Uncertainty Quantification of Aerothermal Flow Simulation Through Low-Density Ablative Thermal Protection Materials

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Introduction



The thermal protection system (TPS) is essential to shield spacecraft and their payload from the severe aerothermal conditions associated with atmospheric entry [1]

2. How? → Aerothermal flow simulations

Develop non-intrusive uncertainty quantification methods around numerical codes for the study of ablation phenomena. Perform uncertainty propagation through the model



Huygens Probe entry into Titan's atmosphere Source: ESA.

Ablative TPS design process





Wind tunnel experiments [2]

Numerical simulations [3]

 Uncertainties coming from experiments and physical models are affecting simulations with unknown impact on predictions!
 Objective: develop a rigorous uncertainty quantification approach for ablative TPS characterization and design

Methodology

Numerical simulations of VKI Plasmatron experiments on ablative thermal protection materials and quantification of uncertain margins





VKI Plasmatron exp. [2]

Plasmatron simulation

Preliminary results

Argo multiphysics and multidimensional CFD tool developed at Cenaero, based on a discontinuous Galerkin method

1. Which uncertainties? → Reassess physicochemical models

- New experiments on pyrolysis and oxidation of phenolic-impregnated carbon ablation
 - Carbon Fiber oxidation [4]



• Phenolic resin pyrolysis [5]







Deterministic simulations!

Ongoing work: review of stochastic inference methods (Bayesian, optimization problem with uncertainties) with application to physico-chemical models



Apply stochastic inference methods to those new experimental results for quantifying uncertainties on physico-chemical models

E.g. pyrolysis gas production model:

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References

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