# Using a second gradient model to simulate the behaviour of concrete structural elements

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#### Abstract

Being a quasi-brittle material, concrete under tensile loading exhibits a strain softening behaviour that cannot be accurately reproduced with classical (without an internal length parameter) continuum mechanics models. An internal length parameter must be introduced to regularize the problem, as in the case of the so called second gradient model. In this approach, an enriched kinematic description of the continuum is adopted considering higher (second) order gradients of the displacements following the work of Cosserat, Toupin, Mindlin and Germain. The model has been developed by Chambon and co-workers and has been mainly used with plasticity constitutive laws to reproduce the non linear behaviour of soils. It is here applied for the first time to concrete and reinforced concrete specimens considering material laws based on the damage mechanics theory. The advantages and limitations of the approach are discussed, and possible improvements towards more realistic responses are suggested.

Keywords: Second gradient, Concrete, Damage, Strain localization

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#### 1. Introduction

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Since the 70's [1],[2],[3] researchers study the strain localization in quasibrittle materials, or more generally in materials exhibiting strain softening. Strain localization zones are clearly observed in experimental tests [4] and it is well known that they cannot be modelled with classical (without an internal length parameter) continuum mechanics models. Analytically, the differential operator becomes hyperbolic and an infinite number of solutions is possible. Numerically, the loss of ellipticity appears as a pathological mesh dependency of the results. These shortcomings are due to the lack of an internal length parameter in the continuum model that characterizes the width of the localization zone [5],[6],[7],[8],[9],[10],[11]. Different approaches exist in the literature to regularize the problem and to obtain objective numerical global (i.e. forces, displacements) and local (i.e. strains, stresses, internal variables) results. The first are important for design purposes and the second to deal for example with durability and crack propagation problems. The different approaches are briefly summarized hereafter (see also [12] for a more detailed literature review):

- Regularisation based on energy: The principle is to keep the same fracture energy dissipated during the formation of cracks whatever the size of the finite element mesh [13],[14],[15],[16]. For this, the post peak behaviour of the adopted constitutive law is changed according to the size of the finite elements. This approach provides global results that may seem independent of the size of the mesh. Nevertheless, the localization zone is necessarily concentrated in one element (as in a classical continuum mechanics model without an internal length parameter) and thus local and global results are not objective. Results are also dependent on the orientation of the finite element mesh.
- Regularisation based on time dependency: Viscous terms are introduced in the model that restore the ellipticity of the differential operator [17]. However, because this method does not introduce an internal length to control the width of the localization zone, severe mesh dependence is avoided for dynamic but not for quasi-static calculations [18].
- Regularisation based on spatial dependency:

- Using a non local integral type variable (i.e. on the damage parameter or on the equivalent strain for constitutive laws based on damage mechanics) [8]. For this integral type model, the interaction between material points across a crack [19] can still pose problems. Similar difficulties also exist for materials presenting a different behaviour in traction and compression (consider for example the interactions across the compression and traction zones for a concrete beam submitted to bending, see also section 4). Different approaches can be found in the literature to deal with these problems, mainly consisting in modifying the adopted weight function either near the boundaries [20] or by introducing a dependence on the stresses [21]. Nevertheless, this last assumption implies that the internal length is no longer a constant material parameter but that it decreases with increasing loading. Furthermore, as is the case for the other regularisation techniques, the ability of the method to reproduce accurately global and local results under size effect needs to be more thoroughly studied [22].

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- Using strain gradients controlling the evolution of the internal variables (i.e. the second gradient of the plastic strain in the consistency condition and/or the flow rule) [7],[23]. This type of model is shown to be equivalent to the integral type model [23].
- Alternatively, the nonlocal variable can be defined via an implicit gradient of the corresponding local variable, and is then the solution of a boundary value problem [24]. This type of model is shown to be equivalent to the integral type model [23].
- By taking into account gradient of internal variables (the damage variable in the case of damage models) in the energy [25], [26], [27]. The gradient term here, acts as a penalization term for the cases of high localisation.
- More recently, strain localization due to damage has been treated using the thick level set approach [28]. The level set separates the undamaged from the damaged zone while the damage variable and its growth is a function of the level set propagation. The force driving the damage front is non-local in the sense that it averages information over the thickness in the wake of the front [28].

A rather natural way of introducing (indirectly) a length parameter in a continuum model is to account for the microstructure of the material. The general class of the so called microstructured models or higher order continuum models allows for the description of the kinematics of the microstructure by using an additional tensor in the displacement field. Higher order continuum theories can be traced back to the works of the Cosserat brothers [29], Toupin [30] and Mindlin [31],[32] and have been generalized and properly formulated by Germain [33],[34] using the virtual power method.

In this article, we choose to work with the second gradient model developed by Chambon and co-workers [35],[36],[37],[38],[39]. This model can be seen as a particular case of a higher order continuum (see section 2) and has been mainly used till now to regularize problems involving strain localization in soils. It is used hereafter to concrete and reinforced concrete elements.

