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2 An inversion method for cometary 3 atmospheres

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30 Abstract

31 Remote observation of cometary atmospheres produces a measurement of the cometary
32 emissions integrated along the line of sight. This integration is the so-called Abel transform of
33 the local emission rate. The observation is generally interpreted under the hypothesis of
34 spherical symmetry of the coma. Under that hypothesis, the Abel transform can be inverted.
35 We derive a numerical inversion method adapted to cometary atmospheres using both
36 analytical results and least squares fitting techniques. This method, derived under the usual
37 hypothesis of spherical symmetry, allows us to retrieve the radial distribution of the emission
38 rate of any unabsorbed emission, which is the fundamental, physically meaningful quantity
39 governing the observation. A Tikhonov regularization technique is also applied to reduce the
40 possibly deleterious effects of the noise present in the observation and to warrant that the
41 problem remains well posed. Standard error propagation techniques are included in order to
42 estimate the uncertainties affecting the retrieved emission rate. Several theoretical tests of the
43 inversion techniques are carried out to show its validity and robustness. In particular, we show
44 that the Abel inversion of real data is only weakly sensitive to an offset applied to the input
45 flux, which implies that the method, applied to the study of a cometary atmosphere, is only
46 weakly dependent on uncertainties on the sky background which has to be subtracted from the
47 raw observations of the coma. We apply the method to observations of three different comets
48 observed using the TRAPPIST telescope: 103P/ Hartley 2, F6/ Lemmon and A1/ Siding
49 Spring. We show that the method retrieves realistic emission rates, and that characteristic
50 lengths and production rates can be derived from the emission rate for both CN and C₂
51 molecules. We show that the retrieved characteristic lengths can differ from those obtained
52 from a direct least squares fitting over the observed flux of radiation, and that discrepancies
53 can be reconciled for by correcting this flux by an offset (to which the inverse Abel transform
54 is nearly not sensitive). The A1/Siding Spring observations were obtained very shortly after
55 the comet produced an outburst, and we show that the emission rate derived from the
56 observed flux of CN emission at 387 nm and from the C₂ emission at 514.1 nm both present
57 an easily-identifiable shoulder that corresponds to the separation between pre- and post-
58 outburst gas. As a general result, we show that diagnosing properties and features of the coma
59 using the emission rate is easier than directly using the observed flux, because the Abel
60 transform produces a smoothing that blurs the signatures left by features present in the coma.
61 We also determine the parameters of a Haser model fitting the inverted data and fitting the
62 line-of-sight integrated observation, for which we provide the exact analytical expression of
63 the line-of-sight integration of the Haser model.

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67 1. INTRODUCTION

68 Comets are relatively small size bodies formed at the early stages of the solar system
69 evolution some 4.6 billions of years ago. They are often considered as potential tracers of
70 conditions prevailing at that time (Ehrenfreund & Charnley, 2000). They mainly consist of an
71 icy water nucleus with other constituents such as carbon monoxide (CO), carbon dioxide
72 (CO₂), and dust. When these bodies escape their reservoirs, mainly the Oort cloud and the
73 Kuiper belt, and approach the sun, they slowly warm up under the effect of solar radiation and
74 their ices start to sublime, releasing water vapor, CO, CO₂, dust and other minor species.
75 This process produces a large, highly rarefied, expanding atmosphere: the coma, surrounding
76 the icy nucleus.

77 The coma is exposed to the sun radiation and in particular to the ultraviolet solar flux,
78 which is capable to trigger photochemical processes such as dissociation and ionization of the
79 gaseous material. Many previous studies focused on the complex photochemistry of the coma
80 from a theoretical and observational standpoint. Among others, Bhardwaj & Raghuram (2012)
81 developed a photochemical model of the coma of comet C/1996 B2 (Hyakutake) to analyze
82 the metastable oxygen O(¹D) and O(¹S) populations and emissions accounting for
83 photodissociation and electron impact dissociation of H₂O, OH, CO and CO₂, as well as the
84 dissociative recombination of ions H₂O⁺, OH⁺, CO⁺ and CO₂⁺ and direct electron impact on
85 oxygen atoms. Loss mechanisms of metastable oxygen were the radiative decay, quenching
86 and reaction with H₂O, CO and CO₂. The densities of the major species of the coma (H₂O,
87 CO, CO₂ and OH) were given by a Haser model (Biver et al., 1999). Bhardwaj & Raghuram
88 (2012) conducted an analysis aimed at matching the observed and computed ratio of the
89 557.7 nm green emission of O(¹S) to the 630.0 and 636.4 nm red emissions of O(¹D), from
90 which they derived the CO₂ abundance and several photochemical parameters. Raghuram &
91 Bhardwaj (2012) also applied the same model with adapted parameters to comet C/1995 Hale
92 Bopp. Bisikalo et al. (2015) developed a model of the photochemistry of O(¹D) and O(¹S)
93 using a Monte Carlo method to solve the Boltzmann equation to retrieve the energy
94 distribution of these species across the expanding coma. They showed that the exothermic
95 nature of the photochemical mechanisms producing metastable oxygen yields a strongly non-
96 thermal distribution of their kinetic energy, which in turn produces a strongly non-gaussian
97 emission line profile.

98 The radial distribution of cometary constituents is often described using a Haser model
99 (Haser, 1957). This model is used for its simplicity and its ability to describe a spherically
100 symmetric expanding coma. It relies on flux conservation and includes the effect of
101 photochemical production and loss of any species in an ad hoc manner, instead of solving for
102 the detailed photochemistry. Simple flux conservation produces a radial profile that varies as
103 $1/r^2$, with r the radial distance:

$$n = \frac{Q}{4\pi r^2 v} \quad (1)$$

104 with n the density of the species considered (H₂O, for example), Q the rate at which the
105 comet's nucleus releases that species, and v the radial outflow speed of the emitting particles.

106 The concentration of a species that gets destroyed by photochemical processes decays
 107 exponentially with time, with a life time τ_p . This life time depends on solar activity,
 108 heliocentric distance etc. and translates into a characteristic length L_p in the expanding coma,
 109 so that the density profile becomes:

$$n_p = \frac{Q_p}{4\pi r^2 v_p} e^{-\frac{r}{L_p}} \quad (2)$$

110 Here, the subscript p stands for “parent”, as we are considering molecules outgassed by the
 111 comet’s nucleus that decompose and produce “daughter” species, and which will be denoted
 112 by subscript d . The production rate of the daughter species is determined by the loss rate of
 113 their parent molecules. Daughter species can in turn be destroyed by photochemical
 114 processes, with a characteristic length L_d . Their density profile in the expanding atmosphere is
 115 then given by

$$n_d = \frac{Q_p}{4\pi r^2 v_d} \frac{L_d}{L_d - L_p} \left(e^{-\frac{r}{L_d}} - e^{-\frac{r}{L_p}} \right). \quad (3)$$

116 The model could even be further complexified to derive the density profile of grand-daughter
 117 species. Expression (1) is however not integrable over \mathbb{R}^3 (accounting for the Jacobian of
 118 spherical coordinates) as $r \rightarrow \infty$, which clearly shows equation (1) does not suffice. The Haser
 119 model also assumes the characteristic length does not vary across the coma and that there
 120 exist only one production and one loss mechanism of the daughter species, which is not
 121 warranted. As the daughter molecules are produced isotropically in a frame of reference
 122 moving with the expanding gas, there is no reason to assume that the expansion velocity of
 123 the different species can largely differ, and a single expansion velocity is generally used.
 124 However, the Haser model neglects molecular diffusion that can influence the density
 125 distribution. Integration of expressions (2) and (3) (multiplied by the appropriate Jacobian)
 126 over \mathbb{R}^3 can be easily carried out analytically, giving $Q_p L_d/v_d$ for the total content of daughter
 127 species particles of the coma. Models of the coma, either idealized using the Haser
 128 approximation or based upon a mechanistic representation such as those of Bhardwaj and
 129 Raghuram (2012), Bisikalo et al. (2015), Combi (1996), Rubin et al. (2011), Weiler (2007,
 130 2012), Combi and Fink (1997) and others have to be compared against observational data.
 131 However, the local densities, which are the natural outputs of the models, cannot be directly
 132 observed remotely, as we discuss in the next section. Moreover, comets are dynamic objects,
 133 and time variations of the activity translate to radial gradients in the density, that are not
 134 accounted for by steady-state models, whatever their degree of sophistication. This is
 135 particularly significant when a comet produces an outburst.

136 Here, we present a method to retrieve the local emission rate from remote sensing
 137 observations of cometary atmospheres. Remote sensing of cometary emission provides only a
 138 line-of-sight integration of the emission rate, also called its Abel transform. We develop a
 139 method that inverts the Abel transform in the special case of cometary atmospheres. Section 2
 140 presents the mathematical developments on which the inverse Abel transform relies. The
 141 result of this inversion must not be confused with a model of the coma. It is rather a direct
 142 processing of the observational data. Fundamentally, the result of the inverse Abel transform

143 of the data contains essentially the same information as the initial line-of-sight integrated
 144 radial profile. In section 3, we present results from numerical tests that were done to validate
 145 the inversion method and highlight its benefits. In section 4, we present the results from
 146 applications of our inverse Abel transform method for three comets. These results are
 147 compared with Haser model fits to the data. Particular attention will be given to an outburst
 148 case. In section 5 we discuss the reach of the results obtained with the inverse Abel transform.
 149 We conclude with a short summary of our results in section 6. Appendix 1 provides additional
 150 analytical results that allow for a further refinement of the inversion method. These results do
 151 not appear to offer a crucial improvement in the case of cometary atmospheres but they could
 152 nevertheless prove useful for planetary atmospheres. Finally, appendix 2 gives the results
 153 needed to perform the exact analytical computation of the Abel transform of a cometocentric
 154 profile described using a Haser model, which is a result that can be used for any study
 155 dedicated to the analysis of observations of comets under the Haser hypothesis.

156 2. THE ABEL TRANSFORM INVERSION

157 A distant observer looking at the coma of a comet has no direct access to the density
 158 profile of the constituents. Excited species relax by emitting photons and the observation
 159 sums up the emission rates along a full line of sight according to the geometry described in
 160 **Figure 1**. If we denote by $n(r)$ the density of an excited atom or molecule (for example) and
 161 by A_{ul} the Einstein transition parameter for spontaneous emission of this excited particle by a
 162 transition from upper state u to lower state l , the emission rate at that radius is given by
 163 $f(r) = A_{ul} n(r)$. In principle, the local density can thus be immediately obtained, if the local
 164 emission rate profile is known. When molecular bands are observed and their spectral
 165 structure remains unresolved (which is generally the case), the characterization of the excited
 166 molecule density based on the emission rate may require a more sophisticated treatment. The
 167 fundamental principle remains nevertheless unchanged: it is possible to relate emission rates
 168 to molecular densities. In the geometrical framework of **Figure 1**, the line-of-sight integrated
 169 emission can be written:

$$F(r_0) = \int_{-\infty}^{+\infty} ds f(s) = 2 \int_0^{+\infty} ds f(s) = 2 \int_{r_0}^{+\infty} dr \frac{r}{\sqrt{r^2 - r_0^2}} f(r) \quad (4)$$

170 where r_0 is the tangent radius, i.e., the distance between the comet's center and the point of
 171 the line of sight closest to this center, f is the quantity to be integrated along the line of sight,
 172 such as the emission rate of a given excited species, or any other quantity. The coma is
 173 supposed to have a spherical symmetry (which allows us to change the integral over s from $-\infty$
 174 to $+\infty$ to the double of the integral from 0 to $+\infty$ and to apply the variable change $s =$
 175 $\sqrt{r^2 - r_0^2}$, which has a jacobian $J = r/\sqrt{r^2 - r_0^2}$. The right-hand side of equation (4) is called
 176 the Abel transform of $f(r)$ (Bracewell, 1999). It has a well-known inverse transform:

$$f(r) = \frac{-1}{\pi} \int_r^{\infty} dr_0 \frac{1}{\sqrt{r_0^2 - r^2}} \frac{dF(r_0)}{dr_0} \quad (5)$$

177 This expression is, however, of little practical usage, as it requires the computation of the
 178 derivative of $F(r_0)$, a difficult task especially when values for F are actually only available
 179 from a limited, discrete set of noisy data. Numerical inversion methods have thus been
 180 derived that use least squares fitting techniques and simple analytical expressions of the direct
 181 Abel transform, that can be obtained when $f(r) = r^n$, for $n \geq -1$. Indeed, let us denote by $I_n(r,$
 182 $r_0)$ the indefinite integral

$$I_n = \int dr \frac{r}{\sqrt{r^2 - r_0^2}} r^n. \quad (6)$$

183 An integration by parts shows that the I_n satisfy a simple recurrence relation:

$$(n + 1)I_n + n r_0^2 I_{n-2} = r^n \sqrt{r^2 - r_0^2}$$

$$I_{-1} = \operatorname{arccosh}\left(\frac{r}{r_0}\right) = \ln\left(\frac{r}{r_0} + \sqrt{\frac{r^2}{r_0^2} - 1}\right) \quad (7)$$

$$I_0 = \sqrt{r^2 - r_0^2}$$

184 I_{-1} and I_0 can be directly obtained from equation (6). Although the recurrence relation
 185 (7) is formally of order 2, it can actually be solved as two joint first order linear recurrences,
 186 one for $n = 2m$ and one for $n = 2m + 1$, starting from I_0 and I_{-1} , respectively. Each I_n is defined
 187 up to an additive constant, which we can take as 0 because we will only use the results to
 188 compute definite integrals (so that the constants cancel out). These results have been used to
 189 derive numerical inversion techniques by several authors to study the emissions of planetary
 190 atmospheres (e.g., Quémerais et al., 2006, Stiepen et al., 2012; Cox et al., 2008) using the
 191 following ideas.

192 Any observation of the line of sight-integrated emission (i.e., brightness) of a given
 193 atmospheric emission will produce a discretized, noisy profile of values obtained for a series
 194 of tangent radii. Such profiles are generally called vertical profiles in the case of a planetary
 195 atmosphere or nucleo-centric profiles in the case of a coma. It then becomes natural to
 196 represent the emission rate profile $f(r)$ as a set of linear segments across well-chosen intervals
 197 that might, for instance but not necessarily, correspond to the set of tangent radii of the
 198 observation. This set of linear segments can be represented as a linear combination of
 199 triangular functions, as shown in **Figure 2**. These triangles $t_k(r)$ can be written as

$$t_k(r) = \frac{r - r_{k-1}}{r_k - r_{k-1}} \chi_{]r_{k-1}, r_k[}(r) + \left(1 - \frac{r - r_k}{r_{k+1} - r_k}\right) \chi_{]r_k, r_{k+1}[}(r)$$

$$= \frac{r - r_{k-1}}{r_k - r_{k-1}} \chi_{]r_{k-1}, r_k[}(r) + \frac{r_{k+1} - r}{r_{k+1} - r_k} \chi_{]r_k, r_{k+1}[}(r), \quad (8)$$

200 where we use the characteristic function $\chi_\Omega(r)$, which takes the value 1 when $r \in \Omega$ and 0
 201 otherwise and where $k = 1, \dots, n$ enumerates the different nodes r_k . The first (second) term of

202 expression (8) can be ignored at $k = 0$ ($k = n$, respectively). Any piecewise linear, continuous
 203 function f can then be written as a linear combination of the t_k :

$$f(r) = \sum_k a_k t_k(r). \quad (9)$$

204 The Abel transform $T_k(r_0)$ of each triangle t_k can be easily computed using I_{-1} , I_0 and I_1 from
 205 equations (6) and (7). The Abel transform (4) is linear for f and we thus have

$$F(r_0) = \sum_k a_k T_k(r_0). \quad (10)$$

206 **Figure 3** shows the Abel transform of a triangular function. In this figure, the Abel
 207 transform $F(r_0)$ is zero for any value of r_0 larger than the upper boundary of the interval over
 208 which the triangle is defined. The Abel transform $F(r_0)$ varies rather smoothly, despite the
 209 discontinuous first derivative of the triangle function. Now, when $f(r)$ has to be estimated
 210 from line-of-sight integrated measurements G_j obtained for a set of radial distances $r_{0,j}$, $j = 1,$
 211 \dots, J (for simplicity, we assume that the $r_{0,j}$ are sorted by increasing r_0) one just has to
 212 minimize the chi-square expression

$$\chi^2 = \sum_{j=1}^J \left(G_j - \sum_k a_k T_k(r_{0,j}) \right)^2 w_j \quad (11)$$

213 using standard linear minimization techniques. The weights w_j will generally be set equal to
 214 the inverse of the variance and they will be the diagonal elements of the inverse of the
 215 variance matrix \mathbf{V}_G of the measured G_j (which we assume do not co-vary). They may also be
 216 set to 1 for unweighted least squares fit. Indeed, under the assumption of homoscedasticity,
 217 the Gauss-Markov theorem states that an optimal estimation of the parameters is provided by
 218 the weighted least squares fitting. The suitable a_k 's are thus obtained by solving the system

$$\mathbf{H} \vec{a} = \vec{b} \quad (12)$$

219 For \vec{a} , with

$$H_{ik} = \sum_{j=1}^J T_i(r_{0,j}) T_k(r_{0,j}) w_j = (\mathbf{T} \mathbf{V}_G^{-1} \mathbf{T}^+)_{ik} \quad T_{ji} = T_i(r_{0,j}) \quad (13)$$

$$b_i = \sum_{j=1}^J G_j T_i(r_{0,j}) w_j$$

220 Many terms of the sums of equation (13) are zero, as $T_k(r_{0,j}) = 0$ for any $r_{0,j} > r_{k+1}$. By solving
 221 system (12), F is adjusted to the set of observations $\{G_j, j = 1, \dots, J\}$. The solutions of the
 222 system, a_k , are then used in expression (9) to construct the adjusted f . The quality of the
 223 solution of such an inverse problem can often be improved by applying Tikhonov
 224 regularization, especially when the problem is ill-conditioned. We outline here the principle
 225 of such regularization; details can be found, e.g., in Press et al. (1992). The key idea behind
 226 the Tikhonov regularization is to modify the quantity that is to be minimized by adding a

