

Using POD and DMD for comparing CFD and experimental results in unsteady aerodynamics

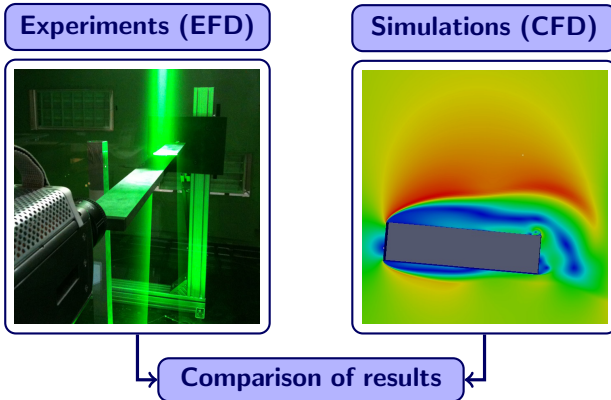
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University of Liege

November 25, 2014

APS-DFD

Motivation

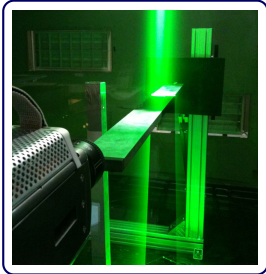


✓ Global quantities

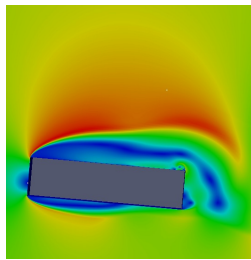
✗ Local quantities

Motivation

Experiments (EFD)



Simulations (CFD)



Comparison of results

- ✓ Global quantities
- ✗ Local quantities

Higher level of comparison is needed

POD and DMD used for comparison

POD finds the most persistent spatial structures ϕ_i^{POD}

$$\mathbf{u}(x, y, t) = \sum_{i=1}^N \underbrace{a_i^{\text{POD}}}_{\substack{\text{amplitude} \\ = \text{energy}}} \underbrace{f_i^{\text{POD}}(t)}_{\substack{\text{time} \\ \text{evolution}}} \underbrace{\phi_i^{\text{POD}}(x, y)}_{\substack{\text{spatial} \\ \text{mode}}}$$

DMD finds single frequency modes ϕ_i^{DMD}

$$\mathbf{u}(x, y, t) = \sum_{i=1}^N \underbrace{a_i^{\text{DMD}}}_{\text{amplitude}} \underbrace{\exp(\lambda_i^{\text{DMD}} t)}_{\substack{\text{time} \\ \text{evolution}}} \underbrace{\phi_i^{\text{DMD}}(x, y)}_{\substack{\text{spatial} \\ \text{mode}}}$$

POD and DMD used for comparison

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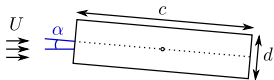
DMD finds single frequency modes ϕ_i^{DMD}

$$\mathbf{u}(x, y, t) = \sum_{i=1}^N \underbrace{a_i^{\text{DMD}}}_{\text{amplitude}} \underbrace{\exp(\lambda_i^{\text{DMD}} t)}_{\substack{\text{time} \\ \text{evolution}}} \underbrace{\phi_i^{\text{DMD}}(x, y)}_{\substack{\text{spatial} \\ \text{mode}}}$$

⇒ Different parts of decompositions are compared

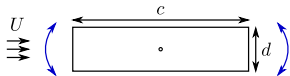
Methodology

Static rectangle



- $Re = Uc/\nu = 10^5$
- $c = 4d$
- $\alpha = 5^\circ$

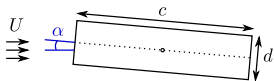
Pitching rectangle



- $Re = 4 \cdot 10^4$
- $k = \pi fd/U = 6.85 \cdot 10^{-2}$
- $\alpha_{MAX} = 10^\circ$