The article is structured as follows: the theoretical framework of the second gradient model and its numerical implementation are at first presented. The objectivity of the numerical results is shown for a 1d concrete specimen and the evolution of the localization zone is discussed. The article ends with a case study, the simulation of a three point bending test on a reinforced concrete beam. Discussion on the numerical results show the advantages and limitations of the approach that should be considered as a first step towards the use of local second gradient models for concrete structures.

# 2. The second gradient model

### 2.1. Theoretical framework

As detailed in the seminal work of Germain [33],[34], using the virtual power method one can choose a field of virtual displacements to describe the proper kinematics of a higher order continuum including its microstructure. The internal stresses, limit conditions and equilibrium equations appear naturally as long as the linear form representing the virtual power is correctly defined and that it respects the principle of material independence.

The second gradient model developed by Chambon et al.[35],[36] can be seen as a particular case of a higher order continuum where up to second gradient terms are adopted and the macro strains are considered equal to

the micro strains. The authors have come to this assumption following experimental results that showed that for the case of geomaterials micro rotations equal macro rotations [40],[41]. They have presented case studies in the framework of plasticity and have shown that this type of model restores mesh objectivity but not the uniqueness of the solution [37],[38],[39],[42].

For the second gradient model, the virtual displacement field must be chosen as a field of continuous and continuously differentiable velocities. According to the general theory for continua with microstructure presented in [37] and assuming that micro strains are equal to macro strains, the virtual work principle equation takes the following form (for any  $\alpha$ ,  $\alpha^*$  defining the virtual quantity). For the sake of simplicity, we neglect hereafter the body force terms and the presentation is done for a 2d continuum:

$$\int_{\Omega} (\sigma_{ij} \frac{\partial u_i^{\star}}{\partial x_j} + \Sigma_{ijk} \frac{\partial^2 u_i^{\star}}{\partial x_j \partial x_k}) \ d\Omega = \int_{\Gamma} (p_i u_i^{\star} + P_i D u_i^{\star}) d\Gamma, \tag{1}$$

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- i,j and k (varying from 1 to 2)
  - $x_i$  the coordinates
- $u_i$  the macro displacements field
  - Dq the normal derivative of any quantity q

$$Dq = \frac{\partial q}{\partial x_k} n_k, \tag{2}$$

- $\sigma_{ij}$  the Cauchy stresses (macro stresses)
  - $\Sigma_{ijk}$  the double stresses
- $p_i$  the classical traction forces
- $P_i$  the double traction forces
  - $\Gamma$  the boundary of  $\Omega$

The Cauchy stress  $\sigma_{ij}$  is, as in classical continua, symmetric, the double stress  $\Sigma_{ijk}$  is symmetric with respect to its indices j and k. Application of the virtual work principle equation (1) and two integrations by parts provide

the balance equation and the boundary conditions. The balance equations become:

$$\frac{\partial \sigma_{ij}}{\partial x_j} - \frac{\partial^2 \Sigma_{ijk}}{\partial x_j \partial x_k} = 0. \tag{3}$$

Assuming that the boundary is regular (which means existence and uniqueness of the normal for every point of the boundary  $\Gamma$  of the studied domain), after one more integration by parts, we get

$$\sigma_{ij}n_j - n_k n_j D\Sigma_{ijk} - \frac{D\Sigma_{ijk}}{Dx_k} n_j - \frac{D\Sigma_{ijk}}{Dx_j} n_k + \frac{Dn_l}{Dx_l} \Sigma_{ijk} n_j n_k - \frac{Dn_j}{Dx_k} \Sigma_{ijk} = p_i, \quad (4)$$

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$$\Sigma_{ijk} n_j n_k = P_i, \tag{5}$$

where  $p_i$  and  $P_i$  are prescribed. The tangential derivative of any quantity  $q_i$  is defined by:

$$\frac{Dq}{Dx_j} = \frac{\partial q}{\partial x_j} - \frac{\partial q}{\partial x_k} n_k n_j.$$

To complete the equations of the problem two constitutive laws have to be introduced linking the static variables macro stresses  $\sigma_{ij}$  and double stresses  $\Sigma_{ijk}$  respectively with the kinematic variables strains (first gradient of the displacements)  $\frac{\partial u_i}{\partial x_j}$  and double strains (second gradient of the displacements). The two constitutive laws are usually supposed decoupled [35],[36]. The first gradient law can be any classical constitutive law (e.g. based on damage mechanics or plasticity). The second gradient law is usually based on linear elasticity.