227 contribution that penalizes a property of the fitted result that is considered as inappropriate.
 228 For example, if the result is expected to be fairly constant, we can add a term proportional to
 229 the square of the first derivative (or its integral) in order to penalize any solution with strong
 230 variations or, if the result is expected to be rather smooth (a special case being close to linear),
 231 we can attenuate possible noisy variations of the fitted function by adding a term proportional
 232 to the square of the second derivative (or its integral), to be represented in discrete form. This
 233 is indeed a way to protect the inversion procedure against the deleterious effects of noise. The
 234 usual method to regularize the fitted function is then to replace equation (12) by

$$(14) \quad (\mathbf{H} + \lambda \mathbf{Q}) \vec{a} = \vec{b}$$

235 where λ can be viewed as a suitable weight applied to the regularization matrix \mathbf{Q} , the other
 236 symbols keeping their original definition. The regularization matrix \mathbf{Q} must now be
 237 determined. Press et al. (1992) provide \mathbf{Q} suitable for equally-spaced observational points.
 238 The derivatives can then be approximated by (forward) finite differences of the fitting
 239 parameters, and the resulting regularization matrices are naturally simple and almost
 240 symmetric. Note that a sophisticated and very accurate method of computation of the
 241 derivatives is indeed not necessary as it would be the case in a solver for differential
 242 equations: we are only searching for an expression that penalizes a property that we consider a
 243 priori should remain small. We adapt the algorithm from Press et al. (1992) to the specific
 244 case of equation (10) in which the derivatives are not estimated by differences of the fitting
 245 parameters. We can write the second derivative of F computed at the observation points $r_{0,i}$
 246 and pack them in a vector:

$$(15) \quad \vec{D} = \left. \frac{\partial^2 F}{\partial r_0^2} \right|_{r=r_{0,j}} = \sum_k a_k \left. \frac{\partial^2 T_k}{\partial r_0^2} \right|_{r=r_{0,j}} = \sum_k S_{jk} a_k = \mathbf{S} \vec{a}$$

247 where the components of the matrix \mathbf{S} are

$$(16) \quad S_{jk} = \left. \frac{\partial^2 T_k}{\partial r_0^2} \right|_{r=r_{0,j}}$$

248 The sum of the squares of the second derivative can then be written in matrix format as

$$(17) \quad D^2 = \vec{D}^+ \vec{D} = \vec{a}^+ \mathbf{S}^+ \mathbf{S} \vec{a} = \sum_{i,j,k} a_k S_{ik} S_{ij} a_j$$

249 We may prefer to compute the integral of the square of the second derivative, which can be
 250 estimated numerically as

$$(18) \quad \int_{r_{0,1}}^{r_{0,J}} dr \left(\frac{\partial^2 F}{\partial r_0^2} \right)^2 \simeq \sum_j \left(\sum_k a_k \left. \frac{\partial^2 T_k}{\partial r_0^2} \right|_{r=r_{0,j}} \right)^2 h_j$$

$$= \sum_j \left(\sum_k a_k \left. \frac{\partial^2 T_k}{\partial r_0^2} \right|_{r=r_{0,j}} \sqrt{h_j} \right)^2,$$

251 where h_j can be taken as $h_j = r_{0,j+1} - r_{0,j}$ (with $h_J = h_{J-1}$) or as any other suitable discretisation
 252 step length. The derivatives of the T_k can be estimated by any suitable mean: analytically or
 253 numerically (using a central difference scheme, for example). We can then define a matrix \mathbf{S}
 254 by

$$S_{jk} = \left. \frac{\partial^2 T_k}{\partial r_0^2} \right|_{r=r_{0,j}} \sqrt{h_j}. \quad (19)$$

255 Formally, computing the sum of the square or the integral of the square of the second
 256 derivative can both be done similarly using expression (17). We now want to obtain the
 257 matrix \mathbf{Q} of equation (14) in order to perform a minimization. We then need to compute the
 258 first derivatives of D^2 with respect to the a_k :

$$\frac{\partial D^2}{\partial a_l} = \frac{\partial}{\partial a_l} \left(\sum_{i,j,k} a_k S_{ik} S_{ij} a_j \right) = 2 \sum_{i,k} a_k S_{ik} S_{il} = 2 \mathbf{S}^+ \mathbf{S} \vec{a}|_l \quad (20)$$

259 so that we can define \mathbf{Q} by

$$\mathbf{Q} = 2 \mathbf{S}^+ \mathbf{S} \quad (21)$$

260 We still have to determine the factor λ in equation (14). We follow Press et al. (1992) and
 261 chose

$$\lambda = \text{Tr}(\mathbf{H}) / \text{Tr}(\mathbf{Q}), \quad (22)$$

262 where $\text{Tr}(\mathbf{A})$ denotes the trace of matrix \mathbf{A} . Note that the factor 2 in equation (21) is simplified
 263 out of equation (14) when adopting this value for λ .

264 The method outlined above is very general and it is not specifically designed for the
 265 case of cometary atmospheres. It was already introduced by Quémerais et al. (2006) for the
 266 study of the atmosphere of planet Mars. We will adapt the inversion method for cometary
 267 atmospheres in two steps: First, we will modify the regularization method, and second, we
 268 will modify the t_k introduced in equation (8).

269 The regularization method proposed occasionally suffers from a severe drawback: if
 270 the observed quantity and its derivatives vary over several orders of magnitude across the
 271 observed atmosphere (and this can be the case in cometary and planetary atmospheres), then
 272 D^2 will be dominated by the largest values, and regularization will become less efficient in
 273 those regions of the atmosphere where the emission rate (for example) is smaller, i.e. where
 274 regularization may be the most needed. We then change the regularization method by
 275 considering the a_k as a list of discrete values of a function $a(r)$, and we regularize the fit by
 276 minimizing its second derivative. Equivalently, we may consider the a_k as a suite and
 277 minimize its second-order discrete difference, with similar results. We can write $h_k = r_{k+1} - r_k$
 278 ($h_n = h_{n-1}$) and use a simple finite difference scheme as an approximation for the second
 279 derivative

$$B_{ki} = B_{0ki} \sqrt{h_k} \quad (28)$$

$$\mathbf{Q}_a = 2 \mathbf{B}^+ \mathbf{B}.$$

295 We now turn to the task of defining new “triangular” elements instead of the
 296 expression given in equation (8). Our purpose is to find an expression that would be more
 297 suitable to the description of the constituents of a cometary atmosphere. In a first
 298 approximation, the Haser model given by equations (1), (2) and (3) for inert, mother and
 299 daughter species, respectively, provides an adequate description of the distributions of these
 300 constituents. We want to derive triangular elements whose Abel transform can be calculated
 301 analytically, in order to reduce the computational cost. The presence of the exponential
 302 function in expressions (2) and (3) severely complicates the analytic computation of the
 303 indefinite integral built from the Abel transform. We can however compute those primitives
 304 for negative powers of r , which points at the Haser model for inert molecules, proportional to
 305 $1/r^2$. We thus define new “triangular” elements, using the triangles t_k from equation (8) as

$$u_k(r) = \frac{t_k(r)}{r^m} \quad m > 0 \quad (29)$$

306 Quite obviously, we will choose $m = 2$ in the case of a cometary atmosphere so that
 307 the $1/r^2$ dependency that appears in the Haser model is explicitly present in the triangular
 308 elements. We write their Abel transforms U_k to use them instead of the T_k in the definitions of
 309 matrices \mathbf{H} and \mathbf{S} in equations (13), (16) and (19). The u_k so defined do always reduce to a
 310 linear combination of negative powers of r (over bounded intervals). Analytical computation
 311 of their Abel transform thus only requires us to know the indefinite integrals of the form

$$L_m = \int dr \frac{r}{\sqrt{r^2 - r_0^2}} r^{-m} \quad (30)$$

312 An integration by parts again shows that the L_m satisfy a recurrence relation:

$$(m - 1)L_m - m r_0^2 L_{m+2} + \frac{\sqrt{r^2 - r_0^2}}{r^m} = 0$$

$$L_0 = \sqrt{r^2 - r_0^2}$$

$$L_1 = \operatorname{arcosh}\left(\frac{r}{r_0}\right) = \ln\left(\frac{r}{r_0} + \sqrt{\frac{r^2}{r_0^2} - 1}\right) \quad (31)$$

$$L_2 = \frac{1}{r_0} \operatorname{arctg}\left(\sqrt{\frac{r^2}{r_0^2} - 1}\right) = \frac{1}{r_0} \arccos\left(\frac{r_0}{r}\right).$$

313 L_0 and L_1 can be directly obtained from equation (30); L_2 is found after the variable change
 314 $s = \sqrt{\frac{r^2}{r_0^2} - 1}$ and using a trigonometric identity to transform the arctangent into an arccosine.
 315 Notice that the recurrence relation cannot be used to deduce L_2 from L_0 as the term in L_2
 316 vanishes for $m = 0$. One possibility to further improve the triangular elements would be to

317 apply an offset, replace r^{-m} by $(r-a)^{-m}$ in expression (29) and to call upon Laurent series
 318 with an offset. This choice could be suitable in the case of a planetary atmosphere, for which
 319 the extent of the emitting layer is small compared with the planet radius. Indefinite integrals
 320 of the form

$$\int dr \frac{r}{\sqrt{r^2 - r_0^2}} t_k(r) \frac{1}{(r-a)^m} \quad 0 < a < r_0 \quad (32)$$

321 can always be reduced to a linear combination of indefinite integrals of the form

$$W_m = \int dr \frac{1}{\sqrt{r^2 - r_0^2}} \frac{1}{(r-a)^m} \quad (33)$$

322 completed with the first elements of the suite of integrals I_n given by equations (6) and (7)
 323 when $m < 3$. These integrals satisfy again a recurrence relation and are also related by a
 324 simple derivative with respect to the parameter a . We will not use that refinement here. We
 325 nevertheless report the analytical results and developments in appendix 1, as some of the
 326 computations could be useful for the studies of planetary atmospheres.

327 In this study, we will only investigate the use of elements t_k and u_k with $m = 2$
 328 (equations (8) and (29)) to represent emission profiles in cometary atmospheres, assuming
 329 spherical symmetry.

330 When uncertainties affecting the observation are known, the weights w_j in equation
 331 (11) can be taken as $1/\sigma_j^2$ i.e., the inverse of the variances affecting the observational points.
 332 As the fitted parameters a_k are obtained by applying formulas of linear algebra, error
 333 propagation techniques can be used to obtain the variance matrix of the a_k 's and the standard
 334 deviation (i.e., the uncertainty) of the fitted profiles. We remind here the standard general
 335 formulas needed to obtain the desired uncertainties. If we denote by \mathbf{V}_G the variance matrix of
 336 the observation (which in our case will be a diagonal matrix $\text{diag}(\sigma_j^2)$) we can obtain the
 337 variance matrix \mathbf{V}_a of the fitted parameters by noting that, formally, they are computed by just
 338 multiplying the observation vector \vec{G} by a matrix \mathbf{M} :

$$\vec{a} = \mathbf{M}\vec{G} \quad (34)$$

339 In this case the variance matrix \mathbf{V}_a can be written in matrix form as

$$\mathbf{V}_a = \mathbf{M}\mathbf{V}_G\mathbf{M}^+. \quad (35)$$

340 Matrix \mathbf{M} is deduced from equations (13) and (14) as:

$$\mathbf{M} = (\mathbf{H} + \lambda \mathbf{Q})^{-1}\mathbf{T}^+\mathbf{V}_G^{-1}. \quad (36)$$

341 The parameter λ can be set to 0 when no regularization is applied. This can, however,
 342 lead to numerical problems when \mathbf{H} is ill-conditioned. In contrast, introducing the
 343 regularization warrants that the problem will be well-conditioned and the inverse matrix will
 344 be computable. Because the Abel transform of a triangular element (**Figure 3**) extends from
 345 the nucleo-centric distance where this element is defined down to $r_0 = 0$, the a_k are expected to

346 co-vary and \mathbf{V}_a will not be diagonal. Its diagonal elements are nevertheless the most important
 347 ones as they determine the (square of the) uncertainties affecting the fitted a_k 's. Once the
 348 covariances and uncertainties affecting the a_k 's have been obtained, standard error
 349 propagation formulas can be used to derive the uncertainties of the fitted F and f from
 350 equations (9) and (10). If we collect the estimated values of F at each $r_{0,j}$ in a vector, the
 351 variances of the F_j are then the diagonal elements of matrix $\mathbf{T}^+\mathbf{V}_a\mathbf{T}$, and a similar expression
 352 can be obtained for the f_k .

353

354 3. THEORETICAL TESTS

355 3.1. Inert species profiles

356 Before analyzing real observations, we apply our method to theoretical nucleo-centric
 357 profiles of the Abel transform F , which is what we use to retrieve the emission rate profile.
 358 We will also check that the inversion method gives appropriate results. **Figure 4** shows the
 359 line-of-sight integrated profile F obtained from an emission rate varying as $1/r^2$, i.e., it is
 360 proportional to the variation of L_2 given in equation (31) between r_0 and ∞ , and thus varies as
 361 $1/r_0$. Panel b shows the emission rate profile f obtained by numerical inversion of F given at a
 362 restricted set of nucleo-centric distances, without regularization, using purely triangular
 363 elements t_k as given by equation (8) (triangles) and elements u_k built by dividing each t_k by r^2
 364 (equation (29), with $m = 2$). The corresponding line-of-sight integrated values are shown in
 365 Figure 4a using the same plotting symbols. At first glance, both methods seem to give a
 366 satisfying inversion, showing that the inversion method correctly retrieves the expected
 367 emission rate. **Figures 4c** and **4d** show the absolute value of the relative difference between
 368 the numerically-inverted profiles and the input local emission rate. Inversion using elements
 369 u_k performs obviously better. This is expected as the chosen elements better match the
 370 emission rate profile corresponding to F . **Figure 5** shows the same as **Figure 4** with
 371 regularization. In the case of purely triangular elements, regularization appears as counter-
 372 productive over this particular profile, while it slightly reduces the absolute deviation from the
 373 correct values in the case of elements consisting in triangles divided by r^2 . Truncation of the
 374 profiles at large nucleo-centric distance is an obvious source of error. Moreover, these profiles
 375 are somewhat artificial: they were built using a set of nucleo-centric radii that are spaced
 376 following a power law, so that the discrete profiles appear as regularly-spaced points in a log-
 377 log diagram. Real data will not resemble those profiles: in general, observations are regularly
 378 spaced versus nucleo-centric distance, and the signal is contaminated by noise.

379 **Figure 6** shows a more realistic (albeit still theoretical) case, using regularly spaced
 380 nucleo-centric bins, and including noise contamination of the Abel transform F . **Figure 6a**
 381 shows the theoretical profile (dashed line) and the noisy profile used as input to the Abel
 382 inversion algorithm (solid line). **Figure 6b** shows the absolute value of the relative difference
 383 between the dashed and solid lines of **Figure 6a**. **Figure 6c** shows the ideal theoretical line-
 384 of-sight (l.o.s.) integrated profile (dotted line) with the l.o.s. integrated profile fitted using
 385 triangular elements divided by r^2 , with and without regularization (long and short dashes,

386 resp.). The uncertainties over the fitted curves that result from noise propagation, are
 387 represented as dark (light) shades for the non-regularized (regularized, respectively) profile.
 388 Figure 6d shows the theoretical local emission rate (dotted line) and the nonregularized (short
 389 dashes) and regularized (long dashes) fitted profiles. Again, the $\pm 1\sigma$ uncertainties over the
 390 fitted profiles are represented as dark (light) shade for the non-regularized (regularized, resp.)
 391 fitted emission rate. Both the regularized and nonregularized fits nearly retrieve the exact
 392 value, but the benefit of regularization clearly appears, as the long-dash curve is smoother and
 393 thus broadly closer to the correct values. This is also reflected by the much smaller
 394 uncertainties affecting those values, especially at large nucleo-centric distance, where F
 395 becomes small. As it can be expected, both the boundary effects and the large simulated noise
 396 impair the quality of the fitted profile near the boundary at 20000 km. **Figures 6e** and **6f** show
 397 the same as **Figures 6c** and **6d**, respectively, except that purely triangular elements were used
 398 in the inverse Abel transform fit. The quality of the results shown in **Figures 6e** and **6f** is
 399 obviously not as high as those from **Figures 6c** and **6d**. Regularization even appears as
 400 counterproductive in this case. This naturally results from the less adapted shape of the
 401 elements used here. It thus clearly appears that the best choice is to use elements u_k with $m = 2$
 402 to study cometary profiles, and to apply the regularization procedure. The regularization used
 403 here aims at minimizing the integral of the square of the second derivative of the fitted a_k
 404 (equations (23) to (28)). Regularization based on the minimization of the sum of the square of
 405 the second order discrete difference of the a_k gives fairly similar results. On the other hand,
 406 regularization based upon the second derivative of the fitted F (not shown) performs worst, as
 407 anticipated above. We note that the argument that we developed to suggest that minimizing
 408 the integral of the second derivative of F might not be the best choice for cometary
 409 atmospheres could also apply to the regularization applied to the fitting parameters obtained
 410 using purely triangular elements. Our best choice finally appears to be to use triangles divided
 411 by r^2 and regularization operating directly on the a_k because the $1/r^2$ multiplication partly
 412 corrects for the drawbacks of the alternative regularization choices.