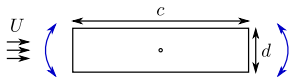
Methodology

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Pitching rectangle



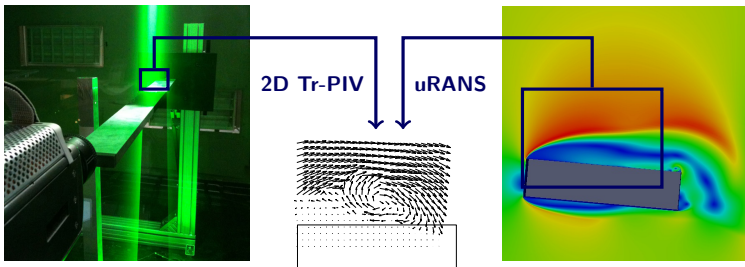
- $Re = 4 \cdot 10^4$
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Data Collection

Decomposition

Comparison

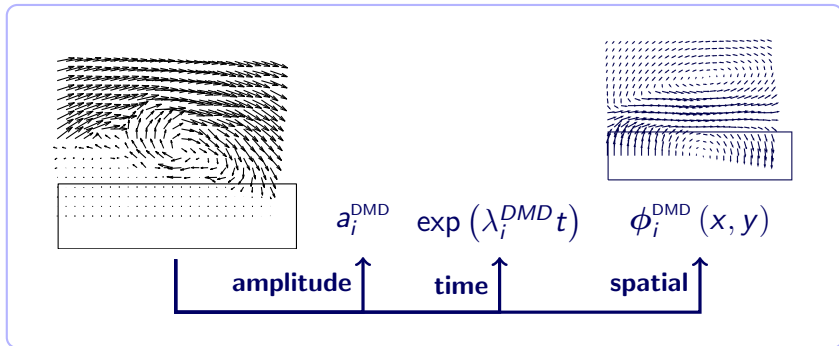
Methodology



Data Collection

1. EFD \rightarrow Tr-PIV
2. CFD \rightarrow uRANS

Methodology



Data Collection

1. EFD \rightarrow Tr-PIV
2. CFD \rightarrow uRANS

Decomposition

1. POD: $\phi_i(x, y)$ and a_i
2. DMD: $\phi_i(x, y)$, a_i and λ_i

Methodology

$$\mathbf{u}(x, y, t) = \sum_{i=1}^N \underbrace{a_i^{\text{DMD}}}_{\text{amplitude}} \underbrace{\exp(\lambda_i^{\text{DMD}} t)}_{\text{time evolution}} \underbrace{\phi_i^{\text{DMD}}(x, y)}_{\text{spatial mode}}$$

Modes shapes comparison

$$\text{MAC}(\phi_i^{\text{EFD}}, \phi_j^{\text{CFD}}) = \left(\frac{\phi_i^{\text{EFD}} \cdot \phi_j^{\text{CFD}}}{\|\phi_i^{\text{EFD}}\| \|\phi_j^{\text{CFD}}\|} \right)^2$$

Amplitudes and frequencies comparison

$$\text{ERR}(a_i^{\text{EFD}}, a_i^{\text{CFD}}) = \frac{\|a_i^{\text{EFD}}\| - \|a_i^{\text{CFD}}\|}{\|a_i^{\text{EFD}}\|}$$

Data Collection

1. EFD → Tr-PIV
2. CFD → uRANS

Decomposition

1. POD: $\phi_i(x, y)$ and a_i
2. DMD: $\phi_i(x, y)$, a_i and λ_i

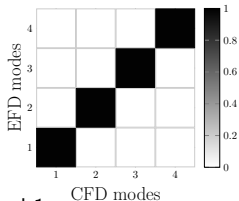
Comparison

1. $\phi_i(x, y)$
2. a_i and λ_i

Methodology

Modal Assurance Criterion

$$\text{MAC}(\phi_i^{\text{EFD}}, \phi_j^{\text{CFD}}) = \left(\frac{\phi_i^{\text{EFD}} \cdot \phi_j^{\text{CFD}}}{\|\phi_i^{\text{EFD}}\| \|\phi_j^{\text{CFD}}\|} \right)^2$$



MAC takes values between 0 and 1

0 if modes are not correlated and \rightarrow 1 as correlation increases

Data Collection

1. EFD \rightarrow Tr-PIV
2. CFD \rightarrow uRANS

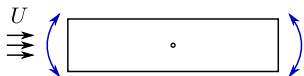
Decomposition

1. POD: $\phi_i(x, y)$ and a_i
2. DMD: $\phi_i(x, y)$, a_i and λ_i

Comparison

1. $\phi_i(x, y)$
2. a_i and λ_i

Oscillating rectangle



- EFD: rectangle undergoing **LCO**
- CFD: pitching frequency and amplitude **imposed** (same as EFD)