Following the work of Mindlin [31],[32], the vector of double stresses for a bidimensional case has eight components and can be found considering the derivative of the strain energy with respect to the second gradient of the strains and symmetry of the tensor [31],[32],[43]. The obtained result is (where  $\alpha^1, \alpha^2, \alpha^3, \alpha^4, \alpha^5$  are five independent constants in the general case of an isotropic material):

$$\begin{cases}
\dot{\Sigma}_{111} \\
\dot{\Sigma}_{112} \\
\dot{\Sigma}_{121} \\
\dot{\Sigma}_{122} \\
\dot{\Sigma}_{211} \\
\dot{\Sigma}_{212} \\
\dot{\Sigma}_{221} \\
\dot{\Sigma}_{222}
\end{cases} = 
\begin{cases}
\alpha^{12345} & 0 & 0 & \alpha^{23} & 0 & \alpha^{12} & \alpha^{12} & 0 \\
0 & \alpha^{145} & \alpha^{145} & 0 & \alpha^{25} & 0 & 0 & \alpha^{12} \\
0 & \alpha^{145} & \alpha^{145} & 0 & \alpha^{25} & 0 & 0 & \alpha^{12} \\
0 & \alpha^{145} & \alpha^{145} & 0 & \alpha^{25} & \alpha^{25} & 0 \\
0 & \alpha^{25} & \alpha^{25} & 0 & \alpha^{34} & 0 & 0 & \alpha^{23} \\
0 & \alpha^{25} & \alpha^{25} & 0 & \alpha^{34} & 0 & 0 & \alpha^{23} \\
\alpha^{12} & 0 & 0 & \alpha^{25} & 0 & \alpha^{145} & \alpha^{145} & 0 \\
\alpha^{12} & 0 & 0 & \alpha^{25} & 0 & \alpha^{145} & \alpha^{145} & 0 \\
0 & \alpha^{12} & \alpha^{12} & 0 & \alpha^{23} & 0 & 0 & \alpha^{12345}
\end{cases} \right] \begin{pmatrix} \dot{\chi}_{111} \\ \dot{\chi}_{112} \\ \dot{\chi}_{121} \\ \dot{\chi}_{211} \\ \dot{\chi}_{211} \\ \dot{\chi}_{221} \\ \dot{\chi}_{221} \\ \dot{\chi}_{222} \end{pmatrix}, (6)$$

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$$\chi_{pqr} = \frac{\partial^2 u_p}{\partial x_q \partial x_r} \tag{7}$$

$$\alpha^{12345} = 2(\alpha^{1} + \alpha^{2} + \alpha^{3} + \alpha^{4} + \alpha^{5}),$$

$$\alpha^{23} = \alpha^{2} + 2\alpha^{3},$$

$$\alpha^{12} = \alpha^{1} + \alpha^{2}/2,$$

$$\alpha^{145} = \alpha^{1}/2 + \alpha^{4} + \alpha^{5}/2,$$

$$\alpha^{25} = \alpha^{2}/2 + \alpha^{5},$$

$$\alpha^{34} = 2(\alpha^{3} + 2\alpha^{4}).$$
(8)

As there is no clear physical definition of the different material constants and in order to simplify the equations the following particular case of the general isotropic form is proposed [39], where only one material parameter B is adopted with:

$$\alpha^{1} = 0$$

$$\alpha^{2} = B$$

$$\alpha^{3} = -B/2$$

$$\alpha^{4} = B$$

$$\alpha^{5} = -B$$

$$(9)$$

leading to:

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$$\begin{pmatrix}
\dot{\Sigma}_{111} \\
\dot{\Sigma}_{112} \\
\dot{\Sigma}_{121} \\
\dot{\Sigma}_{122} \\
\dot{\Sigma}_{211} \\
\dot{\Sigma}_{212} \\
\dot{\Sigma}_{221} \\
\dot{\Sigma}_{222}
\end{pmatrix} = \begin{pmatrix}
B & 0 & 0 & 0 & 0 & B/2 & B/2 & 0 \\
0 & B/2 & B/2 & 0 & B/2 & 0 & 0 & B/2 \\
0 & B/2 & B/2 & 0 & B/2 & 0 & 0 & B/2 \\
0 & 0 & 0 & B & 0 & B/2 & B/2 & 0 \\
0 & -B/2 & -B/2 & 0 & B & 0 & 0 & 0 \\
B/2 & 0 & 0 & -B/2 & 0 & B/2 & B/2 & 0 \\
B/2 & 0 & 0 & -B/2 & 0 & B/2 & B/2 & 0 \\
B/2 & 0 & 0 & -B/2 & 0 & B/2 & B/2 & 0 \\
B/2 & 0 & 0 & -B/2 & 0 & B/2 & B/2 & 0 \\
0 & B/2 & B/2 & 0 & 0 & 0 & 0 & B
\end{pmatrix} \begin{pmatrix}
\dot{\chi}_{111} \\
\dot{\chi}_{112} \\
\dot{\chi}_{122} \\
\dot{\chi}_{211} \\
\dot{\chi}_{212} \\
\dot{\chi}_{221} \\
\dot{\chi}_{221} \\
\dot{\chi}_{222}
\end{pmatrix}. (10)$$

This expression derives from the the strain energy density defined as (considering only the second gradient terms):

$$W_{\chi} = \frac{B}{2} (2\chi_{kii}\chi_{jjk} - \chi_{kii}\chi_{kjj} + 2\chi_{kji}\chi_{kji} - 2\chi_{kji}\chi_{ijk})$$
 (11)

It is shown in [35],[36],[42] and in subsection 3.3 that the material parameter B is directly linked with the size and the evolution of the localization zone.