413 3.2. Disturbed inert species profiles

414 The tests presented up to now used emission rate profiles proportional to $1/r^2$. This
 415 choice does perfectly correspond to the u_k elements used to realize the fits and one may
 416 wonder if these elements would still be appropriate if the emission profile departs from this
 417 best possible case. We thus constructed an emission rate profile consisting of a $1/r^2$ profile to
 418 which a bump (idealized by a Gaussian curve) was added. We carefully performed the l.o.s.
 419 integration numerically (using a very high space resolution and extending the emission rate
 420 profile far beyond 20000 km) and used the inversion method with that l.o.s.-integrated profile
 421 as input. The results are shown in **Figure 7a** and **b**. Regularized inversion with triangles
 422 divided by r^2 is used. The bump added to the profile is indeed retrieved, although the match is
 423 not perfect (such a disturbance of the profile is certainly more severe than any disturbance we
 424 may imagine to find in a real cometary observation). The fitted emission rate becomes
 425 disturbed beyond the bump, because the fitting parameters co-vary and are disturbed by the
 426 bump and by noise. In this extreme test, the propagated noise then becomes a poorer estimator

427 of the uncertainty over the local emission rate profile, and the fitted profile shows erratic
428 oscillations around the correct value.

429 We also performed another important test: the inverse Abel transform of a profile
430 varying as $1/r$ (i.e. for which we expect to retrieve the local emission rate varying as $1/r^2$) to
431 which a constant offset is added. This test is important because cometary observations have a
432 contribution from the background sky, which can often be considered as constant across the
433 whole coma, although some observations have a sky background that varies across the image,
434 especially if the bright moon approaches the field of view. Subtraction of this offset is often a
435 difficult task, and thus a source of uncertainty. The theoretical expression of the inverse Abel
436 transform does however only involve the first derivative of F so that, if it could be applied to
437 real data, it would give a result independent of the constant offset due to the sky background.
438 Unfortunately, real data are noisy, binned over a discrete set of nucleo-centric distances, and
439 spatially limited, so that we have to rely on numerical methods that may be sensitive to the
440 offset. **Figure 7** shows our simulation of an observation contaminated by an offset in panels c
441 and d. The noise applied to the input l.o.s.-integrated emission (F) is not included in the plot
442 for clarity. The constant added to the $\sim 1/r$ l.o.s.-integrated profile has been purposely chosen
443 very large, causing a doubling of F already near $r = 1000$ km. The fitted l.o.s.-integrated
444 profile does not seem to correctly retrieve the augmented profile (the dash-dot-dot-dot line). It
445 rather seems to be offset by a larger amount, with a rapid decrease near the boundary at
446 20000 km. The emission rate profile, however, does more closely correspond to the $\sim 1/r^2$
447 profile, except near the boundary at 20000 km. It is surprising that, despite the ~ 1 order of
448 magnitude contamination of F near 10000 km (already a factor 2 near 1000 km), and despite
449 the erroneous retrieval of F at large nucleo-centric distance, the emission rate is rather
450 correctly retrieved over a broad part of the profile. This stems from the fact that two l.o.s.-
451 integrated profiles differing from each other by only an additive constant have the same
452 inverse Abel transform. The numerical inversion technique developed here is not fully
453 insensitive to the added constant. Consequently, the good strategy to follow when analyzing
454 an observed coma would be to estimate the constant background of the sky as accurately as
455 possible, subtract it from the observed cometary emission and apply the Abel inversion,
456 knowing that the result will be only weakly sensitive to a misestimate of the constant
457 background, across a large portion of the observed coma. This advantage alone can already be
458 seen as a good reason for inverting the l.o.s.-integrated observation and study the emission
459 rate itself. It must be added that all the theoretical tests proposed here were performed using
460 as many fitting elements as pseudo observation points (i.e. $J = K$ and $r_j = r_{0,j}$ in the formalism
461 developed in the preceding section). Other choices are possible and can sometimes give even
462 better results. Quite obviously, least squares fitting is, in principle, a method that is generally
463 used to determine a relatively small number of relevant parameters using a larger number of
464 observations, increasing the number of observation points leading to smaller uncertainties
465 over the fitted parameters. An interesting option is also to use fitting elements centered at
466 nucleo-centric distance larger than that of the last point of the F profile, because the Abel
467 transform of these elements will anyway extend to lower nucleo-centric distance. This choice
468 could be particularly interesting when the signal-to-noise ratio remains very good across the
469 whole observed profile. In principle, regularization could even allow us to “fit” more elements

470 than the number of observation points: the matrix $\mathbf{H} + \lambda\mathbf{Q}$ (equation (14)) would generally not
471 be singular in that case. However, it is illusory to expect to obtain meaningful results using
472 that choice: one can hardly expect to retrieve more information than what stands in the data.
473 The result would rather reflect some kind of additional “information” introduced in the system
474 by the regularization.

475 3.3 Daughter species profiles.

476 Similar tests were conducted for emissions having a radial profile represented by a
477 Haser model for daughter species characterized by realistic scale lengths $L_p = 50000$ km and
478 $L_d = 120000$ km. We found that using fitting elements located at nucleocentric radius larger
479 than that of the outermost point of the simulated observed profile does improve the quality of
480 the fitted emission rate near the outer boundary of the radial range of the observation. When
481 the interval covered by the fitting triangular elements is restricted to that of the radial range of
482 the observation, the emission rate retrieved by the inversion method is overestimated,
483 compared with the expected emission rate following a Haser profile for daughter species. This
484 can be understood as follows: the l.o.s. integration of the emission includes contributions from
485 the emission originating from altitudes above the tangent point. Truncation of the emission
486 rate profile removes contributions to the l.o.s. integration that would be necessary to properly
487 represent the (simulated) observation near the outer boundary of the profile. The least squares
488 fit algorithm compensates for this defect by overestimating the emission rate in the last bins of
489 the adjusted profile. Consequently, considering extra triangular elements beyond the tangent
490 radius of the outermost observation (but still keeping the total number of elements lower than
491 the number of points of the observed profile) introduces contributions that allow for a better
492 retrieval of the emission rate near the outer boundary of the observed, l.o.s. integrated profile.
493 However, beyond some radius, the fitted emission rate can become negative, which does
494 obviously not make any physical sense. Conclusions regarding the emission rate profile at
495 cometocentric radii larger than the tangent radius of the outermost observation can thus not be
496 considered safe and better had to be avoided. Given that the inclusion of those extra bins is
497 not to extend the range of validity of the inverted profile beyond the radius of the last
498 observed point but rather to introduce a few degrees of freedom in the fit procedure to better
499 model the observation at large nucleocentric distance, only a few extra bins suffices to
500 improve the fit. In our test, some 10-15 extra bins extending the grid by some $1/2 - 2/3 L_d$
501 revealed efficient.

502 The tests conducted for the case of daughter species following a Haser model also
503 show that the numerical Abel inversion does, at least partly, remove the effect of a constant
504 background that might contaminate an observed profile. Numerical inversion uses a discrete
505 representation over a truncated profile. One should not expect miracles though and hope the
506 numerical inversion would remove the constant background contamination the way the
507 analytical inverse Abel transform would do over an infinite radial range. There is a benefit in
508 performing the numerical inverse Abel transform, but this benefit is not as large as the
509 theoretical result of equation (5) might suggest.

510 It is common practice in cometary data analysis to determine the parameters of a Haser
 511 model representative of the observation using a least squares fit to the observation. As the
 512 emission rate profile can be estimated using a numerical inverse Abel transform applied to the
 513 data, one may wonder whether it is preferable to adjust the Haser model parameters directly
 514 on the observed profile rather than on the emission rate profile deduced from the inversion.
 515 We test this issue over Haser profiles of known parameters.

516 In a least squares fit procedure applied to a l.o.s. observation, the l.o.s. integration of
 517 the Haser model needs to be computed as well as the derivative with respect to the Haser
 518 parameters. We found the analytical expression of the l.o.s. integral of the Haser model for
 519 mother and daughter species. For least squares fitting purposes, we can express the Haser
 520 model for daughter species as $q_d = 1/L_d$, $q_p = 1/L_p$ and $Y = Q/(4\pi v) L_d/(L_d - L_p)$. One can
 521 equivalently use (Y, q_p, q_d) or (Y, L_p, L_d) as the fitting parameters, and the optimal fit is rapidly
 522 obtained noting the Haser radial profile $h(r)$:

$$\begin{aligned}
 h(r) &= Y(\exp(-q_d r) - \exp(-q_p r))/r^2 \\
 H(r_0) &= 2 \int_{r_0}^{\infty} dr \frac{r}{\sqrt{r^2 - r_0^2}} h(r) = 2 \frac{Y}{r_0} (P(q_d r_0) - P(q_p r_0)) \\
 P(a) &= \int_1^{\infty} dx \frac{x}{\sqrt{x^2 - 1}} \frac{\exp(-a x)}{x^2} \\
 &= \frac{\pi}{2} \left(1 - a(B_K(0, a) S_L(-1, a) + B_K(1, a) S_L(0, a)) \right) \tag{37} \\
 \frac{\partial H(r_0)}{\partial Y} &= \frac{H(r_0)}{Y} \quad ; \quad \frac{\partial H(r_0)}{\partial q_d} = -2 Y B_K(0, q_d r_0) \quad ; \quad \frac{\partial H(r_0)}{\partial q_p} = 2 Y B_K(0, q_p r_0) \\
 \text{or } \frac{\partial H(r_0)}{\partial L_d} &= 2 \frac{Y}{L_d^2} B_K(0, r_0/L_d) \quad ; \quad \frac{\partial H(r_0)}{\partial L_p} = -2 \frac{Y}{L_p^2} B_K(0, r_0/L_p)
 \end{aligned}$$

523

524 where $B_K(n, x)$ is the modified Bessel function of the second kind (Bessel-K) and $S_L(n, x)$ is the
 525 modified Struve function (also called the Struve-L function), which can be evaluated from a
 526 fast-converging series (Abramowitz and Stegun, 1972). Details regarding the calculation of
 527 $P(a)$ can be found in appendix 2 (as well as the series for the S_L function). The Abel transform
 528 of the Haser model for mother molecules can obviously be computed as well using function
 529 $P(a)$ given in equation (37).

530 The Haser parameters of a radial profile for daughter species are estimated by
 531 accounting for a simulated noise and the possible presence of a constant background
 532 contribution in the simulated observation. This simulated radial profile is computed using the
 533 result of equation (37), contaminated by a Poisson noise and a constant offset background,
 534 and inverted using the numerical inverse Abel transform. The Haser parameters $L_{p,los}$ and $L_{d,los}$
 535 are estimated using a Levenberg-Marquardt method applied to the simulated observation,
 536 while $L_{p,em}$ and $L_{d,em}$ are fitted over the emission rate determined by the numerical inverse
 537 Abel transform. When no background is included in the simulated observation, both methods
 538 give similar values for L_p and L_d , although $L_{p,los}$ and $L_{d,los}$ seem to fall somewhat closer to the

539 exact values used as an input. However, when a residual background is present in the
540 simulated profile, it is $L_{p,em}$ and $L_{d,em}$ that seem to be closer to the expected values. The
541 presence of a small positive offset reduces the slope of the simulated l.o.s. integrated profile at
542 large cometocentric distances. This leads to an increase of the fitted $L_{d,los}$ and a reduction of
543 $L_{p,los}$. Because the numerical inverse Abel transform partly removes the effect of the constant
544 offset, the fitted $L_{p,em}$ and $L_{d,em}$ are less disturbed and they fall closer to the exact value.
545 Naturally, if the nucleocentric profile does not rigorously follow a Haser model, only an
546 inversion of the observed profile can estimate the emission rate profile.

547

548 4. APPLICATION TO OBSERVED COMETARY 549 ATMOSPHERES

550 In this section, we will apply the method derived in section 2 and tested in section 3 to
551 real cometary data obtained using the TRAPPIST telescope (Jehin et al., 2011). TRAPPIST is
552 a 60-cm robotic telescope installed in 2010 at La Silla observatory. The telescope is equipped
553 with a 2Kx2K thermoelectrically-cooled FLI Proline CCD camera with a field of view of
554 22'x22' and a plate scale of 1.302"/pix. A set of narrow-band filters isolating the main
555 emission bands in the optical spectrum of comets, i.e., OH, NH, CN, C₃, and C₂, as well as
556 emission-free continuum regions at four wavelengths (Farnham et al., 2000) is permanently
557 mounted on the telescope.

558 The reduction method applied to the TRAPPIST data has been extensively described
559 by Opitom et al. (2015) and will only be briefly summarized here. TRAPPIST images are
560 reduced following a standard procedure using frequently updated master bias, flat and dark
561 frames. The removal of the sky contribution may be problematic for extended objects.
562 However, for the comets considered hereafter, the TRAPPIST field of view was always wide
563 enough to determine the sky contribution from parts of the images free of cometary
564 contribution. We first determine the location of the comet's optocenter in the image (using the
565 Iraf task `imcntr`). Second, we determine the closest distance from the coma optocenter where
566 each image is free of cometary emission, and measure the median sky level at this
567 nucleocentric distance, which is subtracted from the image. We then derive the median radial
568 brightness profile for each image. The use of a median profile eliminates the contribution of
569 background stars. Even though narrowband filters have been carefully designed to isolate
570 specific molecular emissions, they are contaminated by the underlying sunlight reflected by
571 the dust. The dust subtraction is thus a very important step in the data reduction. We use
572 images of the comet in the BC filter (i.e. at 444.9 nm) to obtain the dust radial profile, scale it
573 depending on the contamination in the gas filter, and subtract it from the gas profile.
574 Continuum frames used for the dust subtraction are usually taken during the same hour as the
575 associated frame to avoid changes in the observing conditions or in the rotational state of the
576 comet. Regular observations of narrowband photometric standard stars listed in Farnham et al.
577 (2000) allow us to determine each filter zero point and extinction coefficients used to convert
578 count rates into fluxes.

579 4.1. Estimation of the uncertainties

580 We derive the local rates of various cometary emissions from their l.o.s.-integrated
 581 observations, i.e., from their Abel transform. Estimating the uncertainties affecting the
 582 observations is often difficult. Some of these uncertainties will not have a dramatic effect over
 583 the range of local emission rates that we will estimate: a small misestimate of the sky
 584 background has nearly no effect over the result of the inverse Abel transform, as was
 585 explained in section 3. We thus adopt a rather pragmatic method to estimate the uncertainties
 586 over the observed emission profile. If we note G_j the observation of a given emission,
 587 obtained under a nucleo-centric tangent radius $r_{0,j}$ (all sorted by increasing tangent radius), the
 588 uncertainty σ_j affecting this observation is directly estimated from the neighboring
 589 observations using the following method. First, we smooth the observed radial profile to
 590 obtain the set of numbers G_j^* ($j = 1, \dots, J$). This smoothing is realized using a Savitsky-Golay
 591 filter (Savitsky and Golay, 1964) applied to the logarithm of the G_j 's. This choice is made
 592 because of the fast decrease rate of the l.o.s.-integrated cometary profile: the logarithm of the
 593 G_j 's varies much slower than the original data. One can view the Savitsky-Golay filtering
 594 method as a generalization of the boxcar smoothing. In a boxcar smoothing directly applied to
 595 the data G_j , G_j^* would be the average of the G_i 's over i varying from $j - d$ to $j + d$, the size of
 596 the smoothing "box" being $2d + 1$ elements. This is equivalent to replacing the G_j 's by a
 597 zeroth order polynomial fitting the neighboring elements of G_j . The Savitsky-Golay filter
 598 generalizes this idea: a polynomial of arbitrary degree chosen by the user is fitted over a set of
 599 elements of the array of data centered on G_j , the set having a width $2d + 1$ (chosen by the user
 600 as well). It reduces to a convolution with a kernel (that we will denote $\mathbb{K}_{q,d}$) that depends on
 601 the chosen degree of the polynomial (which we will denote q) and the width over which the
 602 smoothing is realized (namely d). Here, instead of applying the filter directly to the data, we
 603 apply it to the logarithm of the data and compute the exponential of that smoothed set. Once
 604 the smoothed array G_j^* is obtained, we use it to locally de-trend the observed profile G_j and
 605 compute the mean and standard deviation over that restricted interval, as if the de-trended
 606 result gave several estimates of G_j :

$$\begin{aligned}
 G^* &= \exp(\ln(G) \star \mathbb{K}_{q,d}) \\
 m_j &= \frac{1}{2g + 1} \sum_{i=j-g}^{j+g} G_i \frac{G_j^*}{G_i^*} \quad (j = 1, \dots, J) \\
 \sigma_j &= \sqrt{\frac{1}{2g + 1} \sum_{i=j-g}^{j+g} \left(m_j - G_i \frac{G_j^*}{G_i^*} \right)^2} \quad (j = 1, \dots, J)
 \end{aligned} \tag{38}$$

607

608 In equation (38), the operator \star stands for the convolution product and g is a positive integer
 609 which defines the number of adjacent measurements used to estimate the uncertainties over
 610 the G_j 's. It must be chosen sufficiently large to allow for a reasonably meaningful estimation

611 of the uncertainty, but it must also remain small enough so that the set of de-trended
612 measurements $G_i G_j^* / G_i^*$ ($j - g < i < j + g$) can be viewed as several estimates of G_j , which is
613 obviously never strictly true. Moreover, the sums appearing in (38) present problems near the
614 boundaries of the measurements (near $j = 1$ and $j = J$). The sums need to be truncated
615 accordingly, and the denominator amounting to the number of elements actually involved in
616 the sum must be corrected. We performed numerical tests that tend to indicate that the method
617 of equation (38), when applied to a profile typical of a cometary atmosphere (i.e., the Abel
618 transform of a Haser model) with known uncertainties (i.e., a randomly generated noise with a
619 standard deviation proportional to the square root of the profile) tends to somewhat
620 underestimate the uncertainties. In our tests, that bias could be corrected for by applying a
621 safety factor of 1.2 to the estimated σ_j so that the estimated uncertainties better correspond to
622 the known noise used in the numerical test, although one should not expect this nearly unit
623 factor would dramatically influence the results. Yet, there is another subtlety that has to be
624 accounted for in these expressions. The Savitsky-Golay filter reduces to a (numerical)
625 convolution product of the (logarithms of the) G_j 's with an appropriate kernel. Close to the
626 boundaries, and in particular close to the inner boundary (i.e., for $r_{0,j}$ near 0), truncation of the
627 convolution degrades the quality of the smoothed profile G_i^* , leading to unacceptably wrong
628 (over)estimates of σ_j . We correct this problem by scaling σ_j ($j < j_{\text{crit}}$) along the square root of
629 G_j , with j_{crit} being the index of the first j at which the convolution product and the estimates of
630 equation (38) can be carried out without truncation problem: $j_{\text{crit}} = d + |d - g|/2$. This scaling
631 choice makes sense when the uncertainties are mostly due to the Poisson noise affecting the
632 measurements.

