90. 120. t (min)

HD 400X400X1086 Fe 510 e=22.7cm WEAK AX1S MEASURED TEMPERATURES : 17--> ◇ 19--> + 37--> × 39--> □ 19,35 T (DEG.) CALCULATED TEMPERATURES :---**A**3**A** • Δì 1000.0 500.0 * 0.0 ₽ 30. 60. 90. 120. t (min) .

279



280

HD 400X400X1086 Fe 510 e=22.7cm WEAK AXIS MEASURED TEMPERATURES : 18--> \diamond 38--> + 18.38 T (DEG.) CALCULATED TEMPERATURES :---1000.0 500.0 • , \$\$ 0.0 +₽ 30. 120. 60. 90. t (min)





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TEST 5

COLUMN W 360x410x314 - Fe 510

BUCKLING LENGTH 4.14 m

TEST PERFORMED IN GAND



W 360X410X314 Fe 510 e=12.0cm WEAK AXIS THEORETICAL ISO-CURVE : · · × · · T DEG.J EFFECTIVELY MEASURED HEATING CURVE : SIMULATED HEATING CURVE IN CEFICOSS : -----1000.0 500.0 0.0 ∖⊳ 120. 60. 90. 30. t fmin) ;

14,47 W 360X410X314 Fe 510 e=12.0cm WEAK AXIS MEASURED TEMPERATURES : 1 ---> ◆ 8 ---> + 17---> ¥ 24---> □ T (DEG.) CALCULATED TEMPERATURES : --121 Δ 1000.0 500.0 0.0 +₽ 30. 60. 90. 120. t (min)

5.3

289



290



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W 360X410X314 Fe 510 e=12.0cm WEAK AXIS M32 | 49,8 \$ 0 MEASURED TEMPERATURES : 16--> \diamond 32--> + T (DEG.) CALCULATED TEMPERATURES :---1000.0 500.0 0.0 ╊ 30. 60: 90. 120. t (min)

W 360X410X314 Fe 510 e=12.0cm WEAK AXIS 24! \$20 MEASURED TEMPERATURES : 2 --> \diamond 18--> + T DEG.] CALCULATED TEMPERATURES :--1000.0 500.0 0.0 ᡰ₽ 120. 30. 60. 90. t fminl - ;

W 360X410X314 Fe 510 e=12.0cm WEAK AXIS 5, 84 MEASURED TEMPERATURES : 5 ---> ◇ 11---> + 21---> × 27---> □ T (DEG.) CALCULATED TEMPERATURES :---્ર્યોરે € Δ 1000.0. 500.0 0.0 ╊ 30. 60. 90. 120. t (min) .

294









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TEST 6

COLUMN W 360x410x314 - FeE 460

BUCKLING LENGTH 4.14 m

TEST PERFORMED IN GAND







304

6<u>.</u>2

414 W 360X410X314 FeE 460 e=12.0cm WEAK AXIS MEASURED TEMPERATURES : 1 ---> \diamond 8 ---> + 17---> \times 24---> \Box T (DEG.) CALCULATED TEMPERATURES :---18,24 Δ 1000.0. 500.0 . 0.0 ᡰ₽ 120. 30. 60. 90. t (min)



306
W 360X410X314 FeE 460 e=12.0cm WEAK AXIS 4,20 MEASURED TEMPERATURES : 4 --> ♦ 10--> + 20--> ¥ 26--> □ T (DEG.) CALCULATED TEMPERATURES :-----10,26 1000.0 500.0 0.0 ╊ 90. 120. 30. 60. t (min) ، .

307



W 360X410X314 FeE 460 e=12.0cm WEAK AXIS 2448 : 30 MEASURED TEMPERATURES : 2 --> \diamond 18--> + T DEG.J CALCULATED TEMPERATURES :-----Δ 1000.0. 500.0 0.0 ╊ 30. 60. 90. 120. t (min) 1 -



310

W 360X410X314 FeE 460 e=12.0cm WEAK AXIS 0 6,12 MEASURED TEMPERATURES : 6 --> ◇ 12--> + 22--> × 28--> □ T DEG.) CALCULATED TEMPERATURES :---A2,28 1000.0 500.0 0.0 ᡰ₽ 120. 60. 90. 30. t (min)



312

W 360X410X314 FeE 460 e=12.0cm WEAK AXIS MEASURED TEMPERATURES : 14--> ◇ 15--> + 30--> × 31--> □ 15,31 a T (DEG.) CALCULATED TEMPERATURES : 14,30 0 1000.0 ;:: 500.0 0.0 ∖ا 120. 30. 90. 60. t (min) :

W 360X410X314 FeE 460 e=12.0cm WEAK AXIS 4,23 T DEG.J CALCULATED TEMPERATURES :---1000.0. 500.0 0.0 ╊ 30. 60. 120. 90. t (min)

314





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The main parameters to be considered in this research programme, i.e. the geometrical factors (shapes, buckling lengths), steel qualities and coefficients governing the heat exchanges, are presented first.

The temperature-dependent stress-strain relationships of steel as initially existing in the programme Ceficoss have been tested by a simulation of bending tests as described in the literature. It has shown the necessity of improving these laws when pure steel elements have to be calculated.

New improved stress-strain relationships of steel have been carried out and calibrated thanks to transient-state beam tests performed on small, simply supported, steel beams, subjected to a concentrated constant load, and submitted to a controlled temperature increase. These new laws have been established as well for commonly used construction steels and for high-strength steel FeE 460.

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The possibility of taking into account a distribution of residual stresses has been introduced in Ceficoss. The simulation of the six column tests showed that residual stresses have quite a small influence on the fireresistance time of columns. It has been decided, however, to consider systematically a distribution of residual stresses in the calculations.

Practical design tools have finally been implemented and are proposed here in the form of tables as well as diagrams.

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