The second gradient model uses the gradient of kinematic variables. The constitutive equation remains local and so ideas and algorithms used in classical models can be adopted. Formulation of a second gradient extension for any classical model is thus straightforward.

#### 2.2. Numerical implementation

The second gradient of the displacement necessitates the use of  $C^1$  elements in a finite element code. This is avoided by introducing a new field of unknowns  $v_{ij}$  imposed to be equal to the gradient of the displacements using Lagrange multipliers  $\lambda_{ij}$  [39],[44]. The new weak formulation of the problem then becomes:

$$\int_{\Omega} \sigma_{ij} \frac{\partial u_i^{\star}}{\partial x_j} + \sum_{ijk} v_{ij,k}^{\star} \ d\Omega - \int_{\Omega} \lambda_{ij} \left( \frac{\partial u_i^{\star}}{\partial x_j} - v_{ij}^{\star} \right) \ d\Omega = \int_{\Gamma} (p_i u_i^{\star} + P_i D u_i^{\star}) d\Gamma, \tag{12}$$

$$\int_{\Omega} \lambda_{ij}^{\star} \left( \frac{\partial u_i}{\partial x_j} - v_{ij} \right) d\Omega = 0.$$
 (13)

The problem is discretized using a nine (9) nodes finite element, where eight (8) nodes are used for the variables  $u_i$ , four (4) for  $v_{ij}$  and one (1) node at the center for the Lagrange multiplier  $\lambda_{ij}$ , see figure 1. This element has been implemented in the finite element code LAGAMINE (University of Liège) and the problem is solved using the classical Newton-Raphson method [44].

# 3. One-dimensional problem: objectivity of the results and evolution of the localization zone

#### 3.1. Objectivity of the results

The objectivity of the numerical results obtained with a second gradient model has been discussed in the past using constitutive laws for soils

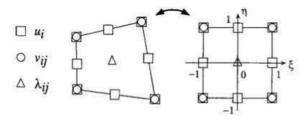


Figure 1: The 2d second gradient finite element, [44].

based on the mathematical theory of plasticity [37],[38],[39]. We study hereafter numerically the objectivity of the results for concrete specimens using constitutive laws based on damage mechanics.

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Let's consider the case of one-dimensional traction applied on a concrete specimen. Figure 2 shows the boundary conditions adopted for the 2d second gradient finite element mesh introduced into the code LAGAMINE. In order to avoid possible 2d effects, the vertical displacements  $u_2$  are considered equal to zero at the upper and lower boundaries along the specimen. The section is considered equal to  $0.1 \times 10^2$  and the length 1m. The right end of the specimen is fixed  $(u_1 = u_2 = 0)$  and the horizontal displacement U is applied at the left end. The additional external double forces are assumed to be zero at both ends. As no global snap-back is expected with these material and geometrical parameters (see [36] equation (38) for a snap-back criterion), the test is simply controlled by the imposed displacement.

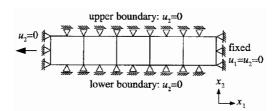


Figure 2: Concrete specimen under 1D traction: Boundary conditions and applied loading.

The constitutive law linking the stresses with the strains is a classical damage mechanics law [45] with an initial slope  $G^{el} = 30 \ GPa$  and a slope at the peak  $G^{tg} = -16.7 \ GPa$  corresponding to a strain  $\epsilon_{D0} = 1.e^{-4}$ , see

figure 3(a). The constitutive law linking the double stresses with the second gradient of the displacements is based on linear elasticity and depends on a single parameter considered equal to B = 0.37 GN, see figure 3(b) and section 2.1. The two constitutive laws are supposed decoupled.

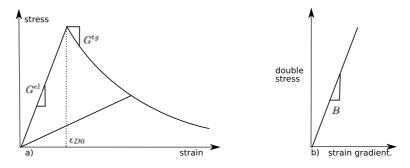


Figure 3: Concrete specimen under 1d traction, constitutive laws: (a) first gradient (stresses Vs. strains), (b) second gradient (double stresses Vs. strain gradients).

Analytically, bifurcation in a 1D problem appears at the peak. In order to visualize different possible solutions, an algorithm of random initialisation of the iterative solver of the equilibrium equation is applied just after the peak [46],[47]. For every step, a full Newton-Raphson involving a numerical consistent tangent stiffness operator for the complete model (i.e. the second gradient terms as well as the classical ones) is used. The results of two meshes with 14 and 50 elements are presented hereafter, [48]. The convergence criterion and convergence rates are detailed in section 3.2.