633 **Figure 8** shows how this method of noise estimation performs when applied to an
634 ideal profile with known uncertainties. We generate a l.o.s.-integrated Haser profile
635 discretized over 500 equally-spaced nucleo-centric distances. We then compute its square root
636 that we use as a standard deviation to generate a Poisson noise to be applied to the ideal Haser
637 profile (dotted line in Figure 8a). We apply a Savitsky-Golay filter (as outlined in equation
638 (38)) using a width of 21 points and a fifth degree polynomial, i.e., with $d = 10$ and $q = 5$
639 (long dashed line in Figure 8a). Obviously, the smoothed profile is a poor estimate of the
640 l.o.s.-integrated emission rate at low nucleo-centric distance. We then compute the local
641 average and standard deviation as explained in equation (38) over 31 neighboring points (i.e.,
642 with $g = 15$), and applying the safety factor of 1.2. The result is shown using the long dashes
643 in panel b. Quite obviously, this estimate of the uncertainty is very wrong near the inner
644 boundary, while it fairly follows the dotted line at larger nucleo-centric distance. We then
645 apply the square root scaling at low nucleocentric distance as explained above to obtain the
646 uncertainties in the part of the profile where the filter-based method does not suffice (short
647 dashed line in Figure 8b). The estimate of the uncertainty is then fairly good all over the
648 profile. Incidentally, the uncertainties are somewhat underestimated at low nucleo-centric
649 distance because the uncorrected method produced a slightly underestimated uncertainty near
650 j_{crit} , but overall, the uncertainties are recovered in an acceptable manner. The method used to
651 estimate the noise level is actually independent of the inverse Abel transform itself. It has
652 been introduced to derive values for the uncertainties in the χ^2 expression (11) and for the
653 error propagation procedure that is used to estimate the uncertainties of the fitting parameters.

654 Obviously, the reliability of any least squares fitting method improves when the uncertainties
 655 are accounted for. Indeed, weighting with adequate uncertainty estimates helps to prevent an
 656 overfitting of the noise affecting the large contributions to the profile (i.e., at low
 657 nucleocentric distance) at the expense of the fitting of physically meaningful signatures that
 658 may arise at large nucleocentric distance where the measured intensity is much smaller. From
 659 that standpoint, a rough estimate of the uncertainties suffices.

660 4.2. Data analysis

661 We first apply our methods to observations of comet 103P/Hartley 2 obtained with
 662 TRAPPIST on November 7, 2010. Comet 103P/Hartley 2 was discovered in 1986. It is a
 663 Jupiter Family comet with a period of 6.47 years. 103P/Hartley 2 is one of the few comets that
 664 have ever been visited by a spacecraft: it was the target of a close flyby by the NASA Deep
 665 Impact space probe on November 4, 2010. In parallel to the flyby, an extensive space-borne
 666 and ground-based campaign was initiated to complement the in-situ observations. The comet
 667 passed within only 0.12au from the Earth two weeks before the flyby, allowing its coma to be
 668 sampled with high precision from the ground. We analyze the emission of molecule CN at
 669 387 nm, i.e., the R branch of the (0-0) band of the $B^2\Sigma^+ - X^2\Sigma^+$ transition. In comets, the CN
 670 radical is predominantly produced by photo-dissociation of molecular HCN (Fray et al., 2005)
 671 (another possibility would be by dissociative recombination of HCN^+ ions). Excitation of the
 672 $B^2\Sigma^+ - X^2\Sigma^+$ system of bands is due to absorption of the solar light and its analysis should
 673 ideally account for the presence of the Fraunhofer bands in the solar spectrum (Arpigny,
 674 1964).

675 **Figure 9** shows the inversion results. The flux was measured at 723 different nucleo-
 676 centric distances and we used 242 triangular elements (equation (29)), i.e., $\sim 1/3$ of the number
 677 of points in the observed flux profile. A few triangular elements were added at radial values
 678 beyond the last point of the observed profile, for the reasons explained at the end of section 3.
 679 Regularization was applied on the integral of the second derivative of the fitting parameters
 680 (equation (28)). Figure 9a shows that the method produces a good fit of the observed flux;
 681 Figure 9b furthermore shows that the emission rate is reconstructed with very small
 682 uncertainties. Please notice that the (differential) flux is given per steradian, so that a factor of
 683 4π is applied after Abel inversion to retrieve the volumetric emission rate. Clearly, the
 684 uncertainties that we retrieve are somewhat underestimated at very large nucleo-centric
 685 distance: the small increase of the emission rate near 90000 km does not seem to be realistic,
 686 and it probably results from a small shoulder seen in the observed flux near that nucleo-
 687 centric distance. We also determined a Haser model by least squares fitting over the emission
 688 rate, using the Levenberg-Marquardt method. Its characteristic lengths are $L_p = 17500$ km and
 689 $L_d = 70100$ km. We deduce the effective production rate Q_{HCN} (assuming that dissociation of
 690 HCN is the only source of CN, which may be an oversimplification) associated with this
 691 profile obtained while comet 103P/Hartley 2 was at a heliocentric distance of $r_H = 1.07$ ua,
 692 moving with a radial velocity of $\dot{r}_H = 3.2$ km/s. We use the g-factor $g_{\text{CN}} = 3.44 \times 10^{-13}$ erg s⁻¹
 693 molecule⁻¹ based on the study of Schleicher (2010), which accounts for both the heliocentric
 694 distance and the radial velocity (important for the Swings effect). Assuming an expansion

695 velocity of 1 km/s, we estimate that $Q_{\text{HCN}} = 2.68 \times 10^{25}$ particles s^{-1} . This number must be
696 considered with care, as the Haser model relies on oversimplified assumptions. We compared
697 these numbers with those obtained by fitting the Haser model directly using the observed flux,
698 again using the Levenberg-Marquardt method. A fast implementation of the fit is possible as,
699 for a Haser model, all the needed quantities can be computed analytically using the results of
700 equation (37).

701 The fit realized directly over the observed flux gives $L_p^{(\text{F})} = 2.16 \times 10^4$ km, $L_d^{(\text{F})} = 4.9 \times 10^4$ km
702 and $Q_{\text{HCN}}^{(\text{F})} = 3.22 \times 10^{25}$ particles s^{-1} , which slightly differ from the values obtained above
703 from the emission rates. The HCN production rates inferred for the different comets
704 considered in this study are listed in **Table 1**. It must be noted that after adding a constant
705 offset of 1.06×10^{-5} erg cm^{-2} s^{-1} sr^{-1} to the observed flux prior to fitting the Haser model, we
706 retrieve very closely the same characteristic lengths and production rates than for the fits
707 realized over the emission rates. This highlights once more that a small offset affecting the
708 observed flux can have significant consequences (although not dramatic in this case): the
709 fitted Haser parameters are sensitive to an offset applied to the observed flux (when the fit is
710 realized directly over the flux), but the Abel-inverted flux (i.e., the emission rate) is nearly
711 insensitive to a small offset. The difference between the values of the fitted parameters may
712 be due to an overestimate of the sky background that was subtracted, which corresponds to an
713 equivalent flux of 6.4×10^{-5} erg cm^{-2} s^{-1} sr^{-1} . It must, however, also be emphasized that comet
714 Hartley 2 may well have an extended source region with a not well-determined size (A'Haern
715 et al., 2011). This is an obvious departure from the hypothesis of the classical Haser model,
716 mostly important near the nucleus, and that influences the fitted characteristic lengths. In
717 addition, the dynamics of the acceleration of the gas produced by the cometary nucleus takes
718 place in a volume extending several tens of kilometers away from the nucleus, as shown by
719 the Monte Carlo simulations of the expanding coma (Tenishev et al., 2001, 2008, Rubin 2001,
720 Combi, 1996). As a consequence, Haser model cannot be valid within some distance from the
721 nucleus, even in the absence of outgassing from dust grains, which further worsens the
722 correspondence with the Haser model when present. On the other hand, the numerical inverse
723 Abel transform does not rely on any particular assumption concerning the shape of the profile
724 of the coma (except for the assumption of spherical symmetry), so that the presence of an
725 extended source or any other signature in the radial profile (providing that it is large enough
726 to be resolved by the observing instrument) does not impinge on the quality of the results of
727 the method itself. Anyway, determining the properties of the coma near a comet's nucleus
728 remains challenging, because a feature needs to fill at least 2-4 pixels of the observed profile
729 to be properly analyzable, due to the limits imposed by the Nyquist theorem. It can also be
730 considered that the presence of a residual, non-constant contribution from the background can
731 never be totally ruled out. This uncertain disturbance can however be expected to be small
732 after subtraction of the estimated background and to mostly affect the radial profiles at large
733 nucleocentric distance. In contrast, uncertainties concerning the centering of the image of the
734 coma are more likely to disturb the radial profile at small nucleocentric distance. Along the
735 same lines, the cumulated effects of flat-field, radial and azimuthal averaging and velocity
736 terms in the outflow can become important at large nucleocentric distance. Both the fitting of
737 the Haser model and the numerical inverse Abel transform will incorporate these effects as if

738 they were physically meaningful contributions, which can somewhat bias the radial profile at
739 large nucleocentric distance.

740 Interestingly enough, the emission rate plotted in log-log scales presents a change of
741 slope near $r = L_p$. This change of slope is less visible in the observed flux, although one can
742 make it out a posteriori, after having first noticed it in the emission rate profile. Comet 103P/
743 Hartley 2 was located at a heliocentric distance of 1.07 au at the time of the observation. The
744 reference characteristic lengths from A'Hearn et al. (1995) for CN are $L_d^* = 2.1 \times 10^5$ km and
745 $L_p^* = 1.3 \times 10^4$ km at 1 ua, to be scaled by the square of the heliocentric distance giving $L_p =$
746 14900 km and $L_d = 240000$ km. The fitted L_p is comparable with the standard reference value,
747 but L_d is quite different. Determining a scale length much longer than the radius range over
748 which the data are available is however a difficult task, and it is not sure it is always possible,
749 especially when noise affects the data (and has to be taken into account for the fit, as it was
750 done here) and when the model does not perfectly match the observation (as it can often be
751 expected from a Haser model).

752 It must be noted here that, when a model is adjusted using a least squares fit with
753 weighting by the inverse of the variances, it is expected that the differences between the data
754 and the fitted curve would be distributed along a Gaussian centered on the fitted curve. It is
755 not exactly the case here: the data are not distributed exactly symmetrically with respect to the
756 fitted flux because the regularization modifies the concept of optimum (the algorithm does not
757 strictly minimize the classical χ^2) and produces a smoother result.

758 Emission of CN at 387 nm from comet C/2012 F6 Lemmon was also observed with
759 ESO-TRAPPIST on February 17, 2013. Comet Lemmon was a very active naked eye comet
760 that reached mag 5 at perihelion, on March 24, 2013. It is a dynamically old, long-period
761 comet following a highly eccentric and inclined orbit. **Figure 10** shows the inversion of its
762 profile. Again, the observed flux is correctly fitted by the method. The emission rate is
763 affected with minor uncertainties only. However, the emission rate does not seem to make
764 sense near the comet's nucleus, although the observed flux is perfectly fitted. This is due to
765 the fact that flux measurements near the nucleus are somewhat more uncertain than the low
766 level of noise affecting it may let suppose. For example, accurate centering (identification of
767 the exact location of the nucleus in TRAPPIST images) is a source of uncertainties, as well as
768 the subtraction of possible contributions from dust, especially if we take into account that the
769 dust profile was obtained separately from the CN profile, so that the centering of both
770 observations of the comet may not perfectly match. The inverted profile offers here a means
771 to diagnose a feature that might have remained unnoticed in the radial profile of the flux:
772 either the first points of the profile are erroneous, or this is a real feature of the radial profile
773 of the emission rate. Indeed, the first points of the profile near the nucleus must be considered
774 with care because, in terms of the Nyquist theorem, information can hardly be obtained at a
775 resolution better than 2-4 pixels. Indeed, the limitations considered in the analysis of the
776 Hartley 2 data also hold in this case, so that conclusions reached regarding the extremes of the
777 radial profile must be considered with care. For a comet such as C/2012 F6 Lemmon, which
778 was very productive, a low emission rate profile near the nucleus could be the signature of
779 significant absorption of the solar UV radiation. Validation of this hypothesis would need a

780 thorough verification, which is beyond the scope of the present study. The emission rate can
 781 again be represented using a Haser model using least squares fitting, giving characteristic
 782 lengths $L_p = 3.11 \times 10^4$ km and $L_d = 2.35 \times 10^5$ km. This last length is comparable (up to a factor
 783 ~ 2) with the radial range of the data used to determine it and should be considered with
 784 caution. F6 Lemmon was located at a heliocentric distance $r_H = 1.01$ au and had a heliocentric
 785 radial velocity $\dot{r}_H = -21.9$ km/s at the time of the observation, giving $g_{\text{CN}} = 4.41 \times 10^{-13}$ erg s⁻¹
 786 molecule⁻¹, from which we estimate the effective production rate $Q_{\text{HCN}} = 8.88 \times 10^{26}$ molecule
 787 s⁻¹. The reference characteristic lengths of CN are $L_d^* = 2.1 \times 10^5$ km and $L_p^* = 1.3 \times 10^4$ km at
 788 1 ua, to be scaled by the square of the heliocentric distance giving $L_p = 1.33 \times 10^4$ km and $L_d =$
 789 2.14×10^5 km. The fitted L_d value is comparable with the standard reference value, while the
 790 shorter L_p values differ by a gross factor of 2. By fitting a Haser model directly on the
 791 observed flux, we find $L_p^{(\text{F})} = 3.8 \times 10^4$ km, $L_d^{(\text{F})} = 1.77 \times 10^5$ km and $Q_{\text{HCN}}^{(\text{F})} =$
 792 1.02×10^{27} particles s⁻¹. These characteristic lengths differ again from those obtained using the
 793 emission rate profile. Again, adding a small offset (2.8×10^{-4} erg cm⁻² s⁻¹ sr⁻¹) to the flux can
 794 bring the fitted lengths closer to those of the emission rate profile, suggesting again the effect
 795 of the sensitivity to the sky background. The background subtracted from this TRAPPIST
 796 image did however correspond to $\sim 2 \times 10^{-5}$ erg cm⁻² s⁻¹ sr⁻¹, an order of magnitude lower than
 797 the needed offset, so that the explanation for the difference must be searched for elsewhere. A
 798 possible explanation could be that the flat flux found near the comet nucleus implies that the
 799 emission rate must, surprisingly, increase with the nucleo-centric distance in the first layers of
 800 the coma. A Haser model cannot reproduce such an emission rate. However, the flat flux is
 801 rather smooth and non-increasing, which is easier to model using a Haser profile. The
 802 inadequacy of the Haser model to represent the coma of comet F6/Lemmon could then be the
 803 origin of the discrepancy. It must also be kept in mind that the anomalous, increasing
 804 emission rate is found within a radius corresponding to only ~ 2 pixels of observation, and the
 805 inferred variation may thus just be an artifact due to the insufficient resolution of the
 806 observation, uncertainties in the centering and the background subtraction, etc. As already
 807 discussed above, the analysis of the data obtained near the comet nucleus is not
 808 straightforward.