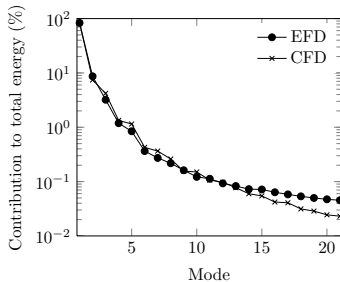
Problem

- No other EFD values than velocities
 - CFD data need to be validated
- ⇒ **Use decomposition methods!**

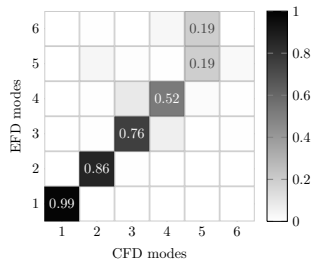
Oscillating rectangle

POD results

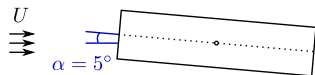
Energy content



MAC matrix



Static rectangle



Global values			
	\bar{c}_l	\bar{c}_d	St
CFD	0.83	0.46	0.136
EFD	0.53	0.45	0.152

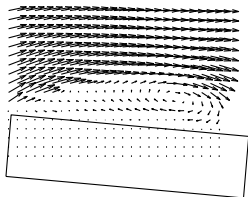
Problem

- \bar{c}_l from CFD far from EFD value
 - Why?
- ⇒ Use decomposition methods!

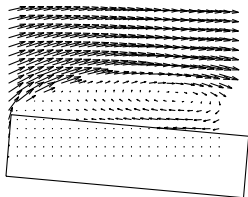
Static rectangle

DMD results

EFD: 2 modes



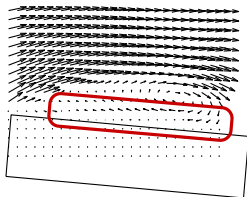
CFD: 2 modes



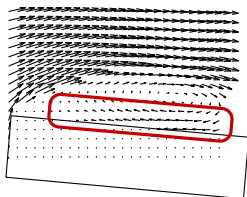
Static rectangle

DMD results

EFD: 2 modes



CFD: 2 modes



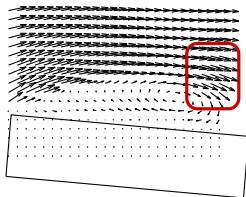
In CFD reconstruction

- More important reverse flow

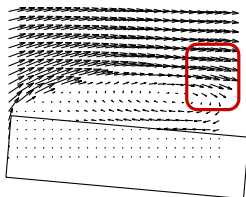
Static rectangle

DMD results

EFD: 2 modes



CFD: 2 modes



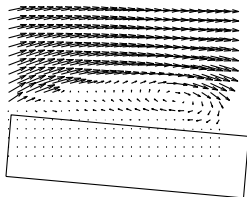
In CFD reconstruction

- More important reverse flow
- Discrepancies on the rear part

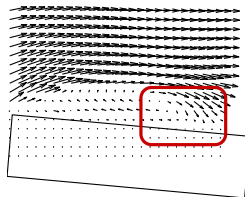
Static rectangle

DMD results

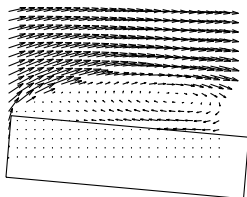
EFD: 2 modes



EFD: 3 modes



CFD: 2 modes



In CFD reconstruction

- More important reverse flow
- Discrepancies on the rear part
- Different dynamic of reattachment

Conclusion and future work

DMD and POD are useful

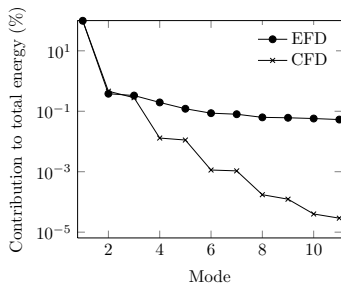
- Compare and validate CFD results
- Highlight and understand potential discrepancies

- Enlarge PIV window to get the rear part
- Apply DMD on lift evolution

Static rectangle

POD results

Energy content



MAC matrix