Figure 4 shows the global force versus displacement curve. The differences just after the peak correspond to distinct converged solutions. At the end of the loading however, both meshes converge to the same solution.

The above remark can be better understood looking at the local results. Figure 5 presents the distribution of the damage variable - varying from 0 (undamaged section) to 1 (damaged section) - just after the peak, at a strain equal to 1.2E-04. Figure 6 presents the distribution of the damage variable at the end of the loading at a strain equal to 2.9E-04. Just after the peak, the mesh with 14 elements converges to a solution with two patterns (a hard part and a soft (localized) part solution). The mesh with 50 elements converges to a three pattern solution (hard - soft (localized) - hard), figure 5. At the end of the loading however, it switches to the same two-pattern solution as the mesh with 14 elements, figure 6. This phenomenon of switching deformation

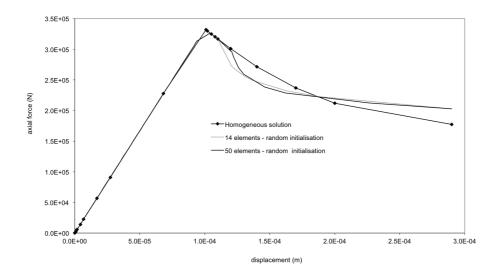


Figure 4: Concrete specimen under 1d traction: Force versus displacement curve.

modes was also found using plasticity models in [49]. At the end of the loading the localization patterns and global curves provided by both meshes become thus identical.

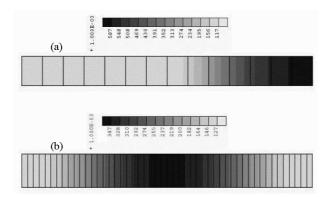


Figure 5: Concrete specimen under 1d traction: Distribution of the damage variable just after the peak (strain equal to 1.2E - 04): (a) 14 element mesh, (b) 50 element mesh.

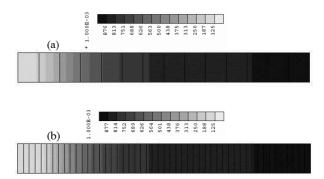


Figure 6: Concrete specimen under 1d traction: Distribution of the damage variable at the end of the loading (strain equal to 2.9E - 04): (a) 14 element mesh, (b) 50 element mesh.

From the above it is obvious that the use of the second gradient model with damage mechanics laws regularizes the problem (mesh independency) but does not restore the uniqueness of the solution for the corresponding boundary value problem. This was also found for constitutive laws based on plasticity [47],[49].

It is also observed in figure 4 that the non homogeneous results at the end present a higher strength. Indeed, and for the same end displacement, the maximum stress is larger than in the homogeneous case (see also [50]). An explanation of this numerical behaviour could be that the stiffness introduced by the second gradient terms does not vanish because of the adopted elastic law. A way to deal with this behaviour is either to introduce a coupling between the first and second gradient material laws (something that can also help to a priori control the evolution of the localisation zone (see [48] and subsection 3.3). It can also be argued that at some point a displacement discontinuity must be introduced because the continuous damage model cannot model correctly a crack. In our case, this should be done by inserting a cohesive zone element since the discontinuity should be introduced before the damage reaches its maximum value. An approach like this is for example proposed for the case of a model with gradient of the internal variables by Cuvilliez et al [51], see also [52],[53],[54].

# 3.2. Numerical convergence

Let  $F^{int}$  and  $F^{ext}$  be the internal and external nodal forces vectors obtained from the finite element discretization of equations (12) and (13). Let

also  $F^{obf}$  be the out of balance forces defined by  $F^{int} - F^{ext}$  and  $F^{react}$  the vector of the forces associated with the fixed degrees of freedom. In the following, the forces associated with the displacement degrees of freedom are noted  $F(u_i)$ , those associated with either the gradient or Lagrange multiplier degrees of freedom are noted  $F(v_{ij}, \lambda_{ij})$ . A norm for the force vectors is here adopted as the sum over the degrees of freedom of the absolute values of the vectors coordinates:

$$\parallel F \parallel = \sum_{\text{dof } i} |F_i| \tag{14}$$

The convergence criterion for the nodal forces is met when:

$$\frac{1}{2} \left( \frac{\frac{\|F^{obf}(u_i)\|}{N_u}}{\frac{\|F^{react}(u_i)\|}{N_u^{react}}} + \frac{\frac{\|F^{obf}(v_{ij},\lambda_{ij})\|}{N_v\lambda}}{\frac{\|F^{react}(v_{ij},\lambda_{ij})\|}{N_v\lambda^{react}}} \right) \le \text{precision}$$
(15)

where  $N_u$ ,  $N_u^{react}$ ,  $N_{v\lambda}$  and  $N_{v\lambda}^{react}$  are the number of degrees of freedom and the number of fixed degrees of freedom for the displacement field and both the gradient and Lagrange multipliers respectively. For the previous computation, the precision was set at  $10^{-11}$ .