809 **Figure 11** shows the emissions of CN molecules at 387 nm and of C₂ at 514.1 nm
 810 from the comet C/2013 A1 Siding Spring on November 11, 2014 observed again with the
 811 TRAPPIST telescope. Comet Siding Spring was discovered at 7.2 au from the Sun on January
 812 3, 2013 and it was soon predicted to have a close encounter with planet Mars on October 19,
 813 2014. The comet has been extensively observed from the ground and from orbiters around
 814 Mars at the time of the encounter. It underwent an outburst that increased the gas production
 815 fivefold within a few days, less than two weeks after its perihelion passage on October 25,
 816 2014 (Opitom et al., 2016). The C₂ 514.1 nm emission belongs to the (0-0) band of the Swan
 817 transition system $d^3\Pi_g - X^1\Sigma_g^+$. Molecular C₂ can be produced by photodissociation of C₂H₄,
 818 C₃H₆ and possibly C₂H₆ in cometary atmospheres (Weiler, 2012; Helbert et al., 2005), and the
 819 514.1 nm emission is fed by absorption of the solar light and is due to (at least at large nucleo-
 820 centric distance) the complex fluorescent equilibrium that includes the transitions $A^1\Pi_u - X$
 821 $^1\Sigma_g^+$, $b^3\Sigma_g^- - a^3\Pi_u$, $d^3\Pi_g - a^3\Pi_u$, $d^3\Pi_g - c^3\Sigma_u^+$, $a^3\Pi_u - X^1\Sigma_g^+$, and $c^3\Sigma_u^+ - X^1\Sigma_g^+$

822 (Rousselot et al., 2000). The comet was located at a heliocentric distance of 1.43 au and had a
 823 heliocentric radial velocity $\dot{r}_H = 5.03$ km/s. The fluorescence g-factors obtained from
 824 Schleicher (2010) for CN and A'Hearn et al. (1982) for C₂ under these conditions are $g_{\text{CN}} =$
 825 $g_{\text{C}_2} = 2.2 \times 10^{-13}$ erg s⁻¹ molecule⁻¹ (both values are incidentally equal). The radial profiles of
 826 the observed fluxes and of the emission rates deduced after Abel inversion are shown in
 827 **Figure 11**. Again, we used three times less u_k elements than the number of bins in the
 828 observed profiles plus a few bins beyond the last observed point, and we applied the
 829 Tikhonov regularization, so that the fitted flux of the C₂ emission is smoother than the
 830 observed flux. Its uncertainties remain small, though. The inverted radial profile for the
 831 emission rate appears to be overestimated. Indeed, the small “bump” that appears in the flux
 832 near $r = 5000$ km does not seem to be real. This feature does however not seem to be dramatic
 833 in the l.o.s.-integrated flux, but it influences the radial profile of the emission rate.
 834 Accordingly, the inverted profile turns out to be a useful tool to diagnose the quality of the
 835 flux profile or perhaps a real phenomenon: indeed, this feature could possibly be attributed to
 836 an underestimate of the contribution from the dust, which was subtracted, and that seems to
 837 become less important beyond ~ 10000 km. Moreover the second pixel of the profile
 838 corresponds to $r = \sim 3000$ km only, and the issue raised above concerning the Nyquist
 839 frequency holds here again, re-emphasizing that it is difficult to draw definite conclusions
 840 from observations obtained close to the nucleus. The Haser model fitted to the radial profile
 841 of the C₂ 514.1 nm emission rate was obtained neglecting the contribution of the points below
 842 10000 km. We find nearly identical values for L_p and L_d : $L_p = 34273$ km and $L_d = 34302$ km
 843 while the effective production rate of the C₂ parents is found to be $Q_{\text{C}_2\text{Hn}} = 4.75 \times 10^{26}$ particles
 844 s⁻¹. The lengths are given with such a high accuracy because, having L_p exactly equal to L_d
 845 would be physically inconsistent. The limit of the Haser model (**equation 3**) for L_d tending to
 846 L_p is proportional to $1/r$, which cannot be integrated over \mathbb{R}^3 . In addition, the Abel transform
 847 (**equation 4**) of such a profile tends to infinity, whatever the value of r_0 . Finding nearly equal
 848 values for L_p and L_d may possibly indicate that there is outgassing from the dust grains.
 849 Combi and Fink (1997) explain that C₂ radial profiles are usually flatter than would be
 850 expected for the photodissociation of a single parent molecule, and can then be more easily
 851 reproduced with a Haser model that has two almost equal scale lengths. Interestingly, the
 852 radial profile of the emission rate of CN has $L_p = 37646$ km and $L_d = 37688$ km. CN would
 853 thus also have nearly equal characteristic lengths, which are above all nearly identical to those
 854 of C₂, thus corroborating the hypothesis of outgassing from grains. However, as we will show
 855 in the next paragraph, such a conclusion cannot be drawn in the case of Siding Spring. The
 856 effective production rate derived from the CN emission rate profile is $Q_{\text{HCN}} =$
 857 4.20×10^{26} particles s⁻¹.

858 There are oscillations that can be seen at large nucleo-centric distance (above
 859 $\sim 1.5 \times 10^5$ km) in the radial profile of the emission rate of C₂ and, to a lesser extent, in the
 860 emission rate profile of CN where a change of slope appears (in the log-log plot of **Figure**
 861 **11b**). These signatures require particular attention. Comet Siding Spring is known to have
 862 produced an outburst shortly before these data were obtained (Opitom et al., 2016). Inverse
 863 Abel transform is particularly adapted to retrieve the radial profile of the emission rate in this
 864 dynamic case, as standard models generally assume steady state. Indeed, both the C₂

865 514.1 nm and the CN 387 nm fluxes show a smooth change of slope around 10^5 km. The
866 radial profile of the CN 387 nm emission rate clearly shows a slope breaking at 1.5×10^5 km.
867 A similar breaking is also seen at the same place in the radial profile of the C_2 514.1 nm
868 emission (see **Figure 12**), especially comparing the emission rate obtained by Abel inversion
869 and the Haser model fitted to the emission rate at nucleocentric distance larger than
870 1.5×10^5 km. Note that C_2 is known to have a shorter lifetime than CN, leading to a smaller
871 characteristic length (A'Haern et al., 1995). This leads to a faster radial decrease of the C_2
872 emission rate compared with CN, as it is easily seen in **Figure 12**, and the signature of the
873 outburst is then harder to detect in the C_2 profile. The oscillation that appears in the C_2
874 emission and peaks at 3×10^5 km may be due to the poorer quality of the observed flux near
875 that nucleo-centric distance, and it is hard to draw conclusions about it. It remains that both
876 the CN and C_2 emission rate profiles show a clear signature of the outburst, seen as a breaking
877 of both profiles near 1.5×10^5 km. The information is of course present in the radial profiles of
878 the observed flux, but the l.o.s. integration smoothens the features present in the emission rate,
879 and it is harder to determine where the junction between the pre- and post-outburst coma is
880 located. The presence of the outburst also casts another light on the characteristic lengths
881 deduced from the fitting of a Haser profile over the emission rate: the observed coma does not
882 comply with the hypothesis of the Haser model, that assumes a fairly constant production rate,
883 and it is hazardous to draw any conclusion over the outgassing mechanisms at play in the
884 coma at that time (although outgassing from grains could make sense right after the outburst,
885 if it were related with an explosive release of matter). In contrast, the numerical inverse Abel
886 transform does not rely on any assumption regarding the functional shape of the radial profile
887 and it can thus account for possible dynamic variations of the production rate of the nucleus
888 and for a possible extended source.

889 5. DISCUSSION

890 We developed an inverse Abel transform method with Tikhonov regularization that
891 specifically accounts for the properties of cometary atmospheres. We used triangular elements
892 matching the density profiles of chemically inert species. However, using more elaborate
893 elements that closely resemble the Haser model for daughter species might have been more
894 appropriate. We had to make a tradeoff between adequacy of the elements and computational
895 efficiency. First of all, the Abel transform of these alternative elements would have been more
896 difficult to compute. Secondly, the least squares fitting on which the method relies would
897 have become non-linear. The impact of more sophisticated triangular elements is difficult to
898 assess. Our theoretical tests tend to show that the elements used in this study have properties
899 that are adequate for the processing of cometary observations.

900 The Abel inversion method calls upon the hypothesis of spherical symmetry of the
901 coma. This assumption is probably never strictly fulfilled, although one may expect that it is
902 valid far from the nucleus. It is difficult to appreciate how large deviations from spherical
903 symmetry can possibly be. Alternatively, one could develop a model under the hypothesis of
904 axial symmetry about the rotation axis of the comet and directly use 2D imaging of the coma
905 to perform an inversion. We conducted a preliminary analysis that suggests such a method

906 could probably be developed and applied when the orientation of the rotation axis is known
907 with sufficient accuracy. However, further developments are required to fully explore the
908 potential of such a method. One may also dream of a method that would produce a 3D
909 tomographic inversion of cometary observation. Such a method would rely on the (inverse)
910 Radon transform which is extensively used in medical imagery, so that an impressive know-
911 how exists about that topic. Such an inversion would however require observations under all
912 possible look directions (i.e. from vantage points distributed in the 4π steradians around the
913 comet). That kind of observation will not be available on a regular basis in a foreseeable
914 future, if it ever becomes available.

915 Application to real cometary observation showed to be efficient in the sense that
916 realistic emission rate profiles could be retrieved from the Abel inversion of the observed flux
917 of radiation. However, comparison between the properties of Haser models fitted over the
918 emission profile and over the observed flux reveals differences in the inferred scale lengths. It
919 is possible to reconcile the numbers by applying a small offset to the observed flux data prior
920 to fitting a Haser model to them, given that the inverse Abel transform applied to noisy data is
921 only weakly sensitive to an offset (which may be related to inaccuracies in the estimate of the
922 sky background). This ad hoc cure may however seem somewhat artificial as it introduces an
923 additional degree of freedom to the problem to reach internal consistency. The independence
924 of the theoretical inverse Abel transform over any applied offset gives nevertheless
925 confidence in the offset explanation of the apparent discrepancies, although an imperfection
926 of the data reduction technique can never be totally ruled out.

927 The inverse Abel transform has proven to be a powerful tool when applied to real
928 observations. It allows an easy diagnosis of the properties of the observation. We were able to
929 identify a possible anomaly in the dust contribution subtracted from the observation of comet
930 A1/ Siding Spring. We were also able to identify a signature in the emission rate profile of
931 comet F6/ Lemmon that may be attributed either to an inaccuracy in the data (possibly due to
932 a problem with the exact identification of the location of the comet nucleus in the TRAPPIST
933 images for example) or that may have a physical explanation, such as significant absorption of
934 the solar UV light by the material of the coma, especially considering that comet F6/ Lemmon
935 was very productive. Whatever the explanation will be, those signatures would have remained
936 unnoticed in the flux profile, while they are patent in the emission rate profile. The analysis of
937 an image of an outburst of comet A1/ Siding Spring with our new method may provide
938 original insight: the separation between the pre- and post-outburst coma could be easily
939 identified in both emission rate profiles from molecules CN and C₂. If consecutive
940 observations can be obtained over timescales of a few hours up to a few days, it would be
941 possible to track the location of that junction versus time, to estimate the velocity at which it
942 propagates in the coma, and to determine at what time the outburst actually takes place at the
943 nucleus.

944 A further consistency check of the fitted parameters can be performed considering the
945 total content of daughter species in the coma. For a Haser density profile with production rate
946 Q and expansion velocity v , the number of particles inside a sphere of radius R centered on
947 the nucleus is

$$N(R) = \frac{Q}{v} \frac{L_d}{L_d - L_p} \left(L_p \exp\left(-\frac{R}{L_p}\right) - L_d \exp\left(-\frac{R}{L_d}\right) + L_d - L_p \right) \quad (39)$$

948 and the fraction of the total number of particles inside of that sphere is obtained by the ratio
 949 $\beta = N(R) / (Q L_d/v)$. When the coma is observed over range R of radii, $N(R)$ can also be
 950 directly obtained from the observation by integrating the flux (given per steradian) over the
 951 observed disc, if the fluorescence g-factor is known:

$$N(R) = 2\pi \int_0^R dr_0 r_0 F(r_0) \frac{4\pi}{g}. \quad (40)$$

952 In the case of comet 103P/Hartley2, the Haser parameters fitted over the emission rate give a
 953 coma content of 1.908×10^{30} CN molecules, 73% of which are contained inside a sphere of
 954 radius given by the maximum radius of the observation. The content of that sphere calculated
 955 from equation (40) is 1.372×10^{30} particles which, when divided by 0.73, gives an estimated
 956 total coma content of 1.88×10^{30} molecules, in excellent agreement with the value derived
 957 from the fitted Haser model. The results provided by the different methods are thus consistent,
 958 and in particular, corroborate the assumption of a Haser density profile, at least as far as the
 959 global properties of the coma are considered. We reached similar conclusions with the
 960 F6/Lemmon observations: both estimates of the CN coma content agree within 0.5%. On the
 961 opposite, in the case of comet Siding Spring, both methods for estimating the coma content
 962 differ by $\sim 11\%$ using the C_2 observation and $\sim 24\%$ using the CN observation, which indicates
 963 that a Haser model cannot be used to represent the density profiles of a coma shortly after an
 964 outburst.

965 The method does not make any assumption about the detailed nature of the
 966 observation (except that it is a cometary observation). It could thus be applied to any
 967 emission, to the study of dust, and it could be adapted to the study of absorption phenomena,
 968 such as star occultation for example, in which the material of the coma or of a planetary
 969 atmosphere absorbs the light emitted by stars depending on the amount of gas present along
 970 the total line of sight. In the case of planetary atmospheres, this technique can be used by
 971 measuring the absorption of sun light aboard an orbiting spacecraft. The method thus appears
 972 to be a promising tool capable of simplifying the analysis of various cometary observations.

973 More sophisticated representations of the density profile of the coma might also be
 974 included in the analysis of the emission rate retrieved after Abel inversion. The vectorial
 975 model of the coma offers a more detailed description of the photochemical processes
 976 responsible for the production of the daughter species, and thus of the destruction of the
 977 mother species. As explained by Festou (1981), inclusion of the vectorial effects has, as a
 978 major consequence, that molecules produced at a given location can end up at another
 979 location which is not necessarily located downstream of the production location. Daughter
 980 molecules are produced isotropically in a reference frame moving with the expanding gas of
 981 the coma. All the points of the coma are thus coupled by diffusive transport. In other words,
 982 the isotropic production of daughter molecules leads to a kind of smoothing of the
 983 composition of the coma. One can thus naturally expect that scale lengths fitted over the
 984 observed coma should be somewhat longer than those we would compute using the

985 photochemical reaction constants, given with an appropriate accuracy from laboratory
986 measurements, using a prescribed profile for the major constituents and neglecting molecular
987 diffusion. One would furthermore expect that these ad hoc fitted lengths would be influenced
988 by the value of the collisional mean free path, which constrains the diffusive transport of the
989 daughter molecules. The numerical Abel inversion method transforms line-of-sight integrated
990 quantities into local quantities. It can, unfortunately, not be used to identify the effect of
991 molecular diffusion without additional processing. The first and second derivatives of the
992 emission rate as a function of the radial distance could possibly provide quantitative
993 information on the effect of diffusive smoothing in relation with the collisional mean free
994 path, something that could probably not easily be done directly using the radial profile of the
995 flux alone. So far neither the feasibility nor the validity of this idea have been tested. The
996 practical implementation of such an analysis would need a reasonable estimate of the collision
997 cross sections required to evaluate the gas kinetic, and validation should rely on detailed
998 modelling of the molecular diffusion inside of the expanding coma (e.g., with a Monte Carlo
999 method or an average random walk technique such as the one developed by Combi and
1000 Delsemme (1980a,b)). This idea could be tested independently of the inversion technique
1001 developed here.

1002 6. CONCLUSIONS

1003 1. We have developed a numerical inverse Abel transform specifically adapted to
1004 cometary atmospheres. Its efficiency is considerably improved in combination with a
1005 Tikhonov regularization. It allows the usage of standard error propagation techniques to
1006 estimate the uncertainties that affect the local emission rates derived from the observed flux of
1007 radiation.

1008 2. The emission rates calculated with our inverse Abel transform are only weakly
1009 sensitive to a constant offset that might result from an inaccurate subtraction of the sky
1010 background with real-world data.

1011 3. We applied our inversion technique to a restricted set of observations of comets and
1012 found that it effectively yields realistic emission rate. The emission rate profiles allow an
1013 easier diagnostic of the characteristics of the observation, such as an erroneous estimate of the
1014 dust subtraction or the identification of a signature possibly attributable to significant UV
1015 absorption by the coma.

1016 4. When we applied our method to an outburst case, we were able to clearly identify
1017 the separation between the pre- and post-outburst parts of the coma, which further illustrates
1018 its efficiency.

1019

1020 APPENDIX 1

1021 In this section, we present the analytical results needed to use triangular elements
1022 $v_k = t_k/(r-a)^m$. Such elements could be useful to realize the inverse Abel transform of

1023 planetary observation. The value of the parameter a can be adjusted to make the elements
 1024 more appropriate for the properties of the observed atmosphere. The computation of the Abel
 1025 transform of elements v_k is necessary to realize the inversion of an observed profile and
 1026 requires the computation of indefinite integrals of the form

$$\int dr \frac{r}{\sqrt{r^2 - r_0^2}} t_k(r) \frac{1}{(r - a)^m}, \quad 0 < a < r_0. \quad (\text{A1.1})$$

1027 These can always be reduced to a linear combination of indefinite integrals of the form

$$W_m = \int dr \frac{1}{\sqrt{r^2 - r_0^2}} \frac{1}{(r - a)^m} \quad (\text{A1.2})$$

1028 completed with the first elements of the suite of integrals I_n given by equations (6) and (7)
 1029 when m is lower than 3. These integrals satisfy a recurrence relation and are also related by a
 1030 simple derivative with respect to the parameter a :

$$\frac{\partial W_m}{\partial a} = m W_{m+1} \quad (m > 0)$$

$$m (r_0^2 - a^2) W_{m+1} = \frac{\sqrt{r^2 - r_0^2}}{(r - a)^m} + (2m - 1)a W_m + (m - 1)W_{m-1}$$

$$W_0 = \operatorname{arcosh}\left(\frac{r}{r_0}\right) = \ln\left(\frac{r}{r_0} + \sqrt{\frac{r^2}{r_0^2} - 1}\right) \quad (\text{A1.3})$$

$$W_1 = \frac{1}{\sqrt{r_0^2 - a^2}} \operatorname{arctg}\left(\frac{ar - r_0^2}{\sqrt{(r_0^2 - a^2)(r^2 - r_0^2)}}\right)$$

$$W_2 = \frac{\partial W_1}{\partial a} = \frac{\sqrt{r^2 - r_0^2}}{(r_0^2 - a^2)(r - a)} + \frac{a}{\sqrt{(r_0^2 - a^2)^3}} \operatorname{arctg}\left(\frac{ar - r_0^2}{\sqrt{(r_0^2 - a^2)(r^2 - r_0^2)}}\right).$$

1031 The recurrence relation can be obtained by multiplying and dividing the integrand by a in
 1032 (A1.2), then replacing the factor a at the numerator by $a - r + r$ in order to make appear W_{m-1}
 1033 and a second indefinite integral that can be reduced by an integration by parts, leading to

$$a W_m = -W_{m-1} + \frac{\sqrt{r^2 - r_0^2}}{(r - a)^m} + m \int dr \frac{r^2 - a^2 + a^2 - r_0^2}{\sqrt{r^2 - r_0^2}} \frac{1}{(r - a)^{m+1}}, \quad (\text{A1.4})$$