A typical convergence profile for the one dimensional traction test is shown in figure 7 starting from the random initialization. Convergence difficulties observed at the first step are a direct result of the random initialization [47]. The first few iterations for this step show a important error. However as soon as a solution is found, a classical convergence rate is recovered.

Convergence difficulties are also encountered seven steps after the random initialization (corresponding to around 70 total iterations). They correspond to a switching deformation mode [49]. Again, a classical convergence rate is recovered once the new localized solution is found.

#### 3.3. Evolution of the localization zone

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The internal length parameter does not appear clearly from the previous equations. However, an analytical solution exists for the 1d problem when a bilinear first gradient law and an elastic second gradient law are adopted [35],[36]. The authors have proven that the solutions are built using patches of different fundamental solutions, consisting - as in section 3 - of hard parts corresponding to the unloading (or elastically loading) pieces and soft parts for the loading pieces of the medium. Various analytical patterns are possible

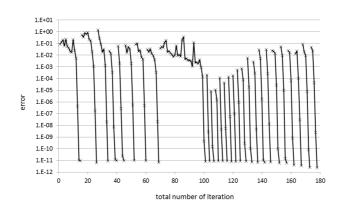


Figure 7: Concrete specimen under 1d traction: Typical convergence profile

(a hard-soft solution, a soft-hard-soft solution...) but their number is finite. This is not the case for a classical continuous mechanics medium without regularization where an infinite number of solutions is possible, [42]. The analytical solution [35],[36] introduces specific ratios for the hard  $(\omega)$  and the soft pattern  $(\eta)$ 

$$\omega^2 = \frac{G^{el}}{R} > 0, \tag{16}$$

$$-\eta^2 = \frac{-G^{tg}}{B} > 0, \tag{17}$$

and a wavelength  $l_s$  given by the following equation:

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$$l_s = 2\pi \sqrt{\frac{-B}{G^{tg}}}. (18)$$

This wavelength is proportional to the ratio of the moduli of the elastic second gradient law and of the softening branch corresponding to the first gradient law. It is only an indicator of the width of the localization zone, it does not provide its exact value. This is due to the fact that it corresponds to a period of the soft part standing alone. The real solution being a patch of the different solutions  $l_s$  is larger than the actual width [42].

Analytical equation (18) is valid for a 1d problem considering a bilinear first gradient law and an elastic second gradient law. Nevertheless, it can be used to estimate the initial length of the localization zone and its evolution

[42]. For concrete structural elements submitted to traction for example, constitutive laws have the general form of figure 3, [45]. As damage increases, the slope  $G^{tg}$  of the stress strain curve diminishes (in absolute value) and the width of the band is thus found increased (see figures 5,6). For the case of compression, where concrete exhibits a more ductile behaviour the localization zone will at first decrease and then increase, [42] (remark: a simplified 1d model to a priori control the evolution of the localization zone constant, decreasing or increasing - is presented in [48] considering a coupling between the first and the second gradient law).

In the following section, an engineering case study is presented: the numerical simulation of a three point bending test on a reinforced concrete beam using the second gradient model.

#### 4. Three point bending test of a reinforced concrete beam

#### 4.1. Experimental configuration

A three point bending test was conducted on a reinforced concrete beam having the following geometrical characteristics: thickness b=200mm, height h=500mm and span 5000mm. The geometry of the beam and informations about the steel reinforcement are shown in figure 8. A vertical cyclic load was applied at the upper part of the beam. Figure 9 presents the positions of six strain gages to monitor the axial strains on the steel bars.

#### 4.2. Finite element discretisation

The three point bending test is modelled hereafter as a two dimensional problem using the second gradient finite element described in section 2.2. Two meshes have been used for the simulations, figure 10. The first mesh consists of 5180 elements, 4148 of which are second gradient elements and 1032 truss elements representing the horizontal reinforcement. The average size of the concrete elements for this mesh is of 0.02 mx 0.035 m. The second mesh consists of 13494 elements with an average size of 0.01 mx 0.017 m for the concrete elements. Concrete and steel elements are supposed to be perfectly bonded (for engineering applications where we are mostly concerned with the behavior of a structure on the ultimate limit state (ULS) this assumption is acceptable [57], [58]). The end nodes at each lower extremity of the beam are blocked vertically; the right node is blocked horizontally. For the finite element calculations, monotonically increased displacements are applied at the upper part of the beam through an elastic plate, which is very stiff

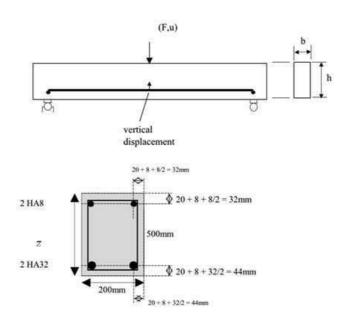


Figure 8: 3 point bending test: beam dimensions and steel reinforcement, [55],[56].