1034 where we have already introduced $-a^2 + a^2$ at the numerator of the integrand. The indefinite
 1035 integral in (A1.4) can now easily be expressed as a combination of W_m by noting that $r^2 - a^2 =$
 1036 $(r - a)(r + a)$. The factor $(r - a)$ can be cancelled with one and we finally retrieve the recurrence
 1037 relation (A1.3). W_0 can be directly derived from (A1.2). W_1 is more difficult to obtain, as it
 1038 cannot be derived from W_0 by simple derivation with respect to a . To obtain the expression
 1039 for W_1 , we first let $x = r/r_0$ and, accordingly, $dx = dr/r_0$ (we also denote $b = a/r_0$). We then
 1040 apply the substitution $x = 1/\cos(u)$, $dx = \operatorname{tg}(u)/\cos(u) du$, which leads to

$$W_1 = \frac{1}{r_0} \int du \frac{1}{1 - b \cos(u)} \quad (b = \frac{a}{r_0}) \quad (\text{A1.5})$$

1041 With the classical substitution $s = \text{tg}(u/2)$, i.e., $\cos(u) = (1-s^2)/(1+s^2)$, $\sin(u) = 2s/(1+s^2)$, $du =$
 1042 $2/(1+s^2) ds$ we get

$$W_1 = \frac{1}{r_0} \int ds \frac{2}{1 - b + (1 + b) s^2} \quad (\text{A1.6})$$

1043 which reduces to an arctangent. After back-substitution of the variable changes, one finds (up
 1044 to an additive constant):

$$W_1 = \frac{1}{r_0} \frac{2}{\sqrt{1 - b^2}} \text{arctg} \left(\frac{bx - 1 + x - b}{\sqrt{(1 - b^2)(x^2 - 1)}} \right). \quad (\text{A1.7})$$

1045 Noting that $2 \text{arctg}(y) = \text{arctg}(2y/(1-y^2)) + \pi$ and $\text{arctg}(y) + \text{arctg}(1/y) = \text{sgn}(y) \pi$, and
 1046 substituting $x = r/r_0$ and $b = a/r_0$, we finally get the expression from (A1.3)

$$W_1 = \frac{1}{\sqrt{r_0^2 - a^2}} \text{arctg} \left(\frac{ar - r_0^2}{\sqrt{(r_0^2 - a^2)(r^2 - r_0^2)}} \right) \quad (\text{A1.8})$$

1047 which is defined up to an additive constant. Derivation with respect to a immediately gives
 1048 W_2 and the recurrence can be started. Care must however be taken when using that recurrence.
 1049 The numerical tests that we performed suggest that it is not always stable. The relations given
 1050 in equation (A1.3) can nevertheless be used to derive analytical expressions of the W_m and
 1051 thus of any integral of the form of expression (A1.1). Based upon these results, the inversion
 1052 method developed above can be adapted for triangular elements of the form

$$v_k = t_k \frac{1}{(r - a)^m}, \quad (\text{A1.9})$$

1053 which have an Abel transform $V_k(r_0)$ for any $r_0 > a$ (all of which can now be calculated from
 1054 the W_0 , W_1 and W_2 above because of the linearity of the Abel transform), to be used instead of
 1055 T_k in the developments of equations (13), (16) and (19). In the case of planetary atmospheres,
 1056 choosing a of the order of the radius of the planet could be appropriate to build triangular
 1057 elements adapted to the observed atmosphere.

1058

1059 APPENDIX 2.

1060 In this section, we provide the detailed developments needed to analytically compute
 1061 the line-of-sight integration of a Haser model for mother and daughter species.

1062 All integrals appearing in the l.o.s. integration of a Haser model for parent and daughter
 1063 species can always be reduced, after the substitution $x = r/r_0$, to integrals of the form:

$$P = \int_1^\infty dx \frac{\exp(-q x)}{x \sqrt{x^2 - 1}}. \quad (\text{A2.1})$$

1064 We first derive P with respect to q :

$$\frac{dP}{dq} = - \int_1^{\infty} dx \frac{\exp(-q x)}{\sqrt{x^2 - 1}} = - \int_0^{\infty} dt \exp(-q \cosh(t)) \quad (\text{A2.2})$$

1065 where we made the variable change $x = \cosh(t)$, $t = \text{arcosh}(x)$, $dt = dx / (x^2 - 1)^{1/2}$. This
 1066 integral can be easily computed with the well-known formula for the modified Bessel
 1067 functions of the second kind, $B_K(n, z)$ (Abramowitz and Stegun, 1972):

$$B_K(n, z) = \int_0^{\infty} dt \exp(-z \cosh(t)) \cosh(n t) \quad (\text{A2.3})$$

1068 to be applied with $n = 0$, so that

$$\frac{dP}{dq} = -B_K(0, q), \quad (\text{A2.4})$$

1069 a result already given by Haser (1957).

1070 Now, we must compute the indefinite integral of $B_K(0, q)$ to retrieve P up to an additive
 1071 constant. We use the following formula, from Olver et al. (2010), and which can also be
 1072 found in the digital version of the NIST handbook of mathematical functions (the Digital
 1073 Library of Mathematical Functions, DLMF) as equation 10.43.2:

$$\begin{aligned} \int dz z^n e^{in\pi} B_K(n, z) \\ = \sqrt{\pi} 2^{n-1} \Gamma\left(n + \frac{1}{2}\right) z \left(e^{in\pi} B_K(n, z) S_L(n-1, z) \right. \\ \left. - e^{i(n-1)\pi} B_K(n-1, z) S_L(n, z) \right) + c, \end{aligned} \quad (\text{A2.5})$$

1074 where $S_L(n, z)$ represents the modified Struve function (also called Struve-L), which can be
 1075 easily computed using a fast-converging series expansion (Abramowitz and Stegun, 1972):

$$S_L(n, z) = \left(\frac{1}{2}z\right)^{n+1} \sum_{k=0}^{\infty} \frac{\left(\frac{1}{2}z\right)^{2k}}{\Gamma\left(k + \frac{3}{2}\right) \Gamma\left(k + n + \frac{3}{2}\right)} \quad (\text{A2.6})$$

1076

1077 If we let $n = 0$ in equation (A2.5), the gamma function can be evaluated as $\Gamma(1/2) = \pi^{1/2}$ and
 1078 expression (A2.5) reduces to

$$\int dz B_K(0, z) = \frac{\pi}{2} z \left(B_K(0, z) S_L(-1, z) + B_K(-1, z) S_L(0, z) \right) + c, \quad (\text{A2.7})$$

1079 so that we can write

$$P = -\frac{\pi}{2} q \left(B_K(0, q) S_L(-1, q) + B_K(-1, q) S_L(0, q) \right) + C. \quad (\text{A2.8})$$

1080 We determine the integration constant C by noting that, when q becomes infinitely large, the
 1081 integrand in (A2.1) becomes zero for any $x \in]1, \infty[$, so that P tends to 0 as well. The limit of

1082 equation (A2.8) for q tending to infinity is computed using the asymptotic developments
 1083 given by Abramowitz and Stegun (1972). For large values of z , noting $B_I(n,z)$ the modified
 1084 Bessel functions of the first kind, we have

$$B_K(n, z) \sim \sqrt{\frac{\pi}{2z}} e^{-z} \left(1 - \frac{4n^2 - 1}{8z} + \dots \right)$$

$$S_L(n, z) \sim B_I(-n, z) + \frac{1}{\pi} \sum_{k=0}^{\infty} \frac{(-1)^{k+1} \Gamma(k + \frac{1}{2})}{\Gamma(n + \frac{1}{2} - k) (\frac{z}{2})^{2k+n+1}} \quad (\text{A2.9})$$

$$B_I(n, z) \sim \sqrt{\frac{1}{2\pi z}} e^{+z} \left(1 - \frac{4n^2 - 1}{8z} + \dots \right)$$

1085 with $|\arg(z)| < 3\pi/2$ when z is complex. For very large values of z , the exponential term in the
 1086 expression of $B_I(n,z)$ will largely dominate the series that appears in the asymptotic
 1087 development of $S_L(n,z)$, so that we can immediately write that

$$\lim_{z \rightarrow \infty} z B_K(0, z) S_L(-1, z) = \lim_{z \rightarrow \infty} z B_K(-1, z) S_L(0, z) = \frac{1}{2} \quad (\text{A2.10})$$

1088 It follows that, in (A2.8), $C = \pi/2$ and we have

$$P = \int_1^{\infty} dx \frac{\exp(-qx)}{x \sqrt{x^2 - 1}} = \frac{\pi}{2} \left(1 - q (B_K(0, q) S_L(-1, q) + B_K(-1, q) S_L(0, q)) \right) \quad (\text{A2.11})$$

1089 It is always possible to compute P numerically, although this integration must be carried out
 1090 with extreme care as the integrand tends to infinity when x approaches 1. The analytical
 1091 expression (A2.11) uses special functions that can be rapidly computed with modern
 1092 computers, with an accuracy that will approach the machine precision. The advantage of
 1093 (A2.11) is thus twofold: it offers a better accuracy and it is faster than numerical integration,
 1094 which is important when P must be evaluated a large number of times, as it is the case in least
 1095 squares fit procedures. The benefit can be expected to be even larger when handling a more
 1096 sophisticated model using similar analytic expressions such as the three-generation Haseer-like
 1097 model (Combi et al., 2004).

1098 For the sake of completeness, we define a suite of integrals of the form

$$D_n = \int_1^{\infty} \frac{\exp(-qx)}{x^n \sqrt{x^2 - 1}} dx . \quad (\text{A2.12})$$

1099 Proceeding by parts, it is easily shown that these integrals satisfy a recurrence of third order
 1100 (letting $U = \exp(-qx)/x^{n+1}$ and $dV = x/(x^2-1)^{1/2} dx$):

$$D_{n+2} = \frac{q}{n+1} D_{n-1} + \frac{n}{n+1} D_n - \frac{q}{n+1} D_{n+1} . \quad (\text{A2.13})$$

1101 Evaluation of any three of the D_n ($n \geq 0$) suffices to start the recurrence, and successive D_n 's
 1102 are also related by a derivative versus parameter q . We already know $D_0 = B_K(0, q)$ (equation

1103 A2.4) and $D_1 = P(q)$ (equation A2.11). Integral D_2 is evaluated proceeding by parts, letting
 1104 $U = \exp(-qx)$ and $dV = dx / (x^2 - 1)^{1/2}$ to find

$$D_2 = q \int_1^\infty \frac{x \exp(-qx)}{\sqrt{x^2 - 1}} dx - q \int_1^\infty \frac{\exp(-qx)}{x \sqrt{x^2 - 1}} dx \quad (\text{A2.14})$$

1105 The first of these integrals is computed using the change of variable $x = \cosh(t)$,
 1106 $t = \operatorname{arcosh}(x)$, $dt = dx / (x^2 - 1)^{1/2}$ and equation (A2.3), while the second integral is given in
 1107 equation (A2.11) so that

$$D_2 = q B_K(1, q) - q P(q) . \quad (\text{A2.15})$$

1108 Recurrence (A2.13) can then be started and all the D_n 's can be computed. These results can be
 1109 used to compute indefinite integrals of $D_n(q)$ and in particular the analytical primitive of $P(q)$,
 1110 an unexpected result. Because $-D_{n-1}$ is the derivative of D_n versus q , the recurrence (A2.13)
 1111 can be transformed in a set of differential equations that admit D_n as solutions. A similar
 1112 remark can be made concerning recurrence (A1.3).

1113

1114

1115

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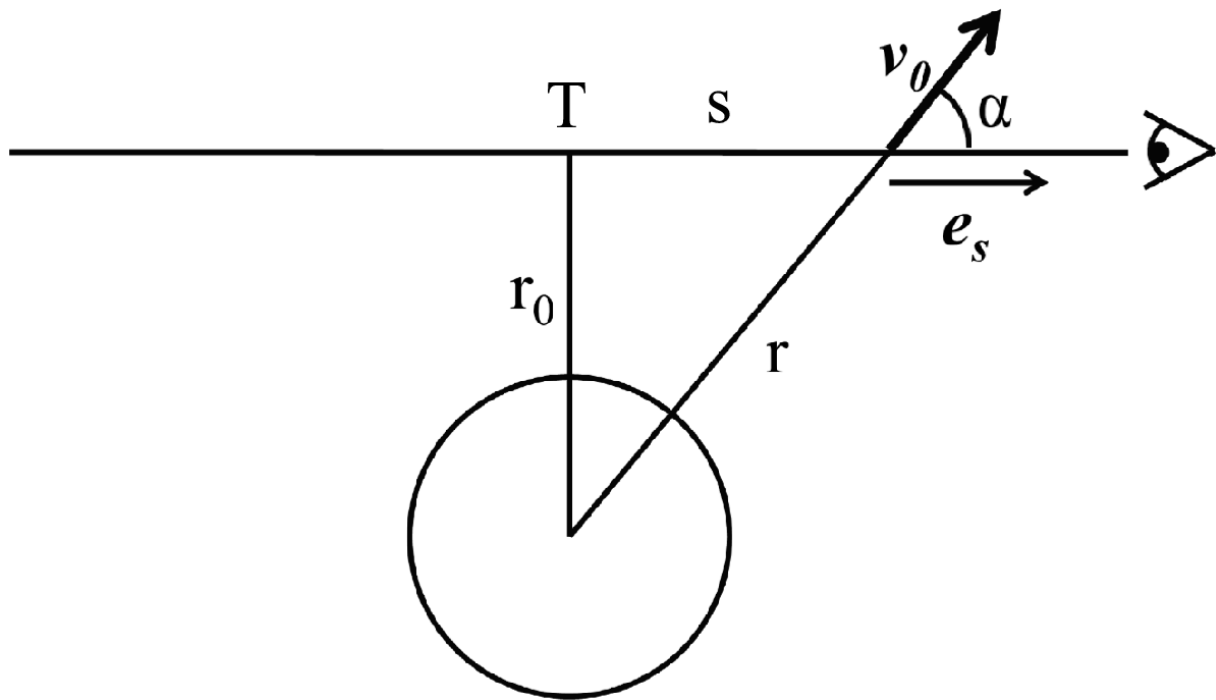
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1220 Figures and captions.

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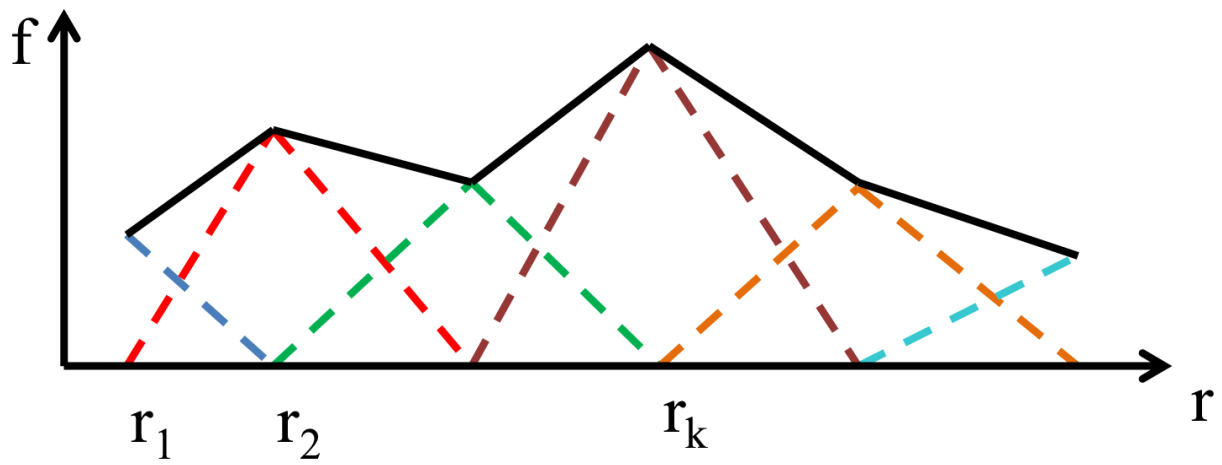
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1223 Figure 1. Geometry of the observation of an expanding coma. A remote observer collects the
1224 light emitted by the gas of the coma, summed up along the line of sight that passes through
1225 the tangent point T, i.e., the point of the line of sight nearest to the comet center. T is at a
1226 distance r_0 of the comet center, while a point of the line of sight is at distance r from the
1227 center. Variable s is counted from point T along the line of sight and can be considered to
1228 vary between $-\infty$ and $+\infty$ when the observer is at great distance. The coma expands radially at
1229 a velocity v_0 . The angle between the expansion direction and the line of sight, noted α ,
1230 changes along the line of sight.

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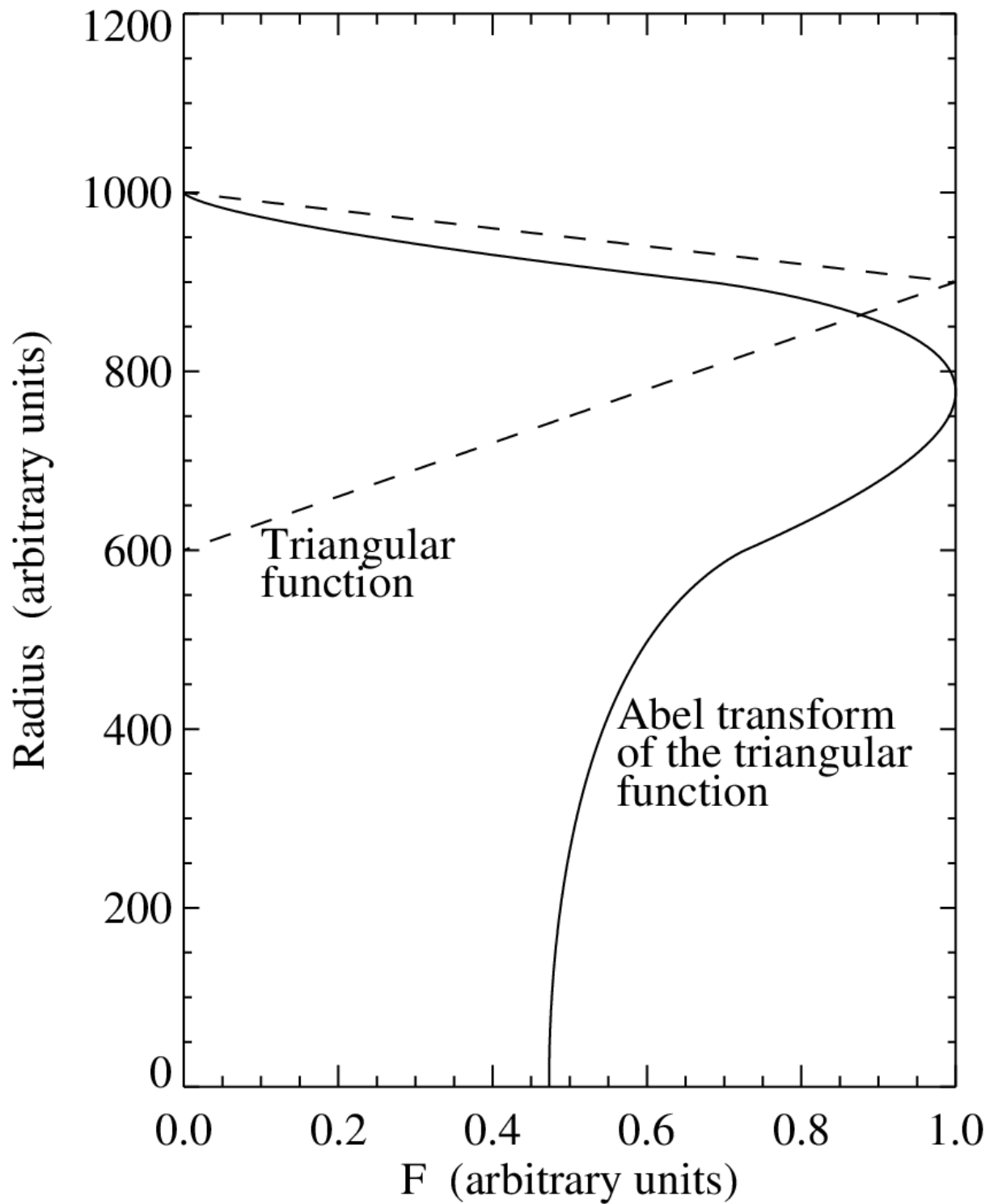


1234

1235 Figure 2. Representation of a function f decomposed into a set of linear segments using a
1236 linear combination of triangular functions. The sum of the colored dash lines triangles gives f ,
1237 represented with the black segments.

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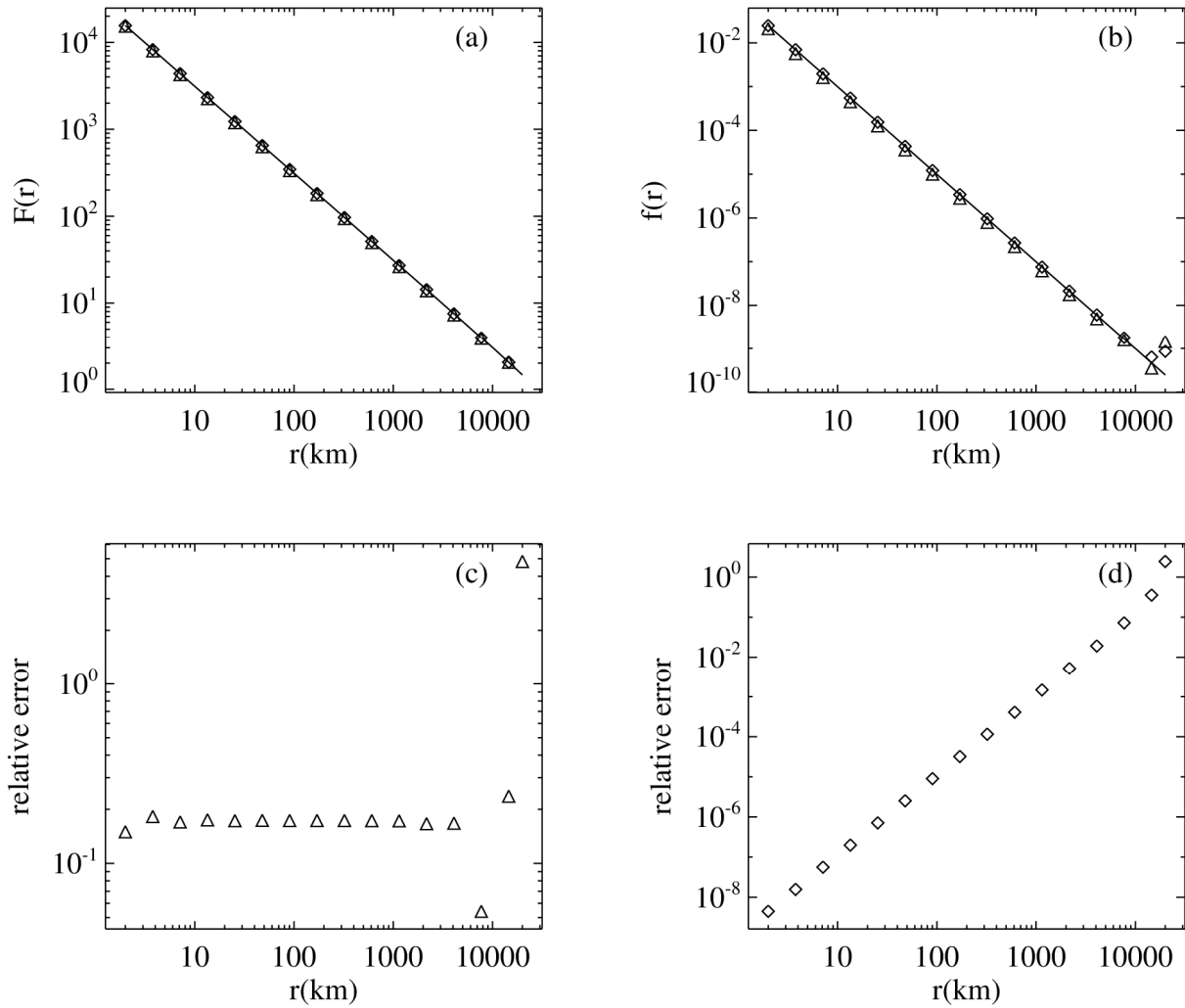
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1241 Figure 3. A triangular function and its Abel transform, both shown in arbitrary units. The peak
 1242 of the Abel transform occurs at a somewhat lower radial distance than that of the summit of
 1243 the triangle.

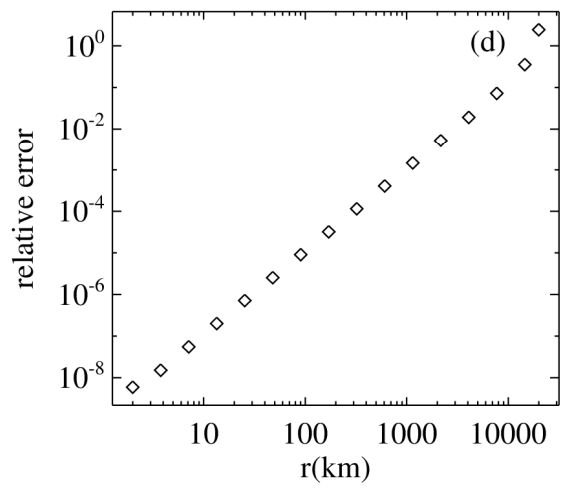
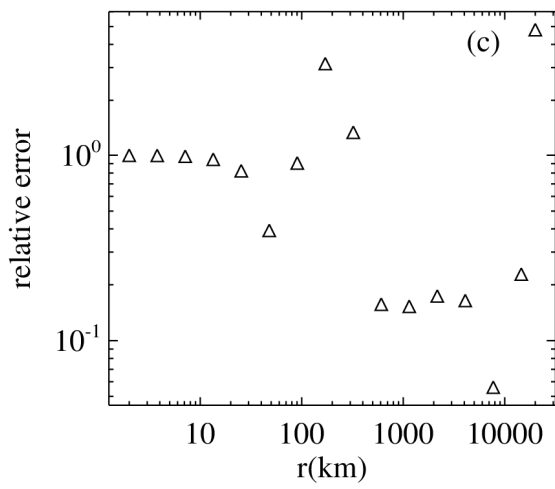
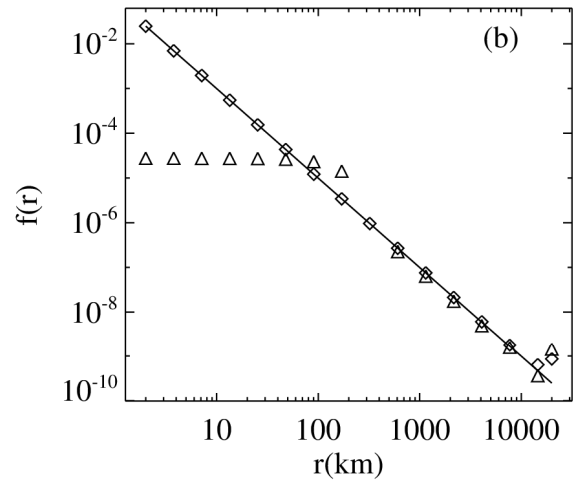
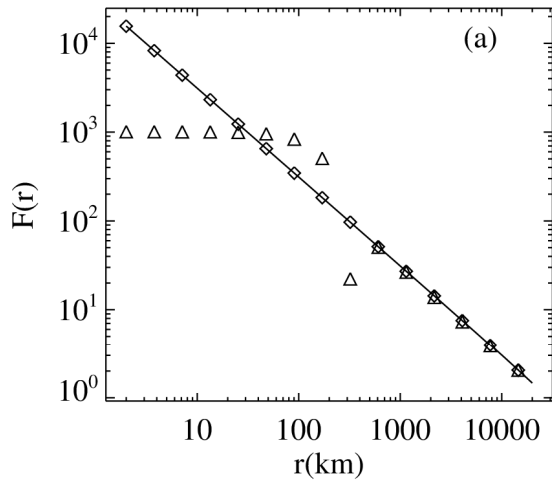
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1246 Figure 4. Abel transform of a theoretical nucleo-centric profile varying as $1/r^2$. Panel (a)
 1247 shows the Abel transform F : the solid line gives the exact analytical values, triangles and
 1248 diamonds show the profiles obtained after inverse transform fitting using purely triangular
 1249 elements and triangular elements multiplied by $1/r^2$ respectively. Panel (b) shows the emission
 1250 rate profile f , of which F is the Abel transform. Triangles show the emission rate profile fitted
 1251 using triangular elements; diamonds represent the profile fitted using triangular elements
 1252 multiplied by $1/r^2$. Panels (c) and (d) show the absolute value of the relative difference
 1253 between the exact and the fitted emission rates obtained using the purely triangular elements
 1254 and the triangular elements multiplied by $1/r^2$, respectively. No regularization was applied for
 1255 these inverse Abel transform fits.

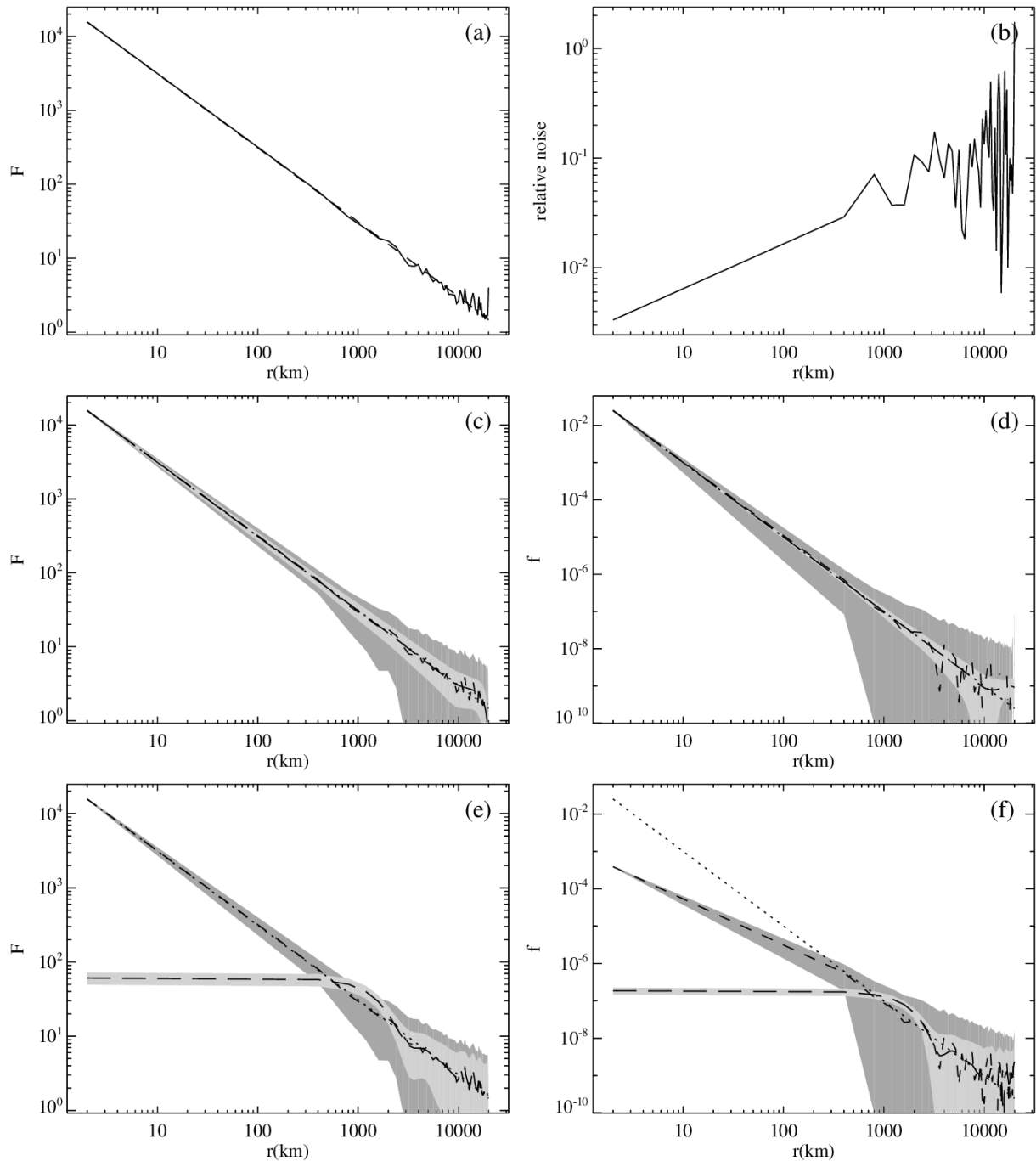
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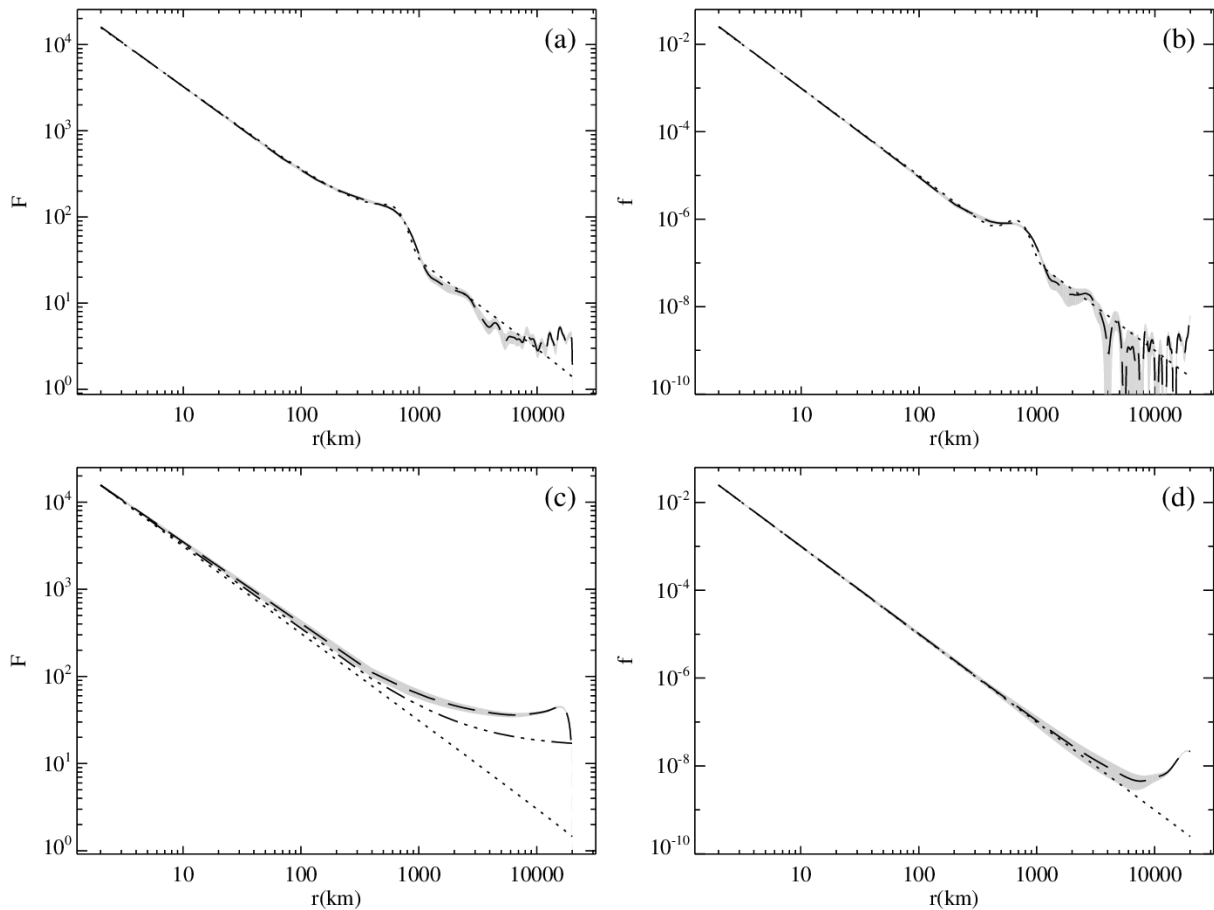
1258 Figure 5. Same as figure 4, but combined with a Tikhonov regularization.