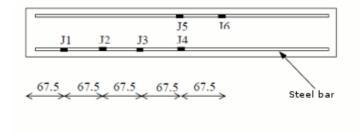


Figure 9: 3 point bending test: Position of the strain gages, [55],[56].

compared to the other materials. At the supports at both ends of the beam and on the upper part, where the displacements are applied, an elastic linear law is introduced to prevent from artificial numerical damage.

# 338 4.3. Material parameters

A classical damage mechanics law is used for the first gradient constitutive law [45]. The concrete material parameters are provided in table 1.

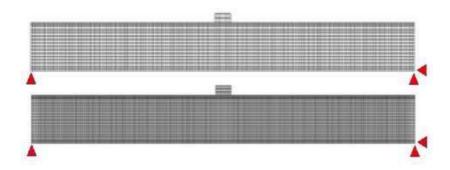


Figure 10: 3 point bending test: finite element meshes and boundary conditions.

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E (GPa)	$\epsilon_{D0}$	$A_t$	$B_t$	$A_c$	$B_c$	β
37.2	9.1E-05	0.7	6800	0.42	780	1.1

Table 1: 3 point bending test: concrete material parameters, [45].

The elastic modulus B for the second gradient constitutive law is considered equal to 1.5 MN. A way to choose this parameter is discussed in section 4.4. The first gradient material law and the second gradient are hereafter considered uncoupled.

An elastic perfectly plastic law is used for the reinforcement. The following parameters are adopted, tables 2 and 3 (where  $\sigma_y$  the yield stress and HA states for high adherence):

E (GPa)	$\sigma_y$ (MPa)	Section $(m^2)$
195	466	16.085E-04 (2HA32)

Table 2: 3 point bending test: steel parameters (lower part).

#### 49 4.4. Choosing the material parameter B

As discussed in section 3.3, the analytical solution provided in equation 18 can be used as an indicator of the initial size of the localization and its

E (GPa)	$\sigma_y$ (MPa)	Section $(m^2)$
195	466	1.0053E-04 (2HA8)

Table 3: 3 point bending test: steel parameters (upper part).

evolution. As damage increases, the slope of the first gradient law decreases (in absolute value) and the width of the band is increased.

The above argument is verified hereafter for the case of the three point bending test: A 1d second gradient calculation is performed on a concrete specimen adopting the parameters of table 1 and an elastic modulus B equal to 1.5 MN. Equation 18 provides an initial width of the localization band (approximately) equal to 15cm. In figure 11, the evolution of the profile of the concrete axial strains for different values of imposed displacements coming from the 1d calculation is provided. At the beginning of the loading the width of the localization zone is (approximately) equal to 15 cm. At the end of the loading it is found (approximately) equal to 30cm. Figure 12 presents the numerical profile of the concrete axial strains for the three point bending test along the red line for an imposed displacement of 6mm. The strain distribution presents several strain bands similar to the one found for the 1d case, figure 11. Comparison is more obvious in figure 13 where the strain distributions coming from the 1d model (dotted lines) and the three point bending test (continuous lines) are plotted. The same "peak strains" (the maximum strain found in the band) are found (see red, green and orange lines in figures 11 and 13).

From the above it is obvious that for a three point bending test a 1d second gradient model can be used to calibrate the parameter B and thus the initial width of the localization zone and its evolution.

#### 4.5. Numerical results

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Figure 14 presents the numerical force versus displacement (measured at the center of the beam) curve compared to the experimental one (note: during the experiment the beam was loaded and unloaded cyclically whereas in the simulation the beam is loaded with a monotonic increasing displacement). At this global level results for both meshes were identical. The force displacement graph exhibits the classical reinforced concrete behaviour in three stages: In the first stage, concrete and steel stay both in the elastic regime; then concrete starts to damage and the slope of the force displace-

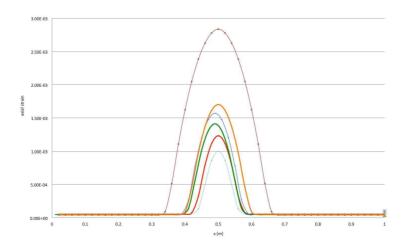


Figure 11: Choosing the material parameter B: 1d second gradient numerical calculation, evolution of the width of the localization zone.

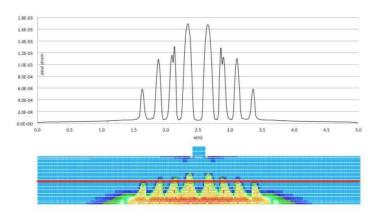


Figure 12: Choosing the material parameter B: 3 point bending test, concrete axial strain profile along the red line obtained numerically for an imposed displacement of 6mm.

ment curve changes. Finally, steel enters in a plastic phase and a second change in the slope appears. The numerical model provides however more stiff results at the last levels of the loading.

Figure 15 shows the evolution of the numerically obtained axial strains at different positions on the reinforcement bars as function of the global force. Results are found comparable to the experimental data [55],[56]. The model is

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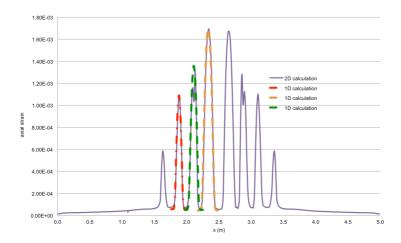


Figure 13: Choosing the material parameter B: comparison of the strain profiles coming from the 1d second gradient model and the three point bending tests.

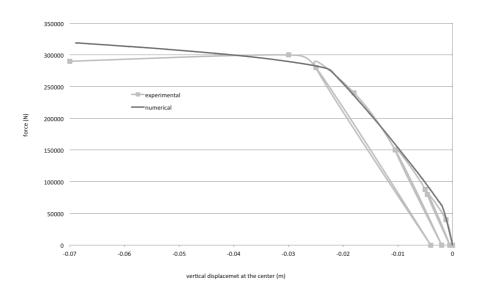


Figure 14: 3 point bending test: force-displacement curve.

able to capture positive and negative strains and thus to distinguish the parts
 of beam in traction or in compression. The strain gage 4 is inside a strain

localization zone, this is why the corresponding strain value is important.

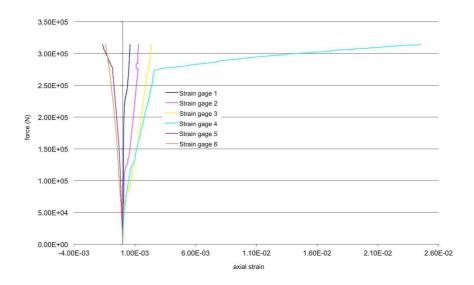


Figure 15: 3 point bending test: force versus axial strains obtained numerically at different positions on the reinforcement bars.

Figure 16 shows the distribution of the damage variable in concrete at different stages of loading and for the two mesh sizes. The damage distributions for the two meshes are similar (mesh objectivity) but not exactly the same. This is due to the fact that the second gradient method doesn't restore the unicity of the solution and small changes in the model can trigger different solutions. In contrast to a classical continuum mechanics model (without any regularization technique), the different solutions are however physically acceptable. For similar bending tests, classical non-local damage models which define an equivalent strain by averaging over a certain distance (material length parameter) have sometimes a tendency to develop artificially damage on the upper, compressed part of the beam (even if the local strain is not high enough to cause compressive damage). This is due to the principle of averaging over an area. There is no such problem with this model as the variables are local.

The damage pattern develops numerically with sudden peaks which experimentally correspond to developing cracks. The crack opening can not

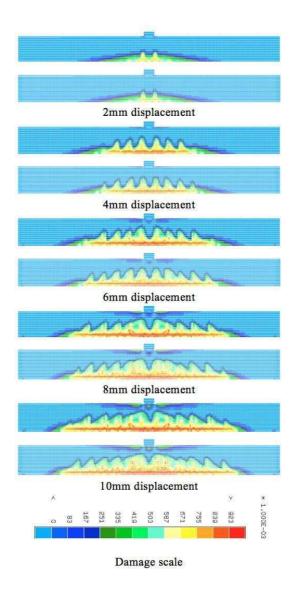


Figure 16: 3 point bending test: distribution of the damage variable in concrete at different steps for the two meshes.

be modelled directly in this simulation as the displacement field remains continuous, but it can be approximated from the damage model by simply measuring the displacement jump between two points located on the opposite sides of a damaged zone. This obviously works only when the damaged bands are clearly separated. The width and separation of the damage bands can be controlled by changing the internal length, which in our case means changing the slope of either or both the first gradient and second gradient constitutive laws.

# 5. Conclusions

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A second gradient model is adopted to simulate the behaviour of plane concrete and reinforced concrete structural elements using a classical damage mechanics law. The contribution is clearly a first step in the modelling of concrete failure with such type of models and the authors tried to highlight the advantages and drawbacks of the approach.

More specifically, objective (mesh independent) global and local results are obtained and damage is localized into bands whose width is controlled by the model parameters. The uniqueness of the solution is not restored and the evolution of the localization zone is discussed. In its actual form, the model provides a higher residual stress and spreading damage/strain fields. Nevertheless, possible solutions are proposed and efforts are currently made for their implementation. These results are encouraging and represent the first steps toward a wider use of second gradient models for concrete structures.

# 431 6. Acknowledgements

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