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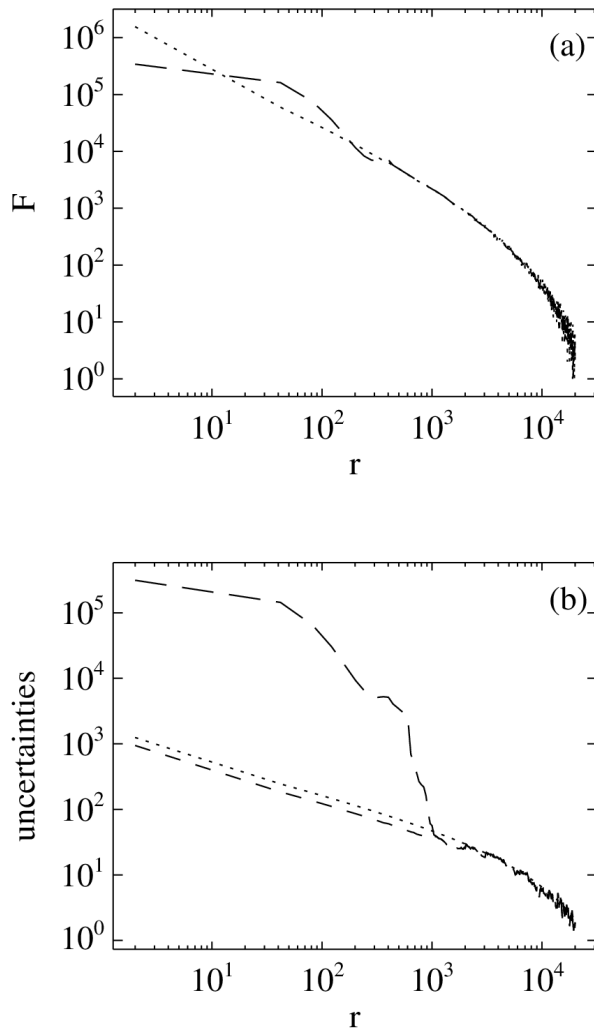
1261 Figure 6. Inversion of a realistic simulated profile including noise and a regular binning. Panel
 1262 (a): line-of-sight integrated profile, i.e., Abel transform of the $\sim 1/r^2$ emission rate. The dashed
 1263 line shows the exact transform, while the solid line shows the noisy values to be used in the
 1264 inverse Abel transform method. Panel (b) shows the absolute value of the relative difference
 1265 between the noisy and the smooth profiles from panel (a). Panel (c) shows the exact Abel
 1266 transform (dotted line) and the values fitted over the noisy profile of panel a, using triangular
 1267 elements divided by r^2 . Short (long) dashes show the fitted profile obtained without
 1268 (respectively with) regularization. The dark (light) grey shade show the $\pm 1\sigma$ interval obtained
 1269 applying error propagation for the unregularized (respectively the regularized) fit. Panels (e)
 1270 and (f) are similar to panels (c) and (d), respectively, using purely triangular elements for the
 1271 fits instead of triangles divided by r^2 .



1272

1273 Figure 7. Inversion of a $\sim 1/r$ l.o.s.-integrated profile modified by a Gaussian disturbance
 1274 (panels (a) and (b)). The ideal disturbed l.o.s.-integrated profile is represented by the dotted
 1275 line in panel (a). The noisy signal actually used as input for the inversion algorithm is omitted
 1276 for clarity. The long dashes show the fitted l.o.s.-integrated profile, the grey shade delimits the
 1277 $1-\sigma$ uncertainty band. Panel (b) shows the local emission rate, the dotted line represents the
 1278 exact profile that we seek to retrieve; the long dashes show the fitted profile with the $1-\sigma$
 1279 uncertainty delimited by the grey shade. Panels (c) and (d) show the results from the inversion
 1280 of a power law profile augmented by a constant offset. In panel (c), the dotted line shows the
 1281 $\sim 1/r$ l.o.s.-integrated profile, the dash-dot-dot-dot line shows the same profile increased by a
 1282 constant amount while the long dashes show the fitted profile with the grey shade delimiting
 1283 the $\pm 1\sigma$ uncertainty band. Panel (d) shows the target $\sim 1/r^2$ emission rate profile, the long
 1284 dashes represent the fitted profile and the grey shade delimits the $\pm 1\sigma$ uncertainty band.

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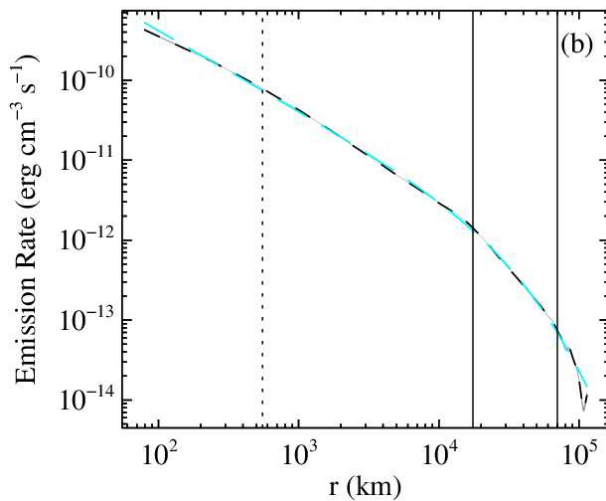
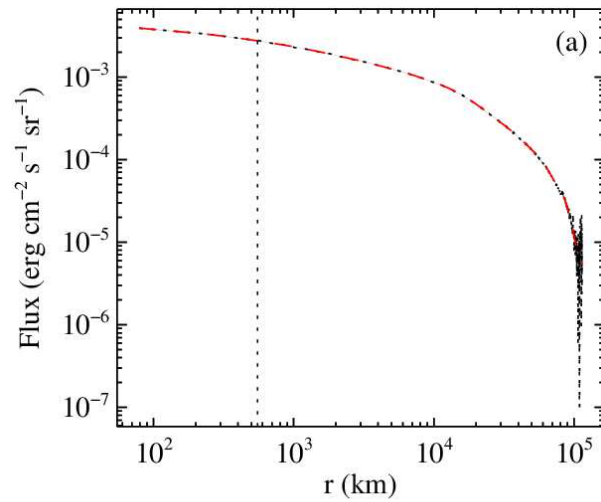


1286

1287 Figure 8. Panel (a): shows a simulated noisy l.o.s.-integrated Haser profile before (dotted line)
 1288 and after smoothing with a Savitsky-Golay filter (long dashes). Panel (b) shows the standard
 1289 deviation used to generated the noise of the profile shown in panel (a) (dotted line), which is
 1290 just the square root of the ideal profile (i.e., before artificial noise contamination). The long
 1291 dashed line shows the standard deviation estimated using the smoothed profile of panel (a)
 1292 and applying the formulas of equation (37) , while the short dashes show the uncertainties
 1293 obtained applying a square root scaling near the inner boundary of the profile.

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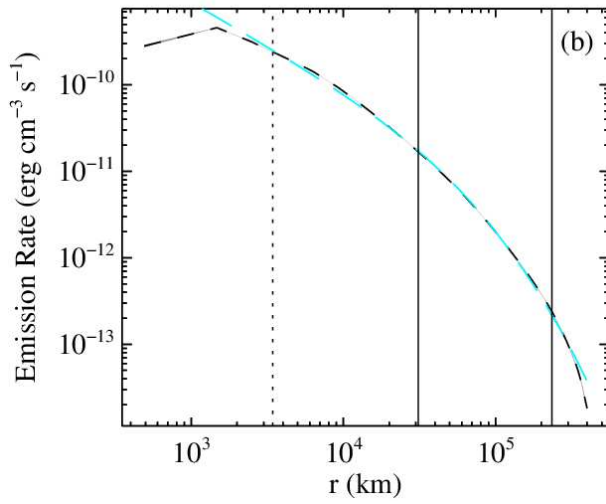
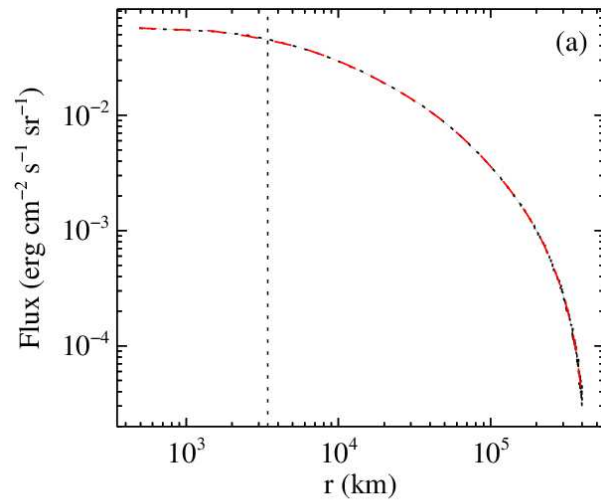


1296

1297 Figure 9. Observation of the CN emission of comet 103P/Hartley2 on 07 November 2010 at
 1298 387 nm. Panel (a) shows the observed flux, i.e., the l.o.s.-integrated data (dotted line). The red
 1299 long dashed curve shows the fitted flux obtained with the inverse Abel transform method;
 1300 uncertainties are shown as grey shades (they are lower than the line thickness in the plot).
 1301 Panel (b) shows the emission rate obtained using the inverse Abel transform of the observed
 1302 flux shown in panel (a) (black short dashes), with the uncertainties indicated by grey shades
 1303 (which are again smaller than the line width in the plot). The blue long dashes show a Haser
 1304 model fitted to the black dashes. It has characteristic lengths $L_p= 17500$ km and $L_d= 70100$
 1305 km, indicated by the vertical solid lines. The vertical dotted lines correspond to the fourth data
 1306 point of the observation.

1307

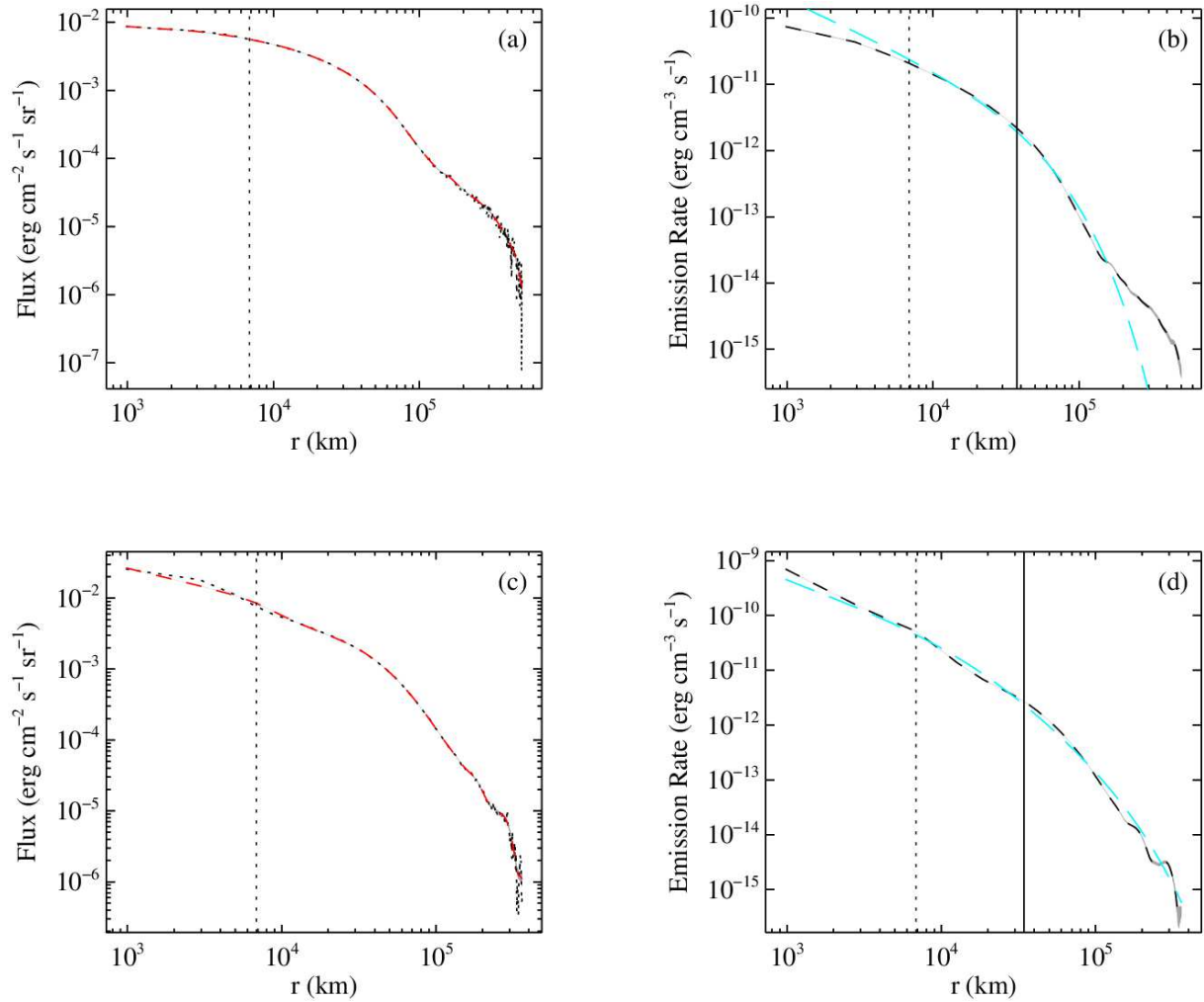
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1309

1310 Figure 10. TRAPPIST observation of the CN emission at 387 nm of comet C2012
 1311 F6/Lemmon on February 17, 2013 (line styles and colors as in **Figure 9**). The characteristic
 1312 lengths of the fitted Haser profile are $L_p = 31100$ km and $L_d = 235000$ km, indicated by the
 1313 vertical solid lines. The vertical dotted lines correspond to the fourth data point of the
 1314 observation.

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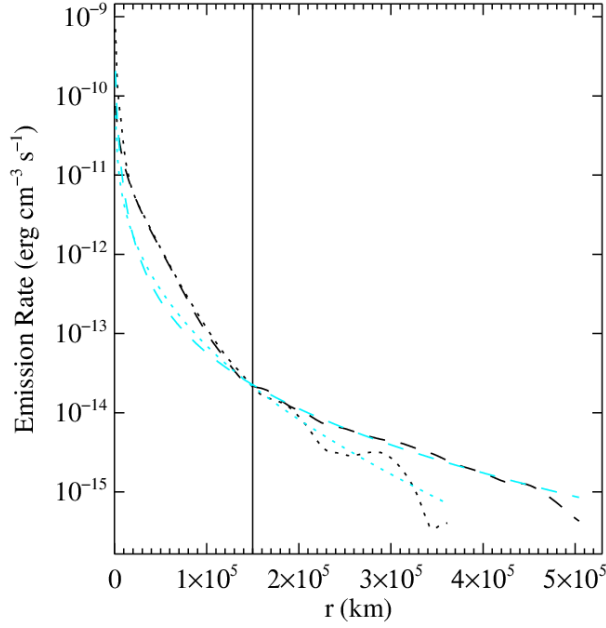


1316

1317 Figure 11. Radial profiles of the emissions of molecules CN at 387 nm (panels (a) and (b))
 1318 and C₂ at 514.1 nm (panels (c) and (d)) from comet C/2013 A1 Siding Spring on November,
 1319 11 2014. Line styles and colors are the same as in **Figure 9**. The characteristic lengths of the
 1320 fitted Haser models are $L_p = 37646$ km and $L_d = 37688$ km for CN and $L_p = 34273$ km and L_d
 1321 $= 34302$ km for C₂. The vertical dotted lines correspond to the fourth data point of the
 1322 observation.

1323

1324



1325

1326 Figure 12. Radial profile of the emission rates of CN at 387 nm (black dashed line) and of C₂
 1327 at 514.1 nm (black dotted line) obtained by inverse Abel transform fitting of the ESO-
 1328 TRAPPIST observation of comet C/2013 A1 Siding Spring on November, 11 2014. The light
 1329 blue lines represent a Haser model fitted to the emission rate of CN (dashed line) and C₂
 1330 (dotted line) at nucleocentric distance larger than 1.5×10⁵ km. The vertical line indicates the
 1331 breaking of both radial profiles as an outburst signature, separating the pre- and post-outburst
 1332 gas.

1333

	Q_{HCN}	$Q_{\text{HCN}}^{(\text{F})}$
103P/ Hartley 2	2.684	3.22
F6/ Lemmon	88.8	102
A1/ Siding Spring	42	

1334

1335 Table 1. Production rates of HCN inferred by least squares fitting of the emission rate (Q_{HCN})
 1336 and observed flux ($Q_{\text{HCN}}^{(\text{F})}$) profiles for comets Hartley 2, Lemmon and Siding Spring (in 10²⁵
 1337 particles s⁻¹). This latter comet experienced an outburst so that the production rate obtained by
 1338 the least squares fitting is of little significance and only Q_{HCN} is listed.

